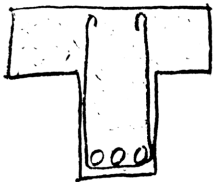
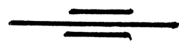
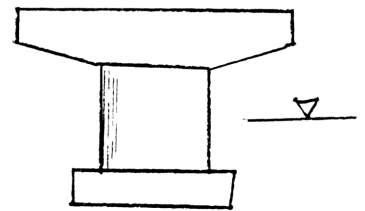
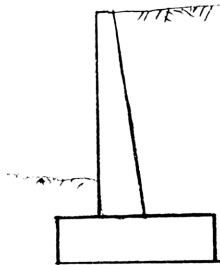
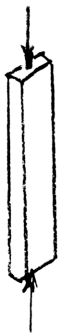


BRIDGE ROM

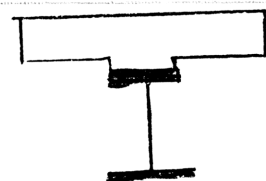
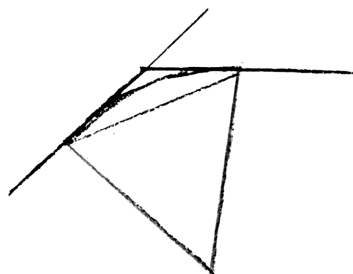
Civil and
Structural Engineering
Software
for the HP-41



$$\frac{b \pm \sqrt{b^2 - 4ac}}{2a}$$



$$PV = \left(\frac{1}{2}\right)^n \cdot FV$$



USER FLAGS

Flag 00	clear	Full output, with intermediate results
	set	Shortened format to save time
Flag 01	clear	COGO programs use North, East
	set	COGO programs use X, Y
Flag 02	clear	BEEP and TONE warnings disabled
	set	BEEP and TONES enabled (in subroutine S)
Flag 03	clear	Geometry programs use angles in decimal degrees (dd)
	set	Geometry programs use angles in Degrees-Minutes-Seconds (DMS)
Flag 04	clear	Structural programs follow AASHTO
	set	Structural programs use ACI/AISC/NDS

DOCUMENTATION

Documentation for most programs follows this pattern:

- 1) OVERVIEW Details softkey and flag use, SIZE requirements, input explanation, and other quick reference information. For use once the user has become familiar with the program.
- 2) EXPLANATION In-depth documentation of how the program works. Explains the program's features, methods, assumptions, and limitations.
- 3) EXAMPLES Hand-worked examples to show how the program works and how it is used.
- 4) LISTING A program listing with the stack configuration traced through every step, for a deeper look at the program.

SECTIONS OF THIS MANUAL

Part A	(yellow)	Contents; Introductory Material
Part B	(white)	Geometry Programs
Part C	(yellow)	Utility Programs and Routines
Part D	(white)	Structural Programs

NOTICE

These programs and their documentation are computational aids only. They are not intended as a substitute for professional judgement and cannot be used as such. It must be emphasized that engineering calculations involve assumptions allowing the engineer to treat real situations as mathematical problems. These programs do not "solve" real situations; they solve mathematical problems. The correlation between the assumptions made and the real situation must be judged by a competent qualified professional, and the results accorded only as much value as the initial assumptions merit.

As much as possible, these programs follow accepted practice in the profession; and every effort has been made to verify their correctness. However, the seller makes no express or implied warranty of any kind concerning these programs or their documentation, including the sample calculations included. Anyone making use of these programs or the material in this documentation does so at his or her own risk and assumes any and all liability resulting from such use.

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D101	CCOL	Rectangular concrete column analysis
D109		Steel design programs
D110	BM	I-beam section properties
D113	SCOL	Steel column analysis
D123	SBCOL	Steel beam-column analysis
D133	COMP	Composite beam analysis
D163	PLG	Built-up plate girder analysis
D205	WBM	Wood beam analysis
D213	WCOL	Wood column and beam-column analysis
D229		Truck moments on simple spans
D243	RW	Concrete cantilever retaining wall analysis
D255	NEO	Elastomeric bearing pad analysis
D269	PIER	AASHTO T-pier loads
D289	FTG	Footing analysis with biaxial bending
D299	SHFTG	Short footing analysis program

LIST OF LABELS

This is a list of all the global alpha labels in the module, in the order of a CAT 2 listing, with the number of bytes necessary to copy the program to main memory (RAM). The labels with the byte counts following them are the first label of that program; i.e. line number 01. Parentheses indicate subroutines which are probably not useful as stand-alone programs.

<u>LABEL</u>	<u>BYTES</u>	<u>DESCRIPTION</u>
<u>PAGE 1</u>		
CHC	515	Horizontal curve routine for coordinate geometry (COGO)
HC		Horizontal curve program
STOR	1186	Stores coordinate points (COGO)
INV		Finds the distance and azimuth between two points
DUP		Duplicates points (copies to other point numbers)
ANGL		Finds the angle subtended by two points about a third
LLI		Sets a point a given distance along a line defined by two other points
PART		Sets a point part way along a line defined by two other points
(XY)		Returns the difference in X and Y of two pts.
(FS)		Appends 'SQ' (square feet) to alpha
(FT)		Appends 'I' (feet) to alpha
DUMP		Dumps (lists) coordinates of desired points
OFPT		Finds the offset of a point from a line defined by two other points
OFAZ/OFBR		Finds the offset of a point from a line defined by a point and an azimuth/bearing
IAZ/IBR		Intersect two lines, each defined by a point and an azimuth/bearing
(SP)		An internal routine to store points
IPT		Intersect two lines, each defined by two points
PRL		Set two points defining a line parallel to a line defined by two other points
LAN		Set a point by backsight angle and distance
LAZ/LBR		Set a point by azimuth/bearing and distance
(BA)		Convert bearing (angle, quadrant) to azimuth
ADJ		Adjust a traverse by compass rule
(P)		Returns the coordinates of a point

ARCS	838	Intersect two arcs (distance distance)
APT		Intersect an arc with a line defined by two points
AAZ/ABR		Intersect an arc with a line defined by a point and an azimuth/bearing
(PM)		An internal routine used to choose between two possible solutions for arc routines
EXTAN/INTAN		Finds point of external/internal tangency for two circles
PTAN		Finds point of tangency of a line through a point to a circle
C3P		Circle through three points
ATRV		Traverse along an arc by arc distance
ATVA		Traverse along an arc by angle subtended
AFRC		Traverse along an arc, a fraction of its subtended angle
FREE		List of free (unused) point numbers
AREA	209	Finds the area bounded by coordinate points
(VC*)	868	"Port of entry" to v. c. routines
VC		Elevations along a vertical curve across a skewed, crowned or superelevated bridge deck
STA		Stations along a vertical curve at a given elevation
SCOL	385	Steel column analysis (axial only)
(SFa)		Steel column allowable stress subroutine
POB		Coordinate geometry stationing along a line

PAGE 2

BWT	184	Reinforcing bar weight calculation/summation
(S)	722	Set-up routine: SIZE check, title block, etc.
(AX)		Appends the integer part of the X register to ALPHA without disturbing the stack
(N)		Prompts for reinforcing stress and concrete strength
(Fc)		Prompts for concrete strength
(Mc)		Figures concrete cracking moment and reinforcing
(U)		Calculates reinforcing by ultimate strength methods
(AS)		Displays area of steel
(IS)		Appends _SQ" (square inches) to contents of ALPHA register
(IN)		Appends " (inches) to ALPHA
(R)		Calculates minimum reinforcing
(K)		Appends "STRS=....KSI" to ALPHA
(KS)		Appends "=....KSI" to ALPHA
(M=)		Moment display
(FK)		Appends "_'K" (foot-kips) to ALPHA
(M)		Moment prompt routine
QUAD		Quadratic equation solution-prompts A, B, C
(Q)		Quadratic equation solution with C, B, A in stack
(L)		Short underline
(LL)		Long underline
(NO)		Appends "number" sign (#) to alpha
(L*)		Line of asterisks
(CL)		Long line of any user-input character or string
(BD)		Prompts for B" and D" (dimensions in inches)
(W)		WSD/LFD prompt and set-up routine
(AA)		AASHTO/ACI (AISC, NDS) display and set-up
(OK)		Displays OK or NG (no good)
NEO	683	Elastomeric (neoprene) bearing pad analysis
WBM	663	Wood beam analysis
WCOL		Wood column and beam-column analysis
(WFc)		Wood compressive stress routine
PLG	1700	Plate girder analysis
(V)		Viewing subroutine
(PV)		Print-or-view subroutine
(SFb)		Unbraced-length bending stress subroutine

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LFAN	1159	Ultimate strength (load factor) concrete beam analysis
Ic		Working stress concrete beam analysis
WSBM	713	Working stress concrete beam design
LFBM		Load factor concrete beam design
CORBL		WSD or LFD concrete corbel design
VST	260	Concrete shear design
(VS)		Concrete shear design subroutine
COMP	1449	Composite steel-concrete beam analysis
BM		I-beam section properties
(SBM)		I-beam section properties subroutine
(AY)		I-beam section properties subroutine
(FY)		Prompts steel yield stress
FIS	158	Decimal feet to Feet-Inches Sixteenths
FIS+		Feet-inches-sixteenths addition
FIS-		Feet-inches-sixteenths subtraction
FTd		Feet-inches-sixteenths to decimal feet
IP	185	Interpolation
ST		Storage register review
(RH)		Reinforcing ratio check

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FTG	714	Footing analysis under biaxial loads
SHFTG		Short footing analysis program
YZ		Stack manipulation: Y<>Z
YT		Stack manipulation: Y<>T
ZT		Stack manipulation: Z<>T
RP		Rectangular-to-polar for flag 03
PR		Polar-to-rectangular for flag 03
FI	324	PPC ROM financial analysis program
SSS	463	Triangle solution: side-side-side
ASA		Triangle solution: angle-side-angle
SAA		Triangle solution: side-angle-angle
SAS		Triangle solution: side-angle-side
SSA		Triangle solution: side-side-angle
(Dd)		Display degrees according to flag 03 convention
(Pd)		Prompt degrees by flag 03
PIER	789	AASHTO pier loads
3S2	459	Type 3S2 truck moment
3S3		Type 3S3 truck moment
T4		Type T4 truck moment
T3		Type T3 truck moment
2S1		Type 2S1 (HS20) truck moment
T2		Type T2 truck moment
CO		Cosine function for flag 03
SI		Sine function for flag 03
TA		Tangent function for flag 03
ACO		Arc-cosine for flag 03
ASI		Arc-sine for flag 03
ATA		Arc-tan for flag 03
CV	364	PPC ROM curve-fitting program
BC		PPC ROM block register-clearing routine
SV		PPC ROM "solve" program for F(x)=0
SSD	748	Stopping-sight-distance vert. curve
RW		Retaining wall loads
KJ	88	Reinf. concrete WSD constants K, J
LPT		COGO: Locate by bearing thru points
PYR		Avg. area for truncated pyramid vol
P1	62	Append "(" to Alpha
P2		Append ")" to Alpha
PP		Append "-" to Alpha
MU		Append Greek letter mu (u) to Alpha
AT		Append "@" (at) to Alpha
AM		Append "&" (ampersand) to Alpha

FOREWORD

A few notes on the use of this module and its documentation:

Starting with the basics, most programs are accessed by keying XEQ, (alpha), "Program Title," (alpha); then just answer the prompts. Exceptions to this are noted in the individual program's documentation, and are usually one of two types: some programs require input with softkeys (e.g. HC, the horizontal curve program); and some are written to be used as functions or sub-routines, requiring input in the stack or specified registers. Again, see the program's documentation. Do NOT use the ENTER key unless directed to by an upward arrow (\nearrow), which occur in prompts for multiple items of input (used mostly by the coordinate geometry routines); just key in the number and hit R/S. Although it was avoided as much as possible, some prompt sequences require values in the stack above the input value; hitting ENTER/ \nearrow changes the necessary stack organization. Consult the program listings for more information.

SOFTKEYS

People who have used HP software on the HP-41, and especially on the HP-97, will notice a difference in the way softkeys are used in this module's programs. (Softkeys are the top two rows of keys, A-J, and the shifted top row, a-e, which in USER mode access local alpha labels (A-J and a-e) in the program the user is in.) On the HP-97, input was accomplished by keying in a number, pushing a softkey, keying in the next input, pushing another softkey, and so on. In most of these programs, instead, pressing a softkey brings the program to a given point in the input sequence; it brings up a prompt for the input rather than processing the (already keyed-in) input directly. This method was chosen because of the uncertainty of blindly pushing softkeys. The tradeoff is that the softkeys are slightly faster once gotten used to. Exceptions to this idea are programs HC, the PPC ROM routines FI and CV, and the "function" routines (CO, SI, FIS, etc.) A few programs (SSD and the triangle solution routines) can be used either way.

FLAGS

The use and understanding of flags, especially flags 00 through 04 (hereinafter called "user flags") is essential to the operation of these programs. Please take the time to read the short section following on flags. If you don't know what flags are or how to use them, please consult your HP-41 Owner's Manual.

DISCLAIMER

Please read and understand the disclaimer on a preceding page. Using these programs and their documentation as a learning aid is fine; using them as a sole source of knowledge is not fine at all. And using them to replace engineering knowledge and judgment is criminal.

DOCUMENTATION

Documentation for most programs starts with an "overview" sheet detailing softkey and flag usage, SIZE requirements, and a brief explanation of input. This is intended as a review and reference sheet once the user is familiar with the program. To get familiar with the program, read the full documentation and worked examples which follow. For a really thorough knowledge of the program, consult the listing and stack trace. Documentation for programs which are intended for use as functions or subroutines may deviate from this standard.

SYNTHETIC PROGRAMMING

The user may notice a few unfamiliar symbols and commands in the program listings, prompts, and output. An understanding of these is not necessary to use the program, but for the curious, see the section "Synthetic Programming and Loose Ends."

THE PROBLEM with COMPUTERS

Finally, a plug. Studies of people using computers have noted four major problems. First, many people panic when first using new programs or machines. Second, many people, even experienced users, feel a loss of control. The computer often seems to be running things; the user can see no flow of progress in a task, cannot change his or her mind, and is shown no mercy for input mistakes. Third, there is often complacency on the part of the user over the quality of the work being done because of boredom and the feeling that the computer is doing the work while the user only tends the machine. And fourth, there is a loss of ability to do the work "the old-fashioned way", by hand, due to simple loss of practice. These problems have had an influence on the writing of this module; the programs have been written to help engineers solve problems, not to do it for them. An attempt was made to keep the user informed of the flow of the solution through the display of intermediate results as in hand calculations, and by the use of units on most prompts and output. The user has control of the programs through the simple power of the R/S key to stop an errant run, and softkey-accessible prompt sequences to

correct and restart it. Hopefully the combination of units on prompts and output, the documentation, worked example calculations, and stack-traced program listings will work to eliminate most of the confusion and keep the user abreast of the method of solution. And almost any program (with the exception of AX, AY, and the neutral axis iteration portion of WSBM) can be traced by single-stepping through it.

FLAGS

Of the fifty-six flags in the HP-41 operating system, only the first thirty or so are accessible to the user. Of these, flags 11 through 17 and 21 through 29 have specific meanings or functions in certain situations or with certain peripherals, and flags 18 through 20 may have on future peripherals. Flags 00 through 10 are reserved for use in users' programs; flags 18 through 20 can usually also be used.

"USER" FLAGS

The status of flags 00 through 04 is always shown in the display; for this reason these flags are used by this module as mode-control or "User" flags. Typically the user will set these as desired and then pretty much leave them alone, just verifying their status at the beginning of each run. The programs in this module refer to flags 00 through 04 constantly but they never change these flags. However, other programs (either plug-in modules or the user's programs) may alter them, so they should be checked at each use. The modes controlled by these flags are:

Flag 00	clear	Full output, with intermediate results
	set	Shortened format to save time
Flag 01	clear	Coord. geometry programs use North, East
	set	Coordinate geometry programs use X, Y
Flag 02	clear	BEEP and TONES (e.g. warnings) disabled in subroutine "S"
	set	BEEP and TONES enabled
Flag 03	clear	Geometry programs use angles in decimal degrees (dd)
	set	Geometry programs use Degrees-Minutes-Seconds (DMS)
Flag 04	clear	Structural programs follow AASHTO
	set	Structural programs follow ACI/AISC/NDS

Flags 05 through 10, 14, 18 through 22, 25 through 27, and 29 are used by many of the programs. These flags are set and cleared as needed by the programs; the user does not need to worry about their status (with the exception of flags 21 and 25--see below) as the programs control it. Flag 14 is used for just one

thing by this module, and flags 06 and 07 are usually used for only one purpose. Flags 21 through 27 and 29 are used for their stated purpose as given in the HP-41 Owner's Handbook. For the convenience of users trying to trace a program, their functions are:

Flag 06	clear	WSD (Working Stress Design)
	set	LFD (Load Factor or Ultimate Strength Design methods)
Flag 07	clear	"OK" in routine OK
	set	"NG" (no good) in routine OK
Flag 14	clear	Functions L and LL print a line (-----)
	set	L and LL print a line of asterisks (***)

Note that flag 14 also has the function of controlling card-reader over-write. As used by these programs it is cleared almost immediately after being set. However, if a card reader is being used there is a risk if the user single-steps through the SF 14 command but stops before the FS?C 14 command which clears it a few steps later.

Flag 21		Set in subroutine S to control the display (stops at AVIEW). If calculator is turned off and then back on again, flag 21 must be set manually to resume without XEQ'ing the program from the beginning. If there is a printer connected and on, this can be neglected.
Flag 22	clear	There has been no numeric input since cleared
	set	There has been numeric input
Flag 25	clear	Normal operation
	set	Error ignore-clear as soon as not needed
Flag 26	clear	BEEP and TONES disabled
	set	BEEP and TONES enabled
Flag 27	clear	Normal mode
	set	USER mode-softkeys and key assignments active
Flag 29	clear	Commas and decimal points not displayed
	set	Commas and decimal points are displayed

SYNTHETIC PROGRAMMING AND OTHER LOOSE ENDS

This section will try to explain some of the mysteries of the programming in this package, and pull a few odds and ends of the module together. It probably should have been called "Loose Ends and Synthetic Programming", because the loose ends will be attempted first.

"BYPASSING" A PROMPT

First, a word used often in the documentation: "bypass." To bypass a prompt means to press the R/S (run/stop) key and thus continue without keying in a number or answering a prompt. Many prompts have default values which are used if the prompt is bypassed; consult the "Input Summary" section of each program's documentation. To check the default, backarrow the prompt. R/S may still be pressed to continue.

YES/NO PROMPTS

Most prompts in these programs ask for numerical input; some, however, require a "yes" or "no" answer to a question. This is explained in each program's Input Summary section. These programs use a simple convention: key in 1 for "yes", 0 (zero) for "no".

NUMERIC INPUT

When keying in a number in response to a prompt, simply key it in and press R/S (or the appropriate softkey for a few programs, e.g. HC and SSD). Do not use the Enter/ key unless directed to by an upward arrow (↑) in the prompt. Although it was avoided as much as possible, some input sequences require values in the stack above the input; hitting Enter/ alters this stack organization. The coordinate geometry programs typically require multiple input; a tradeoff was made for increased speed and convenience at the cost of increased concentration needed to avoid errors.

SOFTKEYS

Softkeys are the top two rows of keys, and the top row shifted, when USER mode is on. These are labelled A through J and a through e; When pressed (in USER mode), they try to find and execute a label of the same letter within the program that the calculator is currently set to. In order to use these keys, it is useful to understand the hierarchy the calculator follows when a softkey is pressed. First, it checks to see if a function or program has been assigned to the key. If not, the calculator looks for the corresponding local

alpha label within the current program. If it finds the label it starts running the program there; if not it executes the function that the key would have executed right away if not in USER mode. If the key wasn't a softkey, the second step is skipped. To summarize:

- 1) Key assignment
- 2) Local alpha label
- 3) Native function

There are a couple of things to be learned from this long-winded discussion. One, it's not a good idea to assign functions or programs to the top two rows of keys because they can't function as softkeys; and two, if trying to execute a native function in USER mode, the search process is maddeningly slow. Turn off USER or assign the function to its own key (yes, contradictions).

LABELS

Having explained everything there is to know about softkey hierarchy, on to labels. Labels too have a hierarchy, easy to remember: CAT 1, CAT 2, CAT 3. When the user executes a label from the keyboard (i.e. by spelling it out), the calculator looks for it first in Catalog 1 (user-written programs), then Catalog 2 (programs in ROM, such as this module), then Catalog 3 (native HP-41 functions). This is useful because a program can be copied down into Main Memory (RAM) from ROM and tailored to the user's wishes (needless to say, at some risk of introducing bugs). Then when executed, the calculator finds and runs the altered version in RAM rather than its predecessor in ROM. It also means that a program in either RAM or ROM can have the same label as a native function, and still be found and executed.

INITIALIZATION SUBROUTINE "S"

Most of the programs use subroutine "S" to initialize the calculator. Among other things, it sets flag 21 for display control, checks the SIZE required, sets flag 26 according to the status of flag 02 to silence or enable BEEPs, and sets the trig mode to DEG. If a printer is attached, "S" also prints a header with the time and date (if a Time Module is present) and gives the opportunity to print a title, which is limited to 24 characters. The time/date line works best with the display set to FIX 4 or less. For keying the title, "S" turns Alpha mode on and then off again; simply key in the title and press R/S. The title may, of course, be bypassed.

XROM vs. XEQ

There may be some confusion about the appearance of XROM instead of XEQ in program listings. XROM means "execute from ROM" (Read Only Memory, i.e. plug-in modules). Many of the programs in this module may be used as subroutines in users' programs. They are entered into a program using the XEQ key, just as for a native HP-41 function or a user-written subroutine. The label type can be told by how it looks in a program listing:

23 COS	Execute native HP-41 Cosine function
24 XEQ "COS"	Execute user program COS from Main Memory
25 XROM "COS"	Execute program COS from ROM

Some program listings in this documentation contain both XEQ and XROM statements. This is due to the revision process; XEQ statements have been revised since the first version of this software was published. However, all these (printed) global XEQ statements were changed during the module manufacturing process and should now (hopefully) list as XROM.

LOGIC SEQUENCES

Another confusing thing which pops up frequently in these programs is the use of double and triple logic sequences. These are used for the conditionals "if A or B then..." or "if A and B then..." For example, to jump to label 14 if either flag 06 or 08 is set, the sequence is:

26 FC? 06	Opposite of test A (Not A)
27 FS? 08	Test B
28 GTO 14	

To do an "and" test requires a third conditional which must never be true. FS? 53 is often used since it always tests clear; if it is certain that the number in X is positive, for example, the conditional X<0? can be used, saving a byte of memory. For example, to jump to label 14 only if flags 06 and 08 are both set:

29 FS? 06	Test A
30 FC? 08	Opposite of test B (Not B)
31 FS? 53	Always false
32 GTO 14	

"SIGN" FUNCTION

Many of these programs use the HP-41 function SIGN, a relatively little-known function. Its main usefulness lies in the fact that it stores the previous contents of X in LastX. It does this faster than most other functions, and uses only one byte.

"PRGM" ANNUNCIATOR

During long program runs with a printer attached, the message or output last displayed will remain in the display instead of being replaced by the "flying goose." However, the PRGM annunciator stays on to verify that the program is running. If this does not happen, the message is a prompt and the program is expecting input.

SYNTHETIC PROGRAMMING

The rest of this article deals with a subject that has come to be called "synthetic programming." Synthetic programming is a collection of techniques that allow the user to synthesize non-standard commands and display characters, and to access and manipulate storage registers normally not available to the user.

SHORT-FORM EXPONENTS

First, short-form exponents. A number such as 1000 may be keyed into a program as 1 E3, saving one byte of memory. However, the leading "1" is not needed. Removing it with synthetic programming methods results in the program line E3, which still means 1000 but now takes only two bytes and also executes faster. Similarly, 100 can be programmed as E2, 10 as E1, and 0.001 as E-1. Also, 1 can be represented by E0 or just E. The replacements for 1 and 10 don't save any bytes; however, they do execute faster. Zero can also be represented by a lone decimal point. Again, this (non-synthetic) representation saves no program space but is faster. To demonstrate any of these, just single-step through them.

NOP's

The conditionals DSE and ISG are often useful as simple decrement/increment functions. To achieve this, these commands are followed with a command which does nothing, or at least nothing harmful. These no-operation (NOP) commands are often short-form local labels or a STO X command. However, the only true NOP which has no effect and also takes only one byte (STO X takes two) is a synthetic command called Text-0. Text strings all have headers which tell the length of the string to follow. A Text-0 command is a header for an alpha string of length zero (it even sounds useless). It prints as " "; it displays as a superscript T (text symbol) with nothing after it.

NON-STANDARD SYMBOLS

The prompts and output descriptors use a few symbols not available on a standard HP-41. These include foot and inch tics (',"), the "number" hash-mark symbol (#), and parenthesis. Some of these have been made available to the non-synthetic programmer through labels FT, IN, NO, and others; these labels simply append the symbol to whatever is in the alpha register. Using these labels actually takes one less byte than the actual synthetic append they execute, although it is a little slower. The HP-41 also has available to its display several other symbols which were not thusly made available but which can be used by anyone willing to learn synthetic programming: brackets [], the Greek symbol lambda (sort of), and an exclamation point. If you don't mind a messy display, there are many symbols which display as a starburst but which print as themselves: nearly the entire Greek alphabet, our lower-case alphabet, as well as many other characters.

SYNTHETIC STORAGE REGISTERS

One of the most useful things available through Synthetic Programming is the ability to access a number of the calculator's internal storage registers for use (within certain strict limitations) as regular storage registers, or to control certain aspects of the machine's behavior. These registers are located at the bottom end (i.e. they have the lowest-numbered addresses) of the calculator's memory. These first registers have been given the name "Status Registers" because their contents are recorded by the card-reader function WSTS. Starting from the bottom, the Status Registers and their functions are as follows:

T]	Stack and Last X
Z		
Y		
X		
L		
M or []	Alpha register
N or \		
O or]		
P or /		
Q		Printer use, internal "scratch"
†		Unshifted key assignments
a]	Subroutine RTN chain and current line number
b		
c		Cold start constant, "curtain"
d		Flag register
e		Shifted key assignments

The first five are the stack and LastX registers, and are accessible through normal means. Synthetic programming is not needed to use these.

Next comes the Alpha "register", actually four registers of seven bytes each. Under usual conditions the Alpha register is 24 bytes long; the last four bytes of P are used for other things by the calculator. Because of this, register P has some very confining limitations on it and finds only infrequent use; see the next paragraph. However, registers M, N, and O have no such limitations; the only caution regarding their use is that any text introduced into Alpha, by the program or the user, wipes out the data stored in these registers. They are therefore well suited to carrying intermediate constants needed only in calculations leading up to a given item of output.

Data stored in register P is altered by, among other things, number entry (but not RCL), any operation that causes a number to be displayed, and SST'ing (single-stepping) through a program. Thus, the few programs which use register P (AX, the Trig functions, AY, and the neutral-axis iteration portion of WSBM) cannot be single-stepped through. If you want to trace through any of these, either put a STOP after the area and R/S to get through it, or copy the program down into main memory and change command using register P to one using a numbered storage register.

The next two registers, Q and T, are used too frequently by the calculator to be of much outside use. Register Q is used occasionally by some advanced synthetic programmers, but it does not appear in these programs.

Registers a and b are the line number and subroutine RTN chain registers. Register b contains the current line number and the first two and a half return addresses pending; register a contains the next three and a half return addresses. Each address, as may be guessed, is two bytes long. Every time a program comes to a RTN, these two registers shift two bytes, and the first RTN address pending becomes the current line number. Register a may thus be used if no XEQ's or RTN's are met and if the subroutine return depth does not exceed two. If an XEQ or RTN occurs, the data in registers a and b is shifted two bytes left or right, trashing the data. Even if it is shifted back by the opposite function, the data is altered. If the program has more than two RTN's pending, there is a worse problem: part of your data gets used for a return address, sending the calculator off into another dimension. The program pointer usually ends up somewhere in the key assignment registers and tries to interpret your key assignments as program steps. (If this happens, press CAT 1 to get back home, then fix the

problem before it happens again.)

Register c contains two important pieces of information: the location of the "curtain", the divider between data and program storage; and a number called the "cold-start constant." When the calculator is turned on, and at certain other times, this number is checked by the processor. If it is anything other than the expected constant, the processor thinks something terrible has happened and panics. MEMORY LOST results. Register c is not used by any of these programs.

Register d contains all the system and user flags, 00-55. An HP-41 register is 7 bytes of 8 bits each; each flag is one bit. A number of things are controlled by different flags: the display setting (FIX, SCI, or ENG, and the number of places displayed), the trigonometric mode (DEG, RAD or GRAD), many print and display functions, and other things. Register d is used in these programs mostly in subroutine AX ("Append X"), which recalls d, changes display setting to FIX 0, clears flag 29 (thus getting rid of the decimal point), appends a number to Alpha using this display configuration, and then returns the display to its former status by re-storing to register d the value previously recalled.

The last Status Register is register e, which contains the shifted key assignments. Like register b, e is not used by any of the programs in this package.

REFERENCES

There are several reference books available on the HP-41 in general and Synthetic Programming in particular. The two best are:

EXTEND YOUR HP-41 by W. Meir-Jedrzejewicz, and
HP-41 SYNTHETIC PROGRAMMING MADE EASY by Keith
Jarett

Both are available from EduCALC Mail Store,
27953 Cabot Road
Laguna Niguel, CA 92677

GEOMETRY

PROGRAMS

BRIDGE DECK VERTICAL CURVE ELEVATIONS

xeq VC (SIZE 010) to find elevations along profile grade
(# OFSTS bypassed or answered zero)
or
(SIZE 017 + number of offsets) to find elevations across
a bridge deck or street
or
(SIZE 023 + number of offsets) to find elevations across
a bridge deck or street using separate crown drops for
each point

SOFTKEYS (USER mode ON)

- c Input new crown information; same as C for profile
grade calculations (# OFSTS = 0)
- d Input new offsets; same as C for profile grade calc's
- e Change base (P.I.) elevation (e.g. for beam seat elev.)
- C New INTERV or STA prompt; access VC from STA
- E Access STA from VC once curve parameters are input

USER FLAGS

Flag 00	clear	Full output
	set	Output of M and HI/LO point skipped; STA display skipped if using INTERV
Flag 02	clear	Warning BEEP disabled
	set	Warning BEEP enabled (used if STA finds OFF CURVE error)
Flag 03	clear	VC prompts SKEW in decimal degrees
	set	VC prompts SKEW in D.MS (for # OFSTS not zero)

INPUT SUMMARY

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
# OFSTS	Number of offsets from center- line, for figuring SIZE needed	Zero(P.G.)
STA PI	Station of the PVI, in feet	No default
EL PI	Elevation of the PVI, in feet	No default
VC L'	Vertical curve length, in feet	No default

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
SL1%	Slope or grade of the back (left) tangent, in percent	No default
SL2% (only if VC L' is not 0)	Slope or grade of the forward (right) tangent, in percent	SL1%

Note: prompts in parenthesis do not appear if the # OFSTS (number of offsets) prompt was answered zero or bypassed.

(SKEW)	Skew of the line across a bridge deck (a "square" bridge has 0° skew)	No default
(CRN1'/')	Slope of the crown from the centerline to a "break" point, in feet per foot. Bypass to enter crown drops (CRN DROP) directly for each offset.	Zero
(TO:)	Distance of the break point from centerline, in feet	12
(CRN2'/')	Slope of the crown from the break point out, in feet per foot	0.020
(OFFSET#1', etc.)	Offset of a desired line from CL in feet (e.g. a beam line)	Zero
(CRN DROP 1, etc.)	Crown drop to the desired line from profile grade; a rise is negative. Only used if CRN1 was input as zero	No default
INTERV	Station interval at which to calculate elevations, in feet	Previous INTERV
STA or POB STA	Station at which to calculate or begin calculating elevations (in feet: 28+50.00 is input 2850)	No default
EL CH (from softkey e)	Change in baseline elevation; up is positive, down is negative	No default

OUTPUT SUMMARY (Items in parenthesis are displayed only if flag 00 is clear)

(M)	The middle ordinate, in feet
(HI/LO @...)	The station of the high or low point, if one exists
(EL)	The elevation of the high or low point

L1, L2, etc.	The elevation at offset #1, etc., on the left side of centerline
CL EL	The elevation at centerline
R1, R2, etc.	The elevation at offset #1 on the right side of centerline

PROGRAM FLAGS

Flag 05	clear set	On or tangent to a vertical curve VC L' = 0 (tangent--no vertical curve)
Flag 06	clear set	High or low point exists; two stations for a given elevation No high or low point; one station for a given elevation
Flag 07	clear set	Left-ahead skew (positive value) Right-ahead skew (negative value)
Flag 08	clear set	Profile grade only Full bridge deck with offsets
Flag 09	clear set	VC STA
Flag 10	clear set	Have crown slope information Have individual crown drop points
Flag 11	clear set	Normal use Portions used as subroutines
Flag 18	clear set	Skewed--figure both sides of crown Crowned with 0° skew, so only figure one side of crown
Flag 19	clear set	Station is on a vertical curve Station is on a tangent

STORAGE REGISTER USE

<u>Register</u>	<u>Value</u>
00	Sta. PI
01	SL1%
02	SL2%
03	Sta. BVC
04	r/200 ($r=g_2-g_1/L$)
05	Sta. EVC

<u>Register</u>	<u>Value</u>
06	Interval
07	El. BVC
08	Input station
09	El. PI

Note: the following registers are not used if only profile grade calculations are being done (i.e. # OFSTS was answered zero).

10	Crown drop
11	tan (skew)
12	Crown break point
13	Crown2 - Crown1
14	Crown1
15	Counter
16	Offset at centerline (zero)
17	Offset #1
18	Offset #2
19	Offset #3
20	Offset #4
21	Offset #5
22	Offset #6
23	Offset #7/Crown drop #1
24	Offset #8/Crown drop #2
	etc.

VERTICAL CURVE SOLUTIONS for ELEVATION

Programs VC and STA solve parabolic, symmetrical (equal-tangent) vertical curves. Both programs work on a single vertical curve and the tangent on either end (assumed to extend indefinitely). They both require input of the PI (point of vertical intersection) station and elevation, the back and forward grades, and the vertical curve length. Stations are expressed in feet, without the "+"; grades are in percent. See the examples for clarification.

VC calculates either the elevation of a given station on the profile grade; or, elevations at specified offsets on a crowned section along a line at a given skew to centerline. It prompts first for an interval, INTERV, on which to figure elevations (typically 50 or 100 feet). If this prompt is answered, the program prompts POB STA (beginning station) and then figures elevations at every interval, displaying the station and its elevation. If the INTERV prompt is bypassed, VC prompts repeatedly for stations (STA) at which to find the elevation. Softkey C accesses these prompts.

Typical applications of VC might be elevations of beam lines on a bridge deck, or gutter and curb lines along a street. For multiple elevations along a line at a given station, in addition to the basic vertical curve data mentioned above, VC requires the number of offsets to be calculated, the skew, crown configuration (see page 3), and the offsets desired. It then calculates the elevations at all the offsets on the left side of centerline, the centerline elevation, and then the elevations on the right side. For a crowned deck with a zero-degree skew, only one side is calculated since the crown is assumed symmetrical.

INPUT DETAILS

"Stations" are input and output in feet. For example, if the PI is at station 28+50.00, answer the STA PI prompt with 2850. Length of vertical curve (VC L') is also in feet.

Tangent grades are input in percent, as mentioned, while crown cross-slopes are input in feet per foot. Don't waste a lot of time looking for a good reason for this discrepancy; there isn't one.

Once the required data is input, VC will either prompt for a beginning station and an interval, or will prompt repeatedly for the station (STA) at which to calculate the elevation or elevations. If the interval prompt (INTERV) is answered with a non-zero number

(let's say, 50) the program will ask for the beginning station (POB STA), and then will figure elevations at fifty-foot intervals from there, displaying first the station and then the elevation or elevations. It will continue doing this until it is stopped; if printing, press R/S to stop it. Then, in USER mode, press softkey C. (If the X key has something assigned to it, key GTO (alpha) C (alpha) instead.) This gets you back to the INTERV prompt.

If the INTERV prompt is bypassed (or answered zero) the programs repeatedly prompt for the station (STA) at which to find an elevation.

ELEVATIONS on TANGENT

If a point is off the vertical curve, VC will append a warning, (T), that the point is on the tangent rather than on the curve. The elevation is figured correctly; the warning is simply a reminder to verify that you're still working on the tangent and not on a preceding or following curve. This is especially important when using VC on sharply skewed bridges; the stations of the offset points are never explicitly displayed, and corners (or ends of long wingwalls) may encroach on a closely preceding or following curve.

If the area you're interested in is entirely on a tangent, input a vertical curve length of zero. The program will only prompt for one grade and the (T) warning will not be displayed, saving annoying display-scrolling time.

ADDITIONAL PARAMETERS CALCULATED

After input of the basic vertical curve parameters (PI station and elevation, grades, and length) the program calculates M (the middle ordinate), and, if there is one (if the grades are of opposite sign), the high or low point station and elevation. If this information is not wanted and you wish to speed execution, set flag 00 and it will not be displayed.

SKEW

In keeping with the standard flag usage of this package, program VC prompts for skew in either decimal degrees (dd) or degrees-minutes-seconds (DMS) depending on whether flag 03 is clear or set. A right-hand-forward skew is defined as negative; a LHF skew is defined as positive. Skews are measured from a line normal to CL Roadway; see the examples.

CROWNS

Crowns are defined by the slope from the centerline to a break point at a user-defined offset, and another slope from the break point on out. Unlike the tangent slopes, which are input in percent, the crown slopes are input in feet per foot per standard practice; see the examples for clarification. Crowns are assumed to be symmetrical about centerline. After the crown information is input, VC asks for offsets (any number can be stored) from the centerline (assumed measured normal to the centerline); these are usually input from the centerline on out, but they need not be in any order.

CROWN "DROP"

If a section is not easily described by a sloping crown, the program offers the option of directly inputting the drop from the profile. Answering the CRN1 (crown one, the slope from centerline to the break point) prompt with zero, or bypassing the prompt, directs the program to prompt for a crown drop in feet after each offset prompt. If the point is above the profile grade (e.g. a curb) the crown drop will be a negative number. When the CRN DROP feature is used, only six offsets can be stored because CRN DROP 1 is stored in the seventh offset's register. Because of this overlap, the size checker at the beginning of the program doesn't work. The minimum SIZE required is (23 + number of offsets), and the user must verify before starting that there are enough storage registers available. Since the maximum number of offsets is six with this feature, a SIZE of at least 029 will always work.

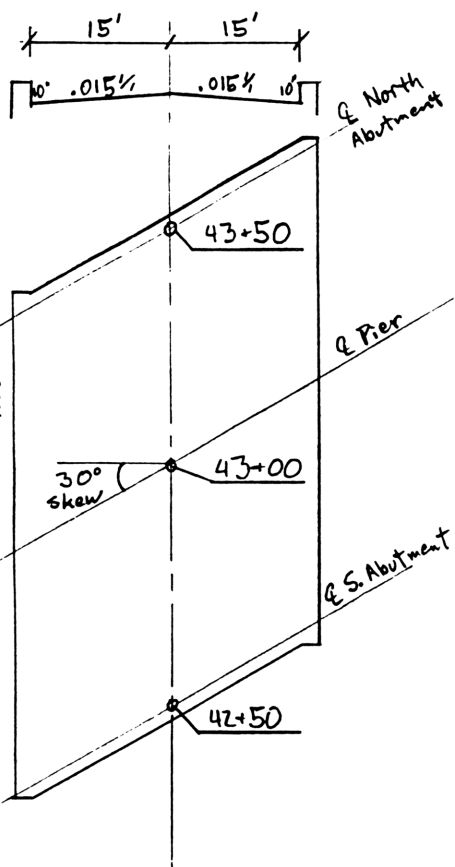
OUTPUT

If the calculator is shut off during a run, and then turned on again and restarted by pressing R/S or a softkey, flags 18 and 19 are cleared. This may affect the (T) warning and the displaying of only one side of a deck for 0° skew, but the numbers are correct. See the "Program Flags" section.

ELEVATION CHANGE

If the user needs to subtract a distance from each elevation, the program can do it directly. Softkey e brings up the prompt EL CH (elevation change). It adds this number to the P.I. elevation (and BVC and EVC elevations) stored at the beginning of the run. Thus, if VC was being used to calculate bridge deck elevations and the user now wants to calculate beam seat elevations (e.g. for a superstructure height of 5.90 feet), press softkey e, answer the prompt with -5.90 (5.90, chs), and press R/S. This feature works with both VC and STA.

VC - Bridge Deck Elevations



XEQ -VC-
OFSTS 2.0000 RUN
SIZE<19 SIZE 019
XEQ -VC-
OFSTS 2.0000 RUN
11:45:34 AM 12/22/88
TITLE:
30d SKEW BRIDGE RUN

STA PI 4,300.0000 RUN
EL PI 879.0000 RUN
VC L' 110.0000 RUN
SL1% 2.5000 RUN
SL2% -3.0000 RUN
M=-0.7563
HI/LO @ 4,295.0000
EL=878.2500

SKEW: L+,R- (dd)
-30.0000 RUN
CRN1' /, .0150 RUN
TO: 20.0000 RUN
CRN2' /, RUN
OFFSET#1' 7.5000 RUN
OFFSET#2' 15.0000 RUN

INTERV 50.0000 RUN
POB STA 4,250.0000 RUN

STA 4,250.0000:
L2 EL=877.3085(T)
L1 EL=877.5291
CL EL=877.7438
R1 EL=877.7240
R2 EL=877.6949

STA 4,300.0000:
L2 EL=878.0217
L1 EL=878.1374
CL EL=878.2438
R1 EL=878.1157
R2 EL=877.9783

STA 4,350.0000:
L2 EL=877.4882
L1 EL=877.4956
CL EL=877.4938
R1 EL=877.2575
R2 EL=877.0152(T)

STA 4,400.0000:
L2 EL=876.0348(T)

$$BVC Sta. = 43 - \frac{1.1}{2} = 42+45.00$$

$$elev. = 879.00 - (.55 \times 2.5) = 877.625$$

$$Elev_x = ax^2 + bx + c ;$$

$$a = \frac{g_2 - g_1}{2L} = \frac{-3 - 2.5}{(1-1) \times 2} = -2.50$$

$$b = g_1 = 2.50$$

$$c = 877.625$$

Find elevations @ E and Offsets of 7.5' and 15' along E South Abutment ; * \Rightarrow off V.C.

OFFSET	$\times \tan(30^\circ)$	STATION	ELEV.	CROWN	ELEV.
15' Left	-4.6603	42+41.3397*	877.5335	.225'	877.3045
7.5' Left	-4.3301	42+45.6699	877.6415	.1125'	877.5291
E	0	42+50	877.7438	0	877.7438
7.5' Right	4.3301	42+54.3301	877.6365	.1125'	877.7240
15' Right	4.6603	42+58.6603	877.9199	.225'	877.6949

Elevations on Bridge Deck with Curbs

Use same vertical curve
and bridge deck; find
elevations with CRN DROP.

Curb height = 10"



Sta. PI = 43+00.00

El. PI = 879.00

V.C. = 110.0'

Crown drops

$$7.5' \times .015' = .1125'$$

$$15' \times .015' = .225'$$

$$15' \times .015' - (10' / 12) = -.6083'$$

OFSTS
3.0000 RUN
10:29:01 AM 12/27/88

TITLE:
BRIDGE DECK RUN

STA PI

4.300.0000 RUN

EL PI
879.0000 RUN

VC L'
110.0000 RUN

SL1%
2.5000 RUN

SL2%
-3.0000 RUN

M=-0.7563'
HI/LO @ 4,295.0000
EL=878.2500

SKEW: L+,R- (dd)
-30.0000 RUN

CRN1' /,
RUN

OFFSET#1'
7.5000 RUN

CRN DROP 1
.1125 RUN

OFFSET#2'
15.0000 RUN

CRN DROP 2
.2250 RUN

OFFSET#3'
15.0000 RUN

CRN DROP 3
-.6083 RUN

INTERV

50.0000 RUN

POB STA
4,250.0000 RUN

STA 4,250.0000:
L3 EL=878.1418(T)
L2 EL=877.3085(T)
L1 EL=877.5291
CL EL=877.7438
R1 EL=877.7240
R2 EL=877.6949
R3 EL=878.5282

STA 4,300.0000:
L3 EL=878.8550
L2 EL=878.0217
L1 EL=878.1374
CL EL=878.2438
R1 EL=878.1157
R2 EL=877.9783
R3 EL=878.8116

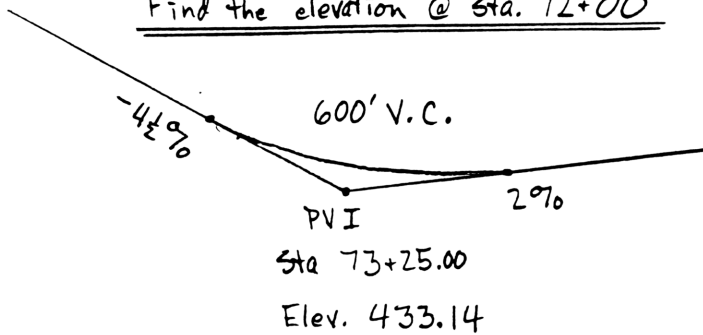
STA 4,350.0000:
L3 EL=878.3215
L2 EL=877.4882
L1 EL=877.4956
CL EL=877.4938
R1 EL=877.2575
R2 EL=877.0152(T)
R3 EL=877.8485(T)

STA 4,400.0000:
L3 EL=876.8681(T)
L2 EL=876.0348(T)
L1 EL=876.0174(T)
CL EL=876.0000(T)
R1 EL=875.7576(T)
R2 EL=875.5152(T)
R3 EL=876.3485(T)

STA 4,450.0000:

Profile Grade Calculation

Find the elevation @ Sta. 72+00



$$BVC \text{ Sta.} = 70+25.00$$

$$\begin{aligned} \text{Elev.} &= \text{El. PVI} - \frac{L}{2} \cdot G_1 \\ &= 433.14 - (3) \cdot (-4.5) = 446.64 \end{aligned}$$

$$EVC \text{ Sta.} = 76+25.00$$

$$\text{Elev.} = \text{El. PVI} + \frac{L}{2} \cdot G_2 = 439.14$$

$$\text{Elev.} = ax^2 + bx + c :$$

$$a = \frac{g_2 - g_1}{2L} = \frac{2 - (-4\frac{1}{2})}{2 \times 6} = .5417$$

$$b = g_1 = -4.50$$

$$c = \text{elev. BVC} = 446.64$$

$$\text{Elev. @ Sta. 72+00 : } x = 72 - 70.25 = 1.75$$

$$\begin{aligned} \text{Elev.} &= .5417 (1.75)^2 - 4.50 \cdot 1.75 + 446.64 \\ &= 440.4239 \end{aligned}$$

$$\begin{aligned} M &= \frac{g_2 - g_1}{2L} \cdot \left(\frac{L}{2}\right)^2 = \frac{(g_2 - g_1)L}{8} \\ &= \frac{(2 - (-4.5)) \cdot 6}{8} = 4.475' \end{aligned}$$

$$\text{Low Pt. Sta. : } 2ax + b = 0$$

$$\frac{2(g_2 - g_1)}{2L} x + g_1 = 0$$

$$\textcircled{B10} \quad x = \frac{-g_1 L}{g_2 - g_1} = \frac{-(-4.5) \times 6}{2 - (-4.5)} = 4.1536 \text{ Sta. ; } (70+25) + (4+15.36) = 74+40.36$$

# OF STS	XEQ "VC"	
11:50:10 AM	12/22/88	RUN
TITLE:		
EXAMPLE		RUN

STA PI	7,325.0000	RUN
EL PI	433.1400	RUN
VC L'	600.0000	RUN
SL1%	-4.5000	RUN
SL2%	2.0000	RUN
M=4.8750'		
HI/LO @	7,440.3846	
EL=437.2938		

INTERV	50.0000	RUN
POB STA	7,150.0000	RUN

STA 7,150.0000:	EL=441.8614	

STA 7,200.0000:	EL=440.4239	

STA 7,250.0000:	EL=439.2572	

STA 7,300.0000:	EL=438.3614	

STA 7,350.0000:	EL=437.7364	

	SORT	
	XEQ C	

INTERV		RUN
STA	7,200.0000	RUN
	EL=440.4239	

STA	7,300.0000	RUN
	EL=438.3614	

Bypass
for only
P.G. calc.
need only
SIZE 010

0 cps- USER not on

VERTICAL CURVE STATIONS

xeq STA (SIZE 010) to find the station or stations at which an elevation occurs

SOFTKEYS (USER mode ON)

- c Same as C for STA
- d Same as C for STA
- e Change base (P.I.) elevation (e.g. for beam seat elev.)
- C Jump to program VC; new INTERV or STA prompt
- E Access STA from VC once curve parameters are input

USER FLAGS

- | | | |
|---------|-------|--|
| Flag 00 | clear | Full output |
| | set | Output of M and HI/LO point skipped |
| Flag 02 | clear | Warning BEEP disabled |
| | set | Warning BEEP enabled (used if STA finds OFF CURVE error) |

INPUT SUMMARY

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
STA PI	Station of the PVI, in feet	No default
EL PI	Elevation of the PVI, in feet	No default
VC L'	Vertical curve length, in feet	No default
SL1%	Slope or grade of the back (left) tangent, in percent	No default
SL2% (only if VC L' is not 0)	Slope or grade of the forward (right) tangent, in percent	SL1%
EL	Elevation at which to calculate the station or stations	No default
EL CH (from softkey e)	Change in baseline elevation; up is positive, down is negative	No default

OUTPUT SUMMARY (Items in parenthesis are displayed only if flag 00 is clear)

(M)	The middle ordinate, in feet
(HI/LO @...)	The station of the high or low point, if one exists
(EL)	The elevation of the high or low point
STA	The station at which the given elevation occurs, if there is only one
STA1	The first station of two at which the given elevation occurs
STA2	The second station; will be the same as the first if at a HI/LO point
OFF CURVE	A warning that the elevation input does not exist on the curve (i.e. is higher than the high point or lower than the low point)

PROGRAM FLAGS

Flag 05	clear set	On or tangent to a vertical curve VC L' = 0 (tangent--no vertical curve)
Flag 06	clear set	High or low point exists; two stations for a given elevation No high or low point; one station for a given elevation
Flag 07	clear set	Elevation is on curve-solution possible Elevation is OFF CURVE
Flag 08	clear set	Profile grade calcs only # OFSTS not zero
Flag 09	clear set	VC STA
Flag 10	clear set	BVC lower than EVC (STA only) EVC lower than BVC (STA only)
Flag 11	clear set	Normal operation Figuring HI/LO elevation

STORAGE REGISTER USE

<u>Register</u>	<u>Value</u>
00	Sta. PI
01	SL1%
02	SL2%
03	Sta. BVC
04	$r/200$ $(r=g_2-g_1/L)$
05	Sta. EVC
06	El. EVC
07	El. BVC
08	Input elevation
09	El. PI

VERTICAL CURVE SOLUTIONS for STATION

Programs VC and STA solve parabolic, symmetrical (equal-tangent) vertical curves. Both programs work on a single vertical curve and the tangent on either end (assumed to extend indefinitely). They both require input of the PI (point of vertical intersection) station and elevation, the back and forward grades, and the vertical curve length. Stations are expressed in feet, without the "+"; grades are in percent. See the examples for clarification.

STA finds the station or stations at which a given elevation occurs. It prompts for an elevation, calculates the station or stations at which it occurs, then returns for another elevation. STA works only along profile grade. If the grades are of opposite sign, STA finds two stations for a given elevation; if not, only one station is found. If no solution is possible, OFF CURVE is displayed. The program correctly solves for points either within the curve or on either tangent. However, unlike VC, no warning (T) is given when a point is on tangent. STA also lacks the INTERV feature that VC has for automatic running.

INPUT DETAILS

As mentioned, "stations" are input and output in feet. For example, if the PI is at station 28+50.00, answer the STA PI prompt with 2850. Length of vertical curve (VC L') is also in feet. Tangent grades are input in percent.

STATIONS on TANGENT

If a point is off the vertical curve, VC will append a warning, (T), that the point is on the tangent rather than on the curve. Program STA does not have this warning feature.

If the area you're interested in is entirely on a tangent, input a vertical curve length of zero. The programs will only prompt for one grade.

ADDITIONAL PARAMETERS CALCULATED

After input of the basic vertical curve parameters (PI station and elevation, grades, and length) the program calculates M (the middle ordinate), and, if there is one (if the grades are of opposite sign), the high or low point station and elevation. If this information is not wanted and you wish to speed execution, set flag 00 and it will not be displayed.

ELEVATION CHANGE

If the user needs to subtract a distance from each elevation, the program can do it directly. Softkey e brings up the prompt EL CH (elevation change). It adds this number to the P.I. elevation (and BVC and EVC elevations) stored at the beginning of the run. Thus, if VC was being used to calculate bridge deck elevations and the user now wants to calculate beam seat elevations (e.g. for a superstructure height of 5.90 feet), press softkey e, answer the prompt with -5.90 (5.90, chs), and press R/S. This feature works with both VC and STA.

"STA" - find stations for an elevation

Find stations at which elev 440.0 occurs.

$$440.00 = .5417(x)^2 - 4.50(x) + 446.64$$

$$.5417(x)^2 - 4.5(x) + 6.64 = 0$$

$$\text{Root 1} = 6.3890 ; \text{Root 2} = 1.9187$$

$$\text{One station is } (70+25) + (1+91.87)$$

$$= \underline{72+16.87}$$

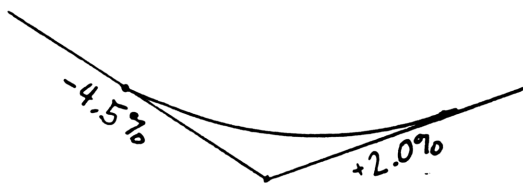
$$\text{Check other station: } (70+25) + (6+38.90)$$

$$= 76+63.90 > \text{EVC sta.}$$

∴ Second point is off curve, on tang.

$$(440.00 - 433.14) / 2\% = 3.430 \text{ stations}$$

$$(73+25) + (3+43.00) = \underline{76+68.00}$$



PI Sta. = 73+25.00
Elev. = 433.14
600' V.C.

From VC,
offset E is
same as
req STA
except uses
existing
vertical
curve data

STA		XEQ E

EL	437.0000	RUN
OFF CURVE		

EL	438.0000	RUN
STA1=7,326.2063		
STA2=7,554.5629		

EL	440.0000	RUN
STA1=7,216.8678		
STA2=7,668.0000		

EL		
	Or req STA	
	XEQ "STA"	
	11:55:29 AM 12/22/88	
TITLE:		
EXAMPLE		RUN

STA PI		
	7,325.0000	RUN
EL PI		
	433.1400	RUN
VC L'		
	600.0000	RUN
SL1%		
	-4.5000	RUN
SL2%		
	2.0000	RUN
M=4.8750'		
HI/LO @ 7,440.3846		
EL=437.2938		

EL		
	438.0000	RUN
STA1=7,326.2063		
STA2=7,554.5629		

EL		
	437.0000	RUN
OFF CURVE		

EL		
	440.0000	RUN
STA1=7,216.8678		
STA2=7,668.0000		

EL		

	X	Y	Z	T	L		X	Y	Z	T	L
01*LBL "VC"						54 STO 01	g ₁				
02 SF 11	Lbl*	X	Y	Z		55 "SL2%"					
03 RDN	x	Y	Z	Lbl*		56 FC? 05					
04 GTO IND T						57 PROMPT					
						58 STO 02	g ₂				
05*LBL "VC"						59 RCL 01					
06 SF 08						60 *	g ₁ *g ₂				
07 CF 09						61 X>0?					
08 CF 11						62 SF 06					
09 CLX						63 RCL 02	g ₂				
10 "OFSTS"						64 LASTX	g ₁				
11 PROMPT						65 -	g ₂ -g ₁				
12 X=0?	only PG?					66 4					
13 GTO 02						67 /					
14 16						68 RCL 06	l/2				
15 +	16*off					69 X#0?					
16 SF 25						70 /	r/2				
17 STO 15						71 STO 04					
18 GTO 00						72 RCL 09	EI PI				
						73 RCL 01	g ₁				
19*LBL "STA"						74 RCL 06	l/2				
20 SF 09						75 *					
21 CF 11						76 -					
						77 STO 07	EI BVC				
22*LBL 02						78 FS?C 11					
23 CF 08						79 RTN					
24 9						80 FC? 05					
						81 FS? 00					
25*LBL 00						82 GTO 00					
26 XROM "S"						83 RCL 02					
27 E3	1000					84 RCL 01					
28 FS? 08						85 -	g ₂ -g ₁				
29 ST/ 15	counter					86 RCL 06	l/2				
						87 *					
30*LBL 07						88 4					
31 CF 05						89 /					
32 CF 06						90 "M="					
33 "STA PI"						91 ARCL X					
34 PROMPT						92 "T"					
35 STO 00	Sta PI					93 AVIEW					
36 STO 03						94 FS? 06					
37 STO 05						95 GTO 00					
38 "EL PI"						96 RCL 03	Sta BVC				
39 PROMPT						97 RCL 01	g ₁				
40 STO 09	EI PI					98 RCL 04	g ₂ -g ₁ /2L				
41 "VC L"						99 50	50	g ₂ -g ₁ /2L	g ₁	Sta BVC	
42 PROMPT						100 /					
43 X=0?						101 /	-(cosθ)				
44 SF 05	tang.					102 -	W/L0 Sta Sta BVC				
45 2						103 "HI/LO @ "					
46 /						104 ARCL X					
47 ST- 03	Sta BVC					105 AVIEW					
48 ST+ 05	Sta EVC					106 SF 11					
49 E						107 X<>Y	Sta BVC Sta				
50 %						108 XEQ 04	EI.				
51 STO 06	l/2					109 "EL="					
52 "SL1%"						110 ARCL X					
53 PROMPT						111 AVIEW					

$$M = \frac{(g_2 - g_1)L}{8}$$

l/2 in Stations
l/200 in feet

	X	Y	Z	T	L		X	Y	Z	T	L
112*LBL 00						166 XROM "AX"					
113 FS? 09						167 "T"					
114 GTO E						168 PROMPT	offset 1,2, etc.				
						169 STO IND 15					
115*LBL c						170 FC? 10					
116 CF 18						171 GTO 06					
117 FC? 08						172 RCL 15	counter				
118 GTO C						173 6					
119 XROM "L"						174 +					
120 CLX						175 RCL 08	1,2,3, etc.				
121 STO 13						176 "CRN DROP "					
						177 XROM "AX"					
122*LBL 09						178 PROMPT	Crn Drop 1, 2, 3, etc. 2, 3, 4, etc.				
123 CF 07						179 STO IND Z					
124 CF 10											
125 "SKEW: L+,R-"						180*LBL 06					
126 XROM "Pd"						181 ISG 15	counter				
127 TAN						182 GTO 06					
128 STO 11	tan(skew)					183 FS?C 11					
129 X=0?						184 RTN					
130 SF 18											
131 X<0?						185*LBL C					
132 SF 07	RNF					186 50					
133 CLX						187 STO 06	interval				
134 "CRN1"/, -						188 XROM "L*"					
135 PROMPT											
136 X=0?						189*LBL 21					
137 SF 10						190 CF 22					
138 X=0?						191 RCL 06	interv				
139 GTO d						192 "INTERV"					
140 STO 14	crn 1					193 X=0?					
141 12						194 PROMPT					
142 "TO: -"						195 FC? 22					
143 PROMPT						196 CLX					
144 STO 12	Break					197 STO 06					
145 .02						198 CLA					
146 "CRN2"/, -						199 X=0?					
147 PROMPT						200 "POB "					
148 RCL 14						201 "TSTA"					
149 -						202 PROMPT					
150 STO 13	Δ crn					203 STO 08	G sta.				
						204 RDN	interv				
151*LBL d						205 X=0?					
152 CLX						206 XROM "LL"					
153 STO 16	offset RCL = 0										
154 RCL 15	counter					207*LBL 32					
155 ABS						208 FC? 08					
156 FRC						209 GTO 10					
157 17						210 RCL 11	tan(skew)				
158 +	17.0--					211 ABS					
159 STO 15						212 FS? 07					
						213 CHS					
160*LBL 06						214 STO 11	tan(skew)				
161 "OFFSET#"						215 E3	1000				
162 RCL 15	counter					216 RCL 15	counter				
163 16						217 ABS					
164 -						218 FRC	.0--				
165 STO 08	1,2,3, etc.					219 *	--				

	X	Y	Z	T	L		X	Y	Z	T	L
220 LASTX	FRC					274 X<>Y	Sta				
221 +						275 RCL 01	g ₁	Sta	BVC		
222 CHS	---a					276 GTO 02					
223 STO 15	counter										
224*LBL 10						277*LBL 03	Sta				
225 *STA *						278 RCL 02	g ₂	Sta	EVC		
226 ARCL 08	Sta.					279*LBL 02					
227 "t:"						280 RCL 09	EI.PI				
228 RCL 06	Interv					281 RCL 00	Sta.PI	EI.PI	g%	Sta	
229 X#0?						282 R↑	Sta	Sta.PI			
230 FS? 00						283 -	ΔSta	EI.PI	g%	g%	
231 FS? 53						284 R↑	g	ΔSta			
232 AVIEW						285 *	ΔEI.				
						286 E2	100				
233*LBL 22						287 /	ΔEI.	EVI			
234 RCL 08	Sta.					288 -	Elev.				
235 FC? 08						289 SF 19					
236 GTO 23						290 GTO 11					
237 RCL 15	counter										
238 ABS						291*LBL 04	Sta.DVC	Sta			
239 6						292 -	ΔSta				
240 +						293 E	1				
241 FS? 10						294 %	Dist, ft.				
242 RCL IND X	Crd Drop					295 ENTER↑					
243 FS? 10						296 ENTER↑					
244 GTO 14						297 RCL 04	r/2	"X"	"X"		
245 RCL IND 15	Offset					298 *	rX/2				
246 RCL 12	Break					299 RCL 01	g ₁	rX/z	X		
247 -						300 +					
248 X<0?						301 *	rX/zgX				
249 .	Dist z					302 RCL 07	EI.DVC				
250 RCL 13	Crown z					303 +	Elev				
251 *	Drop z										
252 RCL IND 15	Offset					304*LBL 11					
253 RCL 14	Crown ₁					305 FS?C 11					
254 *	Drop ₁	Drop z				306 RTN					
255 +	Drop					307 CLA					
						308 FC? 08					
256*LBL 14						309 GTO 11					
257 STO 10	Drop					310 RCL 10	Crown Drop	Elev			
258 RCL 08	Sta					311 -	Elev.				
259 RCL 11	tan(Sta)					312 * R-					
260 RCL IND 15	offset					313 RCL 15	counter				
261 X#0?						314 X<0?					
262 STO 10	(Drop)					315 * L-					
263 *	ΔSta	Sta	Drop			316 ABS					
264 +	NewSta	Drop				317 INT	counter				
						318 16					
265*LBL 23	Sta					319 -	1,2,3, etc	elev.			
266 CF 19						320 X#0?					
267 RCL 05	Sta EVC					321 XROM "AX"					
268 X<>Y	Sta	EVC				322 X#0?					
269 X>Y?						323 "CL"					
270 GTO 03						324 X<>Y	elev.				
271 RCL 03	Sta.DVC	Sta									
272 X<=Y?											
273 GTO 04											

SF19 ⇒ "on tangent"

04 ⇒ Curve

(B19)

325*LBL 11
 326 "F EL=" *clav. 1,2,3, etc.*
 327 ARCL X
 328 FC? 05
 329 FC? 19
 330 FS? 53
 331 "F(T)"
 332 XEQ "PV"
 333 FC? 08
 334 GTO 11
 335 X<>Y *1,2,3, etc.*
 336 X=0?
 337 GTO 00
 338 FS? 18
 339 GTO 11
 340 SIGN
 341 CHS
 342 ST* 11
 343 ST* 15
 344*LBL 00
 345 ISG 15
 346 GTO 22
 347*LBL 11
 348 XROM "L"
 349 RCL 06
 350 X=0?
 351 GTO 21
 352 RCL 08
 353 +
 354 STO 08
 355 GTO 32
 356*LBL e
 357 "EL CH"
 358 PROMPT
 359 ST+ 07
 360 ST+ 09
 361 FC? 09
 362 GTO C
 363 ST+ 06
 364*LBL E
 365 RCL 02
 366 E
 367 %
 368 RCL 05
 369 RCL 00
 370 -
 371 *
 372 RCL 09
 373 +
 374 STO 06

Interv
Sta
New Sta
Δ el.
el. BYC
el. PI
el. EVC
g₂ %
1
g₂
Sta EVC
Sta PI
L/2
Δ EI
el. PI
el. EVC

1,2,3, etc.
tan (skn)
countar
g₂

FC? 05
and
FS? 19

375*LBL 15
 376 XROM "L"
 377 CF 07
 378 "EL"
 379 PROMPT
 380 STO 08
 381 "STA1=" *Elev.*
 382 FC? 05
 383 GTO 08
 384 RCL 01
 385*LBL 13
 386 E
 387 %
 388 RCL 08
 389 RCL 09
 390 -
 391 X<>Y
 392 /
 393 RCL 00
 394*LBL 16
 395 +
 396 ARCL X
 397 AVIEW
 398 GTO 15
 399*LBL 08
 400 FC? 06
 401 GTO 08
 402 CF 10
 403 RCL 06
 404 RCL 07
 405 X>Y?
 406 SF 10
 407 X>Y?
 408 X<>Y
 409 RCL 08
 410 X>Y?
 411 GTO 05
 412 RCL 01
 413 FS? 10
 414 RCL 02
 415 GTO 13
 416*LBL 05
 417 RCL 2
 418 X>Y?
 419 GTO 08
 420 RCL 02
 421 FS? 10
 422 RCL 01
 423 GTO 13

g₁
g
I. Input
EI. PI
Δ EI
g
Δ Dist
Sta PI
Δ Sta
Sta
EI. EVC
EI. BYC
lessor
Input EI
g
Input EI
greater
g

g₁
g
Δ EI
Δ Sta
Sta
lessor
greater
lessor
greater
lessor
greater
lessor
greater

g₁
g
Δ EI
Δ Sta
Sta
lessor
greater
lessor
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lessor
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lessor
greater

g₁
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Δ EI
Δ Sta
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g₁
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Δ EI
Δ Sta
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g₁
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g₁
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lessor
greater
lessor
greater

	X	Y	Z	T	L		X	Y	Z	T	L
424*LBL 08						472*LBL 08	L	Dist 2			
425 RCL 07	EI.BVC					473 RCL 08	Imp EI				
426 RCL 08	Imp EI					474 RCL 09	EI.PI				
427 -	ΔEI					475 -	ΔEI				
428 RCL 01	g_1					476 RCL 02	g_2				
429 RCL 04	r/z	g_1	ΔEI			477 /					
430 SF 25						478 E2					
431 XROM "Q"						479 *	Dist				
432 FC?C 25						480 RCL 00	Sta PI				
433 GTO 01											
434 E2	100					481*LBL 05					
435 *	Dist 1					482 +	Sta				
436 X<>Y						483 "STA2="					
437 LASTX						484 ARCL X					
438 *	Dist 2	Dist 1				485 AVIEW					
439 FC? 06						486 GTO 15					
440 GTO 08											
441 X<=0?						487*LBL 01					
442 X<>Y	Dist					488 SF 07					
443 RCL 03	Sta BVC					489 "OFF"					
444 GTO 16						490 BEEP					
						491 AVIEW					
445*LBL 08						492 GTO 15					
446 X>Y?						493 END					
447 X<>Y	Min	Max									
448 X>0?											
449 GTO 05											
450 CLX											
451 RCL 08	Imp EI	Max									
452 RCL 07	EI.BVC										
453 -	ΔEI										
454 RCL 01	g_1										
455 /											
456 E2	100										
457 *	Dist										
458*LBL 05											
459 RCL 03	Sta BVC										
460 +	Sta										
461 ARCL X											
462 AVIEW											
463 X<>Y	Dist 2										
464 RCL 05	Sta BVC										
465 LASTX	Sta BVC										
466 -	VC L										
467 X<=Y?											
468 GTO 08											
469 CLX											
470 LASTX	Sta BVC	Dist									
471 GTO 05											

FS:06 $\Rightarrow g_1$
and g_2 are
same sign-
only one
station

STOPPING SIGHT DISTANCE

xeq SSD (SIZE 004)

SOFTKEYS (USER mode ON)

- A Input vertical curve length, solve for SSD
- B Input design speed (mph), solve for VC L
- D Input stopping sight distance, solve for VC L

USER FLAGS

Flag 00	clear	Full output
	set	No title block printed

INPUT SUMMARY

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
G1%	Foreslope in percent (up to the right is positive)	No default
G2%	Backslope in percent	No default
L MPH SSD (A) (B) (D)	A "menu" to the softkeys, as shown	No default

OUTPUT SUMMARY (Items in parenthesis are displayed only if flag 00 is clear)

SSD	The available stopping sight distance in feet
VC L	The required vertical curve length in feet

PROGRAM FLAGS

Flag 05	clear	Sag crve
	set	Crown curve

STORAGE REGISTER USE

<u>Register</u>	<u>Value</u>
00	A (G2% - G1%)
01	$(400 + 3.5S)/A$
02	400
03	3.5

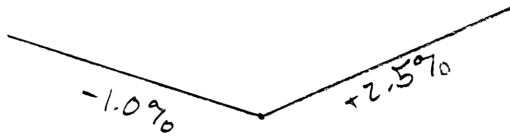
STOPPING SIGHT DISTANCE

Program SSD does stopping sight distance calculations according to the standard AASHTO road design equations and tables (with two exceptions for design speed input, given below). Given the grades, it computes the available stopping distance from the vertical curve length; or the allowable vertical curve length, from either the design speed or stopping sight distance. The program will not backsolve to find the design speed.

The grades are input in response to prompts. Then either the SSD, curve length, or design speed is input using softkeys, as indicated by a crude "menu" in the display. Once the grades are input, they need not be re-input; the program returns to the softkey prompt.

As mentioned, the program will solve a vertical curve length given a design speed. It does this by equating the design speed to a stopping sight distance with an equation derived from the tables in the AASHTO Road Design Manual. This equation gives the correct distance, and thus curve length, for speeds at five mile-per-hour increments, except for 20 and 65 mph (due to a "bump" in the tabular values. Thus, if you have a design speed of either 20 or 65 mph, either input the tabulated SSD values (shown below) or live with a minor error. To compare tabular (AASHTO) values with the program's calculated values:

<u>Design Speed</u>	<u>AASHTO S.S.D.</u>	<u>Program S.S.D.</u>
20	125	100
25	150	150
30	200	200
35	250	250
40	325	325
45	400	400
50	475	475
55	550	550
60	650	650
65	725	750
70	850	850



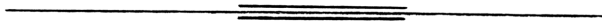
Desire a Stopping Sight Distance
of 450' (70 mph.)

$$L = 2S - \frac{400 + 3.5S}{A}$$

$$= 2 \cdot 450 - \frac{400 + 3.5(450)}{3.5}$$

$$= 735.7'$$

$L < S : \underline{\underline{OK}}$



Design a vertical curve for 55 mph.

$$L = 2S - \frac{1329}{A}$$

55 mph $\rightarrow S = 550'$ from table

$$L = 2 \cdot 550 - \frac{1329}{3} = 657.0'$$

$L > S \Rightarrow$ use other equation

$$L = \frac{S^2 \cdot A}{1329} = \frac{550^2 \cdot 3}{1329} = \underline{\underline{692.8'}}$$

XEQ "SSD"
1:16:00 PM 03/15/89

TITLE:
STOPPING SIGHT DISTANCE
RUN

G1% -1.0000 RUN

G2% 2.5000 RUN

L MPH SSD

850.0000 XEQ D

VC L=735.7143'

L MPH SSD

70.0000 XEQ B

VC L=735.7143'

L MPH SSD

686.0000 XEQ A

SSD=800.2857'

L MPH SSD

XEQ "SSD"
1:17:13 PM 03/15/89

TITLE:
CROWN RUN

G1% 2.0000 RUN

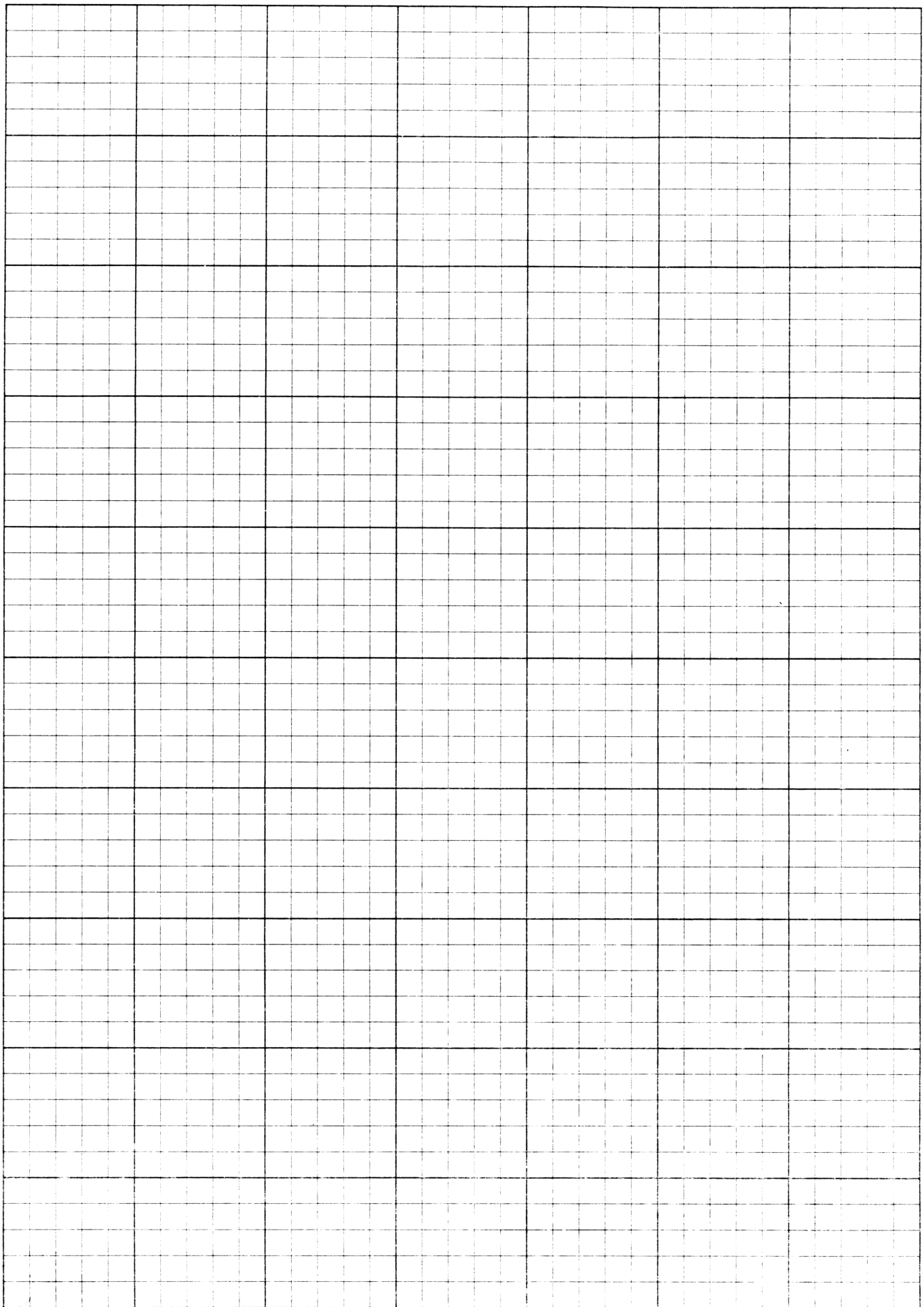
G2% -1.0000 RUN

L MPH SSD

55.0000 XEQ B

VC L=682.8442'

L MPH SSD



	X	Y	Z	T	L		X	Y	Z	T	L
01*LBL "SSD"							D				
02 SF 27							D ²				
03 3							$\frac{3.5D+400}{A}$				
04 XROM "S"							VC L				
05 "G1%"											$L = \frac{D^2 A}{3.5D+400}$
06 PROMPT											if L > D
07 "G2%"											
08 PROMPT	G ₂	G ₁					ZD	D			
09 CF 05							A/1329				<u>Crown</u>
10 X<=Y?											$L = ZD - \frac{1329}{A}$
11 SF 05							ZD - $\frac{1329}{A}$	D			if L < D
12 -	G ₁ - G ₂						D				
13 ABS							D ²				
14 1329							A/1329				
15 FC? 05						A if sag	VC L				$L = \frac{D^2 A}{1329}$
16 SIGN						$\frac{A}{1329}$ if crown					if L > D
17 /											
18 STO 00	A or $\frac{A}{1329}$										
19*LBL 06							VC L				
20 "L MPH SSD"							L	L			
21 PROMPT							A	L	L		
22*LBL B	mph										
23 1.7							AL	L			
24 Y↑X	(mph) ^{1.7}						400	AL	L		
25 40							AL+400	L			
26 /							A		L		
27 .2							ZA		L		
28 +							3.5				
29 INT							ZA-3.5	AL+400	L	L	<u>Sag</u>
30 25							SSD?	L	L	L	$D = \frac{AL+400}{ZA-3.5}$
31 *	SSD										
32*LBL D	SSD						L				
33 "VC L="	D	D					A				
34 ENTER↑	D	D	D				400	A	L	L	
35 ENTER↑	D	D	D				L	400	A	L	
36 ST+ Y	D	ZD	D				400 L	A	L	L	
37 FS? 05							3.5	400 L	A	L	
38 GTO 07							L	3.5	400 L	A	
39 3.5	3.5	D	ZD	D			3.5 L	400 L	A	A	
40 STO 03							A				
41 *	3.5D	ZD	D	D			-A	3.5L	400 L		
42 4 E2	400						SSD				
43 STO 02	3.5D+400										
44 +	A										
45 RCL 00											
46 /											
47 STO 01	$\frac{3.5D+400}{A}$	ZD	D	D		<u>SAG</u>					
48 -	VC L?	D				$L = ZD - \frac{3.5D+400}{A}$					
49 X<=Y?						if L < D					
50 GTO 08											
51 RDN											
52 X↑2											
53 RCL 01											
54 /											
55 GTO 08											
56*LBL 07											
57 RDN											
58 RCL 00											
59 1/X											
60 -											
61 X<=Y?											
62 GTO 08											
63 X<>Y											
64 X↑2											
65 RCL 00											
66 *											
67 GTO 08											
68*LBL A											
69 ENTER↑											
70 "SSD="											
71 RCL 00											
72 FS? 05											
73 GTO 09											
74 *											
75 RCL 02											
76 +											
77 RCL 00											
78 ST+ X											
79 RCL 03											
80 -											
81 /											
82 X>Y?											
83 GTO 08											
84 RDN											
85 RCL 00											
86 RCL 02											
87 R↑											
88 *											
89 RCL 03											
90 R↑											
91 *											
92 R↑											
93 CHS											
94 XROM "Q"											
95 GTO 08											

	X	Y	Z	T	L		X	Y	Z	T	L
						96*LBL 09	$\frac{A}{1329}$	L	L		
						97 /	$\frac{1329L}{A}$				
					<u>Crown</u>	98 SQRT	$\sqrt{\frac{A}{1329}}$	L			
					$SSD = \sqrt{\frac{1329L}{A}}$	99 X<Y?					
					if D < L	100 GTO 08	L				
						101 X<>Y	A/1329				
						102 RCL 00	$1329/A$				
						103 1/X	$L \cdot \frac{1329}{A}$				
						104 +	$L \cdot \frac{1329}{A}$				
					$SSD = \frac{L}{Z} + \frac{1329}{2A}$	105 2	$L \cdot \frac{1329}{2A}$				
					if D > L	106 /					
						107*LBL 08					
						108 X<0?					
						109 CLX					
						110 ARCL X					
						111 XEQ "FT"					
						112 RVIEW					
						113 XROM "L"					
						114 GTO 06					
						115*LBL "RW"					
						:					
						:					
						:					

HORIZONTAL CURVE SOLUTIONS

xeq HC (SIZE 002)

NOTE: The USER FLAGS, OUTPUT SUMMARY, PROGRAM FLAGS, and STORAGE REGISTER USE portions of this documentation, as well as much of the full write-up which follows, pertains to program CHC as well. CHC is also documented in the Coordinate Geometry section.

SOFTKEYS (USER mode ON)

A	Arc Length
B *	Radius
C	Chord length
d *	Degree of curve
D *	Delta angle
E	External
F (x<>y)	Exchange menus (no input)
H (mid)	Middle ordinate
J (tan)	Tangent

At least one item must be an asterisked item, and this item must be input first.

If the degree of curve or delta angle are input, it is up to the user to verify that the input agrees with the status of flag 03, i.e. if the angles are in degrees-minutes-seconds, flag 03 must be set. This is not prompted as in most programs.

USER FLAGS

Flag 00	clear	Full output
	set	Sector, segment, and fillet areas not output
Flag 03	clear	Angular input and output in decimal degrees (dd)
	set	Angles in degrees-minutes-seconds (DMS)

INPUT SUMMARY

Program HC does not use prompts for input as do most of the other programs in this package. The two prompts which are used are intended as crude menus to the softkeys, for which see above.

Note that all the softkeys are active even when their menu is not displayed. For example, to input radius and delta, it is not necessary to bring back the first menu. Simply key in the radius, press softkey B, key in the delta angle, and press softkey D.

OUTPUT SUMMARY (Items in parenthesis are displayed only if flag 00 is clear)

RAD	Radius
DEG	Degree of curve
DEL	Delta, the central angle or angle subtended
ARC L	Arc length
T	Tangent length
CH	Chord length
M	Middle ordinate
E	External dimension
(SECT)	Area of the sector
(SEGM)	Area of the segment
(FIL)	Area of the fillet

PROGRAM FLAGS

Flag 06	clear set	In second menu In first menu
Flag 07	clear set	Don't have radius Have radius--solve after Delta input
Flag 09	clear set	HC CHC
Flag 10	clear set	Don't have Delta Have Delta--solve after radius input
Flag 21	set	Controls display and print behavior
Flag 27	set	Sets USER mode

STORAGE REGISTER USE

<u>Register</u>	<u>Value</u>
00	Radius
01	Delta/2 (decimal degrees stored)

HORIZONTAL CURVE SOLUTIONS

Programs HC and CHC are used for the solution of horizontal (circular) curves. HC computes all the commonly needed parameters, given any (well, almost any) two of them as input. CHC computes these same parameters from the coordinates of the PC, PT and RP (point of curvature or beginning, point of tangency or end, and radius or center point). CHC also computes the azimuths of the incoming and outgoing tangents (PC to PI and PI to PT, respectively, the PI being the point of intersection).

For input, program HC uses the softkeys A, B, C, D, d (shifted), E, F, H, and J in USER mode. USER mode is automatically turned on by the program to facilitate input. Parameters are input by keying the number in and pushing the appropriate key, as follows:

	<u>Key</u>	<u>Parameter</u>
First Menu	A	Arc Length
	B *	Radius
	C	Chord length
	d *	Degree of curve
	D *	Delta angle
	E	External
Second Menu	F (x<>y)	Exchange menus (no input)
	H (mid)	Middle ordinate
	J (tan)	Tangent

The program requires two of these parameters. At least one item must be an asterisked item, and this item must be input first. Radius and degree of curve are not sufficient since they are derivable one from the other; if both are input, HC will ignore the first one and await further input.

MENUS

HC uses menus to prompt input, displaying the above mnemonics (R, DL, etc.) roughly above each key. The first-row parameters are prompted first, then the second-row menu appears. To return to the first-row menu, press softkey F (x<>y). Pressing it again brings back the second-row menu. These menus are only mnemonic aids; the first-row keys are still active when the second-row menu is on, and vice versa.

ANGLES

As with the other geometry programs in this package, HC and CHC format output and expect input for angles to be in decimal degrees (dd) if flag 03 is clear, and in degrees-minutes-seconds (DMS) if flag 03 is set. This is not reflected in the prompts; the user must consider the status of flag 03 when inputting the degree of curve or the delta angle. The output is labeled.

COORDINATE GEOMETRY

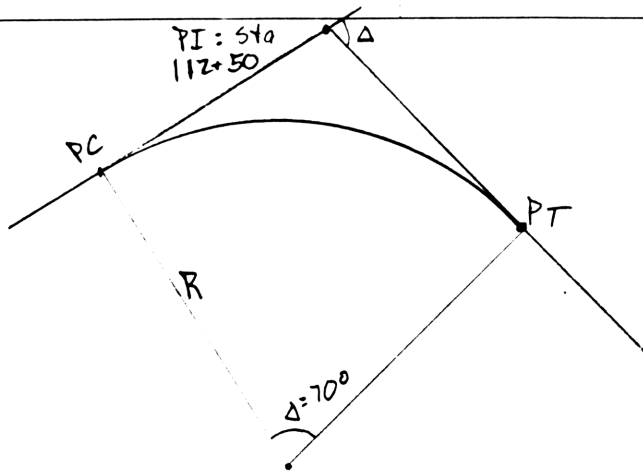
Input for CHC is much simpler (assuming your point coordinates are already stored by other coordinate geometry routines). The program prompts for the point numbers of the PC, RP, and PT. As with most other coordinate geometry routines the numbers are keyed into the stack: key in PC point number, enter, key in radius point number, enter, key in PT point number (don't enter), press R/S. CHC checks the radii PC-RP and PT-RP. The two are first rounded to the accuracy of the display setting and then tested for equality. If they aren't equal, a warning is given; the radius used is from PC to RP. (Thus, don't set the display to FIX 4 if the coordinates are only good to two decimal places.) It also calculates the north azimuths of the incoming and outgoing tangent, again in either D.MS or decimal degrees depending on flag 03. It then drops into the HC output routine, displaying all the parameters listed on the first page and the area of the sector, segment, and fillet. All the linear output is labeled with dimensions of feet and square feet. The programs will also, of course, work with inches, meters, or furlongs.

Both programs set the calculator to DEG (degree) mode upon execution (in subroutine S). If for some reason this is changed by the user during use, errors will result.

If USER mode is set when CHC is run, the softkeys are active, and once pressed (accepting whatever is in the X register as input), the menus will come up as though HC had been executed. However, when all the output has been displayed, the program returns to the CHC prompt. When using HC, all the output need not be viewed; the softkeys are active at any time to accept new input and start over (or to bring up a menu, in the case of softkey F).

FLAGS

If flag 00 is set, the areas of sector, segment, and fillet are not displayed by either HC or CHC; in CHC the distances PC-RP and PT-RP, and the tangent azimuths, are not displayed either.



For a $7^{\circ}30'$ Curve,
solve for missing
parameters and stations

$$D = 7^{\circ}30' ; \Delta = 70^{\circ}$$

$$R = \frac{5729.5780}{D} = 763.944'$$

$$T = R \tan(\Delta/2) = 534.919'$$

$$L = \frac{\Delta}{D} \times 100 = \frac{70}{7.5} \times 100 = 933.33'$$

$$\text{Chord} = 2 \cdot R \sin(\Delta/2) = 876.360'$$

$$M = R - R \cos(\Delta/2) = 138.158'$$

$$E = \frac{R}{\cos(\Delta/2)} - R = 168.659'$$

$$\Delta \text{ Sector} = \frac{\Delta}{360} \cdot \pi R^2 = 356,507.07 \text{ sq'}$$

$$\Delta \text{ Segment} = \text{Sector} - \frac{1}{2} \text{Chord} \times R \cos \frac{\Delta}{2} \\ = 42,300.07 \text{ sq'}$$

$$\Delta \text{ Fillet} = \frac{1}{2} \text{Chord} \times (M+E) - \text{Segment} \\ = 52,141.03 \text{ sq'}$$

SF 03
XEQ "HC"
5:34:58 PM 01/05/89

TITLE:
CURVE EXAMPLE RUN

L R CH d/D E
7.3000 XEQ d
<> MID T
XEQ F
L R CH d/D E
70.0000 XEQ D

RAD=763.9437'
DEG=7.3000 DMS
DEL=70.0000 DMS
ARC L=933.3333'
T=534.9192'
CH=876.3602'
M=138.1577'
E=168.6594'

SECT=356,507.0725 SQ'
SEGM=42,300.0591 SQ'
FIL=52,141.0612 SQ'

L R CH d/D E

$$\text{Sta. PC} = 112+50 - 534.919$$

$$= 107+15.081$$

$$\text{Sta PT} = \text{Sta PC} + L = 116+48.410$$

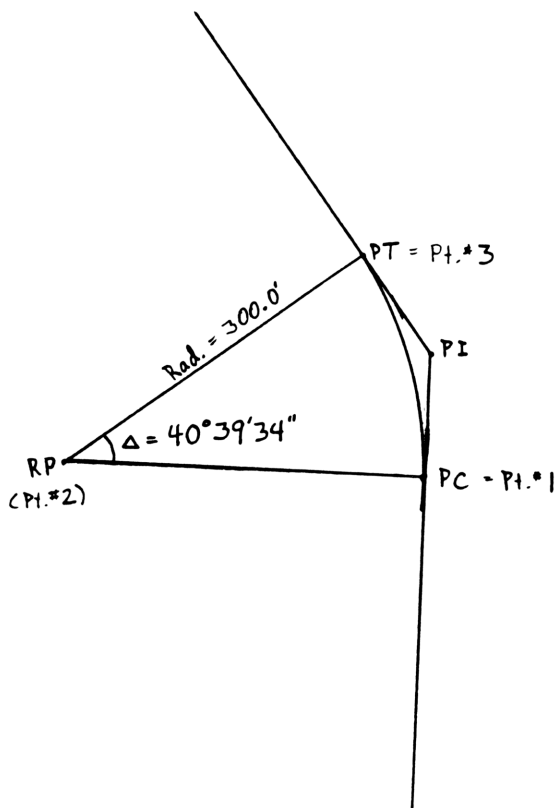
SHORT FORM - SF 00

SF 00
XEQ "HC"
L R CH d/D E
7.3000 XEQ d
<> MID T
70.0000 XEQ D

RAD=763.9437'
DEG=7.3000 DMS
DEL=70.0000 DMS
ARC L=933.3333'
T=534.9192'
CH=876.3602'
M=138.1577'
E=168.6594'

L R CH d/D E

CHC Example



Short version
with Flag 00
set gives
only these
parameters

```

SF 00
XEQ "CHC"

PC↑RP↑PT
1.0000 ENTER↑
2.0000 ENTER↑
3.0000 RUN

RAD=300.0000'
DEG=19.0555 DMS
DEL=40.3934 DMS
ARC L=212.8924'
T=111.1503'
CH=208.4533'
M=18.6873'
E=19.9287'

-----
PC↑RP↑PT
    
```

```

XEQ "STOR"

-----
PT↑NTE
1.0000 ENTER↑
3.103.6749 ENTER↑
9.827.2778 RUN

-----
PT↑NTE
2.0000 ENTER↑
3.106.1477 ENTER↑
9.527.2890 RUN

-----
PT↑NTE
3.0000 ENTER↑
3.299.7338 ENTER↑
9.756.4711 RUN

-----
PT↑NTE
    
```

```

XEQ "CHC"

9:53:34 AM 10/15/87

TITLE:
STERLING ST. RUN
*****
PC↑RP↑PT
1.0000 ENTER↑
2.0000 ENTER↑
3.0000 RUN

PC-RP=300.0000'
PT-RP=300.0000'
PC-PI AZ=0.2820 DMS
PI-PT AZ=-40.1114 DMS
    
```

```

RAD=300.0000'
DEG=19.0555 DMS
DEL=40.3934 DMS
ARC L=212.8925'
T=111.1504'
CH=208.4534'
M=18.6874'
E=19.9288'

-----
SECT=31,933.8705 SQ'
SEGM=2,613.5850 SQ'
FIL=1,411.2467 SQ'

-----
PC↑RP↑PT
    
```

Flag 03
is set
⇒ output
in D.MS

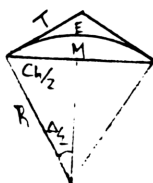
	X	Y	Z	T	L		X	Y	Z	T	L
01*LBL "CHC"						56*LBL 10					
02 E						57 FS? 00					
03 XROM "S"						58 RTN					
04 SF 09						59 "I AZ"					
						60 XROM "Dd"					
05*LBL 08						61 AVIEW					
06 FC? 09						62 RTN					
07 GTO 14											
08 SF 07						63*LBL 09	RP	P+			
09 "PC+RP+PT"						64 XROM "XY"	ΔX	ΔY			
10 PROMPT	PT	RP	PC			65 R-P	d	-AZ			
11 STO 01	PT					66 X<Y	-AZ				
12 RDN						67 CHS	AZ				
13 STO 00	RP	PC		PT		68 X>0?					
14 XEQ 09	AZ _{pc}	d _{pc}				69 RTN					
15 X<> 01	PT					70 360					
16 X<>Y	d _{pc}	PT				71 +					
17 X<> 00	RP	PT				72 RTN	AZ	d			
18 XEQ 09	AZ _{pr}	d _{pr}									
19 X<>Y	d _{pr}					73*LBL "HC"					
20 RCL 00	d _{pc}	d _{pr}	AZ _{pr}			74 CF 09					
21 "PC"						75 SF 27	"User"				
22 XEQ 11						76 E					
23 X<>Y	d _{pr}	d _{pc}	AZ _{pr}			77 XROM "S"					
24 "PT"											
25 XEQ 11						78*LBL 14					
26 "PC*PT"						79 CF 07					
27 X*Y?						80 CF 10					
28 BEEP											
29 X*Y?						81*LBL 07					
30 AVIEW						82 SF 06					
31 X*Y?						83 "L R CH d/D E"					
32 XROM "L"						84 PROMPT					
33 R+											
34 R+	AZ _{pr}					85*LBL d	degree				
35 RCL 01	AZ _{pc}					86 FS? 03					
36 X<=Y?						87 HR					
37 GTO 00						88 E	1				
38 100						89 %	deg/100				
39 -						90 1/X	100/deg				
40 X<>Y						91 R-D	rad.				
41 LASTX											
42 -						92*LBL B					
43 X<>Y	AZ _{pc}	AZ _{pr}				93 STO 00	Rad.				
44*LBL 00						94 SF 07					
45 "PC-PI"						95 FS? 10					
46 XEQ 10						96 GTO 00					
47 X<>Y	AZ _{pr}	AZ _{pc}				97 GTO 06					
48 "PI-PT"											
49 XEQ 10						98*LBL D	Δ				
50 -	Δ					99 FS? 03	Δ				
51 2	$\Delta/2$					100 HR	$\Delta/2$				
52 /						101 2					
53 ABS						102 /					
54 STO 01	$\Delta/2$					103 STO 01					
55 GTO 00						104 SF 10					
						105 FS? 07					
						106 GTO 00					

FS 07
⇒ have radius

FS 10
⇒ have delta

(B35)

	X	Y	Z	T	L		X	Y	Z	T	L
107*LBL F						155*LBL 01	T				
108 FC? 06						156 RCL 00	Rad				
109 GTO 07						157 /	T/R				
						158 ATAN	$\Delta/2$				
110*LBL 06						159 STO 01					
111 CF 06						160 GTO 00					
112 "<" MID T											
113 PROMPT						161*LBL H	M				
						162 FS? 07					
114*LBL A	Arc L					163 GTO 04					
115 R-D	$L \cdot \frac{180}{\pi}$					164 E	1				
116 FS? 07						165 RCL 01	$\Delta/2$				
117 GTO 01	$\Delta/2$					166 COS	$\cos \frac{\Delta}{2}$	1	M		
118 RCL 01	Δ					167 -	$1 - \cos \frac{\Delta}{2}$	M			
119 ST+ X											
120 GTO 03						168*LBL 03					
						169 /	Rad				
121*LBL 01						170 STO 00					
122 RCL 00	Rad					171 GTO 00					
123 ST+ X	2 R										
124 /	$\Delta/2$					172*LBL 04	M				
125 STO 01						173 CHS	-M				
126 GTO 00											
						174*LBL 05	Ecc-M				
127*LBL C	Chord					175 RCL 00	Rad				
128 2						176 +	$R \cdot E/R \cdot \pi$				
129 /	$Ch/2$					177 LASTX	R				
130 FS? 07						178 X<Y?					
131 GTO 02						179 X<>Y	hypotenuse adjacent				
132 RCL 01	$\Delta/2$					180 /					
133 SIN	$\sin \Delta/2$	$Ch/2$				181 ACOS	$\Delta/2$				
134 GTO 03						182 STO 01					
135*LBL 02	$Ch/2$					183*LBL 00					
136 RCL 00	Rad					184 CF 06	"Menu"				
137 /						185 SF 21	"Print"				
138 ASIN	$\Delta/2$					186 E2	100				
139 STO 01						187 FC? 10					
140 GTO 00						188 FS? 07					
						189 X=0?					
141*LBL E	Ext.					190 GTO F					
142 FS? 07						191 RCL 00	Rad	100			
143 GTO 05						192 ADV					
144 RCL 01	$\Delta/2$					193 CF 07					
145 2	$\Delta/4$					194 CF 10					
146 /	$\tan \Delta/4$	E				195 "RAD"					
147 TAN	T					196 XEQ 12					
148 /						197 /	100/R				
						198 R-D	degree				
149*LBL J	Tang					199 "DEG"					
150 FS? 07						200 XROM "Dd"					
151 GTO 01						201 AVIEW					
152 RCL 01	$\Delta/2$										
153 TAN	$\tan \frac{\Delta}{2}$	T									
154 GTO 03											



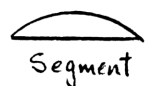
202 RCL 01
 203 ST+ X
 204 "DEL"
 205 XROM "Dd"
 206 AVIEW
 207 RCL 00
 208 *
 209 D-R
 210 "ARC L"
 211 XEQ 12
 212 RCL 01
 213 TAN
 214 RCL 00
 215 *
 216 "T"
 217 XEQ 12
 218 RCL 01
 219 SIN
 220 RCL 00
 221 *
 222 ST+ X
 223 "CH"
 224 XEQ 12
 225 E
 226 RCL 01
 227 COS
 228 -
 229 RCL 00
 230 *
 231 "M"
 232 XEQ 12
 233 RT
 234 RCL 01
 235 2
 236 /
 237 TAN
 238 *
 239 "E"
 240 XEQ 12
 241 XROM "L"
 242 FS? 00
 243 GTO 08

X	Y	Z	T	L
$\Delta/2$				
Δ				
Rad	Δ			
L				
$\Delta/2$				
$\tan \Delta/2$				
R				
T				
$\Delta/2$				
$\sin \Delta/2$				
Rad	$\sin \Delta/2$	T		
$\text{ch}/2$				
ch				
1				
$\Delta/2$				
$\cos \Delta/2$	1	ch	T	
$1 - \cos$				
Rad				
M	ch	T	T	
T				
$\Delta/2$				
$\Delta/4$				
E				

244 RCL 00
 245 X+2
 246 RCL 01
 247 *
 248 D-R
 249 "SECT="
 250 XEQ 13
 251 ENTER+
 252 ENTER+
 253 RCL 01
 254 RCL 00
 255 P-R
 256 *
 257 -
 258 "SEGM="
 259 XEQ 13
 260 RCL 01
 261 TAN
 262 RCL 00
 263 X+2
 264 *
 265 RT
 266 -
 267 "FIL="
 268 XEQ 13
 269 XROM "LL"
 270 GTO 08
 271 LBL 11
 272 RND
 273 FS? 00
 274 RTN
 275 "t-RP"
 276 LBL 12
 277 "t=" "
 278 ARCL X
 279 XROM "FT"
 280 AVIEW
 281 RTN
 282 LBL 13
 283 ARCL X
 284 XROM "FS"
 285 AVIEW
 286 END

X	Y	Z	T	L
Rad				
R^2				
$\Delta/2$				
*				
$\frac{1}{2} \Delta \cdot \frac{1}{2} R^2$				
D-R				
"SECT="				
XEQ 13				
ENTER+				
ENTER+	S	S	S	
RCL 01				
RCL 00	Rad	$\Delta/2$	S	S
P-R	$X/2$	Y	S	S
*	Area			
-				
"SEGM="	Segm	S	S	S
XEQ 13				
RCL 01	$\Delta/2$			
TAN	$\tan \Delta/2$			
RCL 00	R^2	$\tan \Delta/2$	Segm	S
X+2				
*	\Diamond			
RT	S			
-				
"FIL="	Fillet			
XEQ 13				
XROM "LL"				
GTO 08				
LBL 11				
RND				
FS? 00				
RTN				
"t-RP"				
LBL 12				
"t=" "				
ARCL X				
XROM "FT"				
AVIEW				
RTN				
LBL 13				
ARCL X				
XROM "FS"				
AVIEW				
END				

$$\text{Area} = \frac{\Delta}{360} \cdot \pi R^2$$



COORDINATE GEOMETRY PROGRAMS

SIZE (2N+2) for "N" points

xeq STOR to initialize points and store point coordinates
DUMP to list ("dump") points
FREE to find free (CLR'd) points
DUP to duplicate a block of points
INV to find distance and azimuth between two points
LAZ to set (locate) a point by distance and azimuth
LBR to set a point by distance and bearing
LAN to set a point by backsighting on another point
LPT to set a point by distance, on the bearing through two other points
LLI to set a point along a line determined by two other points
ANGL to find the angle between two points about a third
PART to set a point part way between two other points
OFPT to find the offset of a point from a line defined by two other points
OFAZ to find the offset from a line defined by a point and an azimuth
OFBR to find the offset from a line defined by a point and a bearing
IAZ to intersect two lines defined by points and azimuths
IBR to intersect two lines defined by points and bearings
IPT to intersect two lines each defined by two points
PRL to set two points defining a line parallel to a line defined by two other points
ATRV to set a point along an arc by arc length
ATVA to set a point along an arc by angle subtended
AFRC to set a point partway between two points along an arc
ARCS to intersect two arcs
APT to intersect an arc with a line defined by two points
AAZ to intersect an arc with a line defined by a point and an azimuth
ABR to intersect an arc with a line defined by a point and a bearing
INTAN to find the internal tangency points of two circles
EXTAN to find the external tangency points of two circles
PTAN to find the points of tangency of two lines through a point, to a circle
C3P to find a circle through three points
ADJ to adjust a traverse by compass rule
AREA to find the area bounded by points
CHC to calculate horizontal (circular) curve data between two points about a third

SOFTKEYS (USER mode ON)

C Correct a point input error in AREA

None of the other COGO routines makes use of softkeys. However, the softkeys of HC are accessible during use of CHC and should be avoided.

USER FLAGS

Flag 00	clear set	Point protect on Point protect off
Flag 01	clear set	North-East mode X-Y mode
Flag 02	clear set	Warning BEEP disabled in point-protect Warning BEEP enabled
Flag 03	clear set	Angles in decimal degrees (dd) Angles in degrees-minutes-seconds (DMS)

INPUT SUMMARY

A Angle or north azimuth

A1, A2 Azimuth

B Bearing, measured from North or South; between 0° and 90° in one of the four quadrants.

Quad Quadrant of a bearing: 1 = NE
 2 = SE
 3 = SW
 4 = NW

C Center point (radius point)

D Distance

I First point in a series

J Point of beginning: "From point J, ..."

K Backsight or reference point, or last point in a series

L, M Points defining a line

M "Proximity" point; choose the point closest to point M

N, N1, Point number to be solved or found; point number under
N2 which to store the solved coordinates

R, R1, R2 Radius or radii of arcs

OUTPUT SUMMARY

PT Point number stored and displayed

N Northing (flag 01 clear)

E Easting (flag 01 clear)

X X coordinate (flag 01 set)
 Y Y coordinate (flag 01 set)
 D Distance
 AZ North azimuth
 A Angle
 OFFSET Offset from a line

PROGRAM FLAGS

Flag 08	clear set	Don't calculate offset Calculate offset
	clear set	PART LLI
Flag 09	clear set	Quadrant 2 or 3 (in BA) Quadrant 1 or 4
	clear set	INTAN or ATVA EXTAN or ATRV
	clear set	More than two points input in AREA Only one or two points input in AREA
Flag 10	clear set	Softkey C has not been used in AREA Softkey C has been used in AREA
Flag 14	set	Line type control
Flag 21	set	Display control
Flag 26	clear set	BEEP disabled according to flag 02 BEEP enabled according to flag 02
Flag 29	clear	Display of decimal point disabled

STORAGE REGISTER USE

<u>Register</u>	<u>Value</u>
00	Working register
01	Working register
02	Easting (X) coordinate of point #1
03	Northing (Y) coordinate of point #1
04	Easting (X) coordinate of point #2
05	Northing (Y) coordinate of point #2
	Etc.

847

COORDINATE GEOMETRY PROGRAMS

There are thirty-three user programs in the coordinate geometry section of this module (CHC, LPT, and most of those between STOR and AREA, inclusive, in a CAT 2 listing). Each of the routines has its own short description, with a sketch, and input and output information. These individual writeups follow this introduction, which gives an overview of the coordinate geometry group.

All the routines work with points, lines, and curves in two-dimensional (plane) space. The routines will work in an X-Y system if flag 01 is set, and in a North-East system if flag 01 is clear. This may be changed at any time as long as a prompt isn't pending. No matter which orientation is being used, the X (East) coordinate of a point "N" is stored in register (2N); the Y (North) coordinate is stored in register (2N+1). Thus, to store a given number of points "n" requires a SIZE of $2n+2$. The available SIZE is checked when initializing points in STOR.

POINT STORAGE

All points are protected by a point storage subroutine which checks to see if a pair of registers already contain coordinates before it stores a point there. If the user designates a point number which is already used, the program BEEPs and displays USED-NEW PT? If another point number is keyed in, the program tries to store the coordinate as that point. If instead the prompt is bypassed, the existing coordinates are overwritten. The program recognizes as unused, those points which have first been initialized (cleared) in program STOR. Thus, the first thing done at the beginning of a project is usually to initialize points and then store the coordinates of one or more points in STOR. It is important to note that point-protection is disabled if flag 00 is set; and that the BEEP is disabled if flag 02 is clear.

ANGLES

All the COGO (coordinate geometry) routines which use angles recognize the status of flag 03. If flag 03 is clear all angles (input and output) are in decimal degrees. If flag 03 is set, all angles are in degrees-minutes-seconds. This is shown clearly in all prompts and output as (dd) or (DMS).

INPUT FORMAT

Most of these routines require multiple input. Unlike most programs in this module which prompt for input one item at a time, many of the COGO routines require two, three, or four items to be input into the stack at once. This is signified in the prompt by an upward arrow, \nearrow , meaning "Enter \nearrow ." Thus for the STOR prompt PT \nearrow N \nearrow E, key in the point number, enter, key in the northing, enter, key in the easting (don't enter), press R/S. The program uses whatever it finds in the stack. This can produce some shortcuts; for instance, if the coordinates of point #10 are (2,2), simply key in 10, enter, 2, enter, R/S. None of the routines use numbers in the stack above the numbers requested in the initial prompt, so if you realize a mistake was made before R/S is pressed, nothing needs to be cleared; just key in the right numbers. However, if a program has a second prompt (usually "N", asking for the point number to store the coordinates under, or "M", the "proximity" point), important data are carried in the stack; if a mistake is caught, use RDN (roll down) to avoid changing the necessary stack configuration. This rather poor explanation will become clearer with a little use, and perhaps an examination of the program listings with the stack usage traced. If at all uncertain, just re-execute the routine.

TEMPORARY POINTS

As mentioned, each point "N" stores its coordinates in registers 2N and 2N+1. Registers 00 and 01 are the only storage registers used by the programs as working registers, the rest of the work being done in the stack and synthetic registers (about which see the section "Synthetic Programming and Other Loose Ends" at the front of the documentation). This means that if you just want the coordinates of a point displayed but not stored, call it point #0, override the point protection, and it will be temporarily stored until another program needs registers 00 and 01, at which time it will be gone.

PROXIMITY POINT

Many of the routines dealing with arcs find two points instead of just one. The routines use a simple method to determine which one to store; they prompt for a "proximity" point (always prompted as point M) and pick the point closest to it. If both points are wanted, bypass the "M" prompt and they will both be stored. It's especially important in this case to keep a running sketch to designate which point is which.

PRINTER USE

Most routines, upon finishing, return to their initial prompt to go again. The first time each routine is executed, a line of asterisks is printed (that is, if a printer is present) across the paper. If a routine is used repeatedly (for example, LAZ to traverse around an area) it's difficult when reviewing the output to tell which routine is being used. The line of asterisks makes this a little clearer; directly above the last line is the function execution. See the examples for clarification. Using a printer is strongly recommended; tracing down mistakes without one is almost impossible.

LIMITATIONS

It's important to keep in mind the scope of this COGO package. The intended uses typically involve layouts of small structural components or minor land surveying tasks. While a completely "empty" HP-41CV or CX (one with no program storage or key assignments) is capable of storing up to 158 points, anything over thirty or forty points starts to get clumsy. Large complicated layouts can be done, but only with very strong concentration and constant cross-checking, because of some of the following disadvantages:

1) There is no graphics capability. Anyone used to on-screen graphics knows how useful a simple picture can be in terms of ease of use, confidence, and error detection. For anything more than about four points, it is almost essential to have a carefully scaled drawing with point numbers, distances, and azimuths shown. It is also a very good idea to write out the intended procedure, with input, before starting. Just using the routines and getting through the input sequences requires a lot of concentration. Don't try to think your way through the problem at the same time. And, as mentioned, a printer is extremely helpful. Don't let this discourage use of the routines, though; all computer coordinate geometry work was originally done on batch machines with exactly these limitations. They simply require a little caution and concentration.

2) Although stored points are protected from alteration by other COGO routines (if flag 00 is clear), the storage registers they reside in are used quite cavalierly by other programs, wiping out coordinates. However, only a few programs in this module (PIER, PLG, and sometimes FTG) use registers above #30. It may be worthwhile to start with point number 15 or so (or 25 if PIER may be used), leaving the lower-numbered registers for use by other (non-COGO) programs. Program DUP may also be used to duplicate a block of points to a safer (higher-numbered) location for

storage, to be used as re-numbered points or copied back down later.

3) The HP-41 is limited to ten significant figures. It's not a good idea to waste too many of them in front of the decimal point; use of more than four figures before the decimal point is not recommended. Also, intersections involving two lines with not-very-different bearings is inherently inaccurate, with any COGO package.

4) While angles may be input in bearing/quadrant mode, it is clumsy; and all output is expressed as north azimuths. At best, bearings are handled awkwardly by this package. Those used to working with bearings (i.e. surveyors and others who know more about these things than the author) will probably grit their teeth and use azimuths.

5) As stated, all points are two-dimensional. Elevations are not stored.

6) There is no provision for storing descriptors with points. Those used to large-scale professional coordinate geometry programs (the author included) will miss this greatly; good descriptors are second only to interactive graphics in inspiring confidence and ease of use. Both elevations and short description capabilities (one register, six letters) were considered. However, both complicate (swell) the programming immensely; also, each point would require four registers to store instead of just two, halving the available point storage. A realistic assessment of the uses, intended scope, and inherent limitations of this package made both elevations and descriptors seem an expensive luxury.

As mentioned five disadvantages back, this listing is intended to caution, not discourage. A great deal of useful, accurate work has been done on systems with exactly the limitations stated here. People used to the old batch programs will feel right at home using these programs. They simply require a little more care than their \$12,000 siblings.

COMMAND INPUT

STOR I↗K CLR; PT↗N↗E (flag 01 clear)
 I↗K CLR; PT↗X↗Y (flag 01 set)

Clear (initialize) all points from I to K inclusive, then store the North and East (or X and Y) coordinates of points. The I↗K CLR prompt is not repeated; the PT↗... prompt is. If enough points have already been initialized, the I↗K CLR prompt may be bypassed.

—
DUMP I↗K

Dump (display or list) the coordinates of points I through K. Skips free (initialized) points.

—
FREE (No input)

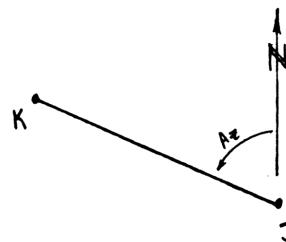
Finds free or unused points (points which have been initialized (CLR'd) in STOR).

—
DUP I↗K↗NEW I

Copies a block of points from I to K, to a new block starting at NEW I. Leaves the original points I through K intact. To just copy one point, key I,↗,↗, NEW I. Point protection is not active with DUP; it will overwrite points with with no warning. The user must also check for overlap.

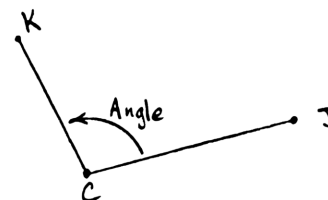
—
INV J↗K

Finds the distance and azimuth between points J and K. The K point of one INV is the default J of the next. Thus to inverse from point #1 to #2, then #2 to #3, then #3 to #4: Xeq (alpha) INV (alpha); key in 1, Enter↗, 2, R/S; 3, R/S; 4, R/S. In addition, the sum of the distances is stored in register 00 for use by program ADJ.



—
ANGL J↗C↗K

Find the angle between points J and K, measured about point C. The distances from J to C and K to C need not be the same.

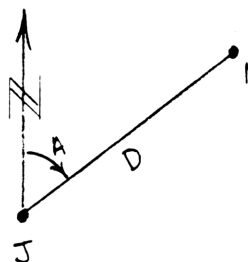


—
Note: the "↗" symbol means "Enter↗." (dd) means "decimal degrees" (flag 03 clear); (DMS) means "Degrees-Minutes-Seconds" (flag 03 set). Clockwise angles are positive.

COMMAND INPUT

LAZ J/D/A (dd or DMS); N

Locate by azimuth. From point J, set point N at distance D and azimuth A.

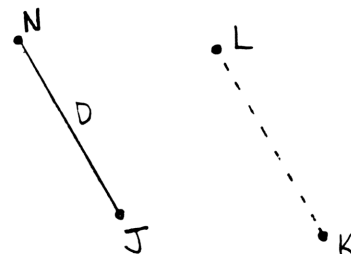


LBR J/D/B (dd or DMS); QUAD; N

Locate by bearing. From point J, set point N at a distance D and bearing B in quadrant QUAD.

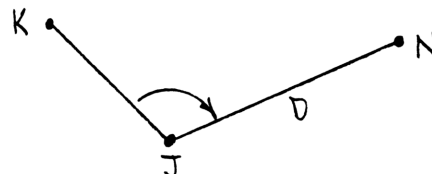
LPT J/D/K/L; N

Locate by point bearing. From point J, set point N at a distance D, at the azimuth of the line through K to L.



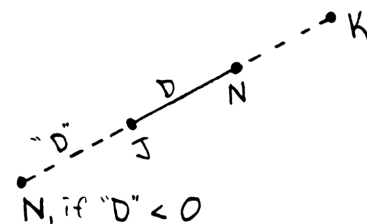
LAN J/K/D/A (dd or DMS); N

Locate by angle. From point J, backsight on point K; turn angle A (clockwise is positive) and set point N at distance D.



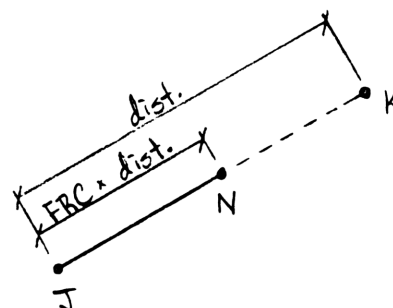
LLI J/K/D; N

Locate on line. From J, set point N a distance D toward point K. A negative D sets point N in the opposite direction of K.



PART J/K/FRC; N

From point J, set point N a fraction FRC of the distance to point K. If FRC is greater than 1, N will be set beyond K. Like LLI, if FRC is negative, N will be set away from rather than toward K from point J.

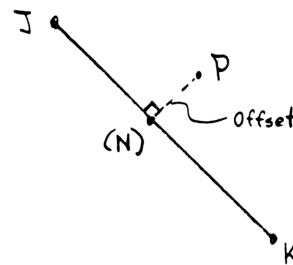


Note: the "↵" symbol means "Enter↵." A semicolon means the items of input are prompted in separate groups. (dd) means "decimal degrees" (flag 03 clear); (DMS) means "Degrees-Minutes-Seconds" (flag 03 set). Clockwise angles are positive.

COMMAND INPUT

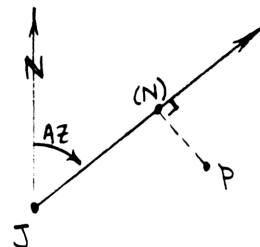
OFPT J/K/P; N

Offset by points. Find the offset of point P from the line through points J and K; set point N on line J-K. The prompt for point N may be bypassed (or answered 0) to avoid setting a point there.



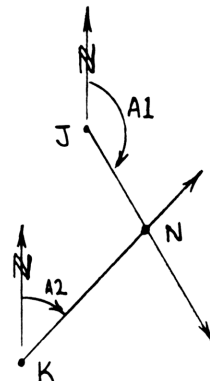
—
OFAZ J/P/AZ (dd or DMS); N

Offset by azimuth. Find the offset of point P from the line through J at azimuth AZ; set point N.



—
OFBR J/P/B (dd or DMS); QUAD; N

Offset by bearing. Find the offset of point P from the line through J at bearing B; set point N.



—
IAZ J/A1/K/A2; N

Intersect by azimuth. Set point N at the intersection of the line through point J at azimuth A1; with the line through point K at azimuth A2.

—
IBR J/B; QUAD; K/B; QUAD; N

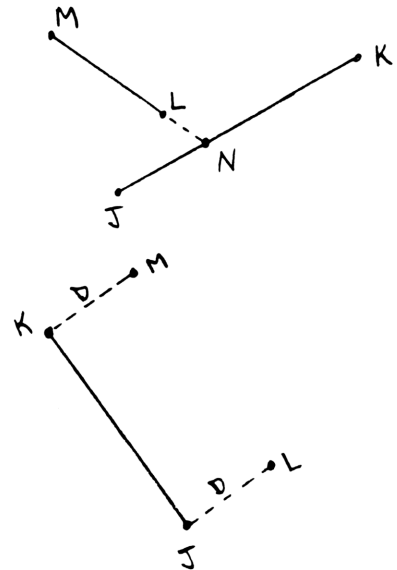
Intersect by bearing. Set point N at the intersection of the line through point J at bearing B; with the line through point K at bearing B.

—
Note: the "↵" symbol means "Enter↵." A semicolon means the items of input are prompted in separate groups. (dd) means "decimal degrees" (flag 03 clear); (DMS) means "Degrees-Minutes-Seconds" (flag 03 set). Clockwise angles are positive.

COMMAND INPUT

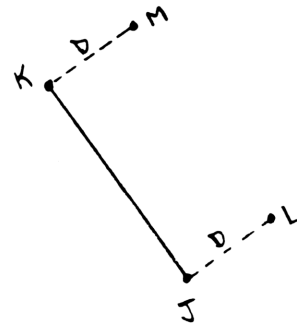
IPT J/K/L/M; N

Intersect by points. Set point N at the intersection of lines J-K and L-M. J and K need not be on opposite sides of line L-M and vice versa.



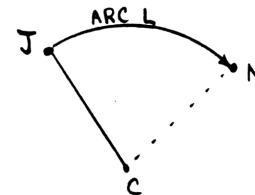
PRL J/K/D; L; M

Parallel line. Set points L and M a distance D from points J and K such that line L-M is parallel to line J-K. From J facing K, D is positive to the right.



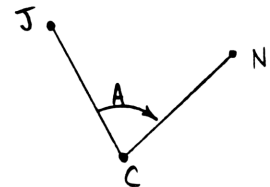
ATRV J/C/ARC L; N

Arc traverse. From point J, about point C, traverse a distance ARC L along the arc to set point N. ARC L is positive measured clockwise.



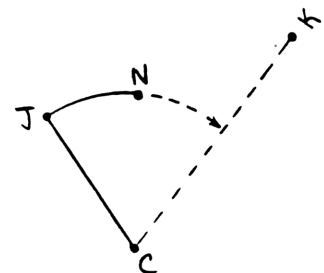
ATVA J/C/A (dd or DMS); N

Arc traverse by angle. From point J, about point C, turn an angle A to set point N on the arc. Angle A is positive measured clockwise. This function is identical to LAN except the distance (here, radius) is set by point J.



AFRC J/C/K/FRC; N

Arc fraction. Set point N a fraction FRC of the way along the arc from J to K about C. If J and K are not the same distance from C, the radius used is from J to C.



Note: the "↵" symbol means "Enter↵." A semicolon means the items of input are prompted in separate groups. (dd) means "decimal degrees" (flag 03 clear); (DMS) means "Degrees-Minutes-Seconds" (flag 03 set).__Clockwise angles are positive.

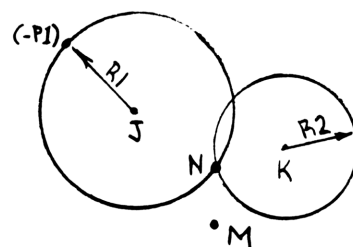
Note: The following arc commands have two possible solutions; the routines will pick the one closest to point M to store as point N (the solved point). If the prompt for M is bypassed, both solutions will be stored as N1 and N2.

If the prompt for a radius is answered as a negative number (e.g. -12) it is taken as point #12 and the radius to be used is the distance from that point to the center point.

COMMAND INPUT

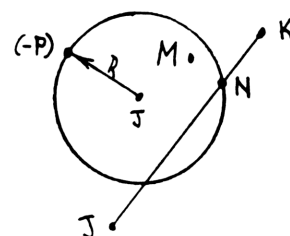
ARCS J/R1/K/R2; M; N (or N1; N2)
 J/-P1/K/-P2; M; N (or N1; N2)

Intersect arcs. Intersect the arc of radius R1 about point J, with the arc of radius R2 about point K, to set point N.



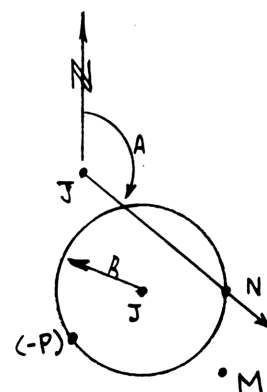
APT J/R/K/L; M; N (or N1; N2)
 J/-P/K/L; M; N (or N1; N2)

Intersect arc by points. Intersect the arc of radius R about point J with the line through points K and L.



AAZ J/R/K/A; M; N; (or N1; N2)
 J/-P/K/A; M; N; (or N1; N2)

Intersect arc by azimuth. Intersect the arc of radius R through point J with the line through point K at azimuth A.



ABR J/R/K/B; QUAD; M; N
 J/-P/K/B; QUAD; M; N

Intersect arc by bearing. Intersect the arc of radius R through point J with the line through K at bearing B in quadrant QUAD.

Note: the "/" symbol means "Enter/." A semicolon means the items of input are prompted in separate groups. (dd) means "decimal degrees" (flag 03 clear); (DMS) means "Degrees-Minutes-Seconds" (flag 03 set).

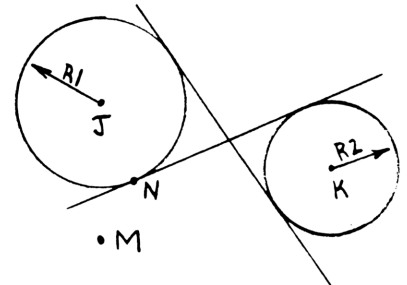
Note: The following three arc commands have two possible solutions; the routines will pick the one closest to point M to store as point N (the solved point). If the prompt for M is bypassed, both solutions will be stored as N1 and N2.

If the prompt for a radius is answered as a negative number (e.g. -12) it is taken as point #12 and the radius to be used is the distance from that point to the center point.

COMMAND INPUT

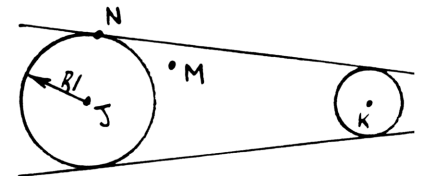
INTAN J/R1/K/R2; M; N
 J/-P1/K/-P2; M; N

Internal tangency. Finds the two points of internal tangency which are on the circle about point J.



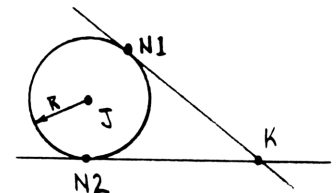
EXTAN J/R1/K/R2; M; N
 J/-P1/K/-P2; M; N

External tangency. Finds the two points of external tangency which are on the circle about point J.



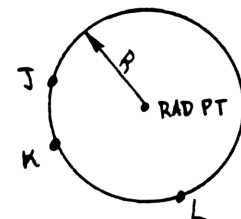
PTAN J/R/K; M; N
 J/-P/K; M; N

Tangents through a point. Finds the points of tangency to the circle about point J of the two lines through point K.



C3P J/K/L; RAD PT

Circle from three points. Finds the radius and sets the center point RAD PT of the circle through any three non-collinear points.



Note: the "/" symbol means "Enter/." A semicolon means the items of input are prompted in separate groups. (dd) means "decimal degrees" (flag 03 clear); (DMS) means "Degrees-Minutes-Seconds" (flag 03 set). Clockwise angles are positive.

COMMAND INPUT

CHC PC/RP/PT

Coordinate horizontal curve. Calculates the following horizontal curve data from the three points (point of curvature, radius point, and point of tangency):

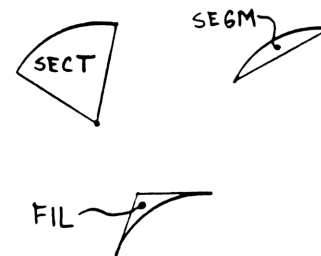
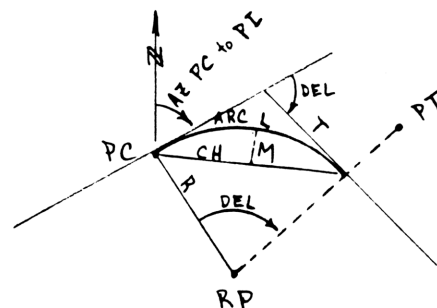
Distance PC to RP
Distance PT to RP
Azimuth of incoming tangent (PC to PI)
Azimuth of outgoing tangent (PI to PT)

RAD Radius, taken as PC to RP
DEG Degree of curve
DEL Delta or central angle
ARC L Arc length
T Tangent length (PC to PI)
CH Chord length
M Middle ordinate (arc to chord)
E External (arc to PI)

SECT Sector area
SEGM Segment area
FIL Fillet area

If the distances PC to RP and PT to RP aren't equal, the program displays a warning and uses the former as the radius. In this case the PT input is used only to find the angle subtended (Delta).

For more information see the documentation for program HC, of which CHC is a part.



Note: the "/" symbol means "Enter/." A semicolon means the items of input are prompted in separate groups. (dd) means "decimal degrees" (flag 03 clear); (DMS) means "Degrees-Minutes-Seconds" (flag 03 set). Clockwise angles are positive.

COMMAND INPUT

ADJ BEG PT; END PT; DIST'; PT#; PT#; PT#; etc.

Traverse adjustment by compass rule. Adjusts all point coordinates of a traverse proportionally to make the END PT close on the BEG PT (i.e. to make the END PT coordinates the same as the BEG PT coordinates).

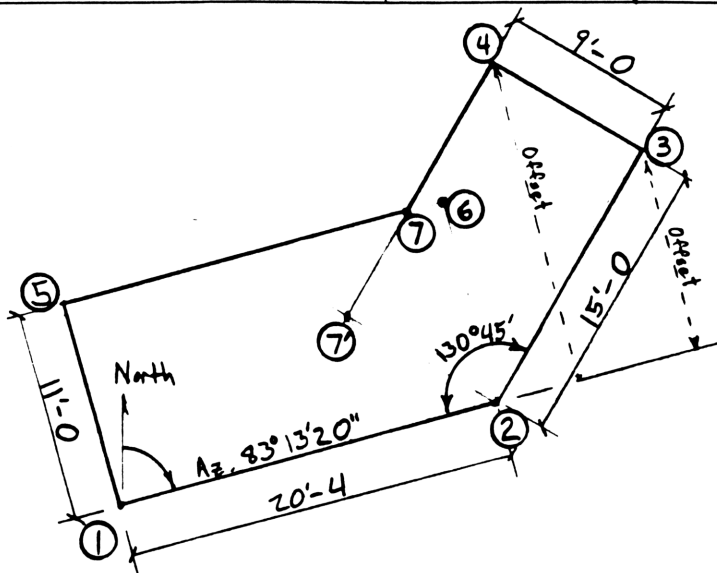
The Σ DIST' prompt asks for the total distance around the traverse, from BEG PT (beginning point, P.O.B.) to END PT. If this distance is known, simply key it in and go on. If not (the usual case), one of three routines may be run before the ADJ adjustment to find the distance. If the points in the traverse are being set by LAZ or LBR from field notes, the distance around the traverse is accumulated (in register 00). If the points are already set, routine INV may be used to inverse around the loop; it also accumulates the distance. If using any of these three, they must run the entire traverse without stopping, since re-executing them zeroes the distance accumulated to start over. Once one of these three routines has been run from the BEG PT to the END PT, xeq ADJ, input the beginning and ending points, then bypass the Σ DIST' prompt (press R/S without keying anything in). The correct value will be inserted.

After the Σ DIST' prompt, ADJ begins prompting for points in the traverse. The first PT# prompt is the second point in the traverse, and so on; continue answering the PT# prompts until you've keyed in the END PT. Respond to the next PT# prompt with a 0 (zero), or just XEQ your next command (typically a DUMP to list the adjusted points). See the example. It is important when keying in the PT#'s not to disturb the stack, which carries needed data. Do not use the Enter/ key; and if an incorrect number is keyed in and R/S has not yet been pressed, use RDN (rolldown) or CLX to remove it before keying in the correct number. Unfortunately, ADJ is very unforgiving of mistakes made during PT# entry, and since the points are altered as one goes around the traverse, it's also impossible to start over. Thus it is a very good idea to duplicate the original coordinates in a safe block of registers with routine DUP in order to provide a method of error recovery. DUP does not use register 00, so it does not interfere with the storage of a Σ DIST' value by LBR, LAZ, or INV as long as one of these three routines is run completely around the traverse first.

—

AREA PT1; PT2; PT3; PT4; etc.

Calculation of area bounded by any number of points (greater than two). Input points bounding the area in a clockwise direction. The same caution applies to the stack and the Enter/ key as for routine ADJ. However, if an error is discovered after pressing R/S but before keying in the next point number, use softkey C (with USER mode on) to correct the error. See the example. To end input and calculate the area, key in 0 (zero).



a) Angles in Degrees-Minutes-Seconds
 ⇒ Set flag 03

Desire North/East format
 ⇒ Clear flag 01

b) Set point *1 : N = 20.00, E = 500.00

STOR 1 20 500 (R/S)

c) Set point *2 @ Dist = 20'-4, Az. 83°13'20"

LAZ 1 20.333 83.1320 ; 2

d) Set point *3 from *2 by backsighting on *1

LAN 2 1 15.0' 130.45 ; 3

e) Set points *5 and *6 parallel to *1-*2 ;
 set points *4 and *7 parallel to *3-*2

PRL 1 2 -11 (to the left) ; 5 ; 6
 2 3 -9 (-) ; 7 ; 4

f) Intersect lines *5-*6 and *4-*7 to set a new point *7:

IPT 5 6 4 7 ; 7

g) Compute distances *5-*7 and *7-*4, and area ;
 and list point coordinates and remaining free points.

SF 03 Set flag
 03 for
 D.M.S

XEQ "STOR"

ITK CLR 1.0000 ENTER↑ Initialize
 10.0000 RUN Pts. *1-10

11:07:25 AM 09/06/88

TITLE:
 ABUTMENT FOOTING

RUN

PT↑N↑E 1.0000 ENTER↑ STOR
 20.0000 ENTER↑ Pt *1
 500.0000 RUN

PT↑N↑E XEQ "LAZ"

J↑D↑A (DMS) 1.0000 ENTER↑ From
 20.3333 ENTER↑ Pt *1,
 83.1320 RUN go 20.33'
 @ Az 83°13'20"

N 2.0000 RUN to set
 Pt *2

PT2: N=22.3997
 E=520.1912

J↑D↑A (DMS) XEQ "LAN"

J↑K↑D↑A (DMS) 2.0000 ENTER↑ From
 1.0000 ENTER↑ Pt *2,
 15.0000 ENTER↑ backsight on *1,
 130.4500 RUN go 15.0'
 turning
 130°45'

N 3.0000 RUN to set
 Pt *3

PT3: N=34.8393
 E=528.5731

J↑K↑D↑A (DMS)

XEQ "PRL"

J↑K↑D

1.0000 ENTER↑

2.0000 ENTER↑

11.0000 RUN

L

5.0000 RUN

PT5: N=9.0769

E=501.2982

M

6.0000 RUN

PT6: N=11.4766

E=521.4894

J↑K↑D

1.0000 ENTER↑

2.0000 ENTER↑

-11.0000 RUN

L

5.0000 RUN

USED-NEW PT?

RUN

PT5: N=30.9231

E=498.7018

M

6.0000 RUN

USED-NEW PT?

RUN

PT6: N=33.3228

E=518.8930

J↑K↑D

2.0000 ENTER↑

3.0000 ENTER↑

-9.0000 RUN

L

7.0000 RUN

PT7: N=27.4288

E=512.7275

M

4.0000 RUN

PT4: N=39.8685

E=521.1093

J↑K↑D

Oops - went
the wrong
way (should
have been
-11.0')

Do over

Bypass to
overwrite

Bypass

XEQ "IPT"

J↑K↑L↑M

5.0000 ENTER↑

6.0000 ENTER↑

4.0000 ENTER↑

7.0000 RUN

N

7.0000 RUN

USED-NEW PT?

RUN

PT7: N=33.0394

E=516.5078

J↑K↑L↑M

XEQ "INV"

J↑K

5.0000 ENTER↑

7.0000 RUN

D=17.9314

AZ=83.1320 DMS

J↑K

4.0000 RUN

D=8.2347

AZ=33.5820 DMS

J↑K

XEQ "OFPT"

J↑K↑P

1.0000 ENTER↑

2.0000 ENTER↑

4.0000 RUN

OFFSET=17.2383'

N

0.0000 RUN

PT0: N=22.7506

E=523.1438

J↑K↑P

1.0000 ENTER↑

2.0000 ENTER↑

3.0000 RUN

OFFSET=11.3635'

N

0.0000 RUN

PT0: N=23.5553

E=529.9142

Overwrite the
old P+7
with the
new one.

From #5
to #7

From #7 (default)
to #4

Find offset
of pt #4 from
line #1-#2

Set "dummy"
point @ offset
location on
line #1-#2 to
obtain coordinates

Offset of
point #3 from
line #1-#2

B56

Note that the first use (the "xeg") of each routine prints with a line of asterisks to make the output easier to follow.

JTK+P
5.0000 ENTER+
6.0000 ENTER+
4.0000 RUN

OFFSET=6.2383'

N

RUN

JTK+P

XROM "AREA"

PT1

1.0000 RUN

PT2

2.0000 RUN

PT3

3.0000 RUN

PT4

4.0000 RUN

PT5

5.0000 RUN

PT6

XEQ C

PT5

7.0000 RUN

PT6

6.0000 RUN

PT7

XEQ C

PT6

5.0000 RUN

PT7

0.0000 RUN

AREA=-315.0119 SQ'

PT1

Offset of
Pt*4 from
line *5-*6

Bypass to avoid
getting any point

Find area

Oops- the fifth
point should be
#7

Oops again- use
softkey "C" to
correct (USER mode on).

End with Pt*0;

Points were
input counter-
clockwise so
area displays
as negative.

XROM "FREE"

FREE PTS:

8

9

10

Have three
points unused.

XEQ "DUMP"

11:17:47 AM 09/06/88

TITLE:

ABUTMENT FOOTING RUN

ITK

1.0000 ENTER+

10.0000 RUN

PT1: N=20.0000

E=500.0000

PT2: N=22.3997

E=520.1912

PT3: N=34.8393

E=528.5731

PT4: N=39.8685

E=521.1093

PT5: N=30.9231

E=498.7018

PT6: N=33.3228

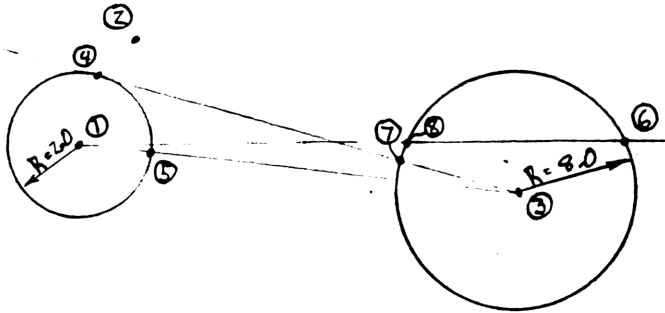
E=518.8930

PT7: N=33.0394

E=516.5078

ITK

Points 8-10
are not
listed since
they are
unused



SF 01

Use (X,Y) rather
than (North, East)

Initialize 9 points

oops

XEQ "STOR"

ITK CLR

```
1.0000 ENTER↑
1.0000 CLX
9.0000 RUN
```

4:57:51 PM 08/30/88

TITLE:

CURVES

RUN

PT↑X↑Y

```
1.0000 ENTER↑
5.0000 ENTER↑
15.0000 RUN
```

PT↑X↑Y

```
2.0000 ENTER↑
6.0000 ENTER↑
18.0000 RUN
```

PT↑X↑Y

```
3.0000 ENTER↑
12.0000 ENTER↑
14.0000 RUN
```

PT↑X↑Y

XEQ "PTAN"

J↑R↑K

```
1.0000 ENTER↑
2.0000 ENTER↑
3.0000 RUN
```

M

2.0000 RUN

N

4.0000 RUN

PT4: X=5.8313

Y=16.8191

J↑R↑K

XEQ "APT"

J↑R↑K↑L

```
1.0000 ENTER↑
2.0000 ENTER↑
1.0000 ENTER↑
3.0000 RUN
```

M

3.0000 RUN

N

5.0000 RUN

PT5: X=6.9799

Y=14.7172

J↑R↑K↑L

```
3.0000 ENTER↑
8.0000 ENTER↑
3.0000 ENTER↑
4.0000 RUN
```

M

4.0000 RUN

N

11.0000 RUN

NONEXISTENT

XEQ "APT"

J↑R↑K↑L

```
3.0000 ENTER↑
8.0000 ENTER↑
3.0000 ENTER↑
4.0000 RUN
```

M

4.0000 RUN

N

7.0000 RUN

PT7: X=4.7238

Y=17.3252

J↑R↑K↑L

XEQ "AAZ"

J↑R↑K↑A (dd)

```
3.0000 ENTER↑
8.0000 ENTER↑
1.0000 ENTER↑
85.0000 RUN
```

M

RUN

N1

6.0000 RUN

PT6: X=19.6673

Y=16.2832

N2

8.0000 RUN

PT8: X=4.0527

Y=14.9171

J↑R↑K↑A (dd)

XEQ "FREE"

FREE PTS:

9

Pt #9 still
availableIntersect the circle about
Point #3, of radius 8.0,
with the line through
points #3 and #4Of the two solutions, pick
the one closest to Pt #4.

Call it Pt #11

Oops - must re-execute -
not enough data registers
to store Pt #11

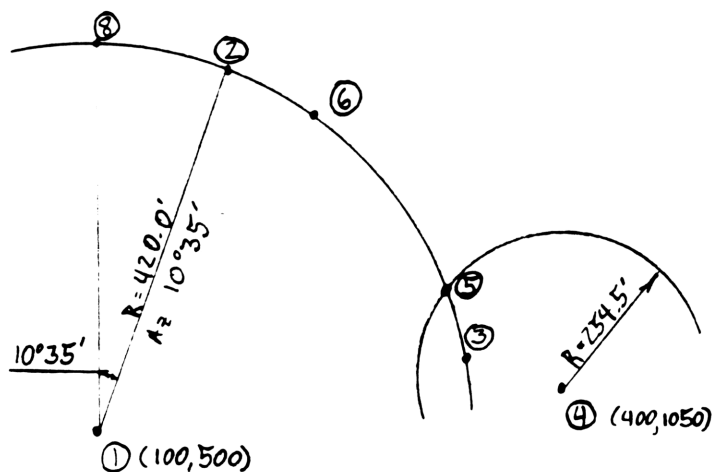
Store as Pt #7

Find the tangent point of the
line through Pt #3, to the circle
about Pt #1 of radius 2.0; of
the two solutions, pick the
one closest to Pt #2.

B58

Arc Example

1/2



a) Set point *1: STOR 1 100 500

b) Set point *2 from *1 @ R=420'
Az = 10°35':

LAZ 1 420 10°35'; 2

c) Set point *3 from *2, 300.0' along the arc about *1 (clockwise = +)

ATRV 2 1 300'; 3

d) Store coordinates of point *4
(bypass initializing sequence): N=400
E=1050

STOR (bypass CLR) 4 400 1050

e) Set point *5 @ intersection of arc about *4 @ R=254.5', with arc about *1 @ R=420.0' (radius of *3) pick the point closest to *2

ARCS 1 3 4 254.5; 2; 5
(3, CHS)

f) Set point *6 1/4 of the way between *2 and *3 along the arc about *1.

AFRC 2 1 3 .25; 6

g) Set point *8, 10°35' counter-clockwise (-) from point *2 about *1

ATVA 2 1 -10.35; 8
(10.35, CHS)

h) Find remaining free points and run out curve data from *3 to *2 about *1

CF 01
SF 03

Work in (North, East)
Work in D.M.S

XEQ "STOR"

ITK CLR

1.0000 ENTER
9.0000 RUN

Initialize
points 1-9

5:30:47 PM 08/30/88

TITLE:
ARCS

RUN

PT↑N↑E

1.0000 ENTER
100.0000 ENTER
500.0000 RUN

Store Pt=1

PT↑N↑E

XEQ "LAZ"

J↑D↑A (DMS)

1.0000 ENTER
420.0000 ENTER
10.3500 RUN

Set Pt=2

N

2.0000 RUN

PT2: N=512.8553
E=577.1395

J↑D↑A (DMS)

XEQ "ATRV"

J↑C↑ARC L

2.0000 ENTER
1.0000 ENTER
300.0000 RUN

Set Pt=3

N

3.0000 RUN

PT3: N=361.4051
E=828.7360

J↑C↑ARC L

359

XEQ "STOR"
 I↑K CLR
 RUN
 PT↑N↑E
 4.0000 ENTER↑
 400.0000 ENTER↑
 1,050.0000 RUN

 PT↑N↑E
 XEQ "ARCS"

 J↑R1↑K↑R2
 1.0000 ENTER↑
 -3.0000 ENTER↑
 4.0000 ENTER↑
 254.5000 RUN
 M
 2.0000 RUN
 N
 5.0000 RUN

PT5: N=398.4577
 E=795.5047

 J↑R1↑K↑R2
 XEQ "AFRC"

 J↑C↑K↑FRC
 2.0000 ENTER↑
 1.0000 ENTER↑
 3.0000 ENTER↑
 .2500 RUN
 N
 6.0000 RUN

PT6: N=492.5884
 E=649.2458

 J↑C↑K↑FRC
 XEQ "ATVA"

 J↑C↑A (DMS)
 2.0000 ENTER↑
 1.0000 ENTER↑
 -10.3500 RUN
 N
 8.0000 RUN

PT8: N=520.0000
 E=500.0000

 J↑C↑A (DMS)
 XEQ "FREE"

 FREE PTS:
 7
 9

XROM "CHC"
 5:45:03 PM 08/30/88

TITLE:
 420 FT RAD. RUN

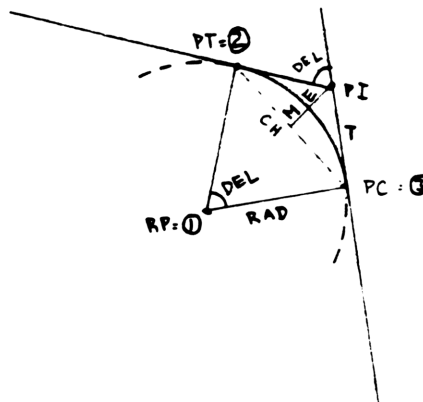
 PC↑RP↑PT

3.0000 ENTER↑
 1.0000 ENTER↑
 2.0000 RUN
 PC-RP=420.0000'
 PT-RP=420.0000'
 PC-PI AZ=-38.2928 DMS
 PI-PT AZ=-79.2500 DMS

RAD=420.0000'
 DEG=13.3831 DMS
 DEL=40.5532 DMS
 ARC L=300.0000'
 T=156.7207'
 CH=293.6630'
 M=26.5022'
 E=28.2871'

 SECT=63,000.0000 SQ'
 SEGM=5,222.1295 SQ'
 FIL=2,822.6734 SQ'

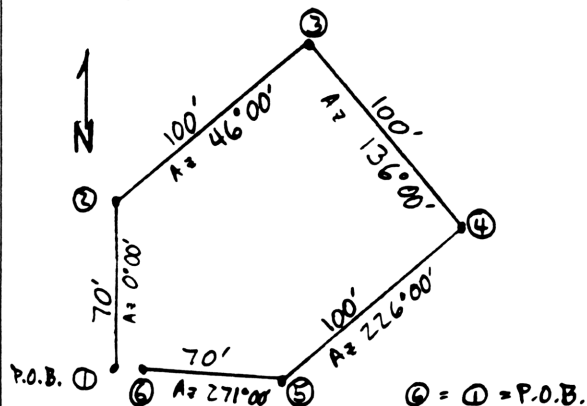
 PC↑RP↑PT



TRAVERSE ADJUSTMENT

1/4

Traverse this circuit and adjust points by the compass rule to close.



Perimeter (ZDIST) = 440.0'

Traversing from ① (30,530) to ⑥ (29.2577, 529.4765)

Error in Northing: $30.0 - 29.2577 = \underline{.7423'}$
 Error in Easting: $530.0 - 529.4765 = \underline{.5235'}$

req STOR

ITK CLR 1, 6

PTAN+E 1, 30, 530

req LBR

JTD+B (dd); QUAD; N

1, 70', 0°; 1; 2

2, 100', 46°; 1; 3

3, 100', 44°; 2; 4

4, 100', 46°; 3; 5

5, 70', 49°; 4; 6

req ADJ

BEG PT 1

END PT 6

ZDIST' (R/S) (default from LBR)

PT# 2

PT# 3

etc.

ITK CLR XEQ "STOR"

1.0000 ENTER↑

6.0000 RUN

SIZE<14

SIZE 014

XEQ "STOR"

ITK CLR

1.0000 ENTER↑

6.0000 RUN

4:37:43 PM 08/29/88

TITLE:

TRAVERSE ADJUSTMENT RUN

PTAN+E

1.0000 ENTER↑

30.0000 ENTER↑

530.0000 RUN

PTAN+E

XEQ "LBR"

JTD+B (dd)

1.0000 ENTER↑

70.0000 ENTER↑

0.0000 RUN

QUAD

1.0000 RUN

N

2.0000 RUN

PT2: N=100.0000

E=530.0000

JTD+B (dd)

2.0000 ENTER↑

100.0000 ENTER↑

46.0000 RUN

QUAD

1.0000 RUN

N

3.0000 RUN

PT3: N=169.4658

E=601.9340

JTD+B (dd)

3.0000 ENTER↑

100.0000 ENTER↑

44.0000 RUN

QUAD

2.0000 RUN

N

4.0000 RUN

PT4: N=97.5319

E=671.3998

Initialize points 1-6

Oops! Not enough room - reSIZE

Try again

Initialize

Title

STOR coord-
inates of P.O.B.

N = 30.0
E = 530.0

Begin traverse

From pt. #1,
go 70.0' at
a bearing of
000° in Quad. 1
(Northeast)
to set point #2

Coord. of
point #2

(B61)

TRAVERSE ADJUSTMENT

2/4

JTD+B (dd)

4.0000 ENTER+
100.0000 ENTER+
46.0000 RUN

QUAD

3.0000 RUN

N

5.0000 RUN

PT5: N=28.0660

E=599.4658

JTD+B (dd)

5.0000 ENTER+
70.0000 ENTER+
89.0000 RUN

QUAD

4.0000 RUN

N

6.0000 RUN

PT6: N=29.2877

E=529.4765

JTD+B (dd)

XEQ "ADJ"

BEG PT

1.0000 RUN

END PT

6.0000 RUN

Σ DIST

RUN

PT#

2.0000 RUN

PT#

3.0000 RUN

PT#

4.0000 RUN

PT#

5.0000 RUN

PT#

6.0000 RUN

PT#

0.0000 RUN

BEG PT

Point #6 didn't quite
close on point #1

Bypass since ΣDIST stored
by LBR run

Begin with first point to
be adjusted - proceed
around traverse

Signify end of
traverse with point
#0

XEQ "DUMP"

7:36:38 AM 08/16/88

TITLE:

ADJUSTED POINTS RUN

ITK

1.0000 ENTER+

6.0000 RUN

PT1: N=30.0000

E=530.0000

PT2: N=100.1133

E=530.0833

PT3: N=169.7410

E=602.1362

PT4: N=97.9690

E=671.7211

PT5: N=28.6650

E=599.9061

PT6: N=30.0000

E=530.0000

ITK

XROM "AREA"

PT1

1.0000 RUN

PT2

2.0000 RUN

PT3

3.0000 RUN

PT4

4.0000 RUN

PT5

5.0000 RUN

PT6

0.0000 RUN

AREA=12,443.9676 SQ'

PT1

Adjusted
Points -
#6 = #1

Find Area
of adjusted
traverse

Signify end of
traverse with
point #0

B62

Traversing from ① around the loop to close @ ⑥. From the azimuths and distances measured, the coordinates of the points are calculated and the error of closure is found:

	North	East
Pt ①	30.00	530.00
Pt ⑥	29.2877	529.4765
error	0.7123'	0.5235'

Adjust the coordinates to close the traverse (by Compass Rule).

TRAVERSE ADJUSTMENT

Point	Dist.	North	Correction to North	Accum.	Adj. N	East	Correction East	Accum.	Adj. Easting	Point
1	70'	30.00	—	0	30.00	530.00	—	0	530.00	1
2	100'	100.00	$\frac{70'}{440'} \times .7123' = .1133'$.1133	100.1133	530.00	$\frac{70'}{440'} \times .5235' = .0833$.0833	530.0833	2
3	100'	169.4654	$\frac{100'}{440'} \times .7123' = .1619'$.2752	169.7410	601.9340	$\frac{100'}{440'} \times .5235' = .1190$.2023	602.1363	3
4	100'	97.5319	.1619'	.4371	97.9680	671.3996	.1190	.3212	671.7210	4
5	100'	29.0660	.1619'	.5990	28.6650	599.4656	.1190	.4402	599.9060	5
6	70'	29.2877	.1133'	.7123	30.00	529.4765	.0833	.5235	530.00	6

TRAVERSE ADJUSTMENT

4/4

If coordinates are
already stored, use
INV to accumulate
distance :

XEQ "INV"

JTK

1.0000 ENTER↑
2.0000 RUN

D=70.0000
AZ=0.0000 dd

JTK

3.0000 RUN

D=100.0000
AZ=46.0000 dd

JTK

4.0000 RUN

D=100.0000
AZ=136.0000 dd

JTK

5.0000 RUN

D=100.0000
AZ=-134.0000 dd

JTK

6.0000 RUN

D=70.0000
AZ=-89.0000 dd

JTK

XEQ "ADJ"

BEG PT

1.0000 RUN

END PT

6.0000 RUN

Σ DIST

—

PT#

2.0000 RUN

PT#

3.0000 RUN

PT#

4.0000 RUN

PT#

5.0000 RUN

PT#

6.0000 RUN

PT#

0.0000 RUN

BEG PT

from #2 to #3
(default "J" is
previous "K")

Last leg is
from #5 to #6

Bypass

First point to
be corrected

Last point to be
adjusted

Signify end with
Pt = 0

XEQ "DUMP"

4:49:00 PM 08/29/88

TITLE:

ADJUSTED POINTS RUN

ITK

1.0000 ENTER↑

6.0000 RUN

PT1: N=30.0000—

E=530.0000

PT2: N=100.1133

E=530.0833

PT3: N=169.7410

E=602.1362

PT4: N=97.9690

E=671.7211

PT5: N=28.6650

E=599.9061

PT6: N=30.0000

E=530.0000

ITK

XROM "AREA"

PT1

1.0000 RUN

PT2

2.0000 RUN

PT3

3.0000 RUN

PT4

6.0000 RUN

PT5

XEQ C

PT4

4.0000 RUN

PT5

5.0000 RUN

PT6

6.0000 RUN

PT7

XEQ C

PT6

0.0000 RUN

AREA=12,443.9676 SQ'

PT1

Oops

Correct with
soft key "C"

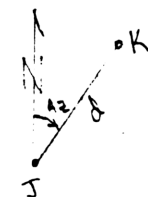
Having a bad
day - end
with Pt = 0
(do not use
Pt = 6 since
it is now
identical to
Point #1)

B64

	X	Y	Z	T	L		X	Y	Z	T	L
						33 "USED-NEW PT#"					
01*LBL "STOR"						54 CF 22					
02 "I+K CLR"						55 FC? 10					
03 CLX						56 FS? 00					
04 PROMPT	K	I				57 FS? 53					
05 X=0?						58 PROMPT	NewPt#	Y	X		
06 GTO 10						59 FS? 22					
07 ENTER↑						60 GTO 30					
08 ST+ X	ZK	K	I			61 X<>Y	X	Y			
09 E	I	ZK	K	I		62 STO IND L	X				ZP
10 +						63 ISG L					
11 ADV	ZK+1	K	I			64 "	"NoP"				
12 XROM "S"	K	I				65 X<>Y	Y	X			
13 E3						66 STO IND L	Y				ZP+1
14 /						67 LASTX	ZP+1				
15 +	I.00K					68 E	I	ZP+1	Y	X	
16 E-5						69 -					
17 +						70 2					
18 ST+ X	ZI.00Z					71 /	P	Y	X	X	
19 E9						72 RTN					
20 CHS	-1x10 ⁹ counter										
						73*LBL 16					
21*LBL 04						74 FC? 22					
22 STO IND Y						75 RTN	Y	X			
23 ISG Y						76 X>0?					
24 GTO 04						77 XEQ 30					
						78 ADV					
25*LBL 10											
26 "PT+NE"	E	N	Pt"			79*LBL 56					
27 FS? 01						80 "PT"					
28 "PT+X+Y"	Y	X	Pt"			81 RCL d	Flags	Pt"	Y	X	If Pt# > 0, store the coords.
29 PROMPT						82 FIX 0					
30 FS? 01						83 CF 29					
31 X<>Y	X	Y	Pt"			84 ARCL Y					
32 X<> Z	Pt"	Y	X			85 STO d					
33 XEQ 30						86 RDN	P	Y	X		
34 XROM "L"						87 FS? 01					
35 GTO 10						88 GTO 00					
						89 "t: N="					
36*LBL 30						90 ARCL Y					
37 SF 21						91 AVIEW					
38 FC? 02						92 " E="					
39 CF 26						93 ARCL Z					
40 ST+ X	ZP	Y	X			94 AVIEW					
41 SIGN	I					95 XROM "L"					
42 RDN	Y	X				96 RTN	P	Y	X		
43 CF 10											
44 RCL IND L	X'										
45 E9											
46 CHS											
47 X=Y?	-1x10 ⁹ clear?	X'	Y	X							
48 SF 10											
49 X≠Y?											
50 BEEP											
51 R↑	Y	X									
52 R↑											

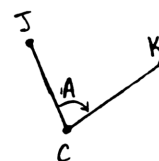
	X	Y	Z	T	L		X	Y	Z	T	L
97*LBL 00						149 RDN					
98 "F: X="	X	-1*10 ⁹	P _F	Y		150 +	ZI .002K+1				
99 ARCL Z						151 X<> 00	ZN .00#				
100 AVIEW						152 STO 01					
101 " Y="						153 E-3	.001				
102 ARCL Y						154 +					
103 AVIEW						155 LASTX	.001	ZN .00#	ZI		
104 XROM "L"						156 *					
105 RTN						157 +	ZI .002N 30#				
106*LBL "INV"						158 SF 25					
107 SF 14						159 REGMOVE					
108 CLX						160 FS?C 25					
109 STO 00						161 GTO 03					
110*LBL 07						162*LBL 76					
111 XROM "L"						163 RCL IND 00	X ₀₁ Y				
112 "J+K"						164 STO IND 01	X ₀₁ Y				
113 PROMPT	K	J				165 ISG 01					
114 STO 01	K					166 "					
115 XROM "XY"	ΔX	ΔY	X _J	Y _J		167 ISG 00					
116 X<>Y	ΔY	ΔX				168 GTO 76					
117 R-P	d	Az				169 GTO 03					
118 ST+ 00						170*LBL "ANGL"					
119 "D="						171 SF 14					
120 ARCL X						172*LBL 11					
121 AVIEW						173 XROM "L"					
122 X<>Y	Az	d				174 "J+K"	K	C	J		
123 "AZ"						175 PROMPT					
124 XROM "Dd"						176 STO 01	K				
125 AVIEW						177 RDN					
126 RCL 01	K	Az	d			178 STO 00	C	J			
127 GTO 07						179 XROM "XY"	ΔX_J	ΔY_J			
128*LBL "DUP"						180 R-P	d _J	θ_J			
129 SF 14						181 RDN	θ_J				
130*LBL 03						182 X<> 01	K				
131 XROM "L"						183 RCL 00	C	K			
132 "I+K+NEW I"						184 XROM "XY"	ΔX_K	ΔY_K			
133 PROMPT	New I	K	I			185 R-P	d _K	θ_K			
134 2	Z	N	K	I		186 RDN	θ_K				
135 ST+ T						187 RCL 01	$\Delta \theta$				
136 ST+ Z						188 -					
137 *						189 "A"					
138 STO 00	ZN	ZK	ZI			190 XROM "Dd"					
139 X<>Y	ZN					191 AVIEW					
140 E	I	ZK	ZN	ZI		192 GTO 11					
141 +						193*LBL "LLI"					
142 STO Y	ZK+1	ZK+1	ZI	ZI		194 XROM "L"					
143 R↑	ZI					195 SF 08					
144 -	*reg.	ZK+1	ZI	ZI		196 GTO 09					
145 E3	1000	*reg.	ZK+1	ZI		197*LBL "PART"					
146 ST/ Z						198 XROM "L"					
147 /	.00#	ZK+1	ZI	ZI		199 CF 08					
148 ST+ 00											

initialize 8.00

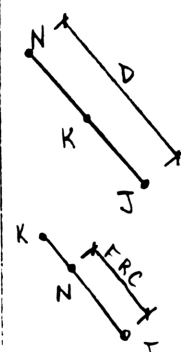


accum. "d"

Use REGMOVE
if available

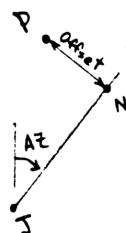
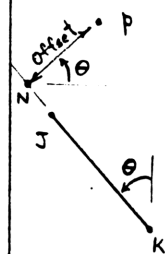


θ_J in R.01

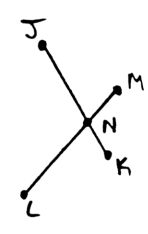
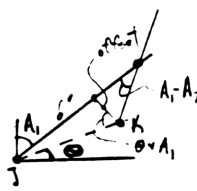


B66

	X	Y	Z	T	L		X	Y	Z	T	L
200*LBL 09											
201 *JTK*											
202 FC? 08											
203 *FRC*	FRC	K	J								
204 FS? 08											
205 *ID*	D	K	J								
206 PROMPT											
207 STO 00											
208 RDN	D or FRC										
209 XROM "XY"	K	J									
210 R-P	ΔX	ΔY	X_J	Y_J							
211 FC? 08	d_K	θ									
212 ST* 00											
213 X(> 00	d_N	θ	X_J	Y_J							
214 P-R	ΔX_N	ΔY_N									
215 X(>Y	ΔX_N	ΔY_N	X_J	Y_J							
216 *N*											
217 XROM "SP"											
218 GTO 09											
219*LBL "XY"	K	J									
220 XROM "P"	Y_K	X_K	J								
221 X(> Z	J	X_K	Y_K								
222 XROM "P"	Y_J	X_J	X_K	Y_K							
223 RDN	X_J	X_K	Y_K	Y_J							
224 -	ΔX										
225 LASTX	X_J	ΔX	Y_K	Y_J							
226 X(> Z	Y_K	ΔX	X_J	Y_J							
227 R↑											
228 -	ΔY	ΔX	X_J								
229 X(>Y	ΔX	ΔY	X_J								
230 LASTX	Y_J										
231 RDN											
232 RTN	ΔX	ΔY	X_J	Y_J							
233*LBL "FS"											
234 *T SQ*											
235*LBL "FT"											
236 *T*											
237 RTN											
238*LBL "DUMP"											
239 E	I										
240 XROM "S"											
241*LBL 02											
242 *ITK*	K	I									
243 PROMPT	1000										
244 E3											
245 /											
246 +	I.00K										
247 ADV											
248*LBL 51											
249 ENTER↑	P_1^*	P_1^*									
250 XROM "P"	Y	X	P_1^*								
251 RCL Z	P_1^*	Y	X								
252 XEQ 56	P_1^*	Y	X								
253 ISG X	P_1^*+1										
254 GTO 51											
255 GTO 02											
256*LBL "OFPT"											
257 XROM "L"											
258*LBL 14											
259 SF 08											
260 *JTK*P*											
261 PROMPT	P	K	J								
262 STO 00	P	K	P								
263 X(> Z	J										
264 STO 01	J										
265 XROM "XY"	ΔX	ΔY									
266 X(>Y	ΔY	ΔX									
267 R-P	d	θ									
268 RCL 01	J										
269 RCL 00	P	J	d	θ							
270 R↑	θ	P	J								
271 XEQ 66											
272 GTO 14											
273*LBL "OFAZ"											
274 XROM "L"											
275*LBL 64											
276 *JTPAZ*											
277 XROM "Pd"	A_Z	P	J								
278 XEQ 66											
279 GTO 64											
280*LBL "OFBR"											
281 XROM "L"											
282*LBL 65											
283 *JTP*											
284 XROM "BA"	A_Z	P	J								
285 XEQ 66											
286 GTO 65											
287*LBL "IAZ"											
288 XROM "L"											
289*LBL 13											
290 CF 08											
291 *JTA1TKTA2*											
292 XROM "Pd"	A_K	K	A_J	J							
293 X(> Z	A_J	K	A_K	J							



	X	Y	Z	T	L		X	Y	Z	T	L
294 FS? 03						344*LBL 00	d	0	X _J	Y _J	
295 HR						345 X<> 00	A ₂	0	X _J	Y _J	d in R.00
296 X<> Z	A _K	K	A _J	J		346 +					
297 XEQ 01						347 LASTX	A ₂	0+A ₂	X _J	Y _J	
298 GTO 13	N	Y _N	X _N	Reg'd		348 X<> 00	d	0+A ₂			A ₂ in R.00
						349 P-R	l	m			
299*LBL "IBR"						350 X<>Y					
300 XROM "L"						351 RDN	l	X _J	Y _J		l in R.00
						352 X<> 00	A ₂				
301*LBL 43						353 RCL 01	A ₁				
302 CF 08						354 -	A ₂ -A ₁				
303 -J						355 CHS	A ₁ -A ₂				
304 XROM "BA"						356 SIN	sin				
305 -K						357 ST/ 00					10000 D in R.00
306 XROM "BA"						358 RDN	X _J	Y _J			
307 XEQ 01						359 RCL 01	A ₁				
308 GTO 43						360 RCL 00	D	A ₁			
						361 P-R	0Y	0X	X _J	Y _J	
						362 GTO 00					
309*LBL 66	A ₂	P	J								
310 SF 08	90	A ₂	P	J		363*LBL 06	A ₂	D	J		
311 90						364 X<>Y	D	A ₂	J		
312 X<>Y	A ₂	90				365 ST+ 00	accum.				
313 -	90-A ₂					366 P-R	0Y	0X	J		
314 CHS	A ₂ -90	P	J	J		367 X<> Z	J	0X	0Y		
315 R↑	J					368 XROM "P"	Y _J	X _J	0X	0Y	
316 LASTX	A ₂	J	A ₂ -90	P							
317 R↑						369*LBL 00					
318 R↑	A ₂ -40	P	A ₂	J		370 -N					
319*LBL 01	A ₂	K	A ₁	J		371*LBL "SP"	0Y	0X	X _J	Y _J	
320 STO 00	A ₂					372 ST+ T	0Y	0X	X _J	Y _N	
321 RDN	K	A ₁	J			373 RDN	0X	X _J	Y _N	0Y	
322 X<>Y	A ₁	K	J			374 +	X _N	Y _N	0Y	0Y	
323 STO 01	A ₁					375 X<>Y	Y _N	X _N			
324 RDN	K	J				376 CF 22					
325 XROM "XY"	0X	0Y	X _J	Y _J		377 PROMPT	N	Y _N	X _N		
326 R-P	d	0				378 GTO 16					
327 FC? 08											
328 GTO 00						379*LBL "IPT"					
329 STO I	d					380 XROM "L"					
330 RDN	0	X _J	Y _J	d							
331 STO a	0										
332 RCL 01	A ₁					381*LBL 18					
333 +	0+A ₁					382 CF 08					
334 RCL I	l					383 -J+K+L+M					
335 P-R	offset	d	X _J	Y _J		384 PROMPT	M	L	K	J	
336 "OFFSET="						385 STO I	M				
337 ARCL X						386 RDN					
338 -t						387 STO 01	L				
339 AVIEW						388 RDN					
340 R↑						389 STO 00	K	J	M	L	
341 R↑	X _J	Y _J				390 XROM "XY"	0X	0Y	X _J	Y _J	
342 RCL a	0					391 X<>Y	0Y	0X	A ₂		
343 LASTX	d	0	X _J	Y _J		392 R-P	d				
						393 X<>Y	A ₂				
						394 X<> I	M				
						395 RCL 01	L	M			

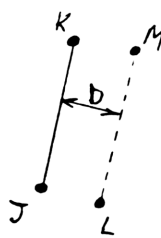
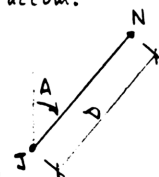


368

A₂ in R.M

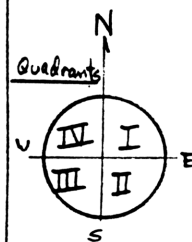
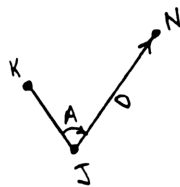
	X	Y	Z	T	L		X	Y	Z	T	L
396 XROM "XY"	ΔX	ΔY				449 LBL "LAZ"					
397 X<Y	ΔY	ΔX				450 XROM "L"					
398 R-P	d	Az				451 CLX					
399 X< 00	K	Az _{LM}				452 STO 00					
400 RCL [Az _{JK}										
401 RCL 01	L										
402 RDN	Az _{JK}	K	Az _{LM}	L		453 LBL 08					
403 XEQ 01	N	Y _N	X _N	R.d		454 "JTD+A"					
404 GTO 18						455 XROM "Pd"	Az	D	J		
						456 XEQ 06	N	Y _N	X _N	Reg "d"	
405 LBL "PRL"						457 GTO 08					
406 XROM "L"											
407 LBL 05						458 LBL "LBR"					
408 "JTD"						459 XROM "L"					
409 PROMPT	D	K	J			460 CLX					
410 STO [D					461 STO 00					
411 RDN	K	J									
412 XROM "XY"	ΔX	ΔY	X _J	Y _J		462 LBL 48					
413 STO 00	ΔX					463 "JTD"					
414 X<Y						464 XROM "BA"	Az	D	J		
415 STO 01	ΔY	ΔX				465 XEQ 06					
416 R-P	d	Az				466 GTO 48					
417 X< [D	Az									
418 P-R	ΔX	$-\Delta Y$				467 LBL "BA"					
419 X<Y						468 "JTB"					
420 CHS	ΔY	ΔX	X _J	Y _J		469 XROM "Pd"	Brg	a	b	c	
421 "L"						470 STO 01	Brg	a	b	c	
422 XROM "SP"	L	Y _L	X _L	R.d		471 RDN	a	b	c		
423 X< Z	X _L	Y _L	L			472 "QUAD"	Q	a	b	c	
424 RCL 00	ΔX					473 PROMPT					
425 RCL 01	ΔY	ΔX	X _L	Y _L		474 R↑	c				
426 "M"						475 STO [Q				
427 XROM "SP"	M	Y _M	X _M	R.d		476 RDN	1	Q	a	b	
428 GTO 05						477 E					
						478 X*Y?					
429 LBL "LAN"						479 CHS	X	±1	Q	a	b
430 XROM "L"						480 ST* 01			Q	a	b
						481 RDN					
431 LBL 12						482 CF 09					
432 "JTD+A"	A	D	K	J		483 1.6					
433 XROM "Pd"	A					484 -	Q-1.6	a	b	b	
434 STO [485 X<0?					
435 RDN	D					486 SF 09	*				
436 STO 01						487 LASTX					
437 ST+ 00	K	J				488 -	Q-3.2				
438 RDN						489 X>0?					
439 XROM "XY"	ΔX	ΔY	X _J	Y _J		490 SF 09	**	140	Q-3.2	a	b
440 R-P	d	θ				491 180					
441 X< [A	θ				492 FS? 09					
442 -	θ-A					493 CLX					
443 RCL 01	D	θ-A				494 ST+ 01	180 or 0				
444 P-R	ΔX	ΔY				495 RDN	Q-3.2				
445 X<Y	ΔY	ΔX	X _J	Y _J		496 E	1				
446 "N"						497 +	Q-2.2	a	b	b	
447 XROM "SP"	N	Y _N	X _N	R.d		498 SIGN	±1				
448 GTO 12						499 CHS	±1	a	b	b	

initialize Reg.00
for distance
accum.



"D" is pos.
to the right
as looking
from J to K

prompt must be
answered; L ≠ 0



Quad I: Az = Brg
II: Az = 180 - Brg
III: Az = Brg - 180
IV: Az = -Brg

* -1 ⇒ Quad II, III or IV

* FS? 09 ⇒ Quad I

** FS? 09 ⇒ Quad III or IV

** -1 ⇒ Quad III or IV

+1 ⇒ Quad I or II

500 FS?C 09
501 ABS
502 RCL 01
503 *
504 RCL [
505 RDN
506 RTN

507*LBL "ADJ"
508 SF 14

509*LBL 53
510 XROM "LL"
511 "BEG PT"
512 PROMPT
513 STO 01
514 "END PT"
515 PROMPT
516 X<> 00
517 "Σ DIST"

518 PROMPT
519 1/X
520 ENTER↑
521 X<> 01
522 X<>Y
523 X<> 00
524 XROM "XY"
525 ST* 00
526 X<>Y
527 ST* 01
528 R↑
529 R↑
530 .
531 SIGN

X Y Z T L

Az a b b
C
Az a b c

Beg Pt

End Pt

Σ Dist?

1/Σ d

Beg Pt

1/Σ d

End Pt

Δ X

Δ X

Δ Y

Δ X

Δ Y

0

1

X Beg

X Beg

-1 over the
X Beg

End Pt in R.00;
default Σ DIST
brought from R.00

1/Σ d in R.01

1/Σ d in R.00

Δ X in P.00

Δ Y in R.01

532*LBL 41
533 RDN
534 "PT#"
535 PROMPT
536 X=0?
537 GTO 53
538 LASTX
539 STO 1
540 RDN
541 XROM "P"
542 X<> L
543 STO [
544 X<> L
545 ST- T
546 X<>Y
547 ST- Z
548 R↑
549 R↑
550 R-P
551 ST+ 1
552 R↑
553 R↑
554 RCL 1
555 ENTER↑
556 X<> 01
557 ST* 01
558 X<> 01
559 ST- IND [
560 CLX
561 RCL 00
562 *
563 DSE [
564 --
565 ST- IND [
566 RCL 1
567 SIGN
568 RDN
569 GTO 41

570*LBL "P"
571 SF 21
572 ST+ X
573 SIGN
574 RDN
575 RCL IND L
576 ISG L
577 --
578 RCL IND L
579 END

X Y Z T L

X₁ Y₁
P₁[#]

0

P₁[#]
Y₂ X₂ X₁ Y₁
ZP+1
ZP+1
Y₂ X₂ X₁ Y₁

X₂ Y₂ Δ X Δ Y
Δ X Δ Y X₂ Y₂
d₁ θ₁ X₂ Y₂

X₂ Y₂
Σ dist
Σ dist
Σ dist
Σ dist

err y
err x
err y
err x

err x
Σ d
Σ dist

*
Σ dist

X₂ Y₂

P a b c

ZP
1

a b c
X a b c

Y X a b

[dist. in 0]

Y in Last X

Σ dist accum.
so far

mult. $\frac{\text{err}}{\Sigma d}$ in R.01

by the distance
accumulated so
far, leaving Δ Y
in R.01

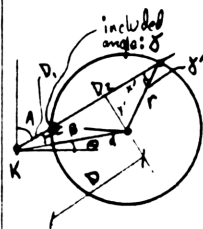
* subtract Δ Y
from the Y coord.
of the Pt.

* subtract Δ X
from the X
coord. of the Pt.

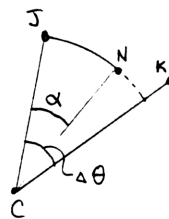
ZP in Last X

ZP+1 in Last X

	X	Y	Z	T	L		X	Y	Z	T	L
100+LBL 05	A	K	R	J		154 XROM "P"	Y _N	X _M	Y _{N-2}	X _{N-2}	
101 STO [A					155 R↑					
102 RDN	Z	R	J	A		156 STO [X _{N-2}				
103 X<>Y	R	Z	J	A		157 R↑					
104 STO \	R					158 STO a	Y _{N-2}				
105 X<>Z	J	K	R	A		159 R↑	X _M	Y _{N-2}	X _{N-2}	Y _M	
106 XROM "XY"	A X	A Y	X _K	Y _K		160 ST- Z	X _N	Y _{N-2}	X _{N-X_N}	Y _M	
107 R-P	d	θ				161 R↑	Y _M	X _M	Y _{N-2}	ΔX _Z	
108 R↑						162 ST- Z	Y _M	X _M	ΔY _Z	ΔX _Z	
109 STO 00	Y _K					163 R↑					
110 R↑						164 R↑	ΔX _Z	ΔY _Z	Y _M	X _M	
111 STO 01	X _K					165 R-P	d _Z	garbage			
112 R↑						166 R↑					
113 R↑	d	θ				167 R↑	Y _M	X _M	d _Z		
114 X<>Y	θ	d				168 RCL 00	Y _{N-1}				
115 RCL [A					169 -	ΔY ₁				
116 +	θ+A	d			θ+A=90-β	170 X<>Y	X _M	ΔY ₁	d _Z	d _Z	
117 X<>Y	d	90-β				171 RCL 01	X _{N-1}				
118 P-R	Y'	D				172 -	ΔX ₁	ΔY ₁			
119 X↑2	(Y') ²					173 R-P	d ₁	garbage	d _Z		
120 RCL \	R					174 X<>Y					
121 ENTER↑	R	R	(Y') ²	D		175 X<> [X _{N-2}	d ₁	d _Z		
122 *	R ²	Y' ²	D	D		176 RDN	d ₁	d _Z		X _{N-2}	
123 X<>Y	Y' ²	R ²				177 "N"					
124 -						178 X=Y?					
125 SORT	X'	D	D	D		179 GTO 01					
126 -	D ₁					180 R↑	X _{N-2}				
127 RCL [A	D ₁	D	D		181 RCL a	Y _{N-2}				
128 R↑	D					182 X<>Y	X _{N-2}	Y _{N-2}			
129 LASTX	X'	D	A	D ₁		183 GTO 10					
130 +	D _Z	A	D ₁								
131 RCL [A	D _Z	A	D ₁		184+LBL 09	X _{N-2}	Y _{N-2}	0		
132 R↑	D ₁	A	D _Z	A		185 X<>Z					
133 P-R	ΔY ₁	ΔX ₁				186 "N1"					
134 X<> 00	Y _K					187 XEQ 10					
135 ST+ 00						188 "N2"					
136 X<>Y	ΔX ₁	Y _K	D _Z	A	Y _{N-1} in R.00	189+LBL 01					
137 X<> 01	X _K	Y _K				190 RCL 00					
138 ST+ 01					X _{N-1} in R.01	191 RCL 01	X _{N-1}	Y _{N-1}			
139 R↑	-										
140 R↑	D _Z	A				192+LBL 10					
141 P-R	ΔY _Z	ΔX _Z	X _K	Y _K		193 .	0				
142 ST+ T	ΔY _Z	ΔX _Z	X _K	Y _{N-2}		194 ENTER↑	0	0	X	Y	
143 RDN	ΔX _Z	X _K	Y _{N-2}	ΔY _Z		195 XROM "SP"					
144 +	X _{N-2}	Y _{N-2}				196 RTN					
145 X<>Y	Y _{N-2}	X _{N-2}									
146+LBL "PM"						197+LBL "EXTAN"					
147 CF 22						198 SF 09					
148 "M"						199 GTO 00					
149 PROMPT	M	Y _{N-2}	X _{N-2}								
150 FC? 22						200+LBL "INTAN"					
151 .	0					201 CF 09					
152 X=0?											
153 GTO 09											



	X	Y	Z	T	L		X	Y	Z	T	L
399*LBL "AFRC"											
400 XROM "L"											
401*LBL 19											
402 "J+C+K+FRC"											
403 PROMPT	FRC	K	C	J							
404 STO I	FRC										
405 RDN											
406 STO 01	X										
407 RDN											
408 STO 00	C										
409 X<>Y	J	C									
410 XROM "XY"	ΔX	ΔY	X_c	Y_c							
411 R-P	d	θ									
412 X<> 00	C	θ									
413 X<>Y	θ	C									
414 X<> 01	K	C									
415 XROM "XY"	ΔX	ΔY	X_c	Y_c							
416 R-P	d	θ_k									
417 RDN	θ_k										
418 RCL 01	θ_j	θ_k	X_c	Y_c							
419 -	$\Delta \theta$										
420 RCL I	Frc										
421 *	α										
422 RCL 01	θ_j										
423 +	θ_N										
424 RCL 00	Rad	θ_N									
425 P-R	ΔX	ΔY									
426 X<>Y	ΔY	ΔX	X_c	Y_c							
427 "N"											
428 XROM "SP"											
429 GTO 19											
430*LBL "FREE"											
431 XROM "L"											
432 SF 21											
433 "FREE PTS:"											
434 AVIEW											
435 .4											
436 STO 00											
437 E9											
438 CHS											
439 STO 01											
440*LBL 03											
441 ISG 00											
442 RCL 00	Pt*										
443 ST+ X	reg*										
444 SF 25											
445 RCL IND X	X_i										
446 FC? 25											
447 XROM "LL"											
448 FC?C 25											
449 STOP											
450 RCL 01	-E9	X_i									
451 X=Y?											
452 GTO 03											
453 RCL 00	Pt*										
454 -											
455 XROM "AX"											
456 AVIEW											
457 GTO 03											
458*LBL 08											
459 X=0?											
460 X>0?											
461 RTN	-Pt*	K	a	b							
462 R+											
463 STO 00	b										
464 R+	a										
465 STO 01											
466 R+	K										
467 STO J											
468 R+	Pt*	K									
469 CHS	ΔX	ΔY	X_k	Y_k							
470 XROM "XY"	Rad.	θ									
471 R-P	b										
472 RCL 00	a										
473 RCL 01	K										
474 RCL J	Rad	K	a	b							
475 R+											
476 END											



Rad in R.00

θ_j in R.01

	X	Y	Z	T	L		X	Y	Z	T	L
01*LBL "AREA"						54 RCL 01	$\Sigma Area$				
02 SF 14						55 2	2				
						56 /	$\Sigma Area$				
03*LBL 08						57 "AREA="					
04 XEQ "L"						58 ARCL X					
05 SF 21						59 XEQ "FS"					
06 SF 09						60 RVIEW					
07 .						61 GTO 08					
08 STO 01	$\Sigma Area$					62*LBL C					
09 "PT1"	$Pt \#1$					63 FS? 09					
10 PROMPT						64 GTO 08					
11 E3						65 LASTX	$1+ZP_{n-1}$				
12 /	.00P					66 X<>Y					
13 "PT2"	$Pt \#2$					67 FS? 10					
14 PROMPT						68 RDN					
15 +	$Pt \#2.00P$					69 FS? 10					
16 ST+ X	$ZP2.00P$					70 GTO 05					
17 E	1					71 SF 10					
18 +	$1+ZP2.00P$					72 R+	$\Delta X \cdot Y_{n-1}$				
19 STO 00						73 ST- 01	$(\Sigma Area)$				
20 3						74 RDN	$1+ZP_{n-1}$				
21 STO I	counter					75 E	1				
22 RDN	$1+ZP2.00P$	0				76 -	ZP_{n-1}				
						77 STO I	counter				
23*LBL 05						78 R+					
24 SIGN	1					79 GTO 05					
25 RDN											
26 RCL I	counter					80*LBL 09	ZP_n	$1+ZP_{n-1}$			
27 "PT"						81 X<>Y	$1+ZP_{n-1}$				
28 XROM "AX"						82 FRC	$.00ZP_{n-2}$				
29 PROMPT	$Pt \#$					83 LASTX	$1+ZP_{n-1}$				
30 CF 10						84 X<>Y	$.00ZP_{n-2}$	$1+ZP_{n-1}$	ZP_n		
31 X<>Y	counter					85 E3	1000				
32 ISG X	counter					86 *	ZP_{n-2}	$1+ZP_{n-1}$	ZP_n	ZP_n	
33 "	"NOP"					87 X<>Z	ZP_n	$1+ZP_{n-1}$	ZP_{n-2}	ZP_{n-2}	
34 STO I	new counter					88 RCL IND X	X_n	X_n	ZP_n	$1+ZP_{n-1}$	
35 RDN	$Pt \#3$					89 RCL IND T	X_{n-2}	X_n	ZP_n		
36 ST+ X	$Z \cdot Pt \#3$					90 -	ΔX	ZP_n	$1+ZP_{n-1}$		
37 LASTX	$1+ZP2$					91 RCL IND Z	Y_{n-1}	ZP_n	$1+ZP_{n-1}$		
38 X<>Y	$Z \cdot P3$	$1+ZP2$				92 *	$\Delta X \cdot Y$	ZP_n			
39 CF 09						93 ST+ 01	(ΣA)				
40 X>0?						94 SIGN	1				
41 XEQ 09	$1+ZP_n$					95 RDN	ZP_n	$1+ZP_{n-1}$			
42 X>0?						96 X<>Y	$1+ZP_{n-1}$	ZP_n			
43 GTO 05						97 E-3	.001	$1+ZP_{n-1}$	ZP_n		
44 RDN	$1+ZP_{n-1}$					98 ST* Y		$(1+ZP_{n-1})$			
45 RCL 00	$1+ZP2.00P$					99 ST- Y		$.00ZP_{n-1}$	ZP_n		
46 FRC						100 X<>L	$\Delta X \cdot Y_{n-1}$				
47 E3	1000					101 RDN	$.00ZP_{n-1}$	ZP_n			
48 *	ZP_1	$1+ZP_{n-1}$				102 +					
49 XEQ 09	$1+ZP_1$					103 E	1				
50 RCL 00	$1+ZP2$					104 +					
51 E	1					105 END	$1+ZP_n$	$.00ZP_{n-1}$			
52 -	$ZP2$	$1+ZP_1$									
53 XEQ 09											

STATIONING of POINTS on a LINE

xeg POB (SIZE $2n + 2$ for "n" points)

SOFTKEYS (USER mode ON)

No softkeys are used by POB

USER FLAGS

Flag 00	clear	Station and offset displayed
	set	Only station displayed
Flag 03	clear	Azimuth input in decimal degrees
	set	Azimuth input in degrees-minutes-seconds

INPUT SUMMARY

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
POB	Point of beginning. <u>Defined as being at station 0+00.</u>	No default
PT ON CL	A second point on the line, used to define the line. Bypass to enter an azimuth from the POB.	No default
AZ	Azimuth of the line from the POB, in decimal degrees (dd) or degrees-minutes-seconds (DMS)	No default
PT	Point for which station and offset are desired	No default

OUTPUT SUMMARY (Items in parenthesis are displayed only if flag 00 is clear)

STA	The station of the point input, in standard "sta + feet" notation.
(OS)	Offset of the point from the centerline, in feet. A negative offset is to the right of the line looking forward on station.

PROGRAM FLAGS

Flag 14	set	Line printing control
Flag 21	set	Printer/aview control

STORAGE REGISTER USE

<u>Register</u>	<u>Value</u>
00	POB
01	AZ
02	Point #1 Northing (Y)
03	Point #1 Easting (X)
04	Point #2 Northing (Y)
05	Point #2 Easting (X)

etc.

STATION and OFFSET of POINTS ON A LINE

Program POB finds the station and offset of coordinate points along a (straight) centerline. These points must be defined by other coordinate geometry routines in the ROM, or by user routines. A point of beginning is used, defined as station 0+00. The line is defined by either another point on the line, or the azimuth of the line. The normal procedure is that points are being created in the coordinate geometry routines which have a known relationship to a centerline. If one of these points (which must be on the centerline) has a known station, the POB is set by projecting backward from that point along the line, a distance which is that point's station in feet. Then all other points can be stationed from this POB.

If flag 00 is set, only the station is output; if flag 00 is clear, both station and offset are given. Stationing is in the standard "sta + feet" notation. The offset is in feet, a positive offset being to the left of centerline looking forward on station (i.e. the POB at your back).

SUBROUTINE USE

Parts of POB are useable as subroutines for other programs. The calling program needs to SF 11, xeq POB with either 2 or 3 in the X-register (to access label 02 or 03 in POB; see the listing). The required input (in the Y-register) is the point number for which stationing is desired (Lb1 02) or the station to be appended to Alpha (Lb1 03). The listing for POB is part of program SCOL.

TRIANGLE SOLUTIONS

(SIZE 002)

SOFTKEYS (USER mode ON)

A	SSS	Three sides known
B	ASA	Two angles and included side known
C	SAA	Two angles and non-included side known
D	SAS	Two sides and included angle known
E	SSA	Two sides and non-included angle known

USER FLAGS

Flag 00	clear set	Full output Only area displayed
Flag 03	clear set	Angles in decimal degrees (dd) Angles in Degrees-Minutes-Seconds (DMS)

INPUT SUMMARY

<u>Prompt</u>		<u>Input</u>	<u>Default</u>
S1/S2/S3	(SSS)	Three sides	None
A1/S2/A3	(ASA)	Two angles and included side	None
S2/A1/A2	(SAA)	Two angles and non-included side	None
S1/A3/S2	(SAS)	Two sides and included angle	None
S1/S2/A1	(SSA)	Two sides and non-included angle	None

OUTPUT SUMMARY (Items in parenthesis are displayed only if flag 00 is clear)

(A1, A2, A3)	Angles, in decimal degrees or DMS
(S1, S2, S3)	Length of sides
AREA	Area of the enclosed triangle

PROGRAM FLAGS

Flag 21	set	Display control
---------	-----	-----------------

STORAGE REGISTER USE

<u>Register</u>	<u>Value</u>
00	Working; S1
01	Working; S2
02	A3
03	S3
04	A2

TRIANGLE SOLUTIONS

There are five programs concerned with solving triangles. Starting with different information, they all calculate and display the internal angles, lengths of the sides, and the area of the solved triangle. The standard triangle notation is used; e.g. side #1 is opposite angle #1.

All of the routines are accessible either by softkey or by alpha label (xeq ____). These softkeys operate a little differently than most softkeys in this module. Instead of bringing up a prompt for the required input, they assume the input is already in the stack; pressing the softkey starts execution of the desired routine. The triangle solution routines, and their softkeys, are:

<u>Key</u>	<u>Label</u>	<u>Description</u>
A	SSS	All three sides known
B	ASA	Two angles and included side known
C	SAA	Two angles and non-included side known
D	SAS	Two sides and included angle known
E	SSA	Two sides and non-included angle known

Each of the routines requires three items of input, keyed into the stack: Item one, enter/, Item two, enter/, Item three, softkey or R/S. If using softkeys, simply key the three items in and press the softkey (making sure USER mode is on). If prompts are desired, execute the desired label from the keyboard, key in the three input items in response to the prompt, and press R/S.

Like any other HP-41 program, in order for the softkeys to be active, the program pointer must be within the program that the local alpha (softkey) labels are in. The easiest way to do this is an alpha GTO; just GTO any of the six global labels listed above. Thereafter, the softkeys will be active.

Routine SSA usually has not one but two possible solutions; both are given. See the examples for clarification.

"SIZE CHECK"

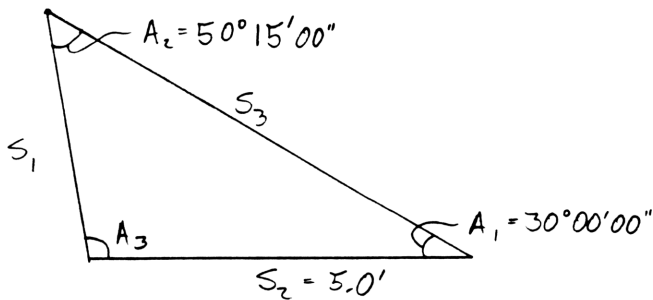
None of the triangle programs use subroutine S. This means that the user must verify that the SIZE is at least 005, and that DEG mode is set.

FLAGS

Like all the geometry programs in this module, the triangle routines recognize the status of flag 03. If it is clear, all angular input and output is in decimal degrees (dd); if set, degrees-minutes-seconds (DMS) is used. The output displays and input prompts are clearly labeled to reflect this. However, if softkeys are used for input, the user must verify that flag 03 matches the desired format.

Flag 00 is used by all of the triangle programs; output of the angles and sides is skipped if flag 00 is set, and only the area of the enclosed triangle is displayed.

Use "XEQ":



(Here, flag 03 was set; angles are input and output in Degrees - Minutes - Seconds, as the prompt and results indicate.)

```
XEQ "SAA"
S2+A1+A2 (DMS)
5.0000 ENTER↑
30.0000 ENTER↑
50.1500 RUN
A1=30.0000 DMS
A2=50.1500 DMS
A3=99.4500 DMS
S1=3.2516'
S2=5.0000'
S3=6.4094'
```

```
AREA=8.0117 SQ'
-----
S2+A1+A2 (DMS)
```

or, Use Local labels (In User Mode)

(same triangle - user must check status of flag 03 to match input format)

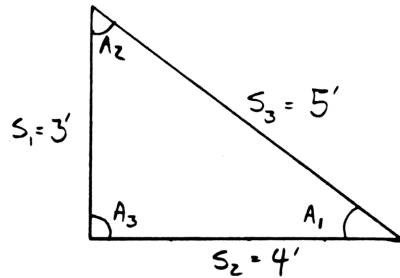
Note: To use key-accessed local labels, three things are required:

- 1) Calculator in User mode.
- 2) No key assignments on the key in question.

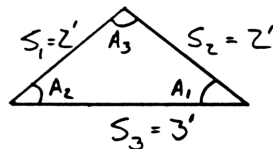
- 3) The calculator is "in" the program; that is, the program pointer is somewhere between the first line of the program and the end. Looking at the program listing, all of the routines (SSS, ASA, etc.) have a common "end" statement. Therefore the easiest way to access a local label is to go to (GTO, not XEQ) any of the global labels within the program (here, "SSS") and then use the local (key-accessed) labels. This only needs to be done once, at first, and then after going to another global label, or PACKing.

```
GTO "SSS"
5.0000 ENTER↑
30.0000 ENTER↑
50.1500 XEQ C
A1=30.0000 DMS
A2=50.1500 DMS
A3=99.4500 DMS
S1=3.2516'
S2=5.0000'
S3=6.4094'
AREA=8.0117 SQ'
-----
```

Use "XEQ":

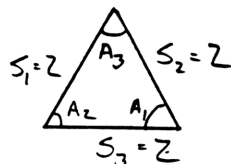


(Here, flag 03 was clear; angles are in decimal degrees)



or Use local label (In User mode):

(Here, flag 03 was set; angles are in Degrees-Minutes-Seconds)



XEQ "SSS"

S1↑S2↑S3
3.0000 ENTER↑
4.0000 ENTER↑
5.0000 RUN
A1=36.8699 dd
A2=53.1301 dd
A3=90.0000 dd

S1=3.0000'
S2=4.0000'
S3=5.0000'

AREA=6.0000 SQ'

S1↑S2↑S3
2.0000 ENTER↑
2.0000 ENTER↑
3.0000 RUN
A1=41.4096 dd
A2=41.4096 dd
A3=97.1808 dd

S1=2.0000'
S2=2.0000'
S3=3.0000'

AREA=1.9843 SQ'

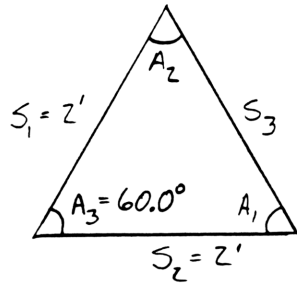
S1↑S2↑S3

2.0000 ENTER↑
2.0000 ENTER↑
2.0000 XEQ A

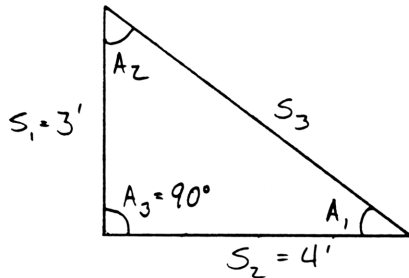
A1=60.0000 DMS
A2=60.0000 DMS
A3=60.0000 DMS

S1=2.0000'
S2=2.0000'
S3=2.0000'

AREA=1.7321 SQ'



(Here, flag 03 was clear. Angles are input and output in decimal degrees.)



```
XEQ "SAS"
S1↑A3↑S2 (dd)
  2.0000 ENTER↑
 60.0000 ENTER↑
  2.0000 RUN
A1=60.0000 dd
A2=60.0000 dd
A3=60.0000 dd
```

```
S1=2.0000'
S2=2.0000'
S3=2.0000'
```

```
AREA=1.7321 SQ'
```

```
-----
S1↑A3↑S2 (dd)
  3.0000 ENTER↑
 90.0000 ENTER↑
  4.0000 RUN
A1=36.8699 dd
A2=53.1301 dd
A3=90.0000 dd
```

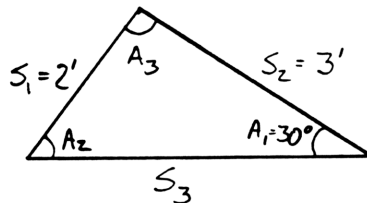
```
S1=3.0000'
S2=4.0000'
S3=5.0000'
```

```
AREA=6.0000 SQ'
```

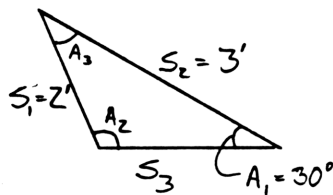
```
-----
S1↑A3↑S2 (dd)
```

"Side-Side-Angle" definitions have two possible solutions. Both are given.

Solution 1:



Solution 2:



```

XEQ "SSA"
S1↑S2↑A1 (DMS)
2.0000 ENTER↑
3.0000 ENTER↑
30.0000 RUN

A1=30.0000 DMS
A2=48.3525 DMS
A3=101.2435 DMS

S1=2.0000'
S2=3.0000'
S3=3.9210'

```

AREA=2.9407 SQ'

```

-----
A1=30.0000 DMS
A2=131.2435 DMS
A3=18.3525 DMS

```

```

S1=2.0000'
S2=3.0000'
S3=1.2752'

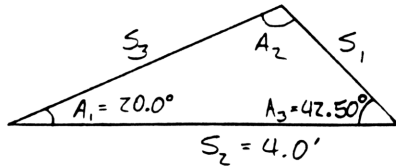
```

AREA=0.9564 SQ'

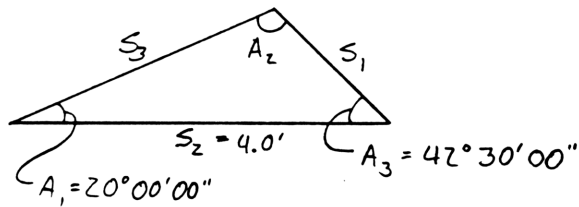
```

-----
S1↑S2↑A1 (DMS)

```



(Flag 03 clear \rightarrow decimal degrees)



(Flag 03 set \rightarrow Degrees - Minutes - Seconds)

```

                                XEQ "ASA"
A1+S2+A3 (dd)
    20.0000 ENTER↑
    4.0000 ENTER↑
    42.5000 RUN
A1=20.0000 dd
A2=117.5000 dd
A3=42.5000 dd

S1=1.5423'
S2=4.0000'
S3=3.0466'

AREA=2.0840 SQ'
-----
A1+S2+A3 (dd)

```

```

                                SF 03
                                XEQ "ASA"
A1+S2+A3 (DMS)
    20.0000 ENTER↑
    4.0000 ENTER↑
    42.3000 RUN
A1=20.0000 DMS
A2=117.3000 DMS
A3=42.3000 DMS

S1=1.5423'
S2=4.0000'
S3=3.0466'

AREA=2.0840 SQ'
-----
A1+S2+A3 (DMS)

```


01*LBL "SSS"
02*LBL 04
03 *S1+S2+S3*
04 PROMPT

05*LBL A
06 STO 03
07 X<>Y
08 STO 01
09 +

10 X<>Y
11 STO 00
12 +
13 2

14 /
15 RCL 01
16 RCL 00
17 RCL 03
18 XEQ 14
19 STO 04

20 RDN
21 RCL 03
22 RCL 00
23 RCL 01
24 XEQ 14
25 STO 02
26 RCL 04
27 XEQ 13
28 XEQ 01
29 GTO 04

30*LBL 14
31 *
32 RDN
33 -

34 X<>Y
35 *
36 LASTX
37 X<>Z
38 /

39 SQRT
40 ACOS
41 ST+ X
42 RTN

43*LBL "ASA"
44*LBL 05
45 *A1+S2+A3*
46 XEQ 02

47*LBL B
48 FS? 03
49 HR
50 STO 02
51 RCL Z
52 FS? 03
53 HR

54 XEQ 12
55 GTO 05

56*LBL 12
57 STO T
58 XEQ 13
59 STO 04
60 SIN

61 X<>Y
62 STO 01
63 X<>Y
64 /
65 RCL 02

66 SIN
67 *
68 STO 03
69 LASTX
70 /

71 X<>Y
72 SIN
73 *
74 STO 00
75 X<>Y

76*LBL 01
77 RCL 04
78 RCL 02

79*LBL 11
80 SF 21
81 FS? 00
82 GTO 00
83 "A1"

84 RT
85 RT
86 XROM "Dd"
87 AVIEW
88 "A2"

89 RT
90 XROM "Dd"
91 AVIEW
92 "A3"
93 RT

94 XROM "Dd"
95 AVIEW
96 ADV
97 "S1="

98 ARCL 00
99 AVIEW
100 "S2="

101 ARCL 01
102 AVIEW
103 "S3="

104 ARCL 03
105 AVIEW
106 ADV

A₁ A₃ S₂ A₁
A₂ S₂ A₁ A₁
A₂
sin A₂
S₂
S₂
S₂ sin A₂

A₃ A₁ A₁ A₁
A₃
sin A₃
S₃
S₂ sin A₂
A₁ S₂ A₁ A₁
sin A₁
S₁

A₁ S₁ A₁ A₁
A₁ A₂ A₁
A₃

A₁ S₂ A₃ A₂
A₂ A₁ S₂ A₃
A₃ A₂ A₁ S₃

A₁ S₂ A₃ A₂
A₂ A₁ S₂ A₃
A₃ A₂ A₁ S₃

A₁ S₂ A₁
A₃ S₂ A₁
A₃ A₁

A₁ A₃ S₂ A₁

	X	Y	Z	T	L		X	Y	Z	T	L
107+LBL 00	A_3	A_2	A_1			159 X↑2	S_1^2				
108 SIN						160 X<>Y	$2S_1S_2\cos A_3$				$S_3^2 = S_1^2 + S_2^2$
109 RCL 00	S_1	$\sin A_3$	A_2	A_1		161 ST+ X					$-2S_1S_2\cos A_3$
110 *		A_1				162 -	S_2^2				
111 RCL 01	S_2	A_2	A_1			163 R↑	S_3^2				
112 *		A_2				164 +	S_3				
113 2			A_2			165 SQR	A_3				
114 /	$\frac{S_1S_2\sin A_3}{Z}$	A_2	A_1	A_1		166 STO 03	$\sin A_3$	S_3			
115 ABS						167 RCL 02	$\frac{\sin A_3}{S_2}$				
116 "AREA="						168 SIN	S_2	$\frac{\sin A_3}{S_2}$	$\frac{\sin A_3}{S_3}$		
117 ARCL X						169 X<>Y	$\sin A_2$				
118 XEQ "FS"						170 /	A_2				
119 AVIEW						171 STO Y	$\frac{\sin A_3}{S_3}$	A_2			
120 XROM "L"						172 RCL 01	S_1				
121 RTN						173 *	A_1				
122+LBL "SAA"						174 ASIN					
123+LBL 07						175 STO 04					
124 "S2↑A1↑A2"						176 X<>Y					
125 XEQ 02	A_2	A_1	S_2			177 RCL 00					
126+LBL C						178 *					
127 FS? 03						179 ASIN					
128 HR						180 XEQ 01					
129 STO 04	A_2					181 GTO 08					
130 X<>Y						182+LBL "SSA"					
131 FS? 03						183+LBL 09					
132 HR	A_1	A_2	S_2			184 "S1↑S2↑A1"					
133 STO 00						185 XEQ 02					
134 XEQ 13	A_3					186+LBL E					
135 STO 02						187 FS? 03	A_1	S_2	S_1		
136 RCL 00	A_1	S_2				188 HR					
137 XEQ 12						189 STO T	$\sin A_1$	S_2	S_1	A_1	
138 GTO 07						190 SIN	S_2				
139+LBL "SAS"						191 X<>Y	$S_2\sin A_1$	S_1	A_1	A_1	
140+LBL 08						192 STO 01	S_1				
141 "S1↑A3↑S2"	S_2	A_3	S_1			193 *	$\frac{S_1\sin A_1}{S_2}$				$\sin A_2$
142 XEQ 02						194 X<>Y	A_2				$= \sin A_1 \frac{S_2}{S_1}$
143+LBL D						195 STO 00	A_1	A_2			
144 STO 01	S_2					196 /	A_3				
145 X<>Y	A_3					197 ASIN	$\sin A_3$	A_1			
146 FS? 03						198 STO 04	S_1				
147 HR						199 X<>Y	$S_1\sin A_3$				
148 STO 02	A_3	S_2	S_1			200 XEQ 13	A_1				
149 COS	$\cos A_3$					201 STO 02	A_3				
150 X<>Y	S_2					202 SIN	$\sin A_3$	A_1			
151 *						203 RCL 00	S_1				
152 LASTX	S_2	$S_2\cos A_3$	S_1			204 *	$S_1\sin A_3$				
153 X↑2						205 X<>Y	A_1				
154 RDN	$S_2\cos A_3$	S_1		S_2^2		206 SIN	$\sin A_1$				
155 X<>Y						207 /	S_3				
156 *						208 STO 03	A_1				
157 LASTX	S_1	$S_1\cos A_3$		S_2^2		209 RDN	area	A_2	A_1	A_1	
158 STO 00						210 XEQ 01	0	A_2			
						211 CLX	$180-A_2$				
						212 XEQ 13	A_2'	A_1			$A_2' = 180^\circ - A_2$
						213 STO 04					

(B91)

	X	Y	Z	T	L		X	Y	Z	T	L
214 X<>Y	A ₁	A ₂	A ₁	A ₁							
215 XEQ 13	A ₃	A ₁	A ₁	A ₁							
216 STO 02											
217 SIN	sin A ₃										
218 X<>Y	A ₁										
219 SIN											
220 /	$\frac{\sin A_3}{\sin A_1}$										
221 RCL 00	S ₁										
222 *	S ₃										
223 STO 03											
224 X<>Y	A ₁										
225 XEQ 01											
226 GTO 09											
227 LBL 13	A ₁	A ₃									
228 +	A ₁ + A ₃										
229 COS	cos										
230 CHS	-cos										
231 ACOS	A ₂					$\cos(180-d) = -\cos d$					
232 RTN											
233 LBL "Dd"											
234 "t="											
235 FS? 03											
236 GTO 03											
237 ARCL X											
238 "t dd"											
239 RTN											
240 LBL 03											
241 HMS											
242 ARCL X											
243 HR											
244 "t DMS"											
245 RTN											
246 LBL "Pd"											
247 XEQ 02											
248 FS? 03											
249 HR											
250 RTN											
251 LBL 02											
252 FS? 03											
253 "t (DMS)"											
254 FC? 03											
255 "t (dd)"											
256 PROMPT											
257 END											

DIMENSION and TRIGONOMETRY ROUTINES

This module contains several short dimension "functions" which work with feet-inches-sixteenths addition, subtraction, and conversions to and from decimal feet. It also contains trigonometry "functions" which take or give angles in either degrees-minutes-seconds (DMS) or decimal degrees (dd) depending on whether flag 03 is set or clear; and useful utility routines.

The "functions" mimic native HP-41 functions as closely as possible. They leave the operand in Last X and are thus reversible. They use no numbered storage registers (i.e. they work with SIZE 00). Binary functions like PR and all unary functions leave the rest of the stack completely undisturbed, while the two binary functions FIS+ and FIS- effect stack rolldown. They are most useful from the keyboard when assigned to a key and executed in USER mode; they may also be executed from programs.

Dimensions

These routines work with dimensions in feet-inches-sixteenths, and are intended to be used with the display mode set to FIX 4. The form used is very similar to that used by the HP-41 function HMS. For example, 12' 2-7/16" is 12.0207. All fractional inches must be in sixteenths. Thus 6' 5-3/4" is 6.0512. Trailing zeroes need not be keyed in; 3'-10 may be keyed in as 3.1 instead of 3.1000.

FIS	Converts decimal feet to feet-inches-sixteenths. Once executed, pressing R/S once resets the display to FIX 6 (for example, to see the next digit in order to round to the nearest eighth of an inch). Pressing R/S (or executing any of the dimension routines) again resets FIX 4 format.
-----	---

FTd	Converts feet-inches-sixteenths to decimal feet. Does not have the FIX 6/FIX 4 feature like FIS.
-----	--

The functions FIS+ and FIS- are for adding and subtracting dimensions directly instead of converting them to decimal feet, adding, and then converting back. Their only drawback is that they are rather slow. It is helpful to realize that when adding dimensions, if there is no carryover from sixteenths to inches or inches to feet, the numbers may be added (or subtracted) directly instead of using FIS+ and FIS-.

Another way to add dimensions directly is to wait until both sixteenths and inches require a carryover, then add 0.8884 to the sum. This effects a carryover automatically. If all this is confusing, just use these two functions. They work far easier if assigned to a key.

FIS+ Adds dimensions in feet-inches-sixteenths. Key in first dimension, enter, key in second dimension, xeq FIS+ (or assign to a key). Stack is rolled down just as for normal addition. Does not have the FIX 6/FIX 4 feature like FIS.

FIS- Subtracts dimensions in feet-inches-sixteenths.

Trigonometry Functions

These functions accept and output angles in degrees-minutes-seconds if flag 03 is set, and in decimal degrees if flag 03 is clear. Thus if flag 03 is clear they are identical to their namesake HP-41 functions (except a little slower). This saves converting angles to decimal degrees to operate on them. All the functions save their operand in Last X and are thus reversible. All of them require the calculator to be set to DEG (degree) mode. Subroutine return depth is limited to two.

CO	Cosine
SI	Sine
TA	Tangent
ACO	Arc cosine
ASI	Arc sine
ATA	Arc tangent
RP	Rectangular to polar conversion
PR	Polar to rectangular conversion

SHORT UTILITY ROUTINES

These routines find use in daily manual calculations. They all use the full stack.

QUAD Quadratic equation solution. The program prompts for A, B, and C, the constants in the quadratic equation $Ax^2 + Bx + C = 0$. It then solves (in subroutine Q) for the two possible values of x, where $x = (-B + \sqrt{B^2 - 4AC})/2A$. If $B^2 - 4AC$ is negative, the calculator flashes DATA ERROR and quits.

IP Linear interpolation or extrapolation.
 IP solves for Y2, where
 X1→ Y1
 X2→ Y2
 and X3→ Y3.

The program prompts for X1, Y1, X3, and Y3, then repeatedly prompts for X2 and solves for Y2. The prompts are X1↵ Y1, X3↵ Y3, and X2; the upward arrow in the first two means "enter↵ ." Thus, key in X1, enter, key in Y1, press R/S. Repeat for X3 and Y3. Then key in X2 and press R/S to get Y2 as many times as you want. The program will continue to solve for Y2 until the X2 prompt is bypassed. The routine has a default value of X1 = 0; if X1 is zero, just key in Y1 and press R/S. The user should note that the entire stack is used; any intermediate calculations during input will cause an error.

PYR This is an extremely simple routine that does part of the work in calculating the volume of a truncated pyramid. It uses the approximate formula

$$V = (A_1 + A_2 + \sqrt{A_1 \times A_2}) \times H/3$$

PYR doesn't actually calculate the volume. It takes as input the dimensions of the top and bottom, assumed to be parallel rectangles. The output, multiplied by the height H, is the volume (or a close approximation). See the examples.

Dimension Routines

Convert to feet:

$$10' - 9 \frac{3}{16} = 10' + (9.1875''/12)$$

$$= 10.7656'$$

Convert to feet-inches-sixteenths:

$$3.1294' = 3' + .1294 \times 12''$$

$$= 3' + 1.55''$$

$$= 3' + 1'' + .55 \times 16$$

$$= 3' + 1'' + 8.85/16ths$$

$$= 3' - 1 \frac{9}{16}$$

$$\text{Subtract: } \begin{array}{r} 9' - 7 \frac{3}{16} \\ - 4' - 9 \frac{1}{2} \end{array} \Rightarrow \begin{array}{r} 8' - 18 \frac{19}{16} \\ - 4' - 9 \frac{8}{16} \\ \hline 4' - 9 \frac{11}{16} \end{array}$$

$$\text{Add: } \begin{array}{r} 9' - 7 \frac{3}{16} \\ 4' - 9 \frac{1}{2} \\ 7' - 6 \frac{5}{8} \\ \hline 20' - 22 \frac{21}{16} \end{array}$$

$$= 21' - 11 \frac{5}{16}$$

```
10.0903
  XROM "FTd"
10.7656  ***

3.1294
  XROM "FIS"
3.0109  ***
      RUN
3.010884 ***
      RUN
3.0109  ***
```

```
9.0703 ENTER↑
4.0908
  XROM "FIS-"
4.0911  ***
```

```
9.0703 ENTER↑
4.0908
  XROM "FIS+"
7.0610
  XROM "FIS+"
21.1105  ***
```

```
9.0703 ENTER↑
4.0908      +
7.0610      +
20.2221     ***

.8884      +
21.1105     ***
```

Quadratic Equation

$$9x^2 + 2x - 33 = 0$$

$$x = \frac{-2 \pm \sqrt{2^2 - 4 \times 9 \times (-33)}}{2 \times 9} = -2.0292$$

$$\text{or } = 1.8070$$

```
          XROM "QUAD"
A          9.0000    RUN
B          2.0000    RUN
C          -33.0000   RUN
R1=1.8070
R2=-2.0292
A
```

A right trapezoidal prism is shown. The base is a right trapezoid with a top width of 4', a bottom width of 8', and a height of 6'. The right side of the trapezoid is the slanted edge, labeled 5'. The depth of the prism is labeled 3'.

[illegible]

End pyramid $\frac{1}{3} \times 6' \times 1' \times 4' = 8.0 \text{ c.f.}$

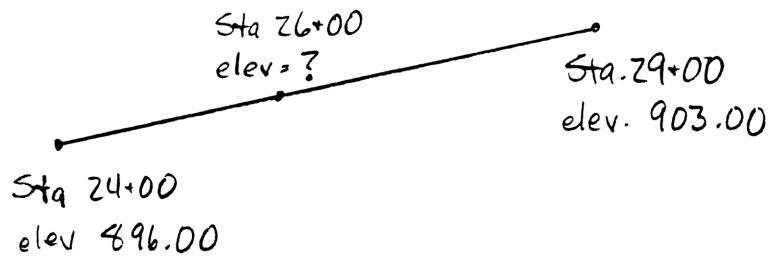
$\checkmark \quad \checkmark \quad \checkmark \quad \checkmark$

Center $\frac{1}{2} \times 4' \times 6' \times 1' = 12.0$

20.0

Using $\frac{1}{2}$ the interval results in $(\frac{1}{2})^2$ the error.

Interpolate/Extrapolate



$$896.0 + \frac{2600 - 2400}{2900 - 2400} \times (903.0 - 896.0)$$

$$= 898.80$$

```

XROM "IP"
X1+Y1
2,400.0000 ENTER↑
896.0000 RUN
X3+Y3
2,900.0000 ENTER↑
903.0000 RUN
X2
2,600.0000 RUN
Y2=898.8000
-----
X2
3,000.0000 RUN
Y2=904.4000
-----
X2
RUN
X1+Y1

```

UTILITY PROGRAMS

FUNCTIONS and UTILITY ROUTINES

This module contains several short routines which were written to mimic or emulate HP-41 functions; several others which are short, useful time savers for everyday work; a number of output, display, and print format functions, and several programming routines to simplify writing and using programs.

In the following write-up, one asterisk (*) signifies the use of register "a", meaning a shortened subroutine return chain. Two asterisks (**) mean part or all of the alpha register is used for "scratch-pad" storage. Three asterisks mean both are used. See the section "Synthetic Programming and Other Loose Ends" in the first part of the documentation for an explanation of these and other things that your Owner's Handbook never told you about your HP-41.

FUNCTIONS

The following routines mimic HP-41 functions as closely as possible. They leave the operand in Last X and are thus reversible. They use no numbered storage registers (i.e. they work with SIZE 00). Binary functions like PR and all unary functions leave the rest of the stack completely undisturbed, while the two binary functions FIS+ and FIS- effect stack rolldown. They are most useful from the keyboard when assigned to a key and executed in USER mode; they may also be executed from programs.

Stack Manipulation

Just like $X \leftrightarrow Y$ or $X \leftrightarrow Z$, these exchange registers in the stack.

YZ	Exchange the Y and Z registers.
YT	Exchange the Y and T registers.
ZT	Exchange the Z and T registers.

Dimensions (**)

These routines work with dimensions in feet-inches-sixteenths, and are intended to be used with the display mode set to FIX 4. The form used is very similar to that used by the HP-41 function HMS. For example, 7' 11-13/16" is 7.1113; 12' 2-7/16" is 12.0207. All fractional inches must be in sixteenths. Thus 6' 5-3/4" is 6.0512.

FIS	Converts decimal feet to feet-inches-sixteenths. Once executed, pressing R/S once resets the display to FIX 6 (for example, to see the next digit in order to round to the nearest eighth of an
-----	---

inch). Pressing R/S (or executing any of the dimension routines) again resets FIX 4 format.

FIS+ Adds dimensions in feet-inches-sixteenths. Key in first dimension, enter, key in second dimension, xeq FIS+ (or assign to a key). Stack is rolled down just as for normal addition. Does not have the FIX 6/FIX 4 feature like FIS.

FIS- Subtracts dimensions in feet-inches-sixteenths.

FTd Converts feet-inches-sixteenths to decimal feet. Does not have the FIX 6/FIX 4 feature like FIS.

Trigonometry Functions (*)

These functions accept and output angles in degrees-minutes-seconds if flag 03 is set, and in decimal degrees if flag 03 is clear. Thus if flag 03 is clear they are identical to their namesake HP-41 functions (except a little slower). They all save their operand in Last X and are thus reversible. All of them require the calculator to be set to DEG (degree) mode. Subroutine return depth is limited to two.

CO	Cosine
SI	Sine
TA	Tangent
ACO	Arc cosine
ASI	Arc sine
ATA	Arc tangent
RP	Rectangular to polar conversion
PR	Polar to rectangular conversion

SHORT UTILITY ROUTINES

These routines find use in daily manual calculations. All of them use the full stack.

QUAD Quadratic equation solution. The program prompts for A, B, and C, the constants in the quadratic equation $Ax^2 + Bx + C = 0$. It then solves (in subroutine Q) for the two possible values of x, where $x = (-B + \sqrt{B^2 - 4AC})/2A$. If $B^2 - 4AC$ is negative, the calculator flashes DATA ERROR and quits.

- IP Linear interpolation or extrapolation.
IP solves for Y2, where
 X1→ Y1
 X2→ Y2
 and X3→ Y3.
- The program prompts for X1, Y1, X3, and Y3, then repeatedly prompts for X2 and solves for Y2. The prompts are X1/ Y1, X3/ Y3, and X2; the upward arrow in the first two means "enter/ ." Thus, key in X1, enter, key in Y1, press R/S. Repeat for X3 and Y3. Then key in X2 and press R/S to get Y2 as many times as you want. The program will continue to solve for Y2 until the X2 prompt is bypassed. See the example for clarification. The routine has a default value of X1 = 0; if X1 is zero, just key in Y1 and press R/S. The user should note that the entire stack is used; any intermediate calculations during input will cause an error.
- ST Storage register review. Displays first the stack and Last X, then all the storage registers from 00 on up, until R/S is pressed or the SIZE limit is reached.

PROGRAMMING, OUTPUT, and DISPLAY

These are used almost exclusively in programs, having very little use from the keyboard. They are included in a user's program by keying (in PRGM mode) xeq (alpha) "Title", (alpha); they list as XROM "Title", which means "execute from ROM."

- AX (*) Append X. Very similar to the ARCL X command. Rounds the contents of the X register and appends the integer portion of that to alpha, but leaves the original X (and the rest of the stack and Last X) undisturbed. AX is very useful, for instance, if the X register contains a counter number. Subroutine returns are limited to two deep when AX is used. AX is one of a few routines in this module that cannot be single-stepped through, because it uses register P (see "Synthetic Programming and Other Loose Ends"). If you want to single-step through it to see how it works, copy it down into RAM and replace every line using the P register with one using a numbered register instead. Since

the P register is used, only 21 characters can be placed in Alpha instead of the 24 usually available.

S Setup routine. Checks the program's SIZE requirements (the largest numbered storage register used must be in X); sets or clears flag 26 to control the beeper according to flag 02; sets flag 21 to control viewing and printer use; and sets DEG mode. If a printer is present, S prompts for a title; if a time module (or a -41CX) is also present, it prints a time and date header first. Typically, a program using S looks like this:

```
01  Lbl "YOUR TITLE"
02  12  (or whatever the
       largest storage
       register used by
       the program is)
03  XROM "S"
04  etc.
```

If your program requires no numbered storage registers, substitute

```
02  1
03  CHS  (change sign)
04  XROM "S"
```

to indicate SIZE 000 required (since if S sees zero, it thinks register 00 is needed and sets the size to 001.

V View. Appends "=" and the contents of the X register to what the program has already (hopefully) placed in alpha, and displays it if flag 00 is clear. If flag 00 is set, V returns without doing anything. V starts at line 392 of program PLG, and is used mostly by that program. The only reasons V wasn't used in more of the module programs are that it makes it impossible to append units to the output, and that while the calculator is displaying something from V, all of the calling program's softkeys are dead (since the calling program doesn't share the same END as V). If the user sees an answer which doesn't look right, the instinctive (well, it's acquired) move to correct something by pressing a softkey gets nowhere.

PV Print or View. If no printer is attached, behaves just like an AVIEW command. However, if a printer is on line, PV prints the contents of Alpha (PRA) without displaying it on the screen. With a long text string, this eliminates the infuriatingly slow scrolling process, speeding up output greatly. Its main disadvantage is the same as V: a program's softkeys are dead while in PV. To use a softkey, SST to return to the calling program first.

L, LL, Line printing commands. L prints a
L*, CL short line of minus signs; LL prints a line all the way across the paper (24 places). L* prints a full-length line of asterisks. CL prints a line of any character (or string of characters) the program has placed in alpha; the number of times the character is repeated, and thus the length of the line, is determined by the number in the X register. This number shouldn't be larger than 24 for one character in alpha.

If flag 14 is set, both L and LL clear it and then jump to L*. Thus, a loop such as the following will print a list of asterisks the first time and a short line (----) thereafter:

```
42 SF 14
43 LBL 01
44 XEQ "L"
45 .
.
49 GTO 01
```

L* and CL change the stack if actually executed full length (i.e. if a printer is plugged in). LL and L also alter the stack if flag 14 is set, sending them to L*. Thus, necessary data in the stack should be stored and then recalled if these are to be used.

OK Appends "OK" to alpha if flag 07 is clear. If flag 07 is set, appends "NG" (no good) and BEEPs.

Pd Prompts for degrees. Appends " (dd)" to Alpha and then prompts if flag 03 is clear. Appends " (DMS)", prompts, and then converts the input to

decimal degrees if flag 03 is set.

Dd Display degrees. If flag 03 is clear, appends an "=", the contents of the X register, and " (dd)". If flag 03 is set, appends "=", the contents of the X register converted to degrees-minutes-seconds, and " DMS". If flag 03 is set, the contents of X are converted to degrees-minutes-seconds and then back to decimal degrees; X is thus unaffected, but Last X contains the angle in degrees-minutes-seconds.

IN Appends " (inches) to the alpha register
IS Appends _SQ" (square inches) to alpha.
FT Appends _' (feet) to alpha.
FS Appends _SQ' (square feet) to alpha.
NO Appends the number symbol (#) to alpha.
P1 Appends "(" to alpha
P2 Appends ")" to alpha
PP Appends "┐" (the perpendicular sign)
MU Appends the Greek "μ" (mu) to alpha
AT Appends "@" (the "at" symbol) to alpha
AM Appends "&" (ampersand) to alpha

M= Appends M=, divides the moment (in the X register) by 12 and appends it and 'K
FK Appends _'K (foot-kips) to alpha.
K Appends _STRS=, the contents of the X register, and _KSI to alpha.
KS Appends _KSI to alpha.
AS Appends A-S=, the contents of the X register, and _SQ"
M Prompts for M, 'K
FY Prompts for F-Y KSI

N Prompts for F-Y or F-S, and Fc', solving for B_1 , $.75P_b$ and M_{cr} , and leaves "N" in Alpha for the calling program to prompt
BD Prompts for B" and D"
W XEQ's "S" and asks WSD?, setting flag 06
AA XEQ's "S" and then displays the contents of the alpha register if flag 04 is set, "AASHTO" if flag 04 is clear

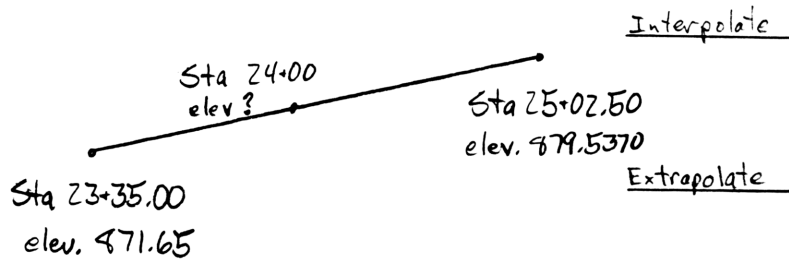
ADDITIONAL PROGRAMMING ROUTINES

The following routines are used by programs in the ROM, and are briefly described here in case they may be of some use.

Mc	Displays $1.2 M_{cr}$ and calculates reinf.
SBM	"Port of entry" to program COMP
R	Displays max/min reinforcing
RH	Displays $.75 P_b$
U	Calculates ultimate strength reinf.
VC*	"Port of entry" to programs VC and STA

Interpolate :

$$\frac{2400 - 2335.0}{2502.5 - 2335.0} (879.537 - 871.65) + 871.65 = 874.7106$$

Start Over

Storage and Stack Review
(here, size 013)

XEQ "IP"

```

X1+Y1
2,335.0000 ENTER↑
871.6500 RUN
X3+Y3
2,502.5000 ENTER↑
879.5370 RUN
X2
2,400.0000 RUN
Y2=874.7106
-----
X2
2,500.0000 RUN
Y2=879.4193
-----
X2
2,600.0000 RUN
Y2=884.1279
-----
X2
3,000.0000 RUN
Y2=902.9626
-----
X2
RUN
X1+Y1

```

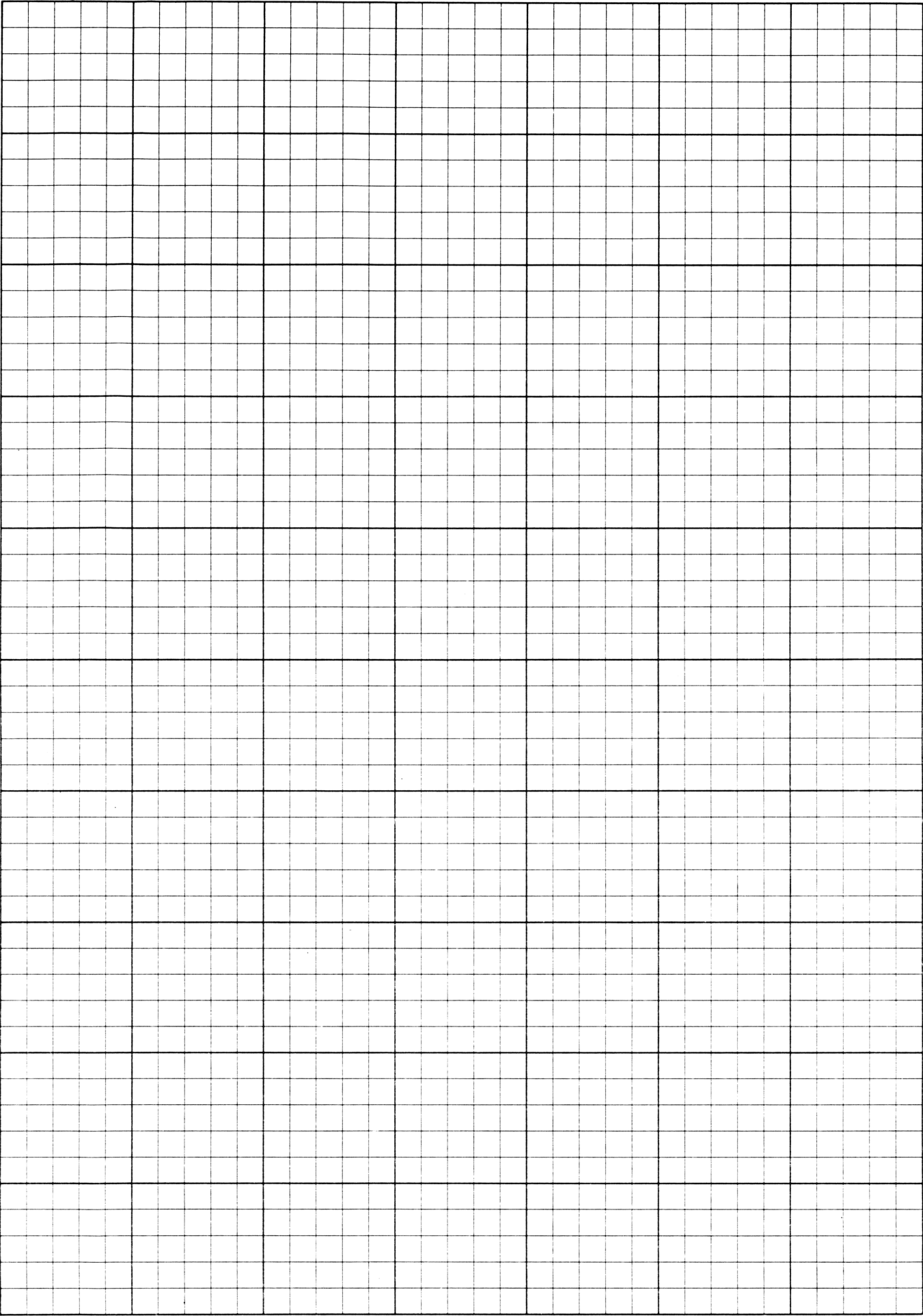
XEQ "ST"

```

X=3.0000
Y=17,591.0646
Z=311.5196
T=1,646.9885
L=165.2249

R.0=3.5000
R.1=0.6667
R.2=0.0000
R.3=23,250.0000
R.4=-0.1111
R.5=23,550.0000
R.6=0.0000
R.7=896.0000
R.8=23,341.3100
R.9=897.0000
R.10=17,591.0646
R.11=37,380.5362
R.12=17,591.0646

```



	X	Y	Z	T	L		X	Y	Z	T	L	
01*LBL "S"	MaxReg	Y	Z	T		55 SF 25	X'	Y	Z	T		SF 25 in case
02 SF 21					for a view	56 RND	L					X-reg. contains
03 SF 26					for BEEP	57 X<> ↑	X	Y	Z	T		Alpha data - for
04 CF 14						58 X<> L	d	Y	Z	T		RND function
05 FC? 02						59 X<> a						X in R.a
06 CF 26						60 ARCL ↑						Arcl rounded X
07 DEG						61 STO d	O	Y	Z	T		
08 SF 25						62 CLX						clear Reg. P
09 STO IND X					SIZE check	63 STO ↑	X	Y	Z	T		L clear Reg. a
10 FC?C 25						64 X<> a						
11 X<0?						65 RTN						
12 GTO 03												
13 E						66*LBL "N"						
14 +						67 "ACI"						Code Display
15 "SIZE<"						68 XEQ 11						
16 XROM "AX"						69 FS? 06						
17 PROMPT						70 GTO 04						
						71 "F-S 20,24"						
18*LBL 03						72 24						
19 RDN	Y	Z	T			73 XEQ 05						
20 FS? 00						74 X<>Y	Z 4	2Fs				
21 RTN						75 -	24 or 16					
22 FC? 55						76 .4						
23 RTN						77 /	60 or 40					Fy in R.01
24 SF 25						78 X<> 01	F5					
25 CLA						79 STO 13						
26 TIME	Time	Y	Z	T		80 GTO 07						
27 FC?C 25												
28 GTO 10						81*LBL 05						
29 ATIME						82 ENTER↑	default	default				
30 "t "						83 PROMPT	F6					
31 RDN	Y	Z	T			84 X>Y?						
32 DATE	Date	Y	Z	T		85 BEEP						
33 ADATE						86 STO 01						
34 SF 25						87 ST+ X	Z F5	default				
35 PRA						88 X<=Y?						
36 RDN	Y	Z	T			89 BEEP						
37 FC?C 25						90 RTN						
38 RTN												
						91*LBL 04						
39*LBL 10	Y	Z	T			92 "F-Y 40,60"						Fy in R.01
40 ADV						93 60						
41 "TITLE:"						94 XEQ 05						
42 AON												
43 PROMPT						95*LBL 07						
44 AOFF						96 CLA						
45 GTO 19												
						97*LBL "Fc"						
46*LBL "AX"	X	Y	Z	T	L	98 4						
47 X<> L	L				X	99 "Fc'KSI"	F6'					
48 STO ↑					Last X in R.P	100 PROMPT	9 65					
49 CLX						101 9						
50 RCL d	d	Y	Z	T	X	102 X<Y?						
51 X<> a	a				"d" in R.a	103 BEEP						
52 X<> L	X	Y	Z	T	a	104 RDN	F6'					
53 FIX 0						105 FS?C 10						
54 CF 29						106 RTN						

	X	Y	Z	T	L		X	Y	Z	T	L
107 ENTER↑						150+LBL "U"	M				
108 ENTER↑						159 STO a	I				
109 SQRT	$\sqrt{f_c'}$	f_c'	f_c'			160 SIGN	M				
110 253	253	$\sqrt{f_c'}$	f_c'	f_c'		161 LASTX	Z				
111 /						162 2					
112 STO 06						163 RCL 00	f_y	Z	M	I	
113 RDN	f_c'					164 /	$\frac{ZM}{f_y}$	I	I	I	
114 4	4	f_c'	f_c'	f_c'		165 *	f_y				
115 -						166 RCL 01	$\frac{f_y}{f_c'}$				
116 X<0?						167 /	qbd^2				
117 CLX						168 RCL 05					
118 CHS						169 /					
119 20	1/05					170 -					
120 /						171 SQRT					
121 .85						172 -					
122 +	β_1					173 RCL 00	f_y				
123 .65	.65	β_1	f_c'	f_c'		174 *	A_s				
124 X>Y?						175 RCL a	M				
125 X<>Y		β_1				176 X<>Y	A_s	M			
126 RDN						177 RTN					
127 STO I	β_1	f_c'	f_c'								
128 LASTX	.45					178+LBL 02					
129 R↑	f_c'	.45	β_1	f_c'		179 - -					
130 *											
131 RCL 01	f_y					180+LBL "AS"					
132 /						181 "A-S="					
133 STO 00	$\frac{f_y}{f_c'}$	β_1	f_c'	f_c'		182 ARCL X					
134 *											
135 87						183+LBL "IS"					
136 *						184 "I SQ"					
137 RCL 01	F_y										
138 LASTX	87					185+LBL "IN"					
139 +	$87 \cdot F_y$					186 "I--"					
140 /	P_{bal}					187 RTN					
141 .75											
142 *	.75 P_{bal}					188+LBL "R"	d	p=b			
143 STO 02						189 *	$A_s \cdot 1.2M_{ur}$				
144 RCL I	β_1					190 LASTX	d				
145 R↑	f_c'	β_1	.75 P_{bal}	f_c'		191 RCL 03	b				
146 "M"						192 *	A_c				
147 RTN						193 .2	200				
						194 RCL 01	F_y				
148+LBL "Mc"	bd^2					195 /	200 F_y				
149 RCL 06	$\sqrt{f_c'}/253$					196 *	$A_s \cdot 200 F_y$				
150 *						197 FC? 04	$A_s \cdot min$				
151 "1.2Mc="						198 X<>Y					
152 ARCL X						199 FS? 00					
153 XROM "FK"						200 RTN					
154 FC? 00						201 XEQ 02					
155 XEQ "PV"						202 FC? 04					
156 12						203 AVIEW					
157 *	1.2 M_{ur}					204 "200/FY="					
						205 ARCL L					
						206 AVIEW					
						207 XEQ 02					
						208 FS? 04					
						209 AVIEW					
						210 ".75Pb="	$A_s \cdot min$				

prepared for
"N" prompt in
calling program

$$\frac{bd^2}{6} \times 1.2 (7.5 \sqrt{f_c'})$$

$$= bd^2 \times \sqrt{f_c'}/253$$

	X	Y	Z	T	L		X	Y	Z	T	L
212*LBL "K"											
213 "† STRS"											
214*LBL "KS"						263*LBL "L"					
215 "†="						264 E					
216 ARCL X						265 GT0 01					
217 "† KSI"						266*LBL "LL"					
218 RTN						267 4					
219*LBL "M="	M ⁿ					268*LBL 01					
220 12						269 RDN					
221 /	M ⁿ					270 FS?C 14					
222 "† M="						271 GT0 19					
223 ARCL X						272 "-----"					
224*LBL "FK"						273 GT0 06					
225 "† 'K"						274*LBL "NO"					
226 RTN						275 "†"					
227*LBL "M"						276 RTN					
228 12						277*LBL "L*"					
229 CF 22						278*LBL 19					
230 "†M, 'K"						279 "*****"					
231 RTN						280 4					
232*LBL "QUAD"						281*LBL "CL"	Counter	X	Y	Z	
233*LBL 08						282 RDN					
234 "A"						283*LBL 06	X	Y	Z	Counter	
235 PROMPT						284 FC? 55					
236 "B"						285 RTN					
237 PROMPT						286 AST0 L					
238 "C"	C	B	A		$Ax^2 + By - C = 0$						
239 PROMPT						287*LBL 00					
240 X<> Z	A	B	C			288 ARCL L					
241 XROM "Q"	R1	R2				289 DSE T					
242 SF 21						290 GT0 00					
243 "R1"						291 SF 25					
244 XROM "V"						292 PRA					
245 X<>Y	R2	R1				293 CF 25					
246 "R2"						294 RTN	X	Y	Z	O	
247 XROM "V"						295*LBL "BD"					
248 GT0 08						296 RCL 03	Prev.B				
249*LBL "Q"	A	B	C			297 "B"	B				
250 ST/ Z						298 PROMPT					
251 ST+ X	Z A	D	C/A			299 ST0 03					
252 /						300 RCL 04	Prev.D				
253 CHS	-B/2A	C/A				301 "D"					
254 ENTER†	B ² /4A ²	-B/2A	-B/2A	C/A		302 PROMPT	D				
255 ENTER†						303 ST0 04					
256 X†2	C/A					304 RTN					
257 R†	$\frac{B^2 - 4AC}{4A^2}$				$\frac{C}{A} = \frac{4AC}{4A^2}$						
258 -											
259 SQRT	$\frac{\sqrt{B^2 - 4AC}}{2A}$	-B/2A	-B/2A								
260 ST- Z			R2								
261 +	R1	R2									
262 RTN											

	X	Y	Z	T	L		X	Y	Z	T	L
						305*LBL "W"					
						306 XROM "S"					
						307 CF 06					
						308 "WSD?"					
						309 E	1				
						310 PROMPT					
						311 X=0?					
						312 SF 06					
						313 RTN					
						314*LBL "AA"					
						315 ASTO Y	Max.Reg	"AKC"	Z	T	
						316 XROM "S"	"AISC"	Z	T		
						317 CLA					
						318 ARCL X					
						319*LBL 11					
						320 FC? 04					
						321 "ARSHTO"					
						322 FC? 00					
						323 AVIEW					
						324 RTN	"AISC"	Z	T		
						325*LBL "OK"					
						326 FS? 07					
						327 "I-NG"					
						328 FC? 07					
						329 "I-OK"					
						330 FS? 07					
						331 BEEP					
						332 END					

	X	Y	Z	T	L		X	Y	Z	T	L
01*LBL "FIS"	Ft	Y	Z	T		48*LBL "FT"	FIS	Y	Z	T	
02 XEQ 01	F.IS	Y			Ft	49 XEQ 01	Ft.				
03 XEQ 04	Y	Z	T	F.IS		50 XEQ 03	Ft.				F.IS
04 XEQ 02	F.IS	Y	Z	T	Ft.	51 XEQ 07	Y	Z	T	Ft.	
05 Rt						52 XEQ 02	Ft.dec	Y	Z	T	F.IS
06 RTN						53 Rt					
07 FIX 6						54 RTN					
08 STOP											
09 FIX 4						55*LBL 05	FIS ₁	FIS ₂	Z	T	
10 STOP						56 XEQ 01	FIS ₁	FIS ₂			
						57 XEQ 03	Ft.dec ₁	FIS ₂			
11*LBL "FIS--"	Ft ₂	Ft ₁				58 X<>Y	FIS ₂	Ft.dec			
12 XEQ 05	Ft ₁	Ft ₂									
13 X<>Y	ΔFt.					59*LBL 03	F.IS	Y			
14 -						60 INT	Ft				
15 GTO 06						61 LASTX					
						62 FRC	.IS	Ft	Y		
16*LBL "FIS+"						63 E2	100	.IS	Ft.	Y	
17 XEQ 05	Ft ₂	Ft ₁				64 *	In.Six				
18 +	Ft					65 FRC	.Six				
						66 .16	.16	.Six	Ft.	Y	
19*LBL 06						67 ST/ Y	.16	.dec	Ft.	Y	
20 XEQ 04	F.IS					68 RDN					
21 RCL \	T	F.IS				69 LASTX	In.				
						70 INT	In.	.dec	Ft.	Y	
22*LBL 02	X	Y				71 +	In.dec				
23 RCL \	T					72 12					
24 RCL]	Z	T	X	Y		73 /	.dec	Ft.	Y	Y	
25 Rt	Y	Z	T	X		74 +	Ft.dec	Y	Y	Y	
26 RTN						75 RTN					
27*LBL 04	Ft					76*LBL 01	X	Y	Z	T	
28 INT	Ft.					77 FIX 4	X				
29 LASTX						78 STO [
30 FRC	.Ft					79 Rt	T				
31 12	12	Frc	Ft.	Y		80 STO \					
32 *	In.					81 Rt	Z				
33 FRC	FRC					82 STO]					
34 .16	.16	FRC"	Ft.	Y		83 Rt	X	Y	Z	T	
35 ST* Y	.16	Sixteenth	Ft.	Y		84 Rt					
36 RDN						85 END					
37 LASTX	In.										
38 INT	In.	Sixteenth	Ft.	Y							
39 +	In.16 th										
40 E2	100										
41 /	.In.16 th	Ft.	Y	Y							
42 +	F.IS										
43*LBL 07											
44 RCL [Ft										
45 SIGN	1										
46 RDN	F.IS	Y	Y	1	Ft in Last X						
47 RTN											

	X	Y	Z	T	L		X	Y	Z	T	L
01*LBL "IP"						50 "Z="					
02 SF 21						51 ARCL Z					
						52 AVIEW					
03*LBL 02						53 "T="					
04 .						54 ARCL T					
05 "X1+Y1"						55 AVIEW					
06 PROMPT	Y ₁	X ₁				56 "L="					
07 X<>Y	X ₁	Y ₁				57 ARCL L					
08 "X3+Y3"						58 AVIEW					
09 PROMPT	Y ₃	X ₃	X ₁	Y ₁		59 ADV					
10 R↑						60 CLX					
11 STO J	Y ₁	Y ₃	X ₃	X ₁		61*LBL 11	counter				
12 -	ΔY	X ₃	X ₁	X ₁		62 "R."					
13 X<>Z	X ₁	X ₃	ΔY	X ₁		63 XROM "AX"					
14 -	ΔX	ΔY	X ₁	X ₁		64 SF 25					
15 /	ΔY/ΔX	Y ₁	X ₁	X ₁		65 "t="					
16 RCL J	Y ₁					66 ARCL IND X					
17 X<>Y	ΔY/ΔX	Y ₁	X ₁	X ₁		67 FC?C 25					
18 R↑	X ₁	ΔY/ΔX	Y ₁	X ₁		68 STOP					
						69 AVIEW					
19*LBL 10						70 ISG X	counter				
20 RDN						71 "	NOP				
21 STO a	Y/X	Y ₁	X ₁	X ₁		72 GTO 11					
22 CF 22											
23 "X2"						73*LBL "RH"	ρ				
24 PROMPT	X ₂					74 SF 21					
25 FC? 22						75 ENTER↑	ρ	ρ			
26 GTO 02						76 "RHO"					
27 R↑	X ₁	X ₂	Y/X	Y ₁		77 XEQ "V"					
28 STO J						78 RCL 02	ΔX/ρ				
29 -	ΔX	Y/X	Y ₁	Y ₁		79 /					
30 *	Y/X					80 .75					
31 X<>Y	Y ₁					81 *	ρ/ρ _b	ρ			
32 +	Y ₂					82 LASTX	.75				
33 RCL J	X ₁					83 X<>Y					
34 LASTX	Y ₁					84 X<Y?					
35 RCL a	ΔY/ΔX					85 FC? 00					
36 R↑	Y ₂	ΔY/ΔX	Y ₁	X ₁		86 X=0?					
37 "Y2="						87 RTN					
38 ARCL X						88 "P/ρb="	ρ/ρ _b	.75	ρ		
39 AVIEW						89 ARCL X					
40 XEQ "L"						90 CF 07					
41 GTO 10						91 X>Y?					
						92 SF 07					
42*LBL "ST"						93 XEQ "OK"					
43 SF 21						94 XEQ "PV"					
44 "X="						95 END					
45 ARCL X											
46 AVIEW											
47 "Y="											
48 ARCL Y											
49 AVIEW											

$$Y_2 = (Y_3 - Y_1) \left(\frac{X_2 - X_1}{X_3 - X_1} \right) + Y_1$$

X<Y?
and
FS:00

PPC ROM PROGRAMS

This package contains, by permission, four programs from the PPC ROM. By way of explanation, PPC is a large user's group dedicated to personal programmable calculators and computers; thus the name. Several years ago they undertook to write a group of math and general utilities programs to fill what they saw as gaps in the HP-41 function set, and to expand its usefulness. These programs were assembled into an 8K ROM module which has become something of a legend, both for its usefulness, and general excellence of programming and documentation.

The programs from the PPC ROM which have been included in this module are:

- FI Financial analysis
- CV Curve fitting (linear, exponential, log, power)
- BC Block clearing of registers
- SV Solution of a function $f(x)=0$ on an interval

Documentation from the original manual has been included. However, three minor changes must be noted.

First, the line numbers of the routine listings in the original documentation do not correspond to the line numbers in this module for BC and SV, due to rearrangement of the routines. Local labels have also been changed in these two routines for the same reason.

Second, BC has been modified to use the Extended Function CLRGX if it is available. This function does the same thing as the original BC but is much faster. However, the stack is affected differently by CLRGX than by BC, so carry data in the stack with caution if using this function.

Third, SV has been modified to follow this package's use of flag 00. To view successive iterations, clear flag 00 instead of setting flag 10 as the original documentation indicates.

FINANCIAL ANALYSIS

Gto (not xeq) FI (SIZE 010)

SOFTKEYS/INPUT SUMMARY (USER mode ON)

<u>Softkey</u>	<u>Input/Function</u>
a	Multiplies contents of X by 12 and stores result as "n"
b	Divides contents and stores result as "%i"
c	Toggles between <u>C</u> ontinuous and <u>D</u> iscrete compounding
d	Toggles between <u>B</u> eginning or <u>E</u> nd of period payment
e	Clears financial registers n, %i, PV, PMT, and FV. Sets the compounding frequency and the payment to the default value of unity. Displays a two-character mnemonic indicator of softkeys c and d (<u>C</u> or <u>D</u> , <u>B</u> or <u>E</u>)
A	Enters or solves for number of periods, n (the total number of payments or compounding periods).
B	Enters or solves for the interest rate. %i is the nom- inal rate for the period implied by the compounding and payment frequency values (usually the nominal annual rate).
C	Enters or solves for the present value.
D	Enters or solves for the periodic payment.
E	Enters or solves for the future value.
H	Enters the compounding frequency if different than the default, 1. CF is the number of times the interest rate is compounded during the period implied by %i. CF is ignored when continuous compounding is specified.
I	Enters the payment frequency when different than the default, 1. PF is the number of payment periods occur- ing during the period implied by %i. When no payments are involved, PF must be set equal to CF.
J	Displays a two-character mnemonic indicator of soft- keys c and d (<u>C</u> or <u>D</u> , <u>B</u> or <u>E</u>)

USER FLAGS

Flag 00	clear	View succeeding approximations of %i
	set	View only final solution when solving for %i

OUTPUT SUMMARY

Output consists of the correct value in the X-register. No output description or units are given. See the examples in the original PPC documentation which follows.

PROGRAM FLAGS

Flag 08	clear	Discrete compounding
	set	Continuous compounding
Flag 09	clear	End of period payments
	set	Beginning of period payments

STORAGE REGISTER USE

<u>Register</u>	<u>Value</u>
00	not used
01	n
02	%i
03	PV
04	PMT
05	FV
06	function call number (indirect label)
07	%i as decimal
08	CF
09	PF

a. the compounding frequency CF is not identical to the payment frequency PF, and/or,

b. Interest is compounded in either discrete intervals or is continuously compounded.

When flag F08 is cleared, the discrete case is selected. When flag F08 is set, the continuous case is selected. F08 is toggled by LBL c.

Solving For n, PV, PMT, or FV.

When a solution for n, PV, PMT, or FV is required, the nominal annual interest rate I, supplied by the user, must first be converted to the effective interest rate per payment period by LBL 07. This rate, i_e , is then used by LBL 01, 03, 04, or 05 respectively to calculate the selected variable. To convert I to i_e the following expressions are used:

$$(19) \quad i_e = (1 + I/CF)^{CF/PF} - 1 \quad (\text{discrete case})$$

$$(20) \quad i_e = e^{(I/PF)} - 1 \quad (\text{continuous case})$$

where:

I = nominal annual interest rate
 i_e = effective interest rate per pmt. period
 CF = compounding frequency per year
 PF = payment frequency per year

Solving for Interest

When a solution for interest is required, LBL 06 (for PMT≠0) or LBL 09 (for PMT=0) produces i_e as the calculated interest value. This value of i_e must then be converted to I using LBL 11. It is the value of I, not i_e which is returned as a percentage to the X-register and register R02.

To convert i_e to I, the following expressions are used:

$$(21) \quad I = CF[(1 + i_e)^{PF/CF} - 1] \quad (\text{discrete case})$$

$$(22) \quad I = \text{LN}[(1 + i_e)^{PF}] \quad (\text{continuous case})$$

The common label, LBL 08

Common to all calculations is LBL 08 which is used to calculate the values of A, A+1, B, and C for use in solving the selected variable. After executing the RTN instruction following LBL 08 the stack and LAST X registers contain the following data values:

Register:	Contents:
LAST X	B
T	A+1
Z	A+1
Y	A
X	C

These values are all calculated using i_e and are then used in equations (8) to (15) as selected.

Routine Listing For:

FI

01*LBL "FI"	74 ABS
02 GTO IND 06	75 RCL 05
03*LBL e	76 ABS
04*LBL 00	77 +
05 E	78 RCL 04
06 STO 00	79 X=0?
07 STO 09	80 GTO 09
08 CLX	81 /
09 STO 01	82 ABS
10 STO 02	83 1/X
11 STO 03	84 LASTX
12 STO 04	85 RCL 01
13 STO 05	86 3
14 GTO 10	87 Y+X
15*LBL c	88 /
16 FC?C 08	89 +
17 SF 08	90 STO 07
18 GTO 10	91*LBL 06
19*LBL d	92 XEQ 08
20 FC?C 09	93 STO 02
21 SF 09	94 RCL 03
22*LBL J	95 +
23*LBL 10	96 STO Z
24 "D"	97 X<>Y
25 FS? 08	98 ST* 02
26 "C"	99 *
27 FC? 09	100 RCL 03
28 "E"	101 +
29 FS? 09	102 RCL 05
30 "B"	103 +
31 ASTO X	104 X<> Z
32 RTN	105 *
33*LBL H	106 RCL 07
34 STO 08	107 FS? 10
35 CF 22	108 VIEW X
36 RTN	109 E
37*LBL I	110 +
38 STO 09	111 /
39 CF 22	112 RCL 01
40 RTN	113 *
41*LBL a	114 RCL 02
42 12	115 RCL 07
43 *	116 /
44*LBL A	117 -
45 FS? 22	118 /
46 STO 01	119 ST- 07
47 FS?C 22	120 RCL 07
48 RTN	121 /
49*LBL 01	122 E2
50 XEQ 07	123 *
51 STO Z	124 RND
52 RCL 05	125 X=0?
53 -	126 GTO 06
54 R+	127 GTO 11
55 RCL 03	128*LBL 07
56 +	129 E
57 /	130 RCL 02
58 LH	131 %
59 RCL 07	132 RCL 08
60 LM1+X	133 RCL 09
61 /	134 FS? 08
62 STO 01	135 X<>Y
63 RTN	136 RDN
64*LBL b	137 /
65 12	138 STO 07
66 /	139 LM1+X
67*LBL B	140 RCL 08
68 FS? 22	141 RCL 09
69 STO 02	142 /
70 FS?C 22	143 *
71 RTN	144 FS? 08
72*LBL 02	145 X<> 07
73 RCL 03	146 E+X-1

Listing continued on page 161.

Routine Listing For: FI	
147 STO 07	196 STO 03
148*LBL 08	197 FS?C 22
149 E	198 RTN
150 RCL 07	199*LBL 03
151 FS? 09	200 XEQ 07
152 ST+ Y	201 *
153 /	202 RCL 05
154 E	203 +
155 RCL 01	204 R+
156 RCL 07	205 /
157 LN1+X	206 CHS
158 *	207 STO 03
159 E+X-1	208 RTN
160 +	209*LBL D
161 LASTX	210 FS? 22
162 RCL 04	211 STO 04
163 R+	212 FS?C 22
164 *	213 RTN
165 RTN	214*LBL 04
166*LBL 09	215 XEQ 07
167 RCL 05	216 X<> L
168 RCL 03	217 *
169 /	218 CHS
170 CHS	219 RCL 03
171 LN	220 R+
172 RCL 01	221 *
173 /	222 RCL 05
174 E+X-1	223 +
175 STO 07	224 X<>Y
176*LBL 11	225 /
177 CLD	226 STO 04
178 RCL 07	227 RTN
179 LN1+X	228*LBL E
180 RCL 09	229 FS? 22
181 *	230 STO 05
182 RCL X	231 FS?C 22
183 RCL 08	232 RTN
184 /	233*LBL 05
185 E+X-1	234 XEQ 07
186 RCL 08	235 RCL 03
187 *	236 +
188 FS? 08	237 *
189 X<>Y	238 RCL 03
190 E2	239 +
191 *	240 CHS
192 STO 02	241 STO 05
193 RTN	242*LBL 12
194*LBL C	243 END
195 FS? 22	

Mathematically, the two sequences produce identical results. However, over the range of numbers typically encountered, the LN1+X and E+X-1 instructions prevent the severe loss of significant digits which occurs in the old sequence at the +1 and -1 steps. Reference 1 provides two examples for accuracy checking, as follows.

Examples for Accuracy Checking

These examples may be used to compare the accuracy of **FI** with other financial programs or calculators.

A.

1. Execute LBL e to clear all data registers
2. Select DISCRETE and END status (DE)
3. Key in the following variables

n = 111.1111111
 %I = 2.222222222
 PV = 333.3333333 (\$)
 PMT = 4.444444444 (\$)

4. Solve for FV
 The displayed FV = -5931.822943
 The true FV = -5931.822944

B.

1. Execute LBL e to clear data registers
2. Select DISCRETE and END status (DE)
3. Key in the following variables

n = 63
 %I = 0.000001610
 PV = 0 (no need to enter this)
 PMT = -1,000,000.00 (\$)

4. Solve for FV
 The displayed FV = 63,000,031.43 (\$)
 The true FV = 63,000,031.44 (\$)

5. Now set FV = 0

6. Solve for PV
 The displayed PV = 62,999,967.55 (\$)
 The true PV = 62,999,967.54 (\$)

The above examples are taken from Reference 3 and are © copyright 1977, Hewlett-Packard Company. Reproduced with permission.

Accuracy Enhancement:

The accuracy has been improved by the use of a new instruction sequence to calculate the A term:

$$(1+i)^n - 1$$

Assuming that n is stored in R01 and i (decimal) is stored in R07, the A term can be calculated in two different sequences, as follows:

Old Sequence:

```

RCL 07    i
1
+         1+i
RCL 01    n
yx       (1+i)n
1
-         (1+i)n - 1

```

New Sequence:

```

RCL 07    i
LN1+X     LN(1+i)
RCL 01    n
*         n*LN(1+i)
E+X-1     (1+i)n - 1

```

Simplified Solution Sequence

1. Solving for n:

- a. Calculate i_e using equations (19) or (20), and LBL 07.
- b. Calculate A, A+1, B, and C using equations (5), (6), and (7), and LBL 08.
- c. Calculate n, using equation (8) and LBL 01.
- d. Store n in R01 and halt.

2. Solving for i:

A. PMT=0

- a. Calculate i_e using equation (9) and LBL 09
- b. Calculate i using equations (21) or (22) and LBL 11.
- c. Store i and halt.

(C20)

FI - FINANCIAL CALCULATIONS

This is a complete financial program that uses the top two rows of keys to either input or solve for the five standard financial values; n , i , PV , PMT , FV .

This highly accurate program extends the capabilities of previous HP financial calculators and programs by adding two new parameters.

1. "CF" The Compounding Frequency can be specified (including continuous compounding) and may be different than the payment frequency.
2. "PF" The Payment Frequency can be specified and may be different from the compounding frequency.

This added facility simplifies the solution of some complex financial problems that are difficult to solve via the standard financial calculator or program. Canadian and European style mortgage problems can now be handled in a simple straightforward manner.

A Beginning/End of period switch is provided and a status display function allows the user to determine the current state of toggle controlled functions. The "CLEAR" financial register function incorporates default parameters that permit the user to operate the program in the same manner as typical financial calculators or programs that do not include the facility to specify different compounding and payment periods. Standard financial sign conventions are used (money paid out is negative, money received is positive).

The $LN1+X$ and $E^{1X}-1$ functions are used in compounding routines instead of Y^{1X} , resulting in more precise answers than are produced by most financial programs and calculators.

BACKGROUND FOR FI

Time Value of Money:

If you borrow money you can expect to pay rent or interest for its use; conversely you expect to receive interest on money you loan or invest. When you rent property, equipment, etc., rental payments are normal; this is also true when renting or borrowing money. Therefore, money is considered to have a "time value". Money available now, has a greater value than money available at some future date, because of its rental value or the interest that it can produce during the intervening period.

Simple Interest:

If you loaned \$800 to a friend with an agreement that at the end of one year he would repay you \$896, the "time value" you placed on your \$800 (principal) was \$96 (interest) for the one year period (term) of the loan. This relationship of principal, interest and time (term) is most frequently expressed as an Annual Percentage Rate (APR). In this case the APR was 12.0% $[(96/800)100]$. This example illustrates the four basic factors involved in a simple interest case. The time period (one year), rate (12.0% APR), present value of the principal (\$800) and the future value of the principal including interest (\$896).

Compound Interest:

In many cases the interest charge is computed periodically during the term of the agreement. For example, money left in a savings account earns interest that is periodically added to the principal and in turn earns additional interest during succeeding interest periods. The accumulation of interest during the investment period represents compound interest. If the loan agreement you made with your friend had specified a "compound interest rate" of 12.0% (compounded monthly) the \$800 principal would have earned \$101.46 interest for the one year period. The value of the original \$800.00 would be increased by 1% the first month to \$808.00 which in turn would be increased by 1% to \$816.08 the second month, reaching a future value of \$901.46 after the twelfth iteration. The monthly compounding of the nominal annual rate (NAR) of 12% produces an effective Annual Percentage Rate (APR) of 12.683% $[(101.46/800)100]$. Interest may be compounded at any regular interval; annually, semiannually, monthly, weekly, daily, even continuously (a specification in some financial models).

Periodic Payments:

When money is loaned for longer periods of time it is customary for the agreement to require the borrower to make periodic payments to the lender during the term of the loan. The payments may be only large enough to repay the interest, with the principal due at the end of the loan period (an interest only loan), or large enough to fully repay both the interest and principal during the term of the loan (a fully amortized loan). Many loans fall somewhere between, with payments that do not fully cover repayment of both the principal and interest. These loans require a larger final payment (balloon) to complete their amortization. Payments may occur at the beginning or end of a payment period. If you and your friend had agreed on monthly repayment of the \$800 loan at 12.0% NAR compounded monthly, twelve payments of \$71.08 for a total of \$852.96 would be required to amortize the loan. The \$101.46 interest from the annual plan is more than the \$52.96 under the monthly plan because under the monthly plan your friend would not have had the use of \$800 for a full year.

Financial Transactions:

The above paragraphs introduce the basic factors that govern most financial transactions; the time period, interest rate, present value, payments, and the future value. In addition, certain conventions must be adhered to; the interest rate must be relative to the compounding frequency and payment periods, and the term must be expressed as the total number of payments (or compounding periods if there are no payments). Loans, leases, mortgages, annuities, savings plans, appreciation, and compound growth are among the many financial problems that can be defined in these terms. Some transactions do not involve payments, but all of the other factors play a part in "time value of money" transactions. When any one of the five (four- if no payments are involved) factors is unknown, it can be derived from formulas using the known factors. This is the function of the FI financial program.

Problem Solving Preliminaries:

Diagram or visualize the positive and negative cash flows and their timing. (See cash flow diagrams)

Clear the financial registers by pressing **e**, unless the problem is a continuation or minor change from the preceding problem. Note that clearing also sets the compounding and payment frequency values to 1.

Check the mnemonic status code for applicability to the current problem and change it by pressing **c** and/or **d** if necessary. The mnemonic status codes are:

CB = Continuous compounding and Beginning of period payments.
 CE = Continuous compounding and End of period payments.
 DB = Discrete compounding and Beginning of period payments.
 DE = Discrete compounding and End of period payments.

Specify the compounding and payment frequencies by entering appropriate values and pressing **H** and/or **I**.

Generalized Cash Flow Diagrams:

Selection of the proper parameters and signs of the factors in a specific financial transaction can often be aided by constructing a cash flow diagram similar to the examples below:

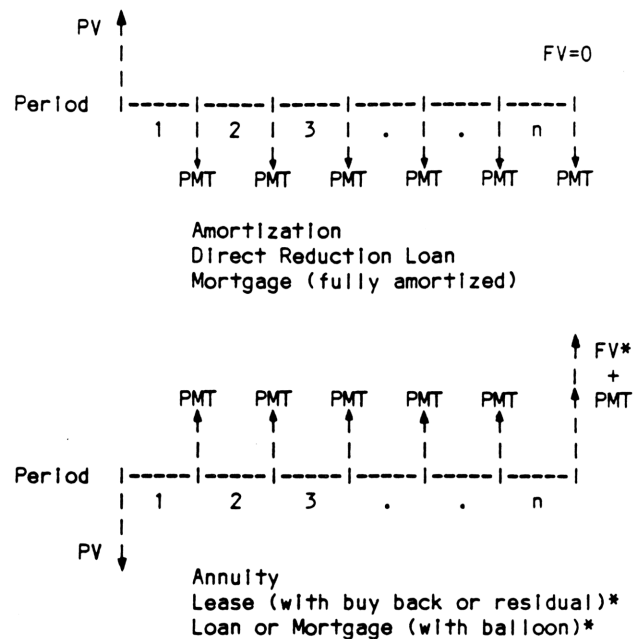
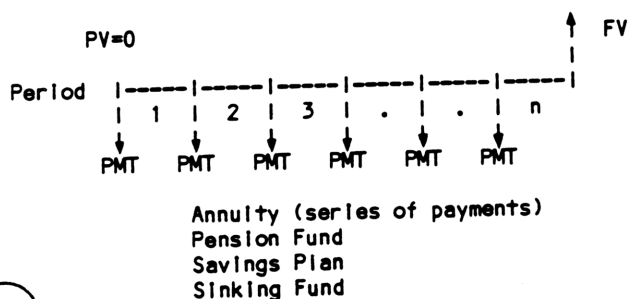
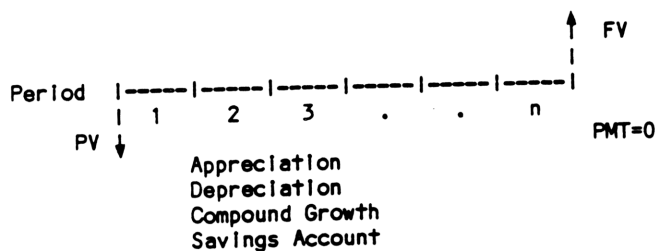
Standard Financial Conventions are:

Money RECEIVED is a POSITIVE value and is represented by an arrow above the line.

Money PAID OUT is a NEGATIVE value and is represented by an arrow below the line.

If payments are a part of the transaction the number of payments must equal the number of periods (**n**).

Payments may be represented as occurring at the end or beginning of the periods.



COMPLETE INSTRUCTIONS FOR **FI**

By manually keying **GT0 "FI"** the program pointer is in the ROM and the following functions become available on the top two rows of the keyboard. A minimum size required for **FI** is SIZE 010.

12 X	12 ÷	CTS./ DISCRETE	BEGIN/ END	CLEAR FI REG.
n	i	PV	PMT	FV
		CF	PF	STATUS

The functions provided are summarized below.

KEY	FUNCTION	(FLAG/REG)
a	Multiplies contents of X by 12 and stores result as n. n=12X	R01
b	Divides contents of X by 12 and stores result as %I. %I=X/12	R02
c	Toggles flag 08 to specify Continuous (F08 set) or Discrete (F08 clear) compounding. Status display shows C or D.	F08
d	Toggles flag 09 to specify Beginning of period payments (F09 set) or End of period payments (F09 clear). Status display shows B or E.	F09
e	Clears financial registers n, %I, PV, PMT, FV, and sets the compounding frequency (R08) and the payment frequency (R09) both to the default value of 1. CF=PF=1. Also displays a two character mnemonic Indicator of the status of F08 and F09.	R01-R05 R08 R09

CTU

A Enters or Solves for number of periods.
n is the total number of payments during the full term of the transaction, or if no payments are made n is the total number of compounding periods.

R01
n

Do:

See:

Result:

e	"DE"	Clear, Discrete/End status
1 A	1.00	n=1
800 CHS C	-800.00	PV=\$800.00
896 E	896.00	FV=\$896.00
B	12.00	APR=%I=12%

B Enters or Solves for the interest rate.**
%I is the nominal rate for the period implied by the compounding and payment frequency values in R08 and R09 (usually the nominal annual rate).

R02
%I

Example 2: Compound Interest. Find the future value of \$800 after one year at a nominal rate of 12% compounded monthly. No payments are specified, so the payment frequency is set equal to the compounding frequency.

C Enters or Solves for the present value.
Use standard financial sign conventions.

R03
PV

D Enters or Solves for periodic payment.
Use standard financial sign conventions.

R04
PMT

Do:

See:

Result:

e	"DE"	Clear, Discrete/End status
12 A	12.00	n=12
H I	12.00	CF=PF=12
12 B	12.00	keyboard Input required to store NAR=12%=%I
800 CHS C	-800.00	PV=\$800.00
E	901.46	FV=\$901.46

E Enters or Solves for future value.
Use standard financial sign conventions.

R05
FV

H Enters the compounding frequency.
CF is the number of times the interest rate is compounded during the period implied by the interest rate %I. When continuous compounding is specified the value in R08 is ignored.

R08
CF

I Enters the payment frequency.
PF is the number of payment periods occurring during the period implied by the interest rate %I. When no payments are involved PF must be set equal to CF. For continuous compounding cases where PMT = 0, set PF = 1.

R09
PF

Example 3: Periodic Payment. Find the monthly end-of-period payment required to fully amortize the loan in Example 2. A fully amortized loan has a future value of zero. Use data retained from Example 2.

Do:

See:

Result:

0 E	0.00	Set FV=\$0.00
D	71.08	PMT=\$71.08

J Displays a two character mnemonic indicator of the status of flags F08 and F09.
C = Continuous D = Discrete
E = End of period B = Beginning of period

** When solving for %I the first guess and succeeding approximations of I (decimal) may be VIEWed during each iteration by setting flag 10.

F10

WARNING: Solutions using or resulting in a zero rate of interest (%I) will cause a "DATA ERROR".

Do:

See:

Result:

e	"DE"	Clear, Discrete/End status
12 H I	12.00	CF=PF=12
13.25 B	13.25	NAR=13.25%=%I
100000 C	100,000.00	PV=\$100,000.00
1125.75 CHS D	-1,125.75	PMT=\$1,125.75
A	360.10	#pmts=n=360.10

MORE EXAMPLES OF FI

In the keystroke solutions shown for each example, the lower case letters a through e represent shifted key functions of keys A through E. Key in the indicated quantities and press the user defined keys as indicated in the "Do" column. Contents of the display at significant points in the solution are shown in the "See" column and are followed by identification in the "Result" column. Before running these examples, perform "MEMORY LOST" and set FIX 2 display mode. A suggestion is to go through all 15 examples, one after the other. Key GTO "FI" and set USER mode.

Example 5: Final Payment. Using the same data as in the preceding example, find the amount of the final payment if n is changed to 360. The final payment will be equal to the regular payment plus any balance (FV) remaining at the end of period number 360.

Do:

See:

Result:

360 A	360.00	Set n=360 exactly
E	-108.87	FV=\$108.87
RCL 04	-1,125.75	Recall PMT
+	-1,234.62	Final PMT=\$1,234.62

Example 1: Simple Interest. Find the annual simple interest rate (%) for an \$800 loan to be repaid at the end of one year with a single payment of \$896.

Example 6: Balloon Payment. On long term loans, small changes in the periodic payments can generate large changes in the future value. If the monthly payment in the preceding example is rounded down to \$1,125.00 how much additional (balloon) payment will be due with the final regular payment?

C13

Do:	See:	Result:
1125 CHS D	-1,125.00	Set PMT=\$1,125.00 even
E	-3,580.00	Additional balloon payment = \$3,580.00

Example 7: Canadian Mortgage. Find the monthly end-of-period payment necessary to fully amortize a 25 year \$85,000 loan at 11% compounded semiannually.

Do:	See:	Result:
e	"DE"	Clear, Discrete/End status
2 H	2.00	CF=2
12 I	12.00	PF=12
25 a	300.00	n=300
11 B	11.00	NAR=11%=\$1
85000 C	85,000.00	PV=\$85,000.00
D	-818.15	PMT=\$818.15

Example 8: European Mortgage. The "effective annual rate (EAR)" is used in some countries (especially in Europe) in lieu of the nominal annual rate commonly used in the United States and Canada. For a 30 year \$90,000 mortgage at 14% (EAR) compute the monthly end-of-period payments. When using an EAR, the compounding frequency (CF) is set to 1.

Do:	See:	Result:
e	"DE"	Clear, Discrete/End status (CF=1 after clearing)
12 I	12.00	PF=12
30 a	360.00	n=360
14 B	14.00	EAR=14%=\$1
90000 C	90,000.00	PV=\$90,000.00
D	-1,007.88	PMT=\$1,007.88

Example 9: Bi-Weekly Savings. Compute the future value of bi-weekly savings of \$100 for 3 years at a nominal annual rate of 5.5% compounded daily. Note: Set status to "DB".

Do:	See:	Result:
e	"DE"	Clear, Discrete/End status
d	"DB"	Discrete/Begin Status
365 H	365.00	CF=365
26 I	26.00	PF=26
3 X A	78.00	n=3x26=78
5.5 B	5.50	NAR=5.5%=\$1
100 CHS D	-100.00	PMT=\$100.00
E	8,489.32	FV=\$8,489.32

Example 10: Present Value - Annuity Due. What is the present value of \$500 to be received at the beginning of each quarter over a 10 year period if money is being discounted at 10% NAR compounded monthly?

Do:	See:	Result:
e	"DB"	Clear, Discrete/Begin status
12 H	12.00	CF=12
4 I	4.00	PF=4
40 A	40.00	n=40
10 B	10.00	NAR=10%=\$1
500 D	500.00	PMT=\$500.00
C	-12,822.64	PV=\$12,822.64

Example 11: Balloon Payment @ n+1. Compute the monthly end-of-period payment on a 3 year \$20,000 loan at 15% NAR compounded monthly, with a \$10,000 balloon payment due at the end of the 37th period. The balloon payment must be discounted one period to make it coincide with the last regular payment. Note: Set status to "DE".

Do:	See:	Result:
e	"DB"	Clear Financial
d	"DE"	Set Discrete/End status
12 H I	12.00	CF=PF=12
3 a	36.00	n=36
15 B	15.00	NAR=15%=\$1
20000 C	20,000.00	PV=\$20,000.00
XEQ 07	0.00	Calculate I as a decimal and leave in R07
10000	10,000.00	Start calculation of 10,000/(1+i) = FV
RCL 07 1 +	1.01	1+i
/ CHS E	-9876.54	FV=\$9,876.54 (discounted)
D	-474.39	PMT=\$474.39

The balloon payment was discounted by executing LBL 07 (XEQ 07) to develop the effective interest rate in R07. The \$10,000 was then divided by (1+i) and entered in E as the discounted future value of the balloon payment.

Example 12: Effective Rate - 365/360 Basis. Compute the effective annual rate (%APR) for a nominal annual rate of 12% compounded on a 365/360 basis used by some Savings & Loan Associations.

Do:	See:	Result:
FIX 3 e	"DE"	Set up display & status
365 A	365.000	n=365
H	365.000	CF=365
360 I	360.000	PF=360
12 B	12.000	NAR=12%=\$1
100 CHS C	-100.000	PV=\$100.00
E	112.935	FV=\$112.94
RCL 03 +	12.935	%APR=12.935%
FIX 2	12.94	Return to normal display

Example 13: Mortgage with "points". What is the true APR of a 30 year, \$75,000 loan at a nominal rate of 13.25% compounded monthly, with monthly end-of-period payments of \$844.33 if 3 "points" are charged? The PV must be reduced by the dollar value of the points and/or any lenders fees to establish an effective PV. Because the payments remain the same the true APR will be higher than the nominal rate.

Do:	See:	Result:
e	"DE"	Clear, Discrete/End status
12 H I	12.00	CF=PF=12
30 a	360.00	n=360
75000 ENTER		PV=\$75,000.00
3 XEQ "%I" - C	72,750.00	PMT=\$844.33
844.33 CHS D	-844.33	
B	13.69	True APR=13.69%

Example 14: Equivalent Payments. Find the equivalent monthly payment required to amortize a 20 year \$40,000 loan at 10.5% NAR compounded monthly, with 10 annual payments of \$5,029.71 remaining. Compute PV of the remaining annual payments, then change n and PF to a

monthly basis and compute the equivalent monthly PMT.

Do:	See:	Result:
e	"DE"	Clear, Discrete/End status (PF=1 after clearing)
12 H	12.00	CF=12
10 A	10.00	n=10
10.5 B	10.50	NAR=10.5%=.1
5029.71 CHS D	-5,029.71	PMT=\$5,029.71
C	29,595.88	PV=\$29,595.88
12 I	12.00	PF=12, set monthly basis
10 a	120.00	n=120 (monthly)
D	-399.35	PMT=\$399.35 (monthly)

Example 15: Perpetuity - Continuous Compounding

If you can purchase a single payment annuity with an initial investment of \$60,000 that will be invested at 15% NAR compounded continuously, what is the maximum monthly return you can receive without reducing the \$60,000 principal? If the interest rate is constant and the principal is not disturbed the payments can go on indefinitely (a perpetuity). Note that the term "n" of a perpetuity is immaterial. It can be any non-zero value. Set status to "CE".

Do:	See:	Result:
e	"DE"	Clear, Discrete/End status (CF=1 after clearing)
c	"CE"	Continuous/End status
12 A	12.00	n=12
I	12.00	PF=12
15 B	15.00	NAR=15%=.15
60000 E	60,000.00	FV=\$60,000.00
CHS 1 X C	-60,000.00	Data entry flag is set so PV is stored as \$60,000.00
D	754.71	PMT=\$754.71

SUPPORTIVE PROGRAMS FOR FI

There are two optional routines provided below to extend the capability of the ROM routine FI. These routines are not located in the ROM, and must be loaded into RAM memory for their execution. They are named LPAS and FAST.

1. LBL LPAS

LBL LPAS "Loan Payments and Amortization Schedule" is really a full program in its own right, although it does use ROM routines FI, CJ, and CP. LPAS extends the capabilities of FI to accommodate "shifted" payment situations, when the first periodic payment does not fall at the beginning (BEGIN) or the end (END) of the first period, but at any date after the effective date. LPAS also provides an amortization schedule as an option.

2. LBL FAST - Reducing Interest Solution Time

LBL FAST is an optional routine used when solving for interest. Its purpose is to provide an initial starting guess for the interest-solving loop which is closer to the exact solution than that provided by LBL FI initial guess. The result is that interest solving execution time is usually shorter.

Don Dewey (5148) produced both supporting programs.

APPLICATION PROGRAM 1 FOR FI

LPAS - Loan Payments and Amortization Schedule

The FI program, like most financial programs and calculators, assumes that the first periodic payment occurs on either the first or last day of the payment period as specified by the beginning of period/end of period switch or toggle. Many financial agreements do not follow this convention. An agreement may call for the regular periodic payments to start earlier or later in order to provide a better match to other cash flow considerations of the borrower or lender. These agreements with "shifted" initial payment dates can be handled by conventional financial programs by computing an effective present value (PV) that compensates for the difference in interest accrued during the irregular first payment period. This computation becomes more complex when the compounding and payment frequencies (CF and PF) are unequal.

Shifting the initial payment date forces a change in the number or amount of the periodic payments or in the amount of the final or balloon payment. However, the participants to an agreement may want to specify the number and/or amount of the regular payments, and adjust the final payment to complete the amortization. Even without a shifted initial payment date or other restrictions the regular periodic payments seldom precisely complete the amortization and the final payment must be adjusted to accomplish this.

For the uninitiated or infrequent user of financial programs, the accommodation of a shifted first payment date and/or the computation of the correct final payment amount can cause problems. The following program easily handles these cases and also takes the drudgery out of computing an amortization schedule.

The LPAS program uses the FI program and the CJ and CP routines in the PPC ROM to expand the capabilities of the FI program to accommodate "shifted" initial payment dates and to compute the number and amount of periodic payments, and the final payment required to amortize a loan or to accumulate a specific future value. The information needed to prepare a loan amortization schedule may also be computed on an optional basis. The extensive capabilities of the FI program are used in their normal manner to define the parameters of a specific problem and to develop the initial solution. Two additional input parameters are provided; the effective date (ED) and the initial payment date (IP). These two dates define the length of the first payment period which need not be equal to the normal payment period implied by the payment frequency value (PF). The initial payment date (IP) also establishes the number of payments that will occur in the first year. The program computes the regular periodic payment and the final payment required to amortize a loan or to accumulate a specified future value over a specified term (n), or the number of payments and the final payment necessary to amortize a loan or to accumulate a specified future value with a specified periodic payment amount.

Conventional loans, mortgages with or without balloon payments, and Canadian or European mortgages are all acceptable to the LPAS program. Cases with payment frequencies of semi-monthly (PF=24) or less, use a 30 day month convention for determining the number of days of shift in the first payment date and the number of payments occurring in the first year. For payment

frequencies greater than semi-monthly (i.e., daily, weekly, or bi-weekly) the actual number of calendar days is used.

LPAS Program - Operation

The LPAS program computes the regular periodic payment and the final payment for both present value (PV*) and future value (FV*) cases. PV* cases involve periodic payments that reduce or amortize a present value. FV* cases involve appreciation or accumulation to a future value. The amortization schedule portion of the program supports PV* cases only. The LPAS program can be used with or without a printer (CF21).

The **FI** program is accessed and used to set the status (CB, CE, DB, DE), the compounding frequency (CF), the payment frequency (PF), the standard financial values (n, %I, PV, PMT, FV) and to solve for any missing financial value. Note: The **FI** program can be accessed by pressing "J" when the LPAS program has control. After entering the normal financial program data, the effective date of the financial agreement (ED) and the date of the initial payment (IP) are entered into the X and Y registers in the form MM.DDYYYY (Y=ED, X=IP). The IP date must not be earlier than the ED date.

The LPAS program is then executed. For easy access the LPAS program should be assigned to a key. The LPAS program was assigned to the X<>Y (F) key in the keystroke solutions in the example programs below. The program computes the regular periodic payment required to maintain the specified interest rate. The computation compensates for any fractional portion of the term and for any deviation from the normal initial payment date. When the program first stops, the computed payment (rounded to two decimal places) is in the PMT register (R04) and is displayed in the X register.

First Stop - The computed payment may be accepted, or a modified payment may be entered and substituted by pressing key "D". To continue the computation, select one of the following two options:

1. By pressing "H" the amortization period is limited to the integer portion of the term (n) and the final or balloon payment is adjusted to complete the amortization.

2. By pressing "J" the term (n) is recomputed to accomplish the amortization with the specified periodic payment with a minimum adjustment to the final or balloon payment. The amortization choice restarts the program and the number of periodic payments and the amount of the final payment are computed. At the second stop the stack contains:

T = number of payments occurring in first year
Z = number of regular periodic payments
Y = amount of the regular periodic payment
X = amount of the combined final and balloon payments

Second Stop - An amortization schedule may be computed by pressing "E" (for PV* cases only) or control may be returned to the **FI** program by pressing "J".

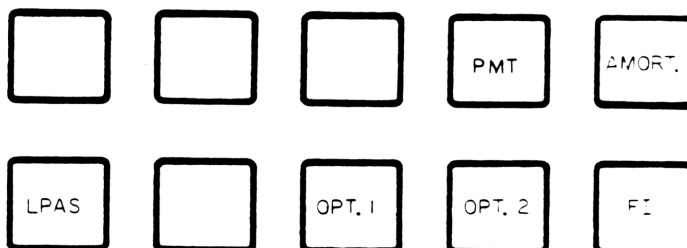
If an amortization schedule is computed and a printer is not available (CF21) the program will stop after computing the values for each year. At each stop the stack will contain:

T = cumulative interest paid
Z = balance outstanding after last PMT for year
Y = interest paid during the year
X = year (YY)

To compute the amortization data for each succeeding year press R/S. Completion of the amortization is indicated by ** in the display. The total interest paid is in the Y register at this final stop.

After completion of the amortization, control may be returned to the **FI** program by pressing "J".

Keyboard Functions: (LPAS Program)



Key	Function	(Flag/Reg)
-----	----------	------------

D	Enter revised periodic payment	"PMT" (R04)
E	Compute amortization data (PV* case only)	(R00-R13)
F	Enter ED and IP dates and compute periodic PMT	
H	Select Option 1 and compute final PMT	(F07)
I	Select Option 2 and compute term n and final PMT	
J	Transfer to FI program and display status	
R/S	Compute amortization data for next year	

Program requirements and limitations. **CJ**, **CP**, **FI** are the **PPC ROM** required routines. LPAS is 655 bytes SIZE=014 Flags 06-10,21,28,29

Acceptable Payment Frequencies (PF) are:

1 = annual	12 = monthly
2 = semi-annual	24 = semi-monthly
3 = tri-annual	26 = bi-weekly
4 = quarterly	52 = weekly
6 = bi-monthly	365 = daily

WARNING: Solutions using or resulting in a zero rate of interest (%I) will cause a "DATA ERROR".

The output of LPAS is printed in three sections separated by horizontal lines. The first section records the original parameters of the case. The second section records the amount and number of regular payments and the final payment necessary to satisfy the options selected. The third section displays the optional amortization schedule.

Examples:

In the keystroke solution for each example, the lower case letters a through e represent shifted functions of keys A through E. Key in the indicated quantities

and press the user defined keys as indicated in the "Do" column. Contents of the display or the printed output at significant points in the solution are shown in the "See" column and are followed by identification in the "Result" column. Use FIX 2 display mode, and assign LPAS to key F (X<>Y).

Example A: Conventional Mortgage. Develop the data for an amortization schedule for a fully amortized 30-year, \$100,000 mortgage at 14.75% NAR compounded monthly with end of period payments of \$1,244.48 with the first payment due on November 1, 1981. Effective date of the loan is September 25, 1981. Use option 1 (H) to limit the amortization period to 360 payments.

Do:	See:	Result:
CLX STO 06	0.00	Store F1 function call
XEQ " F1 "	"DE"	Discrete/End status
12 H I	12.00	CF=PF=12
30 a	360.00	n=360
14.75 B	14.75	NAR=14.75% F1
100000 CHS C	-100,000.00	PV=\$100,000.00
1244.48 D	1,244.48	PMT=\$1,244.48
E	-27.98	FV=\$27.98
9.251981		
ENTER↑		
11.011981 F	1,247.52	PMT, shifted IP \$1,247.52
H	1,248.31	Final PMT=\$1,248.31
E	**	Compute amortization data, see print out below.

```

EXAMPLE A CF=12 PF=12
*****
PV*DE PV -100,000.00
360 PMTS 1,244.48
14.750% FV -27.98
ED 9-25-81 IP 11- 1-81*
*****
359 PMTS 1,247.52
+FINAL PMT 1,248.31
*****
YR INTEREST ENDING BAL
81 2,464. 100,214.
82 14,768. 100,012.
83 14,736. 99,778.
84 14,699. 99,507.
85 14,657. 99,193.
86 14,607. 98,830.
87 14,550. 98,410.
88 14,483. 97,923.
89 14,407. 97,359.
90 14,318. 96,707.
  
```

```

91 14,214. 95,951.
92 14,095. 95,076.
93 13,957. 94,063.
94 13,797. 92,889.
95 13,612. 91,531.
96 13,397. 89,958.
97 13,149. 88,137.
98 12,861. 86,028.
99 12,528. 83,586.
00 2,143. 80,758.
01 11,696. 77,485.
02 11,179. 73,694.
03 10,581. 69,305.
04 9,888. 64,222.
05 9,085. 58,338.
06 8,156. 51,524.
07 7,080. 43,634.
08 5,835. 34,498.
09 4,392. 23,920.
10 2,722. 11,672.
11 804. 0.
** 348,860.
  
```

Note: 2 payments in 1981. The negative amortization during the first two years is due to the delayed first payment date. The asterisk following the IP date indicates a shifted initial payment date.

If a printer is not used when working Example A, after execution the stack will contain the following:

after F	after H	after E*
T= -	T= 2.00	T = 2,464. Σ Int.
Z= -	Z= 359.00	Z = 100,214. E.Bal.
Y= -	Y= 1,247.52	Y= 2,464. Yr.Int.
X= 1,247.52	X= 1,248.31	X= 81. Year

*Press R/S to advance amortization to next year. the end of the amortization is indicated by ** in display.

Example B: Sinking Fund / Savings Plan Starting with an initial deposit of \$3,000 compute the number of bi-weekly deposits of \$200 and the amount of the final deposit needed to accumulate a balance of \$20,000 in an account paying 8% compounded continuously, if the initial deposit (PV) is made on December 1, 1981 and the first bi-weekly deposit (PMT) is made on December 11, 1981. Set the status to CB.

Do:	See:	Result:
J	"DE"	Return to F1
c	"CE"	Set Continuous compounding
d	"CB"	Set Beginning of period payments
e	"CB"	Clear Financial
		Status=Continuous/Beginning
		CF=1 after clearing
26 I	26.00	PF=26
8 B	8.00	NAR=8% F1
3000 CHS C	-3,000.00	PV=\$3,000.00
200 CHS D	-200.00	PMT=\$200.00
20000 E	20,000.00	FV=\$20,000.00
A	72.43	n=72.43
12.011981		
ENTER↑		
12.111981 F	-197.29	PMT, shifted IP \$197.29
200 CHS D	-200.00	Enter revised PMT \$200.00
I	-91.67	Final PMT=\$91.67
E	"FV* ?"	Indicates attempted amortization of FV* case

```

EXAMPLE B CF=1 PF=26
*****
FV*CB PV -3,000.00
72+ PMTS -200.00
8.000% FV 20,000.00
ED 12- 1-81 IP 12-11-81*
*****
71 PMTS -200.00
+FINAL PMT -91.67
  
```

FV*CB = Future Value case with Continuous compounding and Beginning of period payments/deposits. The plus (+) sign following the number of payments indicates that the term includes a fractional payment period as developed from the original specifications.

Example C: Loan with Balloon Payment. Develop the amortization data for a \$500,000 loan at 15% NAR with monthly compounding, to be repaid with 30 monthly end of period payments of \$20,000 and a balloon payment of \$3,225.30 coincident with the final payment. The loan effective date is September 14, 1981 and the first payment is scheduled for October 14, 1981.

Do:	See:	Result:
J	"CB"	Return to F1
		Status from previous example
c	"DB"	Set Discrete compounding
d	"DE"	Set End of period payments
e	"DE"	Clear Financial, final status=
		Discrete/End
12 H I	12.00	CF=PF=12
30 A	30.00	n=30
15 B	15.00	NAR=15% F1
500000 CHS C	-500,000.00	PV=\$500,000.00
20000 D	20,000.00	PMT=\$20,000.00
E	3,225.30	Balloon=\$3,225.30
9.141981		
ENTER↑		

10.141981 F 20,000.00 PMT=\$20,000.00
H 23,225.30 Final + Balloon = \$23,225.30
E ** Compute amortization data, see
print out below.

```

EXAMPLE C CF=12 PF=12
*****
PV*DE PV -500,000.00
30 PMTS 20,000.00
15.000% FV 3,225.30
ED 9-14-81 IP 10-14-81
*****
29 PMTS 20,000.00
+FINAL PMT 23,225.30
*****
YR INTEREST ENDING BAL
81 18,232. 458,232.
82 56,456. 274,688.
83 26,950. 61,638.
84 1,587. 0.
** 103,225.

```

Because the initial payment occurs exactly one month after the loan effective date there is no change in the re-computed PMT.

Example D: Delayed First Payment This example will illustrate the effect of a different repayment plan for the loan defined in Example C. Develop the data for amortizing a \$500,000 loan at 15% NAR with monthly compounding, to be repaid with 60 semi-monthly end of period payments of \$10,000 and a balloon payment coincident with the final payment. The loan effective date is September 14, 1981 and the first payment is scheduled for November 1, 1981.

Do:	See:	Result:
J	"DE"	Return to 1 Status left from Example C
12 H	12.00	CF=12
24 I	24.00	PF=24
60 A	60.00	n=60
15 B	15.00	NAR=15% \Rightarrow 1
500000 CHS C	-500,000.00	PV=\$500,000.00
10000 D	10,000.00	PMT=\$10,000
E	974.25	FV=\$974.25
9.141981 ENTER↑		
11.011981 F	10,268.92	PMT=\$10,268.92
10000 D	10,000.00	Set PMT=\$10,000.00 exactly
H	30,466.27	Final+Balloon=\$30,466.27
E	**	Compute amortization data See print out below

```

EXAMPLE D CF=12 PF=24
*****
PV*DE PV -500,000.00
60 PMTS 10,000.00
15.000% FV 974.25
ED 9-14-81 IP 11-1-81*
*****
59 PMTS 10,000.00
+FINAL PMT 30,466.27
*****
YR INTEREST ENDING BAL
81 12,541. 485,968.
82 60,113. 306,081.
83 31,195. 97,277.
84 3,189. 0.
** 107,038.

```

The total interest on this repayment plan is \$3,813 more than in Example C due to the delayed first payment date and the smaller payments. The borrower has the use of more money for a longer time.

LPAS Program - Equations

All equations assume the use of standard financial transaction sign conventions of money received as positive (+) and money paid out as negative (-).

Notation used:

d = number of days in payment period
 i_e = effective interest rate per payment period
n = integer portion of term n
n = number of payment periods in term
s = number of days first payment is shifted
CF = compounding frequency per year
ED# = effective date - day number
FV = future value after n periods
 FV_m = future value after m periods
 FV_{m-1} = future value after m-1 periods
FV* = future value case
INT = interest for the year
IP# = initial payment date - day number
NP = number of payments in the year
PF = payment frequency per year
PMT = periodic payment
 PMT_f = final payment
PV = present value
 PV_e = effective present value
PV* = present value case

If $|FV| \leq |PV|$, then PV* case
If $|FV| > |PV|$, then FV* case

The initial payment date is "shifted" when: $s \neq 0$

where: $s = IP\# - ED\#$ for beginning of period payments

$s = IP\# - ED\# + d$ for end of period payments

For financial calculations involving a "shifted" first payment date, the present value (PV) must be converted to an effective present value (PV_e) that

is adjusted to compensate for the difference in interest accrued during the irregular first payment period.

$$PV_e = PV(1+i_e)^{(sPF/dCF)}$$

To precisely complete the amortization of a present value or the accrual of a future value, the final payment must be calculated separately from the regular periodic payment. The LPAS program incorporates eight variations of final payment calculations.

$PMT_f = FV_{m-1}$ PV* case, annuity due, Option 1
 $= FV_m$ PV* case, annuity due, Option 2
 $= FV_m + PMT$ PV* case, ordinary annuity, Option 1

APPLICATION PROGRAM FOR:		FI
01*LBL *LPAS*	74 +	
02 STO 10	75 RCL 09	
03 X<>Y	76 /	
04 STO 00	77 INT	
05 0	78 STO 13	
06 XEQ 08	79 FC? 09	
07 CF 06	80 ST+ 10	
08 RCL 03	81 +	
09 ABS	82 LASTX	
10 RCL 05	83 /	
11 ABS	84 INT	
12 X>Y?	85 STO 12	
13 SF 06	86 RCL 10	
14 1	87 RCL 04	
15 XEQ 07	88 X=0?	
16 ASTO X	89 X<>Y	
17 *P*	90 CHS	
18 FS? 06	91 X=0?	
19 *F*	92 *-*	
20 *-V+*	93 FS? 21	
21 ARCL X	94 PRA	
22 *- PV*	95 CLA	
23 RCL 03	96 RCL 08	
24 XEQ 05	97 RCL 13	
25 ADV	98 *	
26 XEQ 12	99 /	
27 RCL 01	100 RCL 09	
28 ENTER†	101 *	
29 INT	102 RCL 07	
30 ARCL X	103 LN1+X	
31 -	104 *	
32 X=0?	105 E+X	
33 *-+*	106 ST* 03	
34 *- PMTS*	107 RCL 06	
35 RCL 04	108 STO 11	
36 XEQ 05	109 4	
37 ADV	110 XEQ 07	
38 FIX 3	111 RND	
39 ARCL 02	112 STO 04	
40 *-+ FV*	113 RTN	
41 RCL 05	114*LBL 01	
42 XEQ 05	115 INT	
43 ADV	116 -100	
44 XEQ 12	117 STO 11	
45 CF 07	118 STO Z	
46 24	119 X<>Y	
47 RCL 09	120 STO 12	
48 X>Y?	121 XEQ 04	
49 SF 07	122 INT	
50 *ED *	123 STO 13	
51 RCL 00	124 XEQ 04	
52 XEQ 01	125 CHS	
53 X<> 10	126 ST* 11	
54 *- IP *	127 FRC	
55 XEQ 01	128 *	
56 ST- 10	129 STO 06	
57 STO 00	130 10	
58 FIX 2	131 X>Y?	
59 SF 28	132 *-0*	
60 SF 29	133 ARCL Y	
61 1	134*LBL 02	
62 ST+ 11	135 FS? 07	
63 STO 12	136 GTO 03	
64 CLX	137 RCL 11	
65 STO 13	138 360	
66 XEQ 02	139 *	
67 RCL 00	140 RCL 12	
68 -	141 30	
69 360	142 *	
70 ENTER†	143 +	
71 6	144 RCL 13	
72 FC?C 07	145 +	
73 CLX	146 RTN	

APPLICATION PROGRAM FOR:		FI
147*LBL 03	220 FC? 09	
148 RCL 11	221 CLX	
149 RCL 12	222 -	
150 RCL 13	223 STO 01	
151 XROM *-CJ*	224 RCL 07	
152 RTN	225 1	
153*LBL 04	226 +	
154 10	227 STO 13	
155 X>Y?	228 RCL 05	
156 *-+*	229 STO 00	
157 ARCL Y	230 5	
158 RDN	231 XEQ 07	
159 LASTX	232 RCL 00	
160 -	233 STO 05	
161 *	234 FC? 06	
162 *-+*	235 CLX	
163 RTN	236 STO 00	
164*LBL 05	237 X<>Y	
165 FIX 2	238 RCL 13	
166 9	239 FS? 09	
167*LBL 06	240 ST/ 00	
168 STO 06	241 X<>Y	
169 X<>Y	242 FC? 09	
170 SF 28	243 GTO 10	
171 SF 29	244 RCL 00	
172 FC? 21	245 -	
173 RTN	246 GTO 11	
174 ACA	247*LBL 10	
175 XROM *-CP*	248 FC? 07	
176 CLA	249 *	
177 RTN	250 RCL 00	
178*LBL 07	251 FC? 06	
179 STO 06	252 CLX	
180 XROM *-FI*	253 -	
181 RTN	254 RCL 04	
182*LBL 08	255 FC? 07	
183 FC? 21	256 CLX	
184 RTN	257 +	
185 24	258*LBL 11	
186 X<>Y	259 RND	
187*LBL 09	260 STO 13	
188 ACCHR	261 1	
189 DSE Y	262 XEQ 00	
190 GTO 09	263 XEQ 12	
191 PRBUF	264 ARCL 10	
192 RTN	265 *- PMTS*	
193*LBL D	266 RCL 04	
194 STO 04	267 XEQ 05	
195 RTN	268 ADV	
196*LBL J	269 *-FINAL PMT*	
197 10	270 RCL 13	
198 STO 06	271 XEQ 05	
199 GTO *-FI*	272 ADV	
200*LBL H	273 RCL 12	
201 RCL 04	274 RCL 10	
202 X=0?	275 RCL 04	
203 SF 07	276 RCL 13	
204*LBL I	277 FIX 2	
205 RCL 01	278 RTN	
206 INT	279*LBL 12	
207 STO 10	280 FIX 0	
208 1	281 CF 28	
209 XEQ 07	282 CF 29	
210 INT	283 RTN	
211 X=0?	284*LBL E	
212 RCL 10	285 *-FV* ?*	
213 FC? 07	286 FS? 06	
214 STO 10	287 PROMPT	
215 RCL 10	288 1	
216 1	289 XEQ 00	
217 FS? 07	290 *-YR INTEREST *	
218 ST- 10	291 *-ENDING BAL*	
219 FS? 07	292 FS? 21	

Listing continued on page 158.

APPLICATION PROGRAM FOR: FI	
293 PRA	340 XEQ 06
294 CF 07	341 RCL 05
295 CLA	342 RND
296 CLX	343 8
297 X<> 12	344 XEQ 06
298 RCL 10	345 ADV
299 X<=Y?	346 RCL 12
300 SF 07	347 RCL 05
301 X<>Y	348 RCL 00
302 STO 01	349 RCL 11
303 -	350 FC? 21
304 STO 10	351 STOP
305*LBL 13	352 FS?C 07
306 XEQ 12	353 GTO 16
307 RCL 11	354 1 E2
308 10	355 RCL 11
309 X>Y?	356 1
310 "+0"	357 +
311 ARCL Y	358 X=Y?
312 RCL 03	359 -
313 RCL 04	360 STO 11
314 RCL 01	361 RCL 10
315 *	362 STO 01
316 +	363 RCL 09
317 STO 00	364 ST- 10
318 CLX	365 X<=Y?
319 STO 05	366 STO 01
320 FC? 07	367 -
321 GTO 14	368 X<=0?
322 RCL 13	369 SF 07
323 ST+ 00	370 GTO 13
324 GTO 15	371*LBL 16
325*LBL 14	372 "***"
326 5	373 RCL 12
327 XEQ 07	374 8
328 FIX 2	375 XEQ 06
329 RND	376 11
330 ST+ 00	377 FS? 21
331 STO 05	378 SKPCHR
332 CHS	379 ADV
333 STO 03	380 "***"
334*LBL 15	381 ASTO X
335 FIX 0	382 FIX 2
336 RCL 00	383 .END.
337 RND	
338 ST+ 12	
339 8	

APPLICATION PROGRAM 2 FOR **FI**

FAST - Reducing Interest Solution Time

When the solution for interest is required for PMT≠0, LBL 02 of **FI** produces an initial guess for the interest which is supplied to the iterative loop starting at LBL 06. In most cases the LBL 02 guess is usually "close" (in the mathematical sense) to the actual solution insuring that the interest solution is found in a reasonably short time.

Unfortunately, there will always exist a problem which will cause the LBL 02 guess to be far enough away from the actual solution to cause the execution time to be long. The optional routine presented below will provide an initial guess which tends to be "closer" to the actual solution than that provided by LBL 02, allowing a shorter execution time for most problems.

In use, the optional routine is executed in RAM memory and produces an initial guess for the interest. The guess is stored in register R07, and control of the calculator is transferred from the FAST routine to LBL 06 of the ROM program **FI**.

For the condition when PMT=0, the routine transfers to LBL 09 of the ROM program for an explicit solution. When solving for n, PV, PMT, or FV, the ROM is used in the usual manner. Don Dewey (5148) produced the mathematical expressions and wrote the program.

LBL FAST INSTRUCTIONS

1. Load the routine below into the calculator memory.
2. Go to LBL **FI** in the ROM.
3. Select desired status and enter known variables in the usual manner.
4. Either a) or b):
 - a) solve for n, I, PV, PMT, or FV in the usual manner.
 - b) Execute FAST to solve for Interest using the optional routine. Do not use LBL B. The interest value is returned in the usual manner.
5. Repeat as needed from step 2.

APPLICATION PROGRAM FOR: FI	
01*LBL "FAST"	27 RCL 01
02 9	28 1
03 STO 06	29 -
04 RCL 04	30 X+2
05 X=0?	31 RCL 04
06 GTO "FI"	32 *
07 6	33 RCL 05
08 STO 06	34 -
09 RCL 05	35 RCL 03
10 RCL 04	36 +
11 RCL 01	37 3
12 *	38 *
13 -	39 /
14 LASTX	40 ABS
15 RCL 05	41 RCL 05
16 +	42 X=0?
17 RCL 03	43 GTO "FI"
18 +	44 RCL 04
19 RCL 01	45 *
20 RCL 03	46 X>0?
21 *	47 GTO "FI"
22 X=0?	48 RND
23 /	49 STO 07
24 ABS	50 GTO "FI"
25 STO 07	51 .END.
26 X<>Y	

EQUATIONS USED IN FAST ROUTINE

If PMT*FV < 0 then FV case.
If PMT*FV ≥ 0 then PV case.

1. PV CASE:

$$I_0 = \left| \frac{n \cdot \text{PMT} + \text{PV} + \text{FV}}{n \cdot \text{PV}} \right|$$

Problem valid only if PV*PMT < 0.

2. FV CASE:

a) For $PV \neq 0$:

$$I_0 = \left| \frac{FV - n \cdot PMT}{3 \cdot [(n-1)^2 \cdot PMT + PV - FV]} \right|$$

b) For $PV = 0$:

$$I_0 = \left| \frac{FV + n \cdot PMT}{3 \cdot [(n-1)^2 \cdot PMT + PV - FV]} \right|$$

FORMULAS USED IN FI

The basic financial equation used in this program was first reported in the Hewlett-Packard Journal of October 1977 (Ref. 3) where the description of its implementation in the HP-92 Financial Calculator was given. In this unique equation, all five financial variables (n , i , PV , PMT , FV) are accounted for, using the simple rule that money paid out is considered negative in sign, while money received is considered positive in sign.

The equation from page 23 of Ref. 3, is:

$$(1) \quad PV \cdot (1+i)^n + PMT \cdot [(1+i)^n - 1]/i + FV = 0$$

Ordinary Annuity and Annuity Due Selection

In its present form, equation (1) is suitable for the ordinary annuity condition, when payments are made at the end of each period. To enable (1) to solve the annuity due condition when payments are made at the beginning of each period, a small modification is required. When this modification is added, equation (1) becomes:

$$(2) \quad PV \cdot (1+i)^n + PMT \cdot (1+X) \cdot [(1+i)^n - 1]/i + FV = 0$$

where $X=0$ for ordinary annuity condition
 $X=1$ for annuity due condition

When flag F09 is cleared, the ordinary annuity condition is selected. When flag F09 is set, the annuity due condition is selected. Flag F09 is toggled by LBL d.

With a simple algebraic rearrangement, (2) becomes:

$$(3) \quad [PV + PMT(1+X)/i] \cdot [(1+i)^n - 1] + PV + FV = 0$$

or

$$(4) \quad (PV + C)A + PV + FV = 0$$

where

$$(5) \quad A = (1+i)^n - 1$$

$$(6) \quad B = (1+X)/i$$

$$(7) \quad C = PMT \cdot B$$

The form of equation (4) simplifies the calculation procedure for all five variables, which are readily

solved as follows:

$$(8) \quad n = \text{LN}[(C-FV)/(C+PV)]/\text{LN}(1+i)$$

n is solved using LBL 01

$$(9) \quad i = [FV/PV]^{1/n} - 1$$

For $PMT=0$, i is solved using LBL 09

For $PMT \neq 0$, i must be solved by iteration

$$(10) \quad PV = -[FV + (A \cdot C)]/(A+1)$$

PV is solved using LBL 03

$$(11) \quad PMT = -[FV + PV(A+1)]/(A \cdot B)$$

PMT is solved using LBL 04

$$(12) \quad FV = -[PV + A(PV + C)]$$

FV is solved using LBL 05

Solution of Interest When $PMT \neq 0$

To solve for interest i when $PMT \neq 0$, an iterative technique must be employed, as equation (1) cannot be explicitly solved for i . This program uses Newton's Method, using exact expressions for the function of i and its derivative. The expressions are:

$$(13) \quad I_{k+1} = I_k - f(I_k)/f'(I_k)$$

where

$$(14) \quad f(i) = A(PV+C) + PV + FV$$

$$(15) \quad f'(i) = n \cdot D \cdot (PV+C) - (A \cdot C)/i$$

where

$$(16) \quad D = (1+i)^{n-1}$$

$$(17) \quad = (A+1)/(1+i) \text{ as calculated by LBL 06}$$

The iterative interest solving loop using equations (13), (14), and (15) starts at LBL 06.

Starting Guess For Interest

To solve for interest using Newton's Method, an initial starting guess must be provided. The program uses the following expression to provide the initial guess, I_0 :

$$(18) \quad I_0 = \left| \frac{PMT}{|PV| + |FV|} \right| + \left| \frac{|PV| + |FV|}{n^3 \cdot PMT} \right|$$

The closer the initial guess I_0 is to the actual solution i , the greater is the probability that the required solution will be obtained, and the shorter is the execution time.

Further Program Refinements

As well as being able to select either an ordinary annuity or annuity due situation, the program also enables solutions to be obtained when

B. PMT≠0

- a. Calculate I_0 using equation (18) and LBL 02.
- b. Calculate A, A+1, B, and C using equations (5), (6), (7), and LBL 08.
- c. Calculate iterative solution, using equations (13), (14), and (15) and LBL 06.
- d. Exit test (1) If error is too large, back to b above
(11) If error acceptable, continue
- e. Calculate I, using equations (21) or (22) and LBL 11.
- f. Store I in R02 and halt.

3. Solving for PV

- a. Calculate I_e using equations (19) or (20) and LBL 07.
- b. Calculate A, A+1, B, and C using equations (5), (6), and (7) and LBL 08.
- c. Calculate PV using equation (10) and LBL 03
- d. Store PV in R03 and halt

4. Solving for PMT

- a. Calculate I_e using equations (19) or (20) and LBL 07.
- b. Calculate A, A+1, B, and C using equations (5), (6), and (7) and LBL 08.
- c. Calculate PMT using equation (11) and LBL 04
- d. Store PMT in R04 and halt.

5. Solving for FV

- a. Calculate I_e using equations (19) or (20) and LBL 07.
- b. Calculate A, A+1, B, and C using equations (5), (6), and (7) and LBL 08.
- c. Calculate FV using equation (12) and LBL 05
- d. Store FV in R05 and halt.

LINE BY LINE ANALYSIS OF **FI**

Lines 01-02 provide access to any **FI** subroutine that begins with a numeric label.

Lines 03-14 clear the financial registers R01-R05 and set CF=PF=1 in R08 and R09. Line 14 is a jump to the status display.

Lines 15-18 toggle flag F08 which controls Continuous/Discrete compounding. Line 018 is a jump to the status display.

Lines 19-21 toggle flag F09 which controls the Begin/End switch.

Lines 22-32 are the status display which shows one of the codes CE, CB, DE, DB.

Lines 33-36 store the number of compounding periods in R08 = CF = compounding frequency.

Lines 37-40 store the number of payment periods in R09 = PF = payment frequency.

Lines 41-43 multiply the X-register by 12 before entering the LBL A routine.

Lines 44-48 either store n in R01 and stop, or drop through to line 049.

Lines 49-63 solve for n via formula (8) and store n in R01.

Lines 64-66 divide the X-register by 12 before entering the LBL B routine.

Lines 67-71 either store %I in R02 and stop, or drop through to line 072.

Lines 72-127 are the major part of the program which solves for %I. Line 79 tests whether %I can be calculated directly if PMT=0. Otherwise lines 73-90 compute the initial guess for %I via formula (18). Lines 91-126 are the recurrence loop for formulas (13)-(17). Formula (14) is complete at line 103 and formula (15) is complete at line 117. Line 127 is a branch to complete the calculation of %I.

Lines 128-165 are a special subroutine. Line 148 provides access to a second entry point within the subroutine. Lines 129-147 calculate formula (19) or (20). Lines 148-165 calculate formulas (5), (6), and (7) which are constants used by other parts of the program.

Lines 166-175 calculate formula (9) for %I when PMT=0.

Lines 176-193 finish the calculation of %I and restore the rate by calculating formula (21) or (22).

Lines 194-198 either store PV in R03 and stop, or drop through to line 199.

Lines 199-208 calculate PV via formula (10) and store PV in R03.

Lines 209-213 either store PMT in R04 and stop, or drop through to line 214.

Lines 214-227 calculate PMT via formula (11) and store PMT in R04.

Lines 228-232 either store FV in R05 and stop, or drop through to line 233.

Lines 233-241 calculate FV via formula (12) and store FV in R05.

Line 242 is provided to allow a running program to stop so the program pointer is in ROM.

NUMERIC LABELS/FUNCTIONS IN THE **FI** PROGRAM

Although **FI** is a complete self-contained program, some users may wish to use some of **FI**'s subroutines in their own programs. The following list gives a correspondence between numeric labels and subroutines to be called as part of **FI** programs. To call a subroutine from one of your own programs, first store the number corresponding to the desired function in data register R06. Then use the instruction XEQ "**FI**" as part of your program. The execution times in seconds for the various subroutines are in parentheses in the following list.

Numeric Label Number In R06	Keyboard Label	Subroutine Function
00	e	Clear R01-R05 and store 1 In R08 and R09 (<1 sec.)
01	A	Solve for n (3.5 sec.)
02	B	Solve for %I (variable)
03	C	Solve for PV (2.5 sec.)
04	D	Solve for PMT (3.3 sec.)
05	E	Solve for FV (3.2 sec.)
12	None	Serves only to restore keyboard functions to top rows of keys

The following special comments apply to **FI** subroutines that would be called from other programs.

First note that labels 01-05 only solve for the indicated variables, whereas the keys A-E perform the double functions of either solving or storing values. If you need to store the value of a financial variable in one of R01-R05 then your program should do that directly. Subroutines for storing are neither necessary nor provided.

The purpose of label 12 at the end of the program is to allow a running program to stop so the program pointer is in ROM and the automatic local label key assignments of the **FI** functions on the top row of keys will be restored to those keys. Otherwise, a running program would normally stop and leave the program pointer in RAM which would make the top row key assignments "disappear".

Numeric labels 07,08,09 & 11 are intended to be internal subroutines within **FI** which would seem to perform no useful purpose outside of **FI**. However, the truly curious PPC member may be able to jump into the middle of **FI** by calling these routines as "hidden functions". See the line by line analysis for the purpose of these functions. Example 11 in the **FI** documentation uses label 07 in this manner.

REFERENCES FOR **FI**

1. W.L. Crowley and F. Rode, "A Pocket-Sized Answer Machine For Business and Finance," Hewlett-Packard Journal, May 1973.
2. R.B.Neff and L. Tillman, "Three New Pocket Calculators: Smaller, Less Costly, More Powerful," Hewlett-Packard Journal, November 1975.
3. Roy E. Martin, "Printing Financial Calculator Sets New Standards For Accuracy and Capability," Hewlett-Packard Journal, October 1977.
4. Greynolds, Aronofsky, Frame, "Financial Analysis Using Calculators," McGraw-Hill, 1980.

For anyone wishing to further his/her knowledge on the subject of financial analysis, and as it applies to calculators, Reference 4 is probably the most definitive book available to date. The 470-page volume assumes the reader has no previous knowledge of financial analysis, and commences at an elementary level, taking the reader through the theory and practice. The main subjects covered are:

- a. Basic Concepts in Compound Interest
- b. Simple Annuities
- c. General Annuities
- d. Continuous Compounding and/or Payments
- e. Variable Cash Flows and Internal Rate of Return
- f. Balloon Annuities Using Present Values
- g. Special Applications

CONTRIBUTORS HISTORY FOR **FI**

Graeme Dennes (1757) is responsible for programming the accuracy enhancements in **FI** and for writing the first substantial version of **FI**, after analyzing the problems associated with the various formulas involved.

The addition of the general annuity capability and an improvement to the initial guess routine were produced by Don Dewey (5148). Don also tested and further debugged the program using literally thousands of test examples. There were more changes made until the program evolved to its final form. Cliff Carrie (834) suggested changes in the use and placement of the local numeric labels. Graeme Dennes and Don Dewey are to be credited with the lion's share of the work on **FI**, both in programming and in the documentation.

FINAL REMARKS FOR **FI**

The first dedicated hand-held financial calculator was the HP-80 (Reference 1). This was later followed by the HP-22 (Reference 2) and HP-27, although they never replaced the HP-80.

As users accepted the new calculators, it wasn't long before they began to demand more in facilities, capabilities and accuracy. These needs were readily satisfied by the HP-92 printing financial calculator (Reference 3).

The HP-92 article provided the inspiration and the starting point for the **FI** program by setting challenges for accuracy and execution times. It also made available for the first time a single unique financial equation which accounts for all five basic financial variables.

Commencing with this single equation, the **FI** program was conceived as a new, original approach to providing a highly accurate and fast financial program for the HP-41C calculator. Both the facilities and mathematical approaches of **FI** have not been used together in any previous HP calculator program. Financial programs are in other HP-41C ROM's but the capability and accuracy justify including the **FI** program as part of the **PPC ROM**.

By applying a reformulation and simplification to the mathematics of the new HP financial equation, the execution times and accuracy have been improved over all previous programs. The execution times when solving for interest rate have been reduced by the use of a simplified, although accurate, Newton's Method for fast convergence, coupled with a routine which produces an initial starting guess "close" to the exact solution.

Previous financial programs used standard instruction sequences to calculate terms of the form

$$(1+i)^n - 1$$

as described elsewhere. Often, several different

terms of this type were used in one program. The calculation of this term created most of the error in those programs. The high accuracy of **FI** was achieved by calculating this term using the LN1+X and E1X-1 Instructions, and calculating it only once per solution.

Thanks to the Hewlett-Packard Company for allowing the reproduction of numerical examples from Reference 3.

FURTHER ASSISTANCE ON **FI**

NOTES

TECHNICAL DETAILS					
XROM: 10, 63	FI SIZE: 010				
<u>Stack Usage:</u> 0 T: used 1 Z: used 2 Y: used 3 X: used 4 L: used	<u>Flag Usage:</u> 04: not used 05: not used 06: not used 07: not used 08: set=continuous clear=discrete 09: set= BEGIN clear=END 10: set=view iterations clear=no display 25: not used				
<u>Alpha Register Usage:</u> 5 M: 2 char. status 6 N: not used 7 O: not used 8 P: not used					
<u>Other Status Registers:</u> 9 Q: not used 10 T: not used 11 a: not used 12 b: not used 13 c: not used 14 d: not used 15 e: not used	<u>Display Mode:</u> FIX 2 recommended <u>Angular Mode:</u> not used <u>Unused Subroutine Levels:</u> 4				
ZREG: not used <u>Data Registers:</u> R00: not used R01: n R02: %I R03: PV R04: PMT R05: FV R06: function call # R07: %I as decimal R08: CF R09: PF N.B. R02 used as temporary store when solving for interest.	<u>Global Labels Called:</u> <table border="1"> <thead> <tr> <th>Direct</th> <th>Secondary</th> </tr> </thead> <tbody> <tr> <td>none</td> <td>none</td> </tr> </tbody> </table> <u>Local Labels In This Routine:</u> A, B, C, D, E, H, I, J a, b, c, d, e 00, 01, 02, 03, 04, 05, 06, 07, 08, 09, 10, 11, 12	Direct	Secondary	none	none
Direct	Secondary				
none	none				
Execution Time: See NUMERIC LABELS section in the FI documentation					
Peripherals Required: none					
Interruptible? yes Execute Anytime? no Program File: FI Bytes In RAM: 324 Registers To Copy: 47	<u>Other Comments:</u>				

CURVE FITTING

Gto (not xeq) CV (SIZE 027)

SOFTKEYS/INPUT SUMMARY (USER mode ON)

Softkey Input/Function
a $\Sigma -$ (subtract from summation)

e Initialize

A $\Sigma +$ (add to summation)

B Solve equation type J, leaving:

<u>Register</u>	<u>Value</u>
X	b
Y	a
Z	r

C Solve for Y value given X (using the equation type and coefficients last solved)

D Solve for X value given Y (using the equation type and coefficients last solved)

E Solve for best equation type, leaving:

<u>Register</u>	<u>Value</u>
X	J (best curve type)
Y	b
Z	a
T	r

OUTPUT SUMMARY

<u>J (Eq. type)</u>	<u>Equation</u>	<u>Description</u>
1	$y = bx + a$	linear
2	$y = a e^{bx}$	exponential ($a > 0$)
3	$y = b \ln(x) + a$	logarithmic
4	$y = a x^b$	power ($a > 0$)

PROGRAM FLAGS

Flag 08 curve type

Flag 09 curve type

Flag 10 $\Sigma -$ function

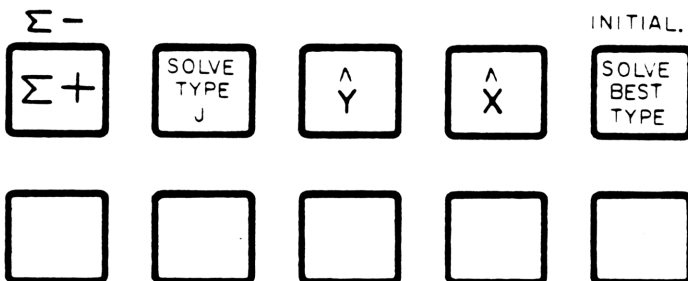
STORAGE REGISTER USE--See original PPC documentation, following

CV - CURVE FIT

This program will determine a curve of best fit to a set of data points. The four standard curve types the program handles are:

1. Linear $y = b*x + a$
2. Exponential $y = a*e^{bx}$ ($a>0$)
3. Logarithmic $y = b*Ln(x) + a$
4. Power $y = a*x^b$ ($a>0$)

The program will compute the coefficients a and b in the equation of one of the above four curve types as well as compute a value r^2 called the coefficient of determination which is a measure of the goodness of fit. Once a set of data has been fit to a given curve type, a prediction may be made for the y -value given a new x -value, or a prediction may be made for the x -value given a new y -value. The functions available on the top row of keys on the keyboard are indicated in the following diagram.



These same functions are referenced in the examples and instructions by enclosing the name of the function on the key in square brackets [].

Example 1: Find the straight line which best fits the following data:
 (1.1, 5.2), (4.5, 12.6), (8.0, 20.0),
 (10.0, 23.0), (15.6, 34.0)
 Then predict y when $x=20$ and predict x when $y=25$.

Plug the **PPC ROM** into the 41C and SIZE 027. GTO "cv" and go into USER mode. This puts the program counter in ROM and makes the curve fit functions available on the top row of keys. Pressing [INITIALIZE] will initialize the program. This clears registers R11 thru R24 so that a new set of data may be entered. In this example the 5 data points will be entered using the [Σ+] key. Key in each pair as x ENTER y and push [Σ+].

Do:	See:
[INITIALIZE]	1.0000
1.1 ENTER 5.2 [Σ+]	2.0000
4.5 ENTER 12.6 [Σ+]	3.0000
8.0 ENTER 20.0 [Σ+]	4.0000
10.0 ENTER 23.0 [Σ+]	5.0000
15.6 ENTER 34.0 [Σ+]	6.0000

All the data has now been entered and the parameters for the curve will be computed next. Since in this example we are interested in a straight line we key 1 ($j=1$) and push [SOLVE TYPE J]. When execution stops the values a , b , and r are available in the stack as:

Z: r and are also stored as R08: b
 Y: a R09: a
 X: b R10: r

For this example:

Z: r=0.999035140.
 Y: a=3.499147270
 X: b=1.972047542

The value r ranges between -1 and $+1$ and is a measure of how well the data fits the given curve type. The sign of r indicates whether the data is positively or negatively skewed. The closer r is to one of the extremes ± 1 the better the fit. For this example the line has positive slope and the fit is extremely good (all sample problems seem to work well).

Having computed the values b and a (these remain stored in R08 & R09 until new data is input) we can determine new points along the line. Key in 20 and push [↑ Y] for the predicted y -value. $y=42.94009811$ when $x=20$. Key in 25 and push [↑ X] for the predicted x -value. $x=10.90280649$ when $y=25$.

COMPLETE INSTRUCTIONS FOR **cv**

(Keyboard Operation)

- 1) Key GTO "cv", SIZE 027 and go into USER mode. The keyboard functions should now be available on the top row of keys.
- 2) Press [INITIALIZE] to initialize the program. This step clears data registers R11 thru R24 inclusive. These registers will be used to accumulate the data for all four curve types. The display will show 1.
- 3) Key in the next data pair (x,y) as x ENTER y and push [Σ+]. Repeat this step for all data pairs. The display will stop with a count of the number of the next data pair to be entered. This feature makes it possible to enter only the y -values when the x -values are consecutive integers which start counting from 1. In this case the display provides the x -values which need not be entered. If an improper data pair has just been input with the [Σ+] key, then immediately pressing R/S will delete the pair. Otherwise an improper or undesired data pair can be deleted by re-entering both x and y and pressing [Σ-].
- 4) As data pairs are entered it is possible that some x or y value is negative or zero. In these cases only one or two of the four curve types may be applied to the data. The four curve types and their respective equations are as follows:

Type J	Name	Equation
1	Linear	$y = b*x + a$
2	Exponential	$y = a*e^{bx}$ ($a>0$)
3	Logarithmic	$y = b*Ln(x) + a$
4	Power	$y = a*x^b$ ($a>0$)

If any x -values are negative or zero then only types 1 & 2 are feasible curves. If any y -values are negative or zero then only types 1 & 3 are feasible curves. If in any data pair both x and y are negative or zero then type 1 is the only feasible curve. The a coefficient must be positive for curve types 2 and 4.

5) After all data pairs have been input the next step is to select the desired curve type. This step can be accomplished in one of two ways. Under either option, the 41C should not be interrupted or else there is a possibility that the data registers will not be returned with their normal contents.

a) To fit a particular curve type, key in the number 1-4 for that type and press [SOLVE TYPE J]. The stack returns with:

Z: r and these parameters R07: j=curve type
Y: a remain stored in R08: b
X: b R09: a
R10: r

Step a) may be repeated at any time for any of the four curve types.

b) If all data input is positive then pressing [SOLVE BEST] will automatically choose the curve of best fit according to the curve type with largest absolute value of r. In this case the stack returns with:

T: r and these parameters R07: j=curve type
Z: a remain stored in R08: b
Y: b R09: a
X: j=best curve type R10: r

6) Predictions for new x or y values may be made only after step 5) has been completed. Predictions for new values are based on the settings of flags F08 and F09 which are automatically set during the fit process in step 5). The status of flags 8 and 9 for the four curve types are as follows.

	Flag 8	Flag 9
1 Linear	clear	clear
2 Exponential	set	clear
3 Logarithmic	clear	set
4 Power	set	set

In general the user need not be concerned with these flag settings, and F08 and F09 are not available for other use and must not be disturbed. To predict y given x, key in x and press [\hat{y}]. To predict x given y, key in y and press [\hat{x}]. In both cases the predicted value is left in the X-register.

7) New data may be added or deleted at any time via the [$\Sigma +$] or [$\Sigma -$] keys. However, step 5) must be performed after updating the data before any new predictions can be made using step 6). The parameters a and b are automatically destroyed after input of new data.

MORE EXAMPLES OF CV

Example 2: Determine whether the following data points are better suited for a logarithmic curve or a power curve. Then re-input the same x values and see how close the program predicts the y values.

(8, 2), (27, 3), (40, 3.2), (50, 3.5), (100, 4.1)

Do:	See:
[INITIALIZE]	1.0000
8 ENTER↑ 2 [$\Sigma +$]	2.0000
27 ENTER↑ 3 [$\Sigma +$]	3.0000
40 ENTER↑ 3.2 [$\Sigma +$]	4.0000
50 ENTER↑ 3.5 [$\Sigma +$]	5.0000
100 ENTER↑ 4.1 [$\Sigma +$]	6.0000

We will now try to fit a logarithmic curve type 3. Key 3 [SOLVE TYPE J]. The program returns with:

Z: 0.997148866 = r
Y: 0.267411352 = a
X: 0.822629796 = b

We next try to fit a power curve which is type 4. Key 4 [SOLVE TYPE J]. The program returns with:

Z: 0.995179948 = r
Y: 1.127479133 = a
X: 0.285458085 = b

Choosing the best r we would assume a type 3 logarithmic curve with the equation:

$$y = (0.822629796) * \ln(x) + 0.267411352$$

Since we just finished the power curve fit, the power curve parameters are still in the machine and hence we must go back and key 3 [SOLVE TYPE J] to return to the logarithmic parameters. Now we can predict the y's using the original x's. The predicted y-values are shown to four decimal places.

Do:	See:
8 [\hat{y}]	1.9780
27 [\hat{y}]	2.9787
40 [\hat{y}]	3.3020
50 [\hat{y}]	3.4856
100 [\hat{y}]	4.0558

Example 3: The following data fits either a linear or exponential curve. Determine which is more appropriate. (2, 12), (-1, 2), (3, 17), (5, 23) Then predict y when x = -10. After solving the above problem add the following as additional data points and resolve the same problem. (-4, 0.713), (2.5, 10.93), (6, 47.53), (10, 254.95)

Do:	See:
[INITIALIZE]	1.0000
2 ENTER↑ 12 [$\Sigma +$]	2.0000
1 CHS ENTER↑ 2 [$\Sigma +$]	3.0000
3 ENTER↑ 17 [$\Sigma +$]	4.0000
5 ENTER↑ 23 [$\Sigma +$]	5.0000

Note that since one of the data points has a negative x the only possible curves to be fit under this program are linear or exponential. For a linear fit key 1 [SOLVE TYPE J]. The program returns:

Z: 0.997577939 = r
Y: 5.520000000 = a
X: 3.546666667 = b

For an exponential fit key 2 [SOLVE TYPE J]. The program returns:

Z: 0.958629344 = r
Y: 3.826163699 = a
X: 0.419923419 = b

Choosing the best r we find a linear fit is more appropriate. Since we just finished the exponential fit, the exponential parameters are still in the machine and hence we must go back and key 1 [SOLVE TYPE J] to return the linear parameters. Now key 10 CHS [\hat{y}] to predict y = -29.94666667 when x = -10.

We next add the additional data points and resolve the problem. (Do not clear the original data). The

display should show 6 after entering the first new data pair below.

<u>Do:</u>	<u>See:</u>
4 CHS ENTER↑ 0.713 [Σ+]	6.0000
2.5 ENTER↑ 10.93 [Σ+]	7.0000
6 ENTER↑ 47.53 [Σ+]	8.0000
10 ENTER↑ 254.95 [Σ+]	9.0000

For a new linear fit key 1 [SOLVE TYPE J]. The data returned is:

Z: 0.765698771 = r
Y: 0.978958100 = a
X: 15.33154618 = b

For a new exponential fit key 2 [SOLVE TYPE J]. The data returned is:

Z: 0.993615263 = r
Y: 3.825595338 = a
X: 0.419945301 = b

Now choosing the best r we see that the new data reflects a change in the curve type. Since the exponential parameters should still be in the machine we can predict y when x = -10. Key 10 CHS [y]. y = 0.057398396.

Example 4: Fit the best curve to the following set of data points.

(1, 2), (2, 2.828), (3, 3.464), (4, 4), (5, 4.472), (6, 4.899), (7, 5.292), (8, 5.657), (9, 6).

In this example the x-coordinates start counting from 1 and are consecutive integers. So we need only input the y-coordinates, but they must be in the proper order. The count in the display will serve as the x-coordinates.

<u>Do:</u>	<u>See:</u>
[INITIALIZE]	1.0000
2 [Σ+]	2.0000
2.828 [Σ+]	3.0000
3.464 [Σ+]	4.0000
4 [Σ+]	5.0000
4.472 [Σ+]	6.0000
4.899 [Σ+]	7.0000
5.292 [Σ+]	8.0000
5.657 [Σ+]	9.0000
6 [Σ+]	10.0000

Since all the data are positive we may use the best fit function to let the program find the best fit among all 4 curve types. Press [SOLVE BEST]. The contents of the stack when the program stops are:

T: 0.999999994 = r
Z: 1.999855865 = a
Y: 0.500043886 = b
X: 4.000000000 = best curve type

This indicates a power curve (type 4) where the equation is of the form:

$$y = (2.00) * x^{0.50} \quad (\text{values rounded to 2 places})$$

APPLICATION PROGRAM 1 FOR **CV**

Curve fit solutions are often more meaningful when the points input are also plotted, superimposed on the plot of the "best fit" or selected equation type. The CVPL program will function exactly as **CV** functions

and, after calculating the parameters a, b, r and r², the program will stop with the prompt: "TO PLOT: R/S" To plot the equation calculated with the points input superimposed, simply press the R/S key. Nothing else need be done to obtain a plot. When accomplished in this way, the default situation, all numbers will be printed with 2 decimal places and the resulting plot will contain 50 plotted points. The detailed instructions include options to print other than 2 decimal places and to plot a smaller or greater number of points. The same key captions used by **CV** are used, plus the shifted keys b, c, and d for the optional features indicated.

Note that this program can also be used without the printer and will function essentially the same as **CV** but with the display labeling the points entered, showing deletions identified as such, and labeling the parameters calculated.

The plotting program takes into account all possibilities: duplicate, identical points; almost identical points that would plot as identical; points with identical x-values but with significantly different y-values; individual single points. Any quantity of duplicate points can be handled. The points are plotted with 4 plotting characters as follows:

a. The equation of type J is plotted using a small square dot (box). One equation point is normally plotted before the first input point and after the last point. If the first input point is close to zero, it will be plotted first.

b. Individual single points are plotted with a large X.

c. Two (or more) essentially identical points are plotted with a double X, two small x's, one above the other.

d. Two (or more) points having essentially the same x-value but having different y-values are plotted with an asterisk located where the largest of the point's (based on x-value) plot should be. If desired the other points not shown for this value of x could be drawn in by hand or more points could be selected for the plot to separate very close x-values.

To simplify the program and reduce the number of registers required to store the data points, both the x- and y-value of a point are stored in one register, using a decimal point to separate them. This limits the magnitude and sign of the numbers to the following: data points must be nonzero, positive numbers and less than 1000 in magnitude. If you need to deal with larger numbers, shift all decimal points before entering them. Note: If the program is used without the printer, or by pressing "NO PLOT" with a printer, none of these restrictions apply and the "data error" message will not be encountered if you try to use negative or large numbers. See the valid use of negative entries in the **CV** instructions, however.

This program was developed originally as a modification to Gary Tenzer's curve fit program, "CFIT" in the PPC JOURNAL, V7N5P46, and was to be published in the JOURNAL as a stand alone program. The program was 691 steps in length (1414) bytes. With the **CV** routine (plus others such as the **S2** sorting routine) the plotting routine was completely re-written to utilize as many of the ROM routines as

possible and the end result is presented below, significantly improved over my earlier version, with 439 steps and using 865 bytes (8 tracks on 4 cards). Most of Gary's displays and labeling are used in this program which partially account for the length of the program. I feel these extras are desirable, specially when using a printer.

Example 1 for CVPL: Use the same problem as Example 1 for **CV**. Find the linear equation for the following data: (1.1, 5.2), (4.5, 12.6), (8.0, 20.0), (10.0, 23.0), (15.6, 34.0).

Then predict y when x=20 and predict x when y=25.

Plug the **PPC ROM** in and using the card reader, read in all 8 sides of the program CVPL. Put the calculator in USER mode. Connect printer and put in MAN mode. Press Initialize (shift E) and the display will tell you to SIZE 038 plus the number of points you plan to input. For this example SIZE 043 (=38 + 5 points). Press R/S to complete initialization of the program. See 1.00 in the display asking for the first point's values. First however, we will select 4 decimal places in the printout so key in 4 and press shift C (for the number of decimal places). See 1.00 again asking for the first point's values. Key in each point exactly as in the **CV** instructions by keying in X ENTER Y

[Σ+]. Keyboard functions assigned to keys are shown in square braces [] below.

Do:	See:
[Initialize]	"SIZE=38+ PTS"
XEQ "SIZE" 043	0.00 (size=38+5)
R/S	1.00 TONE 9
4 [No.Dec.Places]	1.00 TONE 9
1.1 ENTER↑ 5.2 [Σ+]	X1=1.0000, Y1=5.2000
.5 ENTER↑ 12.6 [Σ+]	2.0000 TONE 9
8.0 ENTER↑ 20.0 [Σ+]	X2=4.5000, Y2=12.6000
10.0 ENTER↑ 23.0 [Σ+]	3.0000 TONE 9
15.6 ENTER↑ 34.0 [Σ+]	X3=8.0000, Y3=20.0000
	4.0000 TONE 9
	X4=10.0000, Y4=23.0000
	5.0000 TONE 9
	X5=15.6000, Y5=34.0000
	6.0000 TONE 9

Since we want a linear curve, we key in 1 and push [SOLVE TYPE J]. When execution stops the following will be printed.

```

1: LIN
a=3.4991
b=1.9720
r=0.9990
r↑2=0.9981

```

In the calculator display see "TO PLOT: R/S" If we now press R/S the plot will consist of 50 points. To select plot of 25 points, key in 25 and press [No.Pts.in Plot], the shifted B key. The same display will appear in the calculator (nothing is printed). Before plotting, we will first find the predicted y and x values asked for. Key in 20 and push [ŷ], the C key. Printed (and displayed) see:

"IF X = 20.0000, Y = 42.9401"

, in 25 and push [x̂], the D key and see:

"IF Y = 25.0000, X = 10.9028"

Now press R/S to plot the data. When the plotting is complete, wait for the BEEP before stopping the calculator.

The total time for this example, except for sizing the calculator was 4 min. and 25 sec. The primary consumer of time is normally the plotting, so the number of points selected greatly effects execution time. Often a short plot of 15 points is adequate.

After the BEEP has sounded the completion of the plot you can find other predicted values of x or y, select a different curve type, add points or delete points and see the effect on the new plot.

Σ -	No. POINTS IN PLOT	No. DECIMAL PLACES	NO PLOT	INITIAL.
Σ +	SOLVE TYPE J	^ Y	^ X	SOLVE BEST TYPE

INSTRUCTIONS	INPUT DATA	KEYS	OUTPUT
1. Load cards, slides 1-8 in USER mode			0.00
2. Initialize		shift E	"SIZE=38+PTS"
3. If SIZE inadequate Otherwise go to step 4		XEQ "SIZE"	
4. Complete initialization		R/S	TONE 9 1.00
5. Optional - To print without plotting (including negative or larger numbers)		shift D	1.00
(Note: for new problem with plotting, must CF 24)			
6. Optional - Select no. of decimal places to be printed. Default is 2. Or key in n	n	shift C	TONE 9 1.00
7. Enter data point	X ENTER↑ Y		
		[Σ+]	

Note: where x-values
are same as displayed
of next point, input
only Y and press A

```

XJ=---.----
YJ=---.----
TONE 9
# next point

```

8. If point input is correct go
to step 9. If incorrect,
press R/S to delete the point
just entered.

To delete any previously
entered point, re-enter
exact X & Y values and
press

```

R/S  ***DELETE***
X=---.----
Y=---.----
TONE 9
# next point

```

(C42)

[Σ-] (same)

9. For each new point wait for TONE 9 and repeat step 7. (same as 7)

Note: Program will accept only positive values of X and Y in the range .01-999.99. For numbers outside of range shift decimal before entering. For a zero value use .01. "DATA ERROR" message will be displayed after an invalid entry. This note only applies with printer connected. Any values for X and Y will be accepted without a printer or after pressing "NO PLOT" with a printer. See **CV** instructions regarding acceptable negative numbers.

10. Calculate a,b,r,r²:

a. For "best fit" based on largest ABS value of r: E (typical)

```
"1:LIN"
"a=--.----"
"b=--.----"
"r=--.----"
"r2=--.----"
"TO PLOT: R/S"
```

(Note: r & r² display correctly only on printer. Final caption not shown if printer not connected)

- b. For selected type "j" curve Case:

1: Linear	1 B	
2: Exponential	2 B	(same)
3: Logarithmic	3 B	
4: Power	4 B	

NOTE: Step 10 must be accomplished after all data points have been entered before steps 11, 12, or 13 may be attempted.

11. Optional: select number of points to be plotted (points input plus equation points).

- a. Default value = 50 points no action required.
b. Enter # of desired points

n shift B "TO PLOT: R/S"

12. Project y given x \hat{x} C "IF X = --.----"
"Y = --.----"
"TO PLOT: R/S"

13. Project x given y \hat{y} D "IF Y = --.----"
"X = --.----"
"TO PLOT: R/S"

14. To add additional points to same data, go to step 7.

15. Plot curve and data points R/S Curve and points plotted

The following symbols are used:

- points on curve type "j"
- × data points, no duplicate X or Y value
- × 2 or more data points with the same X and Y values within the plotting tolerance.
- * 2 or more data points with same X-value but different Y-values. Only one of the points is plotted.

BEEP sounds after plot is complete

Note: after plotting wait for BEEP. Then you can add more points, delete points, predict new X or Y values, plot with a different number of points, calculate curve parameters with a different number of decimal places displayed or select a different curve type by going back to the above instructions.

Example 2 for CVPL: This example will demonstrate all four plotting characters described above and show how deletions and points can be added. The initial points are the following:

(70.00, 11.10), (10.40, 71.86), (22.30, 38.71),
(10.50, 73.12), (40.90, 21.73), (4.20, 85.20)
(100.30, 1.34), (41.30, 34.70)

Print with 4 decimal places and solve for the best fit curve. Then find the predicted value of y for X=35 and the predicted value of X for Y=100. Then plot using 30 points in the plot. Size for one additional point to be added. In the following the data in parentheses are not printed.

Do:

See:

[Initialize]	("SIZE=38+PTS")
XEQ "SIZE" 047	(0.00)
R/S to complete initialization	(1.00 TONE 9)
4 [# dec. places]	(1.00 TONE 9)
70 ENTER↑ 11.1 [Σ+]	"X1=70.0000"
	"Y1=11.1000"
	(2.0000 TONE 9)
10.4 ENTER↑ 71.86 [Σ+]	"X2=10.4000"
	"Y2=71.8600"
	(3.0000 TONE 9)
22.3 ENTER↑ 38.71 [Σ+]	"X3=22.3000"
	"Y3=38.7100"
	(4.0000 TONE 9)
10.5 ENTER↑ 73.12 [Σ+]	"X4=10.5000"
	"Y4=73.1200"
	(5.0000 TONE 9)
40.9 ENTER↑ 21.63 [Σ+]	"X5=40.9000"
	"Y5=21.6300"
	(6.0000 TONE 9)

Y5 was entered in ERROR so to delete:

R/S	"DELETE"
	"X=40.9000"
	"Y=21.6300"
	(5.0000 TONE 9)

Now continue entering the correct values

40.9 ENTER↑ 21.73 [Σ+]	"X5=40.9000"
	"Y5=21.7300"
	(6.0000 TONE 9)
4.2 ENTER↑ 85.2 [Σ+]	"X6=4.2000"
	"Y6=85.2000"
	(7.0000 TONE 9)
100.3 ENTER↑ 1.34 [Σ+]	"X7=100.3000"
	"Y7=1.3400"
	(8.0000 TONE 9)
41.3 ENTER↑ 34.7 [Σ+]	"X8=41.3000"
	"Y8=34.7000"
	(9.0000 TONE 9)

Now push E for [SOLVE BEST]

"3: LOG"
"a=132.4456"
"b=-28.2822"
"r=-0.9812"
"r ² =0.9627"
("TO PLOT: R/S")

Find the predicted values:

```
35 [ ^ y ]          "IF X=35.0000"
                      "Y=31.8925"
100 [ ^ x ]          ("TO PLOT: R/S")
                      "IF Y=100.0000"
                      "X=3.1494"
                      ("TO PLOT: R/S")
```

Now select a 30 point plot:

```
30 [# points in plot] ("TO PLOT: R/S")
R/S to plot the data
```

After the BEEP sounds and the plotting is complete, add an additional point (71.1, 11.0), almost the same as point 1, and delete what appears to be the worst fitting point (22.30, 38.71).

```
71.1 ENTER↑ 11.0 [Σ+] "X9=71.1000"
                      "Y9=11.0000"
                      (10.0000 TONE 9)

22.30 ENTER↑ 38.71 [Σ-]
                      ** DELETE **
                      "X=22.3000"
                      "Y=38.7100"
                      (10.0000 TONE 9)
```

Now again solve for the best fit.

```
[SOLVE BEST]          "3: LOG"
                      "a=133.8645"
                      "b=-28.5171"
                      "r=-0.9858"
                      "r↑2=0.9719"
                      ("TO PLOT: R/S")
```

We have slightly improved the fit to a log curve and the parameters a and b have of course changed. Now make a new plot by pressing R/S. After replotting the data, again find the predicted values of y if x=35 and x if y=100.

```
35 [ ^ y ]          "IF X=35.0000"
                      "Y=32.4761"
100 [ ^ x ]          "IF Y=100.0000"
                      "X=3.2789"
```

Looking at the plot, note the value of having the first input point be preceded by a point on the LOG curve. Note the double x at x=11 representing 2 almost identical points X2 and X4. The asterisk at x=41 means 2 or more points have essentially the same x-value but very different y-values. They are X5 and X8 and because X8 has a larger x-value than X5, the asterisk is plotted for Y8.

LINE BY LINE ANALYSIS OF CVPL

Lines 02-11 set up default conditions for 50 point plot and 2 decimal place printout. Lines 14-21 display next point to be input. Lines 22-30 are the delete routine using R/S. Lines 31-70 are the delete routine for later deletion of a point which first combines x and y in a single number as YYYY.XXXXX after rounding to 2 decimal places, then searches stored points registers for the same point. When the point is found a copy of the last point stored is made in that register. Flag F05 prevents display of point number for a delete. Input of new points are added to CV statistical registers (71-118), then x and y values are checked for sign and magnitude and rounded

to 2 decimal places and stored in YYYY.XXXXX format. Lines 86-186 recall full numbers (not rounded) from CV for printing to number of decimal places selected and printout is formatted for input points, deleted x and y, and calculated parameters a, b, r, and r². Lines 187-192 display plotting prompt "TO PLOT: R/S" only if printer connected, so program can be used without printer. Lines 193-200 store the barcoded input plotting symbols. Lines 201-217 exchange registers R07-11 with R33-37 using BE so data needed for CV statistical registers will be saved for later use, not lost when "PRPLOT" in printer ROM uses registers R07-11. Lines 218-236 use BX to find maximum and minimum y values of input points, then increase maximum and decrease minimum y by 25% of range to allow for equation points to be plotted outside of range of input points. Lines 237-241 make Ymin=0 if this value would have become negative after the 25% adjustment. These lines also determine the y-plotting increment used to see if 2 points have essentially same y-value. Lines 244-258 store "CRV" as the curve name for PRPLOT. The next function performed is a reverse of the left and right sides of the decimal point. Points are now stored as XXXX.YYYY (244) and S2 is used to sort the stored points to find maximum and minimum x and for faster plotting (246). Also calculated is the x-plotting increment using the range of x-values and number of points wanted in plot. If the x-minimum is smaller than plotting increment, lines 259-266 make the 1st point plotted the smallest x-value of the points; otherwise the x-minimum is set so one equation point will be plotted first. X-max made large enough that PRPLOT will never stop plot so one equation point can be plotted after largest x-values of input points (267-275). Stop routine initiated when one equation point beyond last point has been plotted. Lines 277-292 restore the statistical registers for CV by XEQ BE, then reverse stored points to original YYYY.XXXXX format (284). Lines 293-296 reset the counter and "BEEP", ready for changes to data, etc. Flags 02 and 00 are used to determine if plotting is complete, lines 329-330. Routine to check stored points to see if they should be plotted at this x-value (297-323), checks +50% of plotting increment from this plotting point. If flag F03 is set (324) at least one point to be plotted here, and still checking for others. Plotting symbol to be used selected (340-360) and stored in R03 for "PRPLOT" to use for plotting. Where 2 input points have essentially the same x-value, checks to see if their y-values are also essentially the same (361-378). Flag F04 is set when 2 points have the same y-values, F01 is set when they have significantly different y-values (375-377). Plotting routines for the 4 curve types are in steps 379-399. The routine to reverse the left and right sides of the stored points (from the decimal point) is LBL 16, steps 400-419. Storage routines for optional selection of number of points in plot and number of decimal places in printout are in steps 425-435. NOTE: The BLSPEC numbers for the plotting characters, if barcodes are not used, are:

```
box: 0, 0, 28, 28, 0, 0
large X: 0, 34, 20, 8, 20, 34, 0
double x: 0, 0, 73, 54, 54, 73, 0
asterisk: 0, 20, 8, 62, 8, 20, 0
```

The ROM routine III can also be used to create the equivalent of these BLSPEC characters.

01*LBL "CVPL"	74 RDN	147 2	220 *	293*LBL 11	366 ISG 30
02*LBL e	75 XROM "CV"	148 GTO 11	221 STO 00	294 FS? 02	367 RCL IND 30
03 4900	76 RCL 08	149*LBL E	222 X<Y	295 GTO 08	368 FRC
STO 29	77 XEQ 14	150 5	223 .01	296 STO [369 -
2	78 1 E3	151*LBL 11	224 *	297*LBL 06	370 1 E3
06 STO 38	79 /	152 FIX IND 30	225 STO 01	298 RCL IND 30	371 *
07 .	80 STO IND 30	153 SF 12	226 -	299 XEQ 00	372 ABS
08 "SIZE=38+ PTS"	81 RCL 09	154 STO 06	227 ABS	300 2	373 RCL 32
09 PROMPT	82 XEQ 14	155 RDN	228 .25	301 /	374 ISG 30
10 STO 06	83 1 E2	156 XROM "CV"	229 *	302 RCL [375 SF 04
11 XROM "CV"	84 *	157 "1: LIN"	230 ST+ 01	303 +	376 X<Y?
12 39.999	85 ST+ IND 30	158 ASTO 01	231 ST- 00	304 X<Y?	377 SF 01
13 STO 30	86*LBL 09	159 "2: EXP"	232 RCL 00	305 GTO 09	378 GTO 06
14*LBL 12	87 SF 12	160 ASTO 02	233 X<0?	306 FS? 03	379*LBL 02
15 RCL 18	88 FIX 0	161 "3: LOG"	234 0	307 GTO 10	380 RCL 34
16 1	89 "X"	162 ASTO 03	235 STO 00	308 GTO 08	381 *
17 +	90 FC? 05	163 "4: PMR"	236 STO 04	309*LBL 00	382 E+X
18 CLA	91 ARCL 18	164 ASTO 04	237 RCL 01	310 INT	383 RCL 35
19 ARCL X	92 FIX IND 30	165 CLA	238 -	311 1 E2	384 *
20 TONE 9	93 "t="	166 ARCL IND 07	239 -62	312 /	385 RTN
21 PROMPT	94 ARCL 08	167 AVIEW	240 /	313 RCL 10	386*LBL 03
22 DSE 30	95 AVIEW	168 PSE	241 STO 32	314 RTN	387 LN
23 SIN	96 PSE	169 "a="	242 "CRV"	315*LBL 09	388*LBL 01
24 SF 10	97 FIX 0	170 ARCL 09	243 ASTO 11	316 X<Y	389 RCL 34
25 6	98 "Y"	171 AVIEW	244 XEQ 16	317 RCL [390 *
26 STO 06	99 FC? 05	172 PSE	245 RCL 25	318 RCL 10	391 RCL 35
27 RCL 08	100 ARCL 18	173 "b="	246 XROM "S2"	319 2	392 +
28 RCL 09	101 FIX IND 30	174 ARCL 08	247 STO 30	320 /	393 RTN
29 XROM "CV"	102 "t="	175 AVIEW	248 RCL 24	321 -	394*LBL 04
30 GTO 08	103 ARCL 09	176 PSE	249 1	322 X<Y?	395 RCL 34
31*LBL a	104 AVIEW	177 "r="	250 -	323 GTO 11	396 Y+X
32 SF 10	105 PSE	178 ARCL 10	251 RCL IND X	324 FS? 03	397 RCL 35
33 6	106 ADV	179 AVIEW	252 INT	325 GTO 10	398 *
34 STO 06	107 FC?C 05	180 PSE	253 RCL 39	326*LBL 08	399 RTN
35 RDN	108 ISG 30	181 "r+2="	254 INT	327 RCL 31	400*LBL 16
36 XROM "CV"	109 GTO 12	182 RCL 10	255 -	328 STO 03	401 RCL 25
RCL 08	110*LBL 14	183 X+2	256 RCL 29	329 FS?C 02	402 STO 30
RND	111 FIX 2	184 ARCL X	257 /	330 SF 00	403*LBL 05
39 1 E3	112 999.99	185 AVIEW	258 STO 10	331 RCL [404 RCL IND 30
40 /	113 X<Y	186 ADV	259 RCL 39	332 GTO IND 33	405 STO 2
41 STO 00	114 RND	187*LBL 07	260 XEQ 00	333*LBL 11	406 FRC
42 RCL 09	115 X<0?	188 FC? 55	261 X<Y	334 FS? 03	407 1 E5
43 RND	116 X<Y?	189 RTN	262 X*Y?	335 GTO 08	408 *
44 1 E2	117 XEQ 17	190 "TO PLOT: R/S"	263 X<Y?	336 SF 03	409 STO Y
45 *	118 RTN	191 CF 12	264 -	337 ISG 30	410 RCL 2
46 ST+ 00	119*LBL C	192 PROMPT	265 ABS	338 GTO 06	411 INT
47 RCL 30	120 SF 03	193 "+++"	266 STO 08	339 SF 02	412 1 E5
48 1	121*LBL D	194 ASTO 26	267 RCL 24	340*LBL 10	413 /
49 -	122 FIX IND 30	195 "a"	268 1	341 1	414 ST+ Y
50 STO 27	123 SF 12	196 ASTO 27	269 -	342 ST- 30	415 RDN
51 39.999	124 STO 28	197 "QAB00"	270 RCL IND X	343 CF 03	416 STO IND 30
52 STO 30	125 3	198 ASTO 28	271 XEQ 00	344 RCL 26	417 ISG 30
53*LBL 13	126 FC? 03	199 "+++"	272 3	345 FS?C 01	418 GTO 05
54 RCL IND 30	127 4	200 ASTO 31	273 *	346 GTO 15	419 RTN
55 RCL 00	128 STO 06	201 7.011	274 +	347 RCL 27	420*LBL 17
56 X=Y?	129 RCL 28	202 ENTER↑	275 STO 09	348 FS?C 04	421 FC? 24
57 GTO 11	130 XROM "CV"	203 33.037	276 XROM "PRPLOT"	349 GTO 15	422 FC? 55
58 ISG 30	131 "IF X="	204 XROM "BE"	277*LBL "CRV"	350 RCL 28	423 RTN
59 GTO 13	132 FC? 03	205 RCL 30	278 FC?C 00	351*LBL 15	424 0
60*LBL 11	133 "IF Y="	206 INT	279 GTO 11	352 CF 04	425 /
61 RCL IND 27	134 ARCL 28	207 STO Y	280 7.011	353 STO 03	426*LBL b
62 STO IND 30	135 AVIEW	208 1	281 ENTER↑	354 RCL IND 30	427 1
63 RCL 27	136 PSE	209 -	282 33.037	355 FRC	428 -
64 STO 30	137 "Y="	210 1 E-3	283 XROM "BE"	356 1 E3	429 100
65*LBL 08	138 FC?C 03	211 *	284 XEQ 16	357 *	430 *
66 SF 12	139 "X="	212 +	285 RCL 24	358 ISG 30	431 STO 29
67 "** DELETE **"	140 ARCL X	213 STO 24	286 INT	359 RTN	432 GTO 07
68 AVIEW	141 AVIEW	214 FRC	287 .999	360 RTN	433*LBL c
SF 05	142 PSE	215 39	288 +	361*LBL 08	434 STO 38
GTO 09	143 ADV	216 +	289 STO 30	362 1	435 GTO 12
11*LBL A	144 GTO 07	217 STO 25	290 FIX IND 30	363 ST- 30	436*LBL d
72 1	145*LBL B	218 XROM "BX"	291 BEEP	364 RCL IND 30	437 SF 24
73 STO 06	146 ENTER↑	219 .01	292 STOP	365 FRC	438 .END.

FORMULAS USED IN **CV**

Linear (Type 1):

$$(1) \quad y = b \cdot x + a$$

$$(2) \quad Y = B \cdot X + A \quad \text{where } Y=y, X=x, A=a, B=b$$

$$(3) \quad x = (y-a)/b$$

Exponential (Type 2):

$$(4) \quad y = a \cdot e^{b \cdot x} \quad (a>0, y>0)$$

$$(5) \quad Y = B \cdot X + A \quad \text{where } Y=\ln(y), X=x, A=\ln(a), B=b$$

$$(6) \quad x = [\ln(y) - \ln(a)]/b$$

Logarithmic (Type 3):

$$(7) \quad y = b \cdot \ln(x) + a \quad (x>0)$$

$$(8) \quad Y = B \cdot X + A \quad \text{where } Y=y, X=\ln(x), A=a, B=b$$

$$(9) \quad x = e^{[(\ln(y) - \ln(a))/b]}$$

Power (Type 4):

$$(10) \quad y = a \cdot x^b \quad (a>0, x>0, y>0)$$

$$(11) \quad Y = B \cdot X + A \quad \text{where } Y=\ln(y), X=\ln(x), A=\ln(a), B=b$$

$$(12) \quad x = e^{[(\ln(y) - \ln(a))/b]}$$

The curve fit program determines the least squares fit for the equation $Y = B \cdot X + A$.

$$(13) \quad B = (\sum XY - (\sum X)(\sum Y)/n) / (\sum X^2 - (\sum X)^2/n)$$

$$(14) \quad A = (\sum Y - (B)(\sum X))/n$$

$$(15) \quad r^2 = \frac{(\sum XY - \sum X \sum Y/n)^2}{((\sum X^2 - (\sum X)^2/n)(\sum Y^2 - (\sum Y)^2/n))}$$

The standard four curve type equations (1), (4), (7), and (10) are all special cases of $Y = B \cdot X + A$ which is equations (2), (5), (8), and (11). Note the distinction between upper and lower case letters. The user inputs and outputs are always in terms of the lower case letters. For example, the data input consists of pairs of the form (x,y) and the coefficients the program determines are a and b. In all four cases $b=B$. The upper case letters are the quantities that the program uses to "conceptually" work with all four curve types simultaneously.

Routine Listing For:

CV

01*LBL "CV"	74 STO 10
02 GTO IND 06	75 RCL 14
03*LBL A	76 RCL 13
04*LBL 01	77 X↑2
05 CF 10	78 RCL 18
06*LBL 06	79 /
07 STO 09	80 -
08 X<>Y	81 STO Z
09 STO 08	82 /
10 ΣREG 13	83 STO 08
11 FC? 10	84 RCL 13
12 Σ+	85 *
13 FS? 10	86 ST- 09
14 Σ-	87 X<>Y
15 RDN	88 RCL 16
16 RCL 08	89 RCL 15
17 ENTER↑	90 X↑2
18 X>0?	91 RCL 18
19 LN	92 ST/ 09
20 ST* Z	93 /
21 RCL 09	94 -
22 X>0?	95 *
23 LN	96 SQRT
24 ST* Z	97 ST/ 10
25 X<>Y	98 XEQ IND 07
26 ΣREG 19	99 8
27 FC? 10	100 ST- 07
28 Σ+	101 RCL 10
29 FS? 10	102 RCL 09
30 Σ-	103 FS? 08
31 R↑	104 E↑X
32 FS? 10	105 STO 09
33 CHS	106 RCL 08
34 ST+ 12	107 RTN
35 R↑	108*LBL 10
36 FS? 10	109 RCL 11
37 CHS	110 X<> 17
38 ST+ 11	111 STO 11
39 X<> Z	112*LBL 13
40 SIGN	113 RCL 21
41 ST+ L	114 X<> 15
42 RCL 08	115 STO 21
43 RCL 09	116 RCL 22
44 X<> L	117 X<> 16
45 RTN	118 STO 22
46 RCL 08	119*LBL 09
47 RCL 09	120 RTN
48*LBL a	121*LBL 11
49 SF 10	122 RCL 12
50 GTO 06	123 X<> 17
51*LBL 8	124 STO 12
52*LBL 02	125*LBL 14
53 CF 08	126 RCL 19
54 CF 09	127 X<> 13
55 STO 07	128 STO 19
56 2	129 RCL 20
57 X<>Y?	130 X<> 14
58 SF 09	131 STO 20
59 /	132 RTN
60 FRC	133*LBL 12
61 X=0?	134 RCL 23
62 SF 08	135 X<> 17
63 8	136 STO 23
64 ST+ 07	137 XEQ 14
65 XEQ IND 07	138 GTO 13
66 RCL 17	139*LBL C
67 RCL 13	140*LBL 03
68 RCL 15	141 FS? 09
69 STO 09	142 LN
70 *	143 RCL 08
71 RCL 18	144 *
72 /	145 RCL 09
73 -	146 FS? 08

Listing continued on page 118.

Routine Listing For: CV	
147 LM	171*LBL E
148 +	172*LBL 05
149 FS? 08	173 .
150 E+X	174 STO 25
151 RTN	175 4
152*LBL D	176 STO 07
153*LBL 04	177*LBL 07
154 FS? 08	178 RCL 07
155 LM	179 XEQ B
156 RCL 09	180 RCL 25
157 FS? 08	181 RCL 10
158 LM	182 ABS
159 -	183 X<=Y?
160 RCL 08	184 GTO 15
161 /	185 STO 25
162 FS? 09	186 RCL 07
163 E+X	187 STO 26
164 RTN	188*LBL 15
165*LBL e	189 DSE 07
166*LBL 00	190 GTO 07
167 11.024	191 RCL 26
168 XROM "BC"	192 XEQ 02
169 E	193 RCL 26
170 RTN	194 END

LINE BY LINE ANALYSIS OF **CV**

Line 02 provides access to all numeric labels within **CV**.

Lines 03-45 perform the function of inputting the next data point. This is the "sigma plus" subroutine. All summations are updated when this routine is called. These summations include sums of

$$x, x^2, y, y^2, xy, \ln(x), \ln(x)^2, \ln(y),$$

$$\ln(y)^2, \ln(x)\ln(y), x\ln(y), y\ln(x).$$

Lines 48-50 perform the "sigma minus" function for deleting a data pair. Note that flag 10 is set and then a jump is made into the "sigma plus" function.

Lines 51-107 calculate the parameters a, b, and r, for the curve type using formulas (13), (14), and (15). b is stored in R08, a is stored in R09 and r is stored in R10.

Lines 108-138 are a series of intertwined subroutines which are called in the curve fit process. These routines simply perform a series of register exchanges which place the proper sums in the sigma registers for the calculation of the parameters a, b, and r depending on the curve type selected. Since the exchange is performed twice (once in line 65 and once in line 98) all registers are returned to their original state.

Lines 139-151 perform the calculation of the predicted y value using formulas (1), (4), (7), and (10).

Lines 152-164 perform the calculation of the predicted value using formulas (3), (6), (9), and (12).

Lines 165-170 perform the initialization for the program by clearing the data registers used to accumulate the sums the program requires.

Lines 171-194 perform the function of selecting the best curve type among the four curve types that the program handles.

NUMERIC LABELS/FUNCTIONS IN THE **CV** PROGRAM

The following list gives a correspondence between numeric labels and subroutines to be called as part of **CV** programs. To call a subroutine function from one of your own programs, first store the number corresponding to the desired function in data register R06. Then use the instruction XEQ "**CV**" as part of your program. The execution times for the more significant subroutines are in seconds and are shown in parentheses.

Numeric Label Number in R06	Keyboard Label	Subroutine Function
00	e	Initialize/clear sigma registers (3.2 sec)
01	A	Sigma Plus function (2.2 seconds)
02	B	Solve Type J (1: 2.9 seconds) (2: 3.5 seconds) (3: 3.4 seconds) (4: 4.3 seconds)
03	C	Predict y
04	D	Predict x
05	E	Solve Best Curve Type (15.7 seconds)
06	a	Sigma Minus function provided F10 is set
09	none	provides simple RTN so no register exchange takes place
10	none	exchange register pairs R11&R17, R15&R21, R16&R22
11	none	exchange register pairs R12&R17, R13&R19, R14&R20
12	none	exchange register pairs R23&R17, R13&R19, R14&R20 R15&R21, R16&R22
13	none	exchange register pairs R15&R21, R16&R22
14	none	exchange register pairs R13&R19, R14&R20

Note that to use the Sigma Minus function you must set flag 10 before calling label 06.

Note that labels 09-14 are not represented by functions on the keyboard. They are only internal subroutines within **CV** which would seem to perform no useful purpose outside of **CV**. However, they are documented here only because they may provide the truly curious PPC member with some "hidden" functions that they will no doubt find some application for. Those who never read this will never know what they are missing.

REFERENCES FOR **CV**

- Gary M. Tenzer (1816) PPC Journal "Curve Fitting Made Easy":

I	V5N1P29
II	V5N3P9
III	V5N6P29
IV	V6N3P16
V	V7N2P20
VI	V7N5P46

CONTRIBUTORS HISTORY FOR **CV**

Curve fitting has long been a topic of great interest among PPC members. Gary M. Tenzer (1816) has written several articles introducing this topic and is responsible among others for the development of curve fit programs for the HP-67 and the HP-41C.

The key equation for this program, $Y = B \cdot X + A$, was taken from a program written by Keith Jarrett (4360). This one equation unifies the four types of curves fit and greatly simplifies previous programs that accomplish the same functions, but less elegantly. John Kennedy (918) did the final coding of the **CV** program. Bill Barnett (1514) provided CVPL.

FINAL REMARKS FOR **CV**

CV is not the most powerful curve fit program ever written for a programmable calculator. Limited space did not allow a more comprehensive routine which would fit up to 16 different types of curves. Only the standard 4 types are present in **CV**. **CV** can be extended by those wishing to take advantage of its subroutines.

FURTHER ASSISTANCE ON **CV**

NOTES

TECHNICAL DETAILS	
XROM: 20, 08	CV SIZE: 027
<u>Stack Usage:</u> 0 T: used 1 Z: used 2 Y: used 3 X: used 4 L: used	<u>Flag Usage:</u> 04: not used 05: not used 06: not used 07: not used 08: used CV type 09: used CV type 10: used Σ -function 25: not used
<u>Alpha Register Usage:</u> 5 M: not used 6 N: not used 7 O: not used 8 P: not used	
<u>Other Status Registers:</u> 9 Q: not used 10 R: not used 11 a: not used 12 b: not used 13 c: not used 14 d: not used 15 e: not used	<u>Display Mode:</u> not used <u>Angular Mode:</u> not used <u>Unused Subroutine Levels:</u> 4
ΣREG: used <u>Data Registers:</u> R00: not used R06: function call # R07: CV type R18: n R08: b, x R19: $\Sigma \ln(x)$ R09: a, y R20: $\Sigma \ln(x)^2$ R10: r R21: $\Sigma \ln(y)$ R11: $\Sigma x \ln(y)$ R22: $\Sigma \ln(y)^2$ R12: $\Sigma y \ln(x)$ R23: $\Sigma \ln(x) \cdot \ln(y)$ R13: Σx R24: n R14: Σx^2 R25: best r R15: Σy R26: best type R16: Σy^2 R17: Σxy	<u>Global Labels Called:</u> <u>Direct</u> <u>Secondary</u> BC none <u>Local Labels In This Routine:</u> A, B, C, D, E, a, e, 00, 01, 02, 03, 04, 05, 06, 07, 09, 10, 11, 12, 13, 14, 15
Execution Time: see NUMERIC LABELS in the CV documentation	
Peripherals Required: none	
Interruptible? yes Execute Anytime? no Program File: CV Bytes In RAM: 292 Registers To Copy: 42	<u>Other Comments:</u> The subroutine which fits a given curve type should be allowed to run to its full conclusion so that a double register exchange is properly completed

SV - SOLVE ROUTINE

This routine is a simple root solving program which will approximate a solution to an equation of the form: $f(x)=0$ using the Secant Method (a simplified form of Newton's Method). **SV** will find only one root at a time. The program requires an initial guess and an initial step size. The output is an x value which most closely makes $f(x)=0$. A flag may be set to display the successive approximations as they converge to the final answer. Convergence depends on the initial guess. Accuracy depends on the display setting.

Example 1: Use **SV** to find the two roots of the quadratic equation $x^2 + 2x - 15 = 0$.

1. Insure a minimum SIZE 010.
2. Select a display setting of SCI 5. The routine will end when two successive approximations are rounded and found to be equal according to the display setting.
3. Set flag F10 to view the successive approximations.
4. The function on the left side of the equation must be programmed as a subroutine. The input to this subroutine, namely x, is assumed to be in the X-register and can be recalled from R07. The output from this subroutine, namely f(x), is also to be left in the X-register. For this example the following routine may be programmed in RAM program memory.

```
01*LBL "FX1"  
02 X↑2  
03 LASTX  
04 2  
05 *  
06 +  
07 15  
08 -  
09 RTN
```

5. The name of the global label "FX1" should be stored in R06. Go into alpha mode and key "FX1" ASTO 06.
6. The initial guess (nonzero) is to be entered along with an initial step size which may be zero or may be a small number compared to x. If a 0 step size is entered then the program will calculate the first step as 1% of the initial guess x. The program will also accept a non-zero value as the initial step size. For most applications and for this example use 0 as the initial step size. Choose x=7 as the initial guess for x. Key 0 ENTER↑ 7
7. XEQ "**SV**". The following consecutive approximations will be displayed.

```
7.00000+00  
6.93000+00  
3.98682+00  
3.30024+00  
3.03190+00  
3.00115+00  
3.00000+00
```

The final solution is returned after about 8 seconds. The true answer is exactly x=3. Since the above quadratic has two roots we will key in another initial guess to search for the other root. This time we will guess x = -10. Key 0 ENTER↑ 10 CHS and XEQ "**SV**". The following sequence of numbers will be displayed.

```
-1.00000+01  
-9.90000+00  
-6.36872+00  
-5.47003+00  
-5.06539+00  
-5.00360+00  
-5.00003+00  
-5.00000+00
```

The true answer this time is exactly x = -5.

COMPLETE INSTRUCTIONS FOR **SV**

(Keyboard Operations):

To calculate a root of $f(x)=0$:

- 1) Select SIZE. The minimum size required by **SV** is SIZE 010. The storage requirements for constants and coefficients associated with the function f(x) may dictate a larger size.
- 2) Select display setting. The display setting will generally determine when **SV** ends. If an exact solution is found then **SV** will end on the next iteration, otherwise **SV** rounds the last two approximations and ends if those rounded values are equal. In general, a display setting of SCI n will produce (optimistically) a solution correctly rounded to n+1 significant figures. A display mode of SCI or ENG is generally preferred to a FIX mode.
- 3) Specify display option. Flag 10 controls a display option. If F10 is set then the successively calculated approximations will be displayed. In this manner the user may view the progress of the iterations. This is especially recommended in the **SV** routine since **SV** may fail to converge if the initial guess is too far away from an actual root. Even when the values stabilize they may oscillate and it is a simple matter for the user to manually stop the program. If F10 is clear only the final x-value is returned.
- 4) Program the function $f(x)=0$. The function f(x) represented by one side of the equation must be programmed as a subroutine in program memory which starts with a global label name and ends with a RTN or END instruction. The label name should be of six or less characters and should be stored in R06. The input x and the output f(x) are both assumed to be in the X-register. The input x may also be recalled from R07. Since global label search begins from the bottom of program memory, it is advisable to place f(x) near the bottom of program memory. The f(x) program should not use registers R06-R09 and should not disturb flag F10.
- 5) Store the global label name from step 4) (six or less characters) in R06. The function subroutine call will be made via an XEQ IND 06 instruction.
- 6) Specify initial step size and initial guess. **SV** requires two input values. The first input is the step size which the program uses to determine the approximation for the derivative at the initial guess. The second input is the initial guess and is used as the starting x value by the program. The closer the initial guess is to the true solution the quicker the solution is found. Do not use 0 as an initial guess.

If zero is entered as the initial step size then the program will automatically calculate 1% of x as the actual step size. A zero step size should prove adequate for the majority of applications. However, the user may enter a non-zero step size which may be finer or coarser than 1% of the initial x.

These two values are keyed in as:

step size ENTER↑ guess

7. XEQ "SV". If F10 is set the program will display the consecutive approximations. If a printer is plugged in and turned on these approximations will be printed. The final solution will be left in the X-register when the program ends.

MORE EXAMPLES OF SV

Example 2: Solve $f(x) = x^3 - x - 1 = 0$.

- 1) SIZE 010 minimum
- 2) Set display mode as SCI 6.
- 3) Set flag F10 to view the approximations.
- 4) Key the following routine for f(x) into program memory:

```
LBL*FX2
ENTER↑
X↑2
1
-
*
1
-
RTN
```

- 5) Key "FX2" in the alpha register and press ASTO 06.
- 6) Key in the initial step size as 0 and key in the initial guess as x=4. Key 0 ENTER↑ 4.
- 7) XEQ "SV".

The following sequence of approximations will be displayed:

```
4.000000+00
3.960000+00
2.731772+00
2.226515+00
1.780222+00
1.522190+00
1.382556+00
1.333776+00
1.325185+00
1.324722+00
1.324718+00
```

The final solution is returned after about 13 seconds. The solution is correct to the digits displayed.

Example 3: Solve $f(x) = x^3 - 3x^2 + 4 = 0$.

- 1) SIZE 010 minimum
- 2) Set display mode as SCI 4.
- 3) Set flag F10 to view the approximations.
- 4) Key the following routine for f(x) into program memory:

```
LBL*FX3
X↑2
LAST X
3
-
*
4
+
RTN
```

- 5) Key "FX3" in the alpha register and ASTO 06.
- 6) Key in an initial step size of 0 and an initial guess of -2. Key 0 ENTER↑ 2 CHS
- 7) XEQ "SV".

The following approximations will be displayed:

```
-2.0000+00
-1.9800+00
-1.3283+00
-1.1289+00
-1.0229+00
-1.0018+00
-1.0000+00
```

The final answer is returned after about 8 seconds.

The true solution is exactly $x = -1$.

The following examples contain abbreviated instructions.

Example 4: The following equation is known as Kepler's equation:

$$x - E \cdot \sin(x) - m = 0$$

and plays an important role in astronomy and astrodynamics (space travel). It can be programmed as follows:

```
LBL*FX4
ENTER↑
SIN
RCL 01      (Note: R01=E)
*
-
RCL 02      (Note: R02=m)
-
RTN
```

Set RADIANS angle mode. The equation can now be solved for any values of E (R01) and m (R02). When $E=0.2$ and $m=0.8$ the function has only one root (9.64334-01) which can be found with any initial guess.

Example 5: SV can be used to find maxima and minima of a function by solving for zeros of its derivative.

For example, if $f(x) = \sin(x)/x$ then the derivative $f'(x) = [x \cdot \cos(x) - \sin(x)]/x^2$. Zeros of f' occur where the numerator is 0. Consequently, solutions can be found by applying SV to the following function which represents the numerator $g(x) = x \cdot \cos(x) - \sin(x)$

```
LBL*GX5
ENTER↑
COS
*
RCL 07
SIN
-
RTN
```

Assuming SCI 6 display mode and RADIANS angle mode. Store "GX5" in R06. Key in small initial guesses using a step size of 0 and SV will find the first few roots as 0, $\pm 4.49341+00$, $\pm 7.72525+00$. (Note that instead of taking the derivative algebraically, the ROM routine FD might be used).

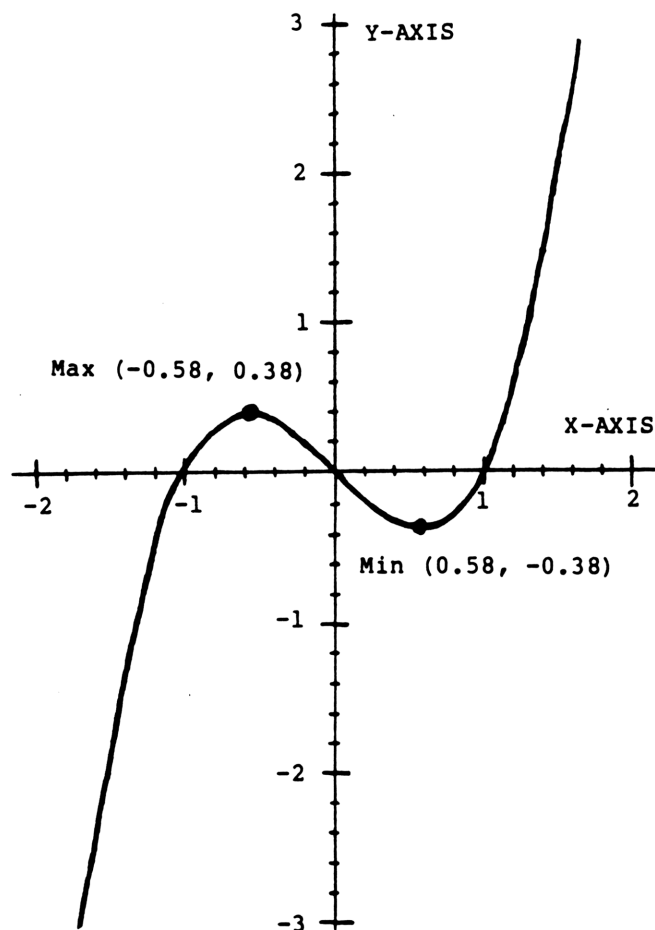
Example 6: Use **SV** to find all three roots of the cubic equation:

$$x*(x^2-1) = 0$$

The following should be programmed:

```
LBL*FX6
ENTER↑
X↑2
1
-
*
RTN
```

A sketch of the graph of this function will prove useful in understanding how the initial guesses determine which root is found.



GRAPH OF $Y = X^3 - X = X(X^2 - 1) = X(X+1)(X-1)$

The initial step size is 0 for each of the guesses suggested below. An initial guess greater than 0.6 for this example will find the root $x=1.0$, while a guess between -0.49 and $+0.49$ will find the root at $x=0.0$. However, guesses too close to the local peaks of the function at $x=\pm 1/\sqrt{3}$, where the slope of the tangent line is zero, lead to oscillations that fail to converge. Try an initial guess of $x=0.58$ to observe this behavior. This example also illustrates that the root found is not necessarily the one nearest to the initial guess. Try $x=0.52$ which finds the root $x = -1.0$.

Example 7: $f(x) = x*LN(x) - 1.2 = 0$

Display mode: SCI 6
Initial step size = 0
Initial guess = 3
See:

```
3.000000+00
2.970000+00
1.998929+00
1.902018+00
1.888327+00
1.888087+00
```

Result: 1.888087 after about 8 seconds

Example 8: $f(x) = 3*x - \cos(x) - 1 = 0$

Display mode: SCI 6
Use RADIANS angle mode
Initial step size = 0
Initial guess = 2
See:

```
2.000000+00
1.980000+00
6.159990-01
6.078243-01
6.071024-01
6.071016-01
```

Result: 6.071016-01 after about 10 seconds

Example 9: $f(x) = x^2 + 4*\sin(x) = 0$

Display mode: SCI 6
Use RADIANS angle mode
Initial step size = 0
Initial guess = -4
See:

```
-4.000000+00
-3.960000+00
-2.210789+00
-2.037220+00
-1.946227+00
-1.934406+00
-1.933758+00
-1.933754+00
```

Result: -1.933754 after about 12 seconds

Example 10: $f(x) = x^4 - 26x^2 + 49x - 25 = 0$

Display mode: SCI 6
Initial step size = 0
Initial guess = 5
See:

```
5.000000+00
4.950000+00
4.310588+00
4.077156+00
3.927333+00
3.883035+00
3.876065+00
3.875777+00
3.875775+00
```

Result: 3.875775 after about 14 seconds

Continuing this same example for another root:

Initial step size = 0

Initial guess = -10

See:

```
-1.000000+01
-9.900000+00
-7.959395+00
-7.135530+00
-6.438967+00
-6.086301+00
-5.945260+00
-5.917701+00
-5.915863+00
-5.915842+00
```

Result: -5.915842+00 after about 14 seconds

FURTHER DISCUSSION OF **SV**

SV is not a sophisticated root solver and is subject to all the difficulties and error traps that confront all other root solvers. Limited space in the **PPC ROM** did not allow protection schemes to detect or rectify possible trouble areas. The method used, strictly speaking, is the Secant Method, however, it can be considered a form of Newton's Method where a numerical approximation is used for the derivative. A secant line is used to approximate the true tangent line. If **SV** fails to converge then another initial guess must be tried. **SV** can be effective as a subroutine in a program provided the user has knowledge of the range of appropriate values for the given function.

The display setting will help control the accuracy of the final result. When in SCI n display mode the final answer will (usually) be accurate to n+1 significant digits. However, sometimes this is not the case and the final answer will not be as accurate as the display setting would indicate. Every floating point operation in a computational process can give rise to rounding error which, once generated, may then be increased in subsequent operations.

For example, let $f(x) = x^2 - 6x + 9$ and use **SV** to solve for $f(x)=0$. In FIX 9 the answer **SV** returns may be 3.000030072 which is accurate to five digits only. The true solution is a double root at $x=3$. Thus the display setting has not determined the accuracy in this example. This is basically caused by using only ten digits internally in the calculator, causing each and every calculation to have its solution rounded to ten digits. The root solver itself cannot then be held to ransom when the $f(x)$ routine is affected by rounding errors.

This example highlights the action of **SV** when the secant line is horizontal, that is, when $f(a)=f(b)$ where a and b are successive approximations. This situation may occur at multiple roots where the first derivative shares a root with the original function. (When the first derivative is zero the tangent line is horizontal). In general, do not select a display n value any larger than necessary. The use of SCI or ENG display modes are generally preferred to the FIX display mode. And do not blindly accept any solution given by this or any other root-solving program. Any potential real solution can be validated by applying the $f(x)$ subroutine to see how close $f(x)$ really is to 0.

(C52)

Because the HP-34C calculator's SOLVE routine uses a similar method as **SV** (with many refinements), users are urged to study Reference 5. In that informative article some of the problems of root solving and a description of the mathematics of the secant method are discussed. Reference 1 provides a broad background to the subject. Further information may be found in most university and college libraries.

THE SELECTION OF A METHOD FOR **SV**

The oldest known method of root-solving is the Method of False Position, or Regula Falsi. Commencing with two estimates, lying on either side of the actual root, inverse linear interpolation is applied to produce a new estimate. Here, a linear function (a straight line) is used to approximate the true function $f(x)$ over the interval of interest. This can be seen to be "reasonable" approximation so long as the interval is "small". As the two estimates must always straddle or bracket the root, convergence is always guaranteed. Of course, after calculating the new estimate, a decision has to be made as to which previous estimate is to be discarded.

The Method of False Position has a unity order of convergence, making the iteration time reasonably long. However, the solution is always obtained.

A method similar to False Position is the Secant Method. Although the mathematics of the two methods are identical, the difference lies in the fact that the Secant Method uses the approximations in strict sequence. Thus, the bracketing of the root is no longer necessary, and a secant is used to approximate the function $f(x)$ over the interval of interest. Then, inverse linear interpolation is applied to produce the next estimate of the root. The Secant Method's order of convergence is approximately 1.62, which is higher than the Method of False Position, but convergence to the root is no longer guaranteed.

Both the Method of False Position and the Secant Method are in the class of two-point iterative methods, which, while the order of convergence is not high, nevertheless have high stability.

Another popular method derived from calculus using a Taylor Series is commonly known as the Newton-Raphson, or Newton's Method. In this method a tangent line to the function is used to determine the direction and amount of displacement to move from the current estimate to the new estimate. Newton's Method belongs to the class of one-point iterative methods. Newton's Method is the official and familiar name of tangent sliding philosophy, and has an order of convergence of two.

The mercurial properties of Newton's Method arise from its use of derivative information gathered at one point. Both the function and its derivative must be evaluated at each iteration. (This also results in a greater programming effort for two functions are really being evaluated). The time for an iteration is longer than the False Position and Secant Methods, which both require only one function evaluation per iteration. However, the convergence rate of Newton's Method is greater than both.

Comparison of Methods

A. Method of False Position.

Advantages:

1. Convergence is guaranteed.
2. Only one function evaluation is needed at each iteration.

Disadvantages:

1. Low (unity) order of convergence.
2. A decision is needed as to which estimate to discard to insure root-bracketing occurs.
3. The root-bracketing requirement prevents its use at multiple, even-order roots.

B. The Secant Method.

Advantages:

1. Medium (approx. 1.62) order of convergence.
2. Estimates are used in strict sequence, so no decision is needed on which estimate to discard.
3. Only one function evaluation is needed at each iteration.
4. Can be very stable.

Disadvantages:

1. Convergence is not always guaranteed.
2. May have difficulty at multiple even-order roots.

C. Newton's Method.

Advantages:

1. High (2.0) order of convergence.

Disadvantages:

1. Convergence is not always guaranteed.
2. Both the function and its derivative must be evaluated at each iteration, which increases the time per iteration.
3. The derivative must be known explicitly.
4. Can be very unstable.
5. Has difficulty at multiple, even-order roots, where the function and its first derivative both have the same root.

Summary

From the above comparison, the Secant Method was considered to be the optimum algorithm, and was selected for **SV**. It combines a reasonable rate of convergence, a (usually) stable two-point step, uses the calculated approximations in strict sequence, and does not require evaluation of the derivative, dispensing with the need to provide the derivative explicitly.

ROOT SOLVING DIFFICULTIES - A PRIMER

One of the most frequently occurring problems in scientific work is to find the values of x for an expression $f(x)$ which will make $f(x)=0$. Those values of x are called the roots of the equation $f(x)=0$. The function may be given explicitly as, for example, a polynomial, or as a transcendental function. In rare cases it may be possible to obtain the exact roots by algebraic manipulations. In general, however, we can hope to obtain only approximations to the roots relying on some iterative computational procedure to produce those approximations.

In the year 1225 Leonardo of Pisa studied the equation:

$$f(x) = x^3 + 2x^2 + 10x - 20$$

and was able to produce the root of 1.368808107.

Nobody knows by what method Leonardo found this value, but it was a remarkable achievement for his time.

Simply put, all we require are those values of x which will make $f(x)=0$. It should be easy, so why all the fuss? The basic difficulty stems from the fact that our root solving methods tend only to use the function expression to numerically evaluate $f(x)$ and have no analytical knowledge of the function. If they did, better starting guesses and better exit criteria could be selected. More importantly, the best root solving method to use for a particular function could be selected.

Easy though root solving may seem, and when coupled with one of the popular methods, e.g., Newton's Method, Secant Method, Bisection Method, it may come as a surprise to know that the search for the perfect root solver is no less difficult than the search for the Holy Grail! For every root finding method put forward, a situation can be provided which will cause the method to fail and deny us the solution.

What can be done? A root solver is simply a computational process which uses known facts (data) to calculate a better approximation to the root. The basic difference among root solving methods is the way in which the known facts are used to calculate the improved estimate.

If we know situations that cause our root solver to fail to find the root, we can provide assistance by enhancing the basic method with strategies to detect these problem situations and allow an escape from them. How many difficulties may confront a root solver? How many strategies do we design into it? One, five or one hundred? If we limit our root solver to solve only a specific class of problems we may be able to implement some strategies to overcome the typical problems encountered by that class.

Starting Values

All root solvers require starting values whether they be entered manually as in **SV** or use some set of values, e.g., 1 and 10. If the starting value is not "close" to the root, our selected method may step away from the root, making the problem worse.

Therefore the accuracy of the starting values (guesses) can be seen to be as important as the selected root solving method and its built-in strategies. Do we have strategies for determining an approximation to the root we seek? In some cases the answer is yes.

Exit Criteria.

When the root solver has located a root, it exits the iteration process, gives us the answer and stops. How does the root solver know when it really has found a root? Our built-in exit criteria must apply certain tests to the known facts and assess if a root has been found. Again, situations can be provided to fool the exit tests, and cause execution to stop when really no root has been found (i.e., no root exists).

Should we have several exit tests built into the program? Under what conditions should each be invoked? If only the answers were simple!

Numerical Instability and round off errors.

Some root solvers can exhibit instability in certain

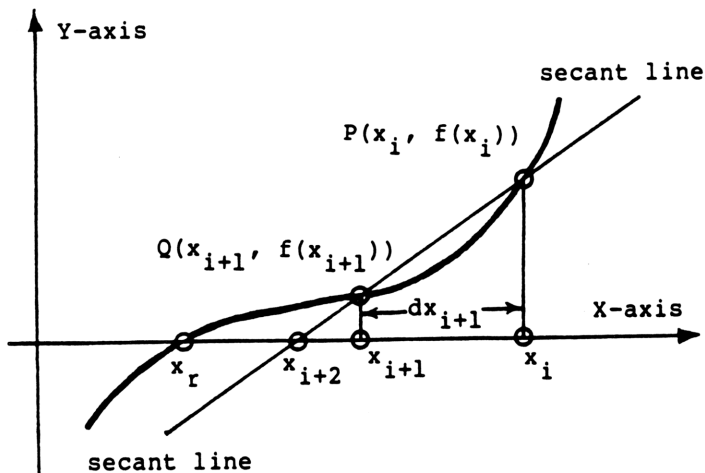
conditions. Do we know how many of these conditions exist for any one root solver? How do we overcome these problems? As all floating point calculations are rounded to ten digits by the calculator a further difficulty arises due the propogation of these errors.

Because of the enormous difficulties which may confront a root solving process, **SV** has been written as an elementary routine, with no refinements or strategies included. It is left to the user to provide such strategies, by observing the behavior of the approximations, and taking action should divergence occur. A little experience with particular problems will provide guidance. Readers are urged to consult Reference 5.

FORMULAS USED IN **SV**

Let $f(x)$ be the function whose root x_r we desire.

x_{i+1} and x_i are the two previous approximations.



GRAPH OF GENERAL FUNCTION WITH SECANT LINE SHOWING APPROXIMATION x_{i+2} FOLLOWING x_i AND x_{i+1}

The slope of the secant line through points P and Q is given by:

$$\frac{[f(x_{i+1}) - f(x_i)]}{[x_{i+1} - x_i]}$$

Letting $dx_{i+1} = x_{i+1} - x_i$ we have as the equation of the secant line through points P and Q:

$$y - f(x_{i+1}) = \frac{[f(x_{i+1}) - f(x_i)]}{dx_{i+1}} * (x - x_{i+1})$$

Hence,

$$(1) \quad x_{i+2} = x_{i+1} + dx_{i+2}$$

where

$$(2) \quad dx_{i+2} = \frac{(dx_{i+1}) * f(x_{i+1})}{f(x_i) - f(x_{i+1})}$$

The initial value x_0 is input by the user and dx_0 is usually taken as a small fractional part of x_0

For example, if we assume $f(x_{-1})=0$ and $dx_0=(.01)x_0$ then $x_1 = (.99)x_0$

Execution halts when x_{i+2} and x_{i+1} are rounded and found to be equal.

Routine Listing For: SV	
91*LBL C	108 ST* 09
92*LBL "SV"	109 ST- 08
93 STO 07	110 RCL 09
94 E	111 RCL 08
95 %	112 X*0?
96 RCL Z	113 /
97 X=0?	114 STO 09
98 X<>Y	115 X<> 07
99 STO 09	116 ST+ 07
100 CLST	117 RND
101*LBL 04	118 RCL 07
102 RCL Z	119 RND
103 STO 08	120 X*Y?
104 RCL 07	121 GTO 04
105 FS? 10	122 RCL 07
106 VIEW X	123 RTN
107 XEQ IND 06	

LINE BY LINE ANALYSIS OF **SV**

Lines 91-101 initialize the program by storing the initial guess X_0 in R07 and store the initial step size in R09. Note that if the user has input 0 as the initial step size then lines 94 & 95 and lines 97 & 98 calculate and select the value $0.01 * X_0$ as the actual step size.

Lines 101-121 are the main loop in the program. At LBL 04 X, Y, and T are assumed to be scratch and $f(X_i)$

is assumed to be in Z. The next approximation is calculated via formula (1) and is stored in R07 (line 116). Next, the two most recent approximations are rounded and tested for equality in line 120. A branch is then made back to LBL 04 unless the rounded values are equal.

Lines 122 & 123 recall the final solution and end the routine.

REFERENCES FOR **SV**

1. Forman S. Action, NUMERICAL METHODS THAT WORK, Harper and Row, New York, 1970
2. S. D. Conte and C. de Boor, ELEMENTARY NUMERICAL ANALYSIS, McGraw-Hill, 1972
3. John Kennedy (918) PPC JOURNAL "Method of Successive Bisections" V5N8P19. See also V6N5P10
4. Chris Stevens, PPC JOURNAL, V5N8P45, Sept.-Oct. 1978
5. William M. Kahan, "Personal Calculator Has Key To Solve any Equation $f(x)=0$ ", Hewlett-Packard Journal, December 1979.

CONTRIBUTORS HISTORY FOR **SV**

John Kennedy (918) wrote the **SV** program for the HP-41C from a previous HP-25 program. Graeme Dennes (1757) and Richard Schwartz (2289) made suggestions for improvements in the accuracy and overall program operation. Harry Bertucelli (3994) suggested register usage to allow **SV** to be used with **IG**. Graeme Dennes(1757) and Iram Weinstein (6051) contributed to the documentation of **SV**.

FINAL REMARKS FOR **SV**

SV needs improvement in almost all areas. **SV** is only a basic routine designed to be used primarily from the keyboard where the user may watch the convergence (or lack thereof) and take action to halt **SV** and make a new guess. **SV** lacks all of the sophistication of the SOLVE function on the HP-34C calculator.

FURTHER ASSISTANCE ON **SV**

NOTES

TECHNICAL DETAILS

XROM: 20, 10

SV

SIZE: 010 minimum

Stack Usage:

- 0 T: used
- 1 Z: used
- 2 Y: used
- 3 X: used
- 4 L: used

Flag Usage:

- 04: not used
- 05: not used
- 06: not used
- 07: not used
- 08: not used
- 09: not used
- 10: displays successive approximations when F10 is set
- 25: not used

Alpha Register Usage:

- 5 M: not used
- 6 N: not used
- 7 O: not used
- 8 P: not used

Other Status Registers:

- 9 Q: not used
- 10 I: not used
- 11 a: not used
- 12 b: not used
- 13 c: not used
- 14 d: not used
- 15 e: not used

Display Mode:

SCI n recommended
controls accuracy

Angular Mode:

not used, but may be
required by function

Unused Subroutine Levels:

4

ΣREG: not used

Data Registers:

R00: not used

R06: function LBL name

R07: point x_1

R08: $f(x_1)$

R09: dx_1

R10: not used

R11: not used

R12: not used

Global Labels Called:

Direct	Secondary
function	none
LBL in R06	

Local Labels In This Routine:

C, 04

Execution Time: variable, depends on function,
display setting, and initial guess.

Peripherals Required: none

Interruptible? yes

Execute Anytime? no

Program File: **IG**

Bytes In RAM: 51

Registers To Copy: 43

Other Comments:

BC - BLOCK CLEAR

This block clear routine may be used to store zeros in a block of registers. **BC** uses the complete form of the general block control word bbb.eeell and can thus be used to clear blocks of consecutive registers or can be used to skip over registers within a block.

Example 1: Use **BC** to clear registers R05-R15.

Before clearing these registers we will first use the block increment routine **BI** to load them with data. (The **PPC ROM** routines **BI** and **BV** are extremely convenient for loading and viewing blocks of registers). Key 5.015 ENTER↑ 1 ENTER↑ and XEQ "**BI**". The consecutive integers from 1-11 should now be loaded in R05-R15 inclusively. To convince yourself, clear flags F09 and F10 and key 5.015 XEQ "**BV**". The block view routine **BV** should run through the registers and show the contents as described.

Now to clear these registers, simply key 5.015 and XEQ "**BC**". If you again use **BV** to view these registers you won't see anything because they have been cleared.

Example 2: Use **BC** to clear every 5th register starting with R07 and ending with R102.

Assuming you have the available memory and that the current size is at least 103, simply key in 7.10205 and XEQ "**BC**". The registers R07, R12, R17, R22, R27, R32, R37, R42, R47, R52, R57, R62, R67, R72, R77, R82, R87, R92, R97, and finally R102 should all contain zero. These registers may be inspected by keying in 7.10205 and XEQ "**BV**". Since **BV** skips over registers which are zero, none of the above registers will show up in the display; only a series of short TONES will be heard as **BV** runs through the registers.

COMPLETE INSTRUCTIONS FOR **BC**

- 1) The only input to **BC** is the block control word which is of the form bbb.eeell.
- 2) **BC** contains an internal ISG loop that is controlled by bbb.eeell. **BC** stores and uses this block control word in the Last X register. The Y, Z, and T registers are all preserved by **BC**. The X register contains 0 when **BC** ends. The following shows the stack contents on input/output from **BC**.

Input to **BC**:

T: T
Z: Z
Y: Y
X: bbb.eeell

L: L

Output from **BC**:

T: T
Z: Z
Y: Y
X: 0

L: final control word

MORE EXAMPLES OF **BC**

Example 3: Use **BC** to clear registers R13-R49 inclusive.

Key 13.049 and XEQ "**BC**".

Example 4: Use **BC** to clear the even numbered registers from R02-R100.

Key 2.10002 and XEQ "**BC**".

Example 5: Use **BC** to clear the odd numbered registers from R01-R99.

Key 1.09902 and XEQ "**BC**".

Routine Listing For: BC	
208*LBL "BC"	
209 SIGN	
210 CLX	
211*LBL 13	
212 STO IND L	
213 ISG L	
214 GTO 13	
215 RTN	

LINE BY LINE ANALYSIS OF **BC**

BC is a very short routine. The SIGN function at line 209 is used to store the block control word in LAST X which is used as an ISG counter in the loop in **BC**.

CONTRIBUTORS HISTORY FOR **BC**

The **BC** routine and documentation were written by John Kennedy (918) based on the suggestion from Richard Schwartz (2289) that where possible, the block routines should make full use of the block control word.

FINAL REMARKS FOR **BC**

The one area of improvement for a future **BC** routine would be greater speed.

FURTHER ASSISTANCE ON **BC**

NOTES

NOTES

TECHNICAL DETAILS

XROM: 20, 43

BC

SIZE: depends on block used

Stack Usage:

0 T: not used
1 Z: not used
2 Y: not used
3 X: 0
4 L: ISG counter

Flag Usage:

04: not used
05: not used
06: not used
07: not used
08: not used
09: not used
10: not used

Alpha Register Usage:

5 M: not used
6 N: not used
7 O: not used
8 P: not used

25: not used

Other Status Registers:

9 Q: not used
10 F: not used
11 a: not used
12 b: not used
13 c: not used
14 d: not used
15 e: not used

Display Mode:

not used

Angular Mode:

not used

Unused Subroutine Levels:

5

ZREG: not used

Data Registers:

R00:

R06: the data
R07: registers used
R08: are those
R09: defined by
the block

R10:

R11:

R12:

Global Labels Called:

Direct

none

Secondary

none

Local Labels In This Routine:

13

Execution Time: depends on block size, clears approximately 7.8 registers per second

Peripherals Required: none

Interruptible? yes

Execute Anytime? no

Program File: M

Bytes In RAM: 18

Registers To Copy: 61

Other Comments:

No special SIZE requirement is necessary provided the data block already exists

CSA

	X	Y	Z	T	L		X	Y	Z	T	L
						51*LBL 8					
01*LBL "CV"						52*LBL 02					
02 GTO IND 06						53 CF 08					
						54 CF 09					
03*LBL A						55 STO 07					
04*LBL 01						56 2					
05 CF 10						57 X<Y?					
						58 SF 09					
06*LBL 06						59 /					
07 STO 09						60 FRC					
08 X<>Y						61 X=0?					
09 STO 08						62 SF 08					
10 ΣREG 13						63 8					
11 FC? 10						64 ST+ 07					
12 Σ+						65 XEQ IND 07					
13 FS? 10						66 RCL 17					
14 Σ-						67 RCL 13					
15 RDN						68 RCL 15					
16 RCL 08						69 STO 09					
17 ENTER↑						70 *					
18 X>0?						71 RCL 18					
19 LN						72 /					
20 ST* Z						73 -					
21 RCL 09						74 STO 10					
22 X>0?						75 RCL 14					
23 LN						76 RCL 13					
24 ST* Z						77 X↑2					
25 X<>Y						78 RCL 18					
26 ΣREG 19						79 /					
27 FC? 10						80 -					
28 Σ+						81 STO Z					
29 FS? 10						82 /					
30 Σ-						83 STO 08					
31 R↑						84 RCL 13					
32 FS? 10						85 *					
33 CHS						86 ST- 09					
34 ST+ 12						87 X<>Y					
35 R↑						88 RCL 16					
36 FS? 10						89 RCL 15					
37 CHS						90 X↑2					
38 ST+ 11						91 RCL 18					
39 X<> Z						92 ST/ 09					
40 SIGN						93 /					
41 ST+ L						94 -					
42 RCL 08						95 *					
43 RCL 09						96 SQRT					
44 X<> L						97 ST/ 10					
45 RTN						98 XEQ IND 07					
46 RCL 08						99 8					
47 RCL 09						100 ST- 07					
						101 RCL 10					
48*LBL a						102 RCL 09					
49 SF 10						103 FS? 08					
50 GTO 06						104 E↑X					
						105 STO 09					
						106 RCL 08					
						107 RTN					

	X	Y	Z	T	L		X	Y	Z	T	L
100*LBL 10						152*LBL D					
109 RCL 11						153*LBL 04					
110 X<> 17						154 FS? 08					
111 STO 11						155 LN					
						156 RCL 09					
112*LBL 13						157 FS? 08					
113 RCL 21						158 LN					
114 X<> 15						159 -					
115 STO 21						160 RCL 08					
116 RCL 22						161 /					
117 X<> 16						162 FS? 09					
118 STO 22						163 E+X					
						164 RTN					
119*LBL 09											
120 RTN						165*LBL e					
						166*LBL 00					
121*LBL 11						167 11.024					
122 RCL 12						168 XEQ "BC"					
123 X<> 17						169 E					
124 STO 12						170 RTN					
125*LBL 14						171*LBL E					
126 RCL 19						172*LBL 05					
127 X<> 13						173 .					
128 STO 19						174 STO 25					
129 RCL 20						175 4					
130 X<> 14						176 STO 07					
131 STO 20											
132 RTN						177*LBL 07					
						178 RCL 07					
133*LBL 12						179 XEQ B					
134 RCL 23						180 RCL 25					
135 X<> 17						181 RCL 10					
136 STO 23						182 ABS					
137 XEQ 14						183 X<=Y?					
138 GTO 13						184 GTO 15					
						185 STO 25					
139*LBL C						186 RCL 07					
140*LBL 03						187 STO 26					
141 FS? 09											
142 LN						188*LBL 15					
143 RCL 08						189 DSE 07					
144 *						190 GTO 07					
145 RCL 09						191 RCL 26					
146 FS? 08						192 XEQ 02					
147 LN						193 RCL 26					
148 +						194 RTN					
149 FS? 08											
150 E+X											
151 RTN											

	X	Y	Z	T	L		X	Y	Z	T	L
195*LBL "BC"											
196 SF 25											
197 CLRGX											
198 FS?C 25											
199 RTN											
200 SIGN											
201 CLX											
202*LBL 16											
203 STO IND L											
204 ISG L											
205 GTO 16											
206 RTN											
207*LBL "SV"											
208 STO 07											
209 E											
210 %											
211 RCL Z											
212 X=0?											
213 X<>Y											
214 STO 09											
215 CLST											
216*LBL 08											
217 RCL Z											
218 STO 08											
219 RCL 07											
220 FC? 00											
221 VIEW X											
222 XEQ IND 06											
223 ST+ 09											
224 ST- 08											
225 RCL 09											
226 RCL 08											
227 X=0?											
228 /											
229 STO 09											
230 X<> 07											
231 ST+ 07											
232 RND											
233 RCL 07											
234 RND											
235 X*Y?											
236 GTO 08											
237 RCL 07											
238 END											

STRUCTURAL PROGRAMS

Structural Engineer's Creed

Hope is stronger than truth.

Truth is stronger than steel.

I hope this thing stands up.

REINFORCED CONCRETE DESIGN/ANALYSIS

"In the field of structural design the effort to get intelligence through standardization has been carried pretty far. In reinforced concrete, for example, it has been necessary to set up elaborate standards. Out of this work came a narrowly circumscribed standardization of procedures, which is called "the theory of reinforced concrete" and to which unfortunate students are exposed. Few will question that the standardized theory of reinforced concrete is perhaps as complicated a bit of nonsense as has been conceived by the human mind. It does, however, work pretty well as a check on undiscriminating unintelligence."

Hardy Cross, Engineers And Ivory Towers

There are six programs and several subroutines in this module concerned with reinforced concrete. The programs are:

WSBM	Working stress rectangular beam design
LFBM	Load factor (ultimate strength) rectangular beam design
Ic	Working stress cracked-section beam analysis
LFAN	Load factor beam analysis with cracking and fatigue check
VST	Shear reinforcing design (WSD or LFD)
CORBL	Concrete corbel design (WSD or LFD)

The subroutines are listed on the next page. In addition, a listing for a rectangular concrete column analysis program is included for use in RAM. Shortage of space and a re-evaluation of needs prevented it from being included in this version of BRIDGEROM.

FLAGS

These programs follow AASHTO if flag 04 is clear, and ACI if it is set. There isn't a lot of difference between the two (there isn't any difference for the LFD versions of VST and CORBL); what differences there are are outlined in the individual program documentation.

All the programs follow the conventions for the "user flags" 00 and 02. Shortened output results if flag 00 is set. Flag 02 is also active in these programs. Warning BEEPs are enabled if flag 02 is set;

they are silenced if it is clear. The programs all access the input subroutines N/Fc, which have warnings if the concrete and steel strengths are outside a certain range. Individual programs also have other warnings written in. Flags 01 and 03 have no effect on these programs.

CONCRETE COLUMN ANALYSIS

As noted, listings for CCOL and its subroutine CC are provided for the user to key into RAM. Space constraints made it impossible to include program CC in this version of BRIDGEROM or to provide a full concrete column analysis program.

CONSTANTS and STRESSES

The constant B is calculated (in subroutine N) as 0.85 for F_c' of 4000 psi or less, decreasing by 0.05 for each 1000 psi gain in concrete strength, with a minimum value of 0.65 for F_c' of 8000 psi or more.

An idiosyncrasy resulting from the use of both working stress and ultimate strength methods is that the reinforcing strength is prompted differently for working stress (F-S 20,24) than for ultimate strength (F-Y 40,60); but the concrete prompt (F_c' KSI) always intends the ultimate concrete strength to be input. For working stress designs it is then factored down according to the status of flag 04.

SUBROUTINES

All of the following subroutines (except KJ) are accessed by the reinforced concrete programs and may find some use in user-written programs. KJ is a stand-alone program which may be useful for hand calculations and as a subroutine in other programs. For storage register and input requirements, see the individual program listings with stack trace.

- Mc Calculates and displays 1.2 times the concrete cracking moment based on gross section and concrete modulus of rupture, $7.5 / f_c'$. Falls into subroutine "U."
- U Calculates the area of reinforcing needed for a given moment by ultimate strength methods, leaving the moment in the Y register. The moment should include the load factor but not the undercapacity factor ϕ . Falls into display routine AS.
- AS Displays the area of steel in square inches.
- N Prompts for reinforcing steel F-S (for WSD, FC 06) or F-Y (for LFD, FS 06). If by WSD, calculates

F-Y. Falls into routine Fc.

Fc Prompts for concrete strength f_c' . If flag 10 is set, it returns; if clear, calculates B_1 , $1/m$ ($.85 f_c'/F_y$), and $.75 \text{ Rho}_{bal}$; and leaves "N" in the alpha register for prompting by the calling program.

R Calculates and displays minimum reinforcing areas.

RH Calculates and displays maximum reinforcing areas.

VS "Port of entry" into shear reinforcing program VST

KJ Given $(\rho \times N)$, calculates the working stress constants K and J for a singly reinforced rectangular section, using the formulae

$$K = \sqrt{(pn)^2 + 2pn} - pn \quad \text{and} \quad J = 1 - K/3$$

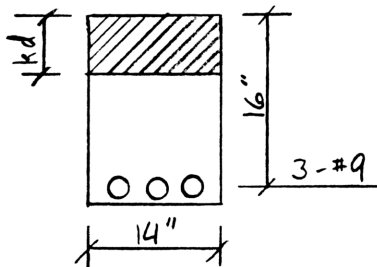
For these simple sections, it's often simpler to use KJ than the more comprehensive program Ic, solving for the missing variable:

$$M = A_s f_s J d/12$$

$$A_s = (M \times 12)/(f_s J d)$$

$$f_c = (2M \times 12)/(K J b d^2)$$

Example



$$f_c' = 4.0 \text{ ksi}$$

$$n = 8$$

$$f_s = 24 \text{ ksi}$$

$$\rho = 3.0 \text{ in}^2 / 14 \times 16 = .0134$$

$$\rho n = .0134 \times 8 = .1071$$

$$k = \sqrt{2\rho n + (\rho n)^2} - \rho n = .3680$$

$$\text{N.A. position} = kd = .368 \times 16 = 5.89 \text{ inches}$$

$$j = 1 - k/3 = .8773$$

$$\begin{aligned} \text{Allowable } M &= 3.0 \text{ in}^2 \times 24 \text{ ksi} \times .8773 \times 16 \text{ in} / 12 \\ &= 84.22 \text{ in-k} \end{aligned}$$

01*LBL "KJ"
 02 ENTER↑
 03 ENTER↑
 04 ENTER↑
 05 *
 06 +
 07 +
 08 SQRT
 09 X<>Y
 10 -
 11 ENTER↑
 12 SIGN
 13 LASTX
 14 3
 15 /
 16 -
 17 X<>Y
 18 RTN
 19*LBL "LPT"
 20 XEQ "L*"
 21*LBL 01
 22 "J↑D↑K↑L"
 23 PROMPT
 24 R↑
 25 STO 01
 26 R↑
 27 STO 1
 28 ST+ 00
 29 R↑
 30 R↑
 31 XEQ "XY"
 32 R-P
 33 X<> 1
 34 P-R
 35 RCL 01
 36 XEQ "P"
 37 "N"
 38 XEQ "SP"
 39 GTO 01

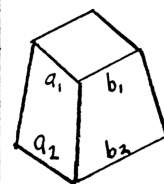
X	Y	Z	T	L
p n				
p n	p n	p n	p n	
$(pn)^2$				
$pn + 2pn$				
\sqrt{pn}				
p n				
K				
K	K	p n	p n	
1				
K				
3	K	1	K	
J	K	K	K	
K	J	K	K	
L	K	D	J	
J				
D				
L	K	D	J	
ΔX	ΔY			
d	0			
D	0			
$\Delta X'$	$\Delta Y'$			
J				
Y_J	X_J	ΔX	ΔY	

$$K = \sqrt{(pn)^2 + 2pn} - pn$$

$$J = 1 - K/3$$

40*LBL "PYR"
 41 *
 42 RDN
 43 *
 44 ST* Z
 45 R↑
 46 +
 47 R↑
 48 SQRT
 49 +
 50 3
 51 /
 52 RTN
 53*LBL "P1"
 54 "1(-"
 55 RTN
 56*LBL "P2"
 57 "1)"
 58 RTN
 59*LBL "PP"
 60 "11"
 61 RTN
 62*LBL "MU"
 63 "1μ"
 64 RTN
 65*LBL "AT"
 66 "1θ"
 67 RTN
 68*LBL "AM"
 69 "1&"
 70 END

X	Y	Z	T	L
a ₁	b ₁	a ₂	b ₂	
A ₁	a ₂	b ₂	A ₁	
a ₂	b ₂	A ₁	A ₁	
A ₂		A ₁ A ₂	A ₁ A ₂	
A ₁	A ₂		A ₁ A ₂	
A ₁ +A ₂				
$\frac{A_1 A_2}{\sqrt{A_1 A_2}}$				
$\frac{A_1 A_2}{\sqrt{A_1 A_2}}$				
$\frac{A_1 A_2}{\sqrt{A_1 A_2}}$				



$$Vol. = \frac{ht.}{3} \times (A_1 + A_2 + \sqrt{A_1 A_2})$$

REINFORCED CONCRETE BEAM DESIGN

xeq WSBM (SIZE 015) for working stress beam design
xeq LFBM (SIZE 011) for load factor beam design

SOFTKEYS (USER mode ON)

D New beam dimensions.

USER FLAGS

Flag 00	clear	Full output
	set	Only area of steel and warnings viewed
Flag 02	clear	Warning BEEP disabled
	set	Warning BEEP enabled
Flag 04	clear	AASHTO
	set	ACI

INPUT SUMMARY

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
F-S 20,24 or F-Y 40,60	Reinforcing or yield stress for WSBM or LFBM, respectively, in ksi	24 or 60
Fc'KSI	Ultimate concrete strength, ksi	4.0 ksi
N (WSBM only)	Modular ratio E_{st}/E_c	No default
B"	Beam width, inches	Previous B
D"	Depth to reinforcing, inches.	Previous D
H"	Full beam depth (height), inches	Previous H
M, 'K	Design moment in foot-kips. By- pass (or enter zero) to access shear design subroutine.	Zero

OUTPUT SUMMARY (Items in parenthesis are displayed only if flag 00 is clear)

(AASHTO or ACI)	States which code is being followed. If not using a printer, press R/S to continue.
(1.2Mc)	1.2 times the concrete cracking moment, for AASHTO's minimum reinforcing calculation
(200/FY)	ACI's minimum reinforcing ratio
(A-S)	The minimum area of steel based on one of the above two criteria
(.75Pb)	The maximum reinforcing ratio, 75% of rho-balanced
(M/BD ²)	The moment over bd^2 , for checking hand calculations. Is actually M/Obd^2 for LFBM.
(Y or a)	The depth to the neutral axis (WSBM) or the depth of the compression block (LFBM), inches
(OVER)	(WSBM only) The design is over-reinforced by working stress rules (general practice discourages but does not prohibit this)
(RHO)	The required reinforcing ratio
P/Pb	The required reinforcing ratio divided by the balanced reinforcing ratio. Design may still be acceptable if compr. steel is present.
A-S	The required area of steel
4/3 A-S	4/3 times the area of steel required for strength, if A-S is less than the minimum

Note: shear input and output are detailed in documentation for program VST.

PROGRAM FLAGS

Flag 06	clear set	WSBM LFBM
Flag 07	clear set	Design is physically possible Design requires a neutral axis depth greater than the depth of the member
Flag 09	clear set	WSBM or LFBM CORBL
Flag 10	clear	Subroutine N runs full length

STORAGE REGISTER USE

<u>Register</u>	<u>WSBM</u>	<u>LFBM</u>
00	$.85 F_c' / F_y$	$.85 F_c' / F_y$
01	F_y	F_y
02	$.75 P_b$	$.75 P_b$
03	B (width)	B
04	D (depth)	D
05	$\phi b d^2$	$\phi b d^2$
06	$\sqrt{F_c'} / 253$	$\sqrt{F_c'} / 253$ (for 1.2Mcr calc.)
07	H (member height)	H
08	Min. A_{st}	Min. A_{st}
09	N (E_{st} / E_c)	Not used
10	D/3 (trial Y); temporary $.75 P_b$	Temporary $.75 P_b$ while in shear design subroutine
11	Trial Y	
12	$(.40 \text{ or } .45) F_c'$	
13	F_{st} (20 or 24 ksi)	
14	R_{ideal}	

REINFORCED CONCRETE BEAM DESIGN

Programs WSBM and LFBM calculate the amount of tension reinforcing necessary to resist a given moment (service or factored, respectively). They specify the amount of reinforcing steel needed for singly reinforced rectangular beams. The programs assume the design reinforcing to be in a single layer, and they do not recognize compression reinforcing. T-beams can be accommodated only if the neutral axis falls within the flange. For the purpose of checking this, the programs display the neutral axis depth, Y , in inches (WSBM), or the depth of the compression block, " a " (LFBM), if flag 00 is clear.

AASHTO/ACI MINIMUM A_s

Both programs design by AASHTO if flag 04 is clear, and by ACI if flag 04 is set. The main differences are in the allowable concrete compression stress for WSBM, and the minimum amount of reinforcing required. The programs calculate a maximum reinforcing ratio of $.75p_b$; $1.2M_{cr}$, AASHTO's cracking moment criterion for minimum reinforcing; $200/F_y$, ACI's minimum reinforcing ratio; and the minimum reinforcing based on one of these, depending on the status of flag 04. These are all displayed if flag 00 is clear. The reinforcing designed is checked against these minimum and maximum values and warnings are displayed if they are violated. Neither ACI nor AASHTO requires the maximum of $.75p_b$ for working stress designs. However, it is checked for WSBM as well as LFBM as a matter of good design practice. The AASHTO minimum reinforcing is calculated (with $\phi = 0.9$) for $1.2M_{cr}$ based on ultimate strength concepts for both WSBM and LFBM; although the present AASHTO is rather vague on this, the Commentary to the 1982 Interim Specifications, where it was introduced, indicates this more clearly. Logically, all designs should have the same minimum and maximum steel requirements.

The programs are written to accept beam dimensions and material properties, calculate the above-mentioned (ad infinitum) requirements, then enter a flexural design loop. A design moment is prompted; the programs calculate M/bd^2 , the neutral-axis or compression block depth, the required reinforcing ratio, and the corresponding amount of steel. If flag 00 is clear, all these are displayed (useful for checking hand calculations); if flag 00 is set, only the area of steel is displayed. The maximum and minimum requirements are checked next. If the design is above the maximum, the programs display " $P/P_b=...$ "; 0.75 is the maximum allowed. If the design is less than minimum, the program calculates and displays $4/3$ the design area

of steel. If either program finds that a reinforcing ratio over 0.1 is required, the program displays NG (no good) and returns to the M,'K prompt.

OVER-REINFORCED BEAMS

If flag 00 is clear, WSBM displays a notice, "OVER," if its design yields an "over-reinforced" beam by working stress concepts, i.e. a design governed by the allowable concrete stress. This is acceptable by general practice but may produce deflection problems. It is possible to have an over-reinforced beam which has less than $.75p_b$, especially when designing by AASHTO. With its lower allowable concrete stress ($.40F_c'$), AASHTO will sometimes force an over-reinforced design that would have been under-reinforced by ACI.

STRESSES

WSBM has a couple of idiosyncracies concerning the allowable stresses input. First, although the input steel stress is a working stress (usually 20 or 24 ksi), the concrete stress to be input at the F_c' KSI prompt is the ultimate strength of the concrete, usually 3 or 4 ksi. The factor of .40 (AASHTO) or .45 (ACI) is applied by the program. Second, the input subroutine N calculates the reinforcing yield stress from the working stress input: $F_y = 5 \times (F_{st} - 12 \text{ ksi})$, giving 40 and 60 ksi from 20 and 24, respectively. If the designer is using a different allowable stress (e.g. some tank wall designs are done at 14 or 16 ksi to hold down cracking), the calculated maximum and minimum reinforcing ratios (which are based on ultimate strength concepts and yield stresses) will be invalid.

LOAD and UNDER-CAPACITY FACTORS

The designer must be careful to input the correct moment to LFBM. The moment must have the load factor(s) already applied; however, it must not include the undercapacity factor ϕ , which is applied by the program ($\phi = 0.90$). See the examples.

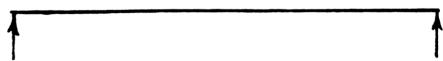
SHEAR DESIGN

Both LFBM and WSBM access a shear reinforcing design subroutine by bypassing the M,'K prompt (or keying in zero). This subroutine uses the beam dimensions and material properties input into the flexural design programs; the only new input required is the area of the stirrups and the shear. See the documentation for program VST. While in this subroutine, softkeys from the calling program are useless. VST has softkeys A and D; if either of these is accidentally used while in the shear subroutine, it will not be able to return to the calling program.

Reinforced Concrete Beam Design

1/10

Design a beam for the following conditions:



Span = 20'-0", simple
DL = 2400 #/ft (heavy industrial)
LL = 3600 #/ft

$f'_c = 4000$ psi
 $f_y = 60$ ksi
 $n = 8$

Moment

$$DL = 2.40 \times 20^2 / 8 = 120.0 \text{ k}$$
$$LL = 3.60 \times 20^2 / 8 = 180.0 \text{ k}$$

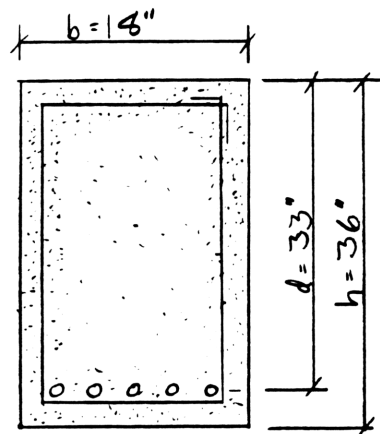
Shear

@ Reaction: $DL = 240 \times 20' / 2 = 24.0 \text{ k}$
 $LL = 3.60 \times 20 / 2 = 36.0 \text{ k}$

(Note: take @ reaction since beam depth for 'd/2' is unknown)

@ 1/4 Pt.: $DL = 2.40 \times 5' = 12.0 \text{ k}$
 $LL = \frac{7.5'}{20'} \times (15' \times 3.60) = 20.25 \text{ k}$

Try this configuration:



Design by:

- 1) AASHTO W.S.D.
- 2) ACI W.S.D. ("Alternate")
- 3) AASHTO L.F.D.
- 4) ACI L.F.D. ("Strength")

1) and 2) use "WSBM"

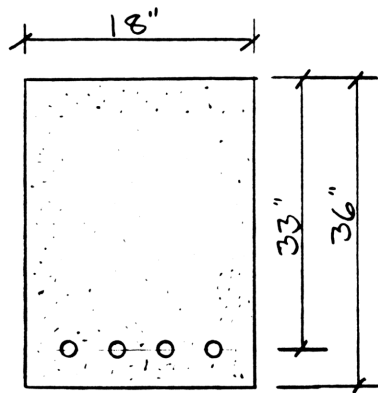
3) and 4) use "LFBM"

Calculate the amount of reinforcing necessary for the following loads:

$$\begin{array}{rcl} \text{DL} & 120' \cdot \text{ft} & \text{(including selfwt.)} \\ \text{LL} & 180' \cdot \text{ft} & \\ \hline & 300' \cdot \text{ft} & \end{array}$$

$$f'_c = 4000 \text{ psi}; n = 8$$

$$f_y = 24 \text{ ksi (Grade 60)}$$



Calculate Min. Steel

$$M_{cr} = S \times f_r$$

$$f_r = 7.5 \sqrt{f'_c} = 7.5 \sqrt{4000}$$

$$= 474.34 \text{ psi}$$

$$S = \frac{bh^2}{6} = 18 \times 36^2 / 6 = 3888.0 \text{ in}^3$$

$$1.2 M_{cr} = 1.2 (3888 \times 474.34) / 12$$

$$= 184.42' \cdot \text{ft}$$

Calc. reinf. by ult. strength methods:

$$\rho = \frac{1}{m} \left(1 - \sqrt{1 - \frac{2mR_u}{f_y}} \right)$$

$$m = \frac{f_y}{.45 f'_c} = \frac{60}{.45 \times 4} = 17.647$$

$$R_u = \frac{M}{\phi b d^2} = \frac{184.42 \times 12}{.9 \times 18 \times (33)^2}$$

$$= .1254$$

$$\rho = \frac{1}{17.647} \left(1 - \sqrt{1 - \frac{17.647 \times .1054 \times 2}{60 \text{ ksi}}} \right)$$

$$= .0021$$

$$\text{Req'd } A_{s, \min} = \rho b d = .0021 \times 18 \times 33$$

$$= 1.266 \text{ in}^2$$

Calculate Max. Steel

$$\rho_{bal} = \frac{.85 \beta_1 f'_c}{f_y} \left(\frac{87}{87 + f_y} \right)$$

$$\text{for } f'_c = 4000 \text{ psi}, \beta_1 = .85$$

$$.75 \rho_b = .75 \times \frac{.85 \times .85 \times 4}{60} \times \left(\frac{87}{87 + 60} \right)$$

$$= .0214$$

$$\text{Max. } A_s = \rho b d = 12.70 \text{ in}^2$$

Note: neither AASHTO nor ACI require that $A_s < .75 \rho_b$ for working stress design. However, it is included as a matter of good design practice. The reasons for limiting the reinf. to $.75 \rho_b$ under L.F.D. are no less valid under W.S.D.; the beam doesn't know how it was designed.

Calculate Ideal R

$$R_{ideal} = \frac{1}{2} (.40 f'_c) \cdot j \cdot k$$

$$k = \frac{.40 f'_c}{(f_{sr}/n + .40 f'_c)}$$

$$= \frac{1.6}{(24 \text{ ksi}/8 + 1.6)} = .3478$$

$$j = 1 - \frac{k}{3} = .8841$$

$$R_{ideal} = \frac{1}{2} (1.60 \text{ ksi}) \cdot .3478 \cdot .8841$$

$$= .2460 \text{ ksi}$$

Calculate Req'd Reinforcing

$$\text{Actual } R = \frac{M}{bd^2} = \frac{300' \cdot k \cdot 12}{18 \cdot 33^2}$$

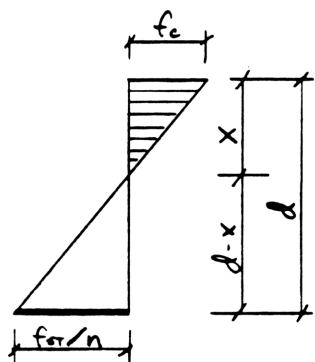
$$= .1837 \text{ ksi}$$

$$R_{act} < R_{ideal} \Rightarrow \text{under-reinforced}$$

\therefore Reinf. @ 24 ksi ; Concrete stress unknown

$$M = C \cdot (d - x/3)$$

$$\text{Max. concrete stress } f_c = \frac{(f_{sr}/n) \cdot x}{d - x}$$



$$C = \frac{1}{2} \cdot f_c \cdot x \cdot b$$

$$\therefore 300' \cdot k \cdot 12 = \frac{1}{2} \left(\frac{(24/8) x}{33 - x} \right) \cdot 18 x \cdot (33 - \frac{x}{3})$$

$$3600' \cdot k \cdot (33 - x) = \frac{1}{2} (3x) (18x) (33 - \frac{x}{3})$$

$$9x^3 - 891x^2 - 3600x + 118,800 = 0$$

solving the cubic by iteration,

$$x = 10.1435'' \text{ (Neutral Axis)}$$

$$C = \frac{1}{2} \left(\frac{3 \cdot 10.1435''}{33 - 10.1435''} \right) \cdot 10.1435'' \cdot 18''$$

$$= 121.54 \text{ k}$$

$$\text{Req'd } A_s = 121.54 \text{ k} / 24 \text{ ksi}$$

$$= \underline{5.0643 \text{ in}^2}$$

$$\text{Check: } 5.06 < 12.70 \text{ in}^2 \text{ (}.75 p_o)$$

$$5.06 > 1.27 \text{ (Min)}$$

OK

USE 4 - #10

Check spacing

$$18'' - 2(2'' + \frac{3}{4}'') - \overset{\text{say}}{1\frac{1}{2}''} = 11''$$

clear stirrup bar

$$11\frac{1}{3}'' \text{ spe. } - 1\frac{1}{2}'' \text{ bar} = 2.17'' \text{ clear}$$

OK

Note: WSBM does not check spacing or bar placement.

(DB)

Calculate Reinforcing
needed for 450 k on
same section.

$$R = \frac{450 \times 12}{18 \times 33^2} = .2755 \text{ ksi}$$

$$R_{act} > R_{ideal} \Rightarrow \text{Over-reinf.}$$

$$\therefore \text{Conc. stress} = .40 f'_c; f_{st} < 24 \text{ ksi}$$

$$M = C \times (d - \frac{x}{3})$$

$$450 \times 12 = \frac{1}{2} (1.60 \text{ ksi}) \times 18 \times (33 - \frac{x}{3})$$

$$5400 \text{ k} = 475.20 x - 4.80 x^2$$

solving the quadratic directly,

$$x = 13.0960''$$

$$C = \frac{1}{2} \times 1.60 \text{ ksi} \times 18'' \times 13.0960''$$

$$= 188.58 \text{ k}$$

$$f_s = n f_c \left(\frac{d - x}{x} \right)$$

$$= 8 \times (.4 \times 4 \text{ ksi}) \times \left(\frac{33 - 13.0960''}{13.0960''} \right)$$

$$= 19.454 \text{ ksi}$$

$$\text{Req'd } A_s = \frac{188.58 \text{ k}}{19.454} = \underline{9.6937 \text{ in}^2}$$

$$\text{Check: } 9.69 \text{ in}^2 < 12.70 \text{ in}^2 (.75 p_b)$$

$$9.69 > 1.27 \text{ (min)}$$

$$\underline{\text{OK - USE 7-}\#11} \quad (A_s = 10.72 < 12.70)$$

Check Spacing:

$$18'' - 2 \left(2'' + \frac{3}{4}'' \right) - 1\frac{1}{2}'' = 11''$$

cl. st. bar

$$\frac{11''}{6 \text{ sp}} - 1\frac{1}{2}'' = .33'' \text{ clear}$$

bar

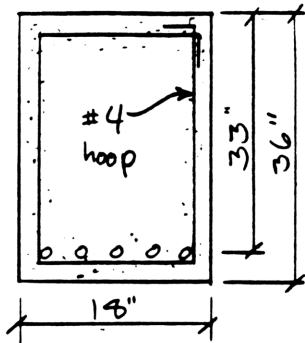
No Good

Will need 2 layers of reinforcing. Results of calculation (9.69 in²) no longer valid. Can use program "Ic" to check capacity of two-layer estimate.

Note: WSBM does not check spacing or bar placement. As in this case, the area of steel calculated by WSBM may be physically impossible to place.

From the preceding:

$$\begin{aligned} f'_c &= 4000 \text{ psi} \\ f_y &= 60 \text{ ksi} \\ f_{\sigma} &= 24 \text{ ksi} \end{aligned}$$



@ 1/4 Point

$$V_s = 32.25 - 35.69$$

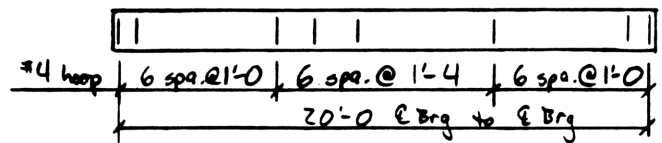
$$V_s < 0 \Rightarrow \text{Min. Steel}$$

$$\frac{1}{2} V_c = 17.84 \text{ k}$$

$$32.25 > 17.84 \Rightarrow \text{Need Stirrups}$$

$$\begin{aligned} \text{Max. Spa.:} \\ d/2 = 33/2 = 16.5" \quad \leftarrow \text{Governs} \\ \text{Max} = 24" \end{aligned}$$

Use this layout:



@ Reaction:

$$\begin{aligned} \text{DL} &= 24.0 \text{ k} \\ \text{LL} &= 36.0 \text{ k} \\ \Sigma &= 60.0 \text{ k} \end{aligned} \quad (\text{service})$$

@ 1/4 Point:

$$\begin{aligned} \text{DL} &= 12.0 \text{ k} \\ \text{LL} &= 20.25 \text{ k} \\ \Sigma &= 32.25 \text{ k} \end{aligned}$$

$$V_c = .95 \sqrt{4000} \cdot 18 \cdot 33 = 35.69 \text{ k}$$

@ Reaction,

$$V_s = 60.0 - 35.69 = 24.31 \text{ k}$$

Use stirrups - #4 hoop

$$A_v = 2 \cdot 20 = .40 \text{ in}^2$$

$$\text{Spa.} = \frac{A_v f_y d}{V_s} = \frac{.40 \text{ in}^2 \cdot 60 \text{ ksi} \cdot 33 \text{ in}}{24.31 \text{ k}} = 13.03 \text{ in}$$

Check Maximums

$$\begin{aligned} 2 \sqrt{f'_c} b d &= 2 \sqrt{4000} \cdot 18 \cdot 33 \\ &= 75.14 \text{ k} \end{aligned}$$

$$V_6 < 2 \sqrt{f'_c} \Rightarrow \text{Spa. OK - AASHTO 4.15.5.3.9}$$

$$V_s < 4 \sqrt{f'_c} \Rightarrow \text{OK - AASHTO 4.15.5.3.9}$$

$$A_{v-\min} = \frac{50 \text{ ksi} \cdot \text{Spa}}{f_y} \quad \text{AASHTO 4.19.1.2}$$

$$\therefore \text{Max Spa.} = \frac{.40 \cdot 60,000}{50 \cdot 18 \text{ in}}$$

$$= 26.67 \text{ in} \quad \text{OK}$$

XEQ "WSBM"

6:33:12 PM 12/09/88

TITLE:

TEST 1 RUN

AASHTO

F-S 20,24

24.0000 RUN

Fc, KSI 4.0000 RUN

N 8.0000 RUN

B" 18.0000 RUN

D" 33.0000 RUN

H" 36.0000 RUN

1.2Mc=184.4111 'K

A-S=1.2656 SQ"

200/FY=0.0033

.75Pb=0.0214

M, 'K 300.0000 RUN

M/BD2=183.6547

Y=10.1435"

RHO=0.0085

P/Pb=0.2991-OK

A-S=5.0643 SQ"

M, 'K 450.0000 RUN

M/BD2=275.4821

OVER

Y=13.0960"

RHO=0.0163

P/Pb=0.5725-OK

A-S=9.6937 SQ"

M, 'K 100.0000 RUN

M/BD2=61.2182

Y=6.2049"

RHO=0.0027

P/Pb=0.0955-OK

A-S=1.6165 SQ"

Design by
AASHTO1.2 Mcrack
Min. Reinf.Over-
reinforced

M, 'K

60.0000 RUN

M/BD2=36.7309

Y=4.8884"

RHO=0.0016

P/Pb=0.0565-OK

A-S=0.9563 SQ"

4/3 A-S=1.2751 SQ"

M, 'K

RUN

SHEAR

Vc=35.6895 K

A-V

.4000 RUN

V, K

60.0000 RUN

SPA=13.0314"

V, K

32.2500 RUN

SPA=16.5000" (MAX)

V, K

120.0000 RUN

SPA=3.7575"

V, K

15.0000 RUN

.5Vc>V

V, K

RUN

A-V

RUN

B"

As < Min. As
so display 4/3 AsBypass to
access shear
design

Bypass

Bypass to
return to
WSBM - new
beam dimensions

Short Format Example

7/10

CF 04
SF 00

Clear flag 04 → AASHTO
Set flag 00 → Short Format

XEQ "WSBM"

No title block

F-S 20,24

24.0000 RUN

Fc'KSI

4.0000 RUN

N

8.0000 RUN

B"

18.0000 RUN

D"

33.0000 RUN

H"

36.0000 RUN

No minimums or maximums

M, 'K

300.0000 RUN

A-S=5.0643 SQ"

Only A_s output

M, 'K

450.0000 RUN

A-S=9.6937 SQ"

M, 'K

600.0000 RUN

P/Pb=1.6172-NC

A-S=27.3837 SQ"

Warnings still output

M, 'K

60.0000 RUN

A-S=0.9563 SQ"

4/3 A-S=1.2751 SQ"

Minimums still output

M, 'K

Calculate the reinforcing required by ACI for moments of $300' \cdot k$ and $450' \cdot k$ on the same section.

$$f'_c = 4000 \text{ psi}; n = 8$$

$$f_{st} = 24 \text{ ksi (Grade 60)}$$

$$b = 18"$$

$$d = 33"$$

Calculate Min. Steel

$$\rho_{min} = \frac{200}{f_y} = \frac{200}{60,000} = .0033$$

$$\text{Min. } A_s = .0033 \times 18" \times 33" = \underline{1.98 \text{ in}^2}$$

Calculate Max. Steel

$$\text{As for AASHTO, } .75 \rho_b = .0214$$

$$\text{Max. } A_s = .0214 \times 18" \times 33" = \underline{12.70 \text{ in}^2}$$

Calculate Ideal R

$$R_{ideal} = \frac{1}{2} (.45 f'_c) \times j \times k$$

$$k = \frac{.45 \times 4.0}{(24 \text{ ksi}/n + .45 \times 4.0)} = .3750$$

$$j = 1 - \frac{1}{3} = .6750$$

$$R_{ideal} = \frac{1}{2} (1.8 \text{ ksi}) \times .375 \times .675$$

$$= \underline{.2953 \text{ ksi}}$$

Calculate Req'd Reinf. for $M = 300' \cdot k$

$$\text{Actual } R = \frac{300 \times 12}{18 \times 33^2} = .1837 \text{ ksi}$$

$$.1837 < .2953 \Rightarrow \text{under-reinf.}$$

\therefore Calculations are as before:

$$x = 10.1435"$$

$$c = 121.54 \text{ k}$$

$$\text{Req'd } A_s = \underline{5.0643 \text{ in}^2}$$

$$\text{check: } 5.06 < 12.70 \text{ (.75 } \rho_b)$$

$$5.06 > 1.98 \text{ (min)}$$

$$\underline{\underline{\text{OK - USE 4-}\#10}}$$

Calculate Req'd Reinf. for $M = 450' \cdot k$

$$R = \frac{450 \times 12}{18 \times 33^2} = .2755 \text{ ksi}$$

$$.2755 < .2953 \Rightarrow \text{under-reinforced}$$

(was over-reinforced by AASHTO)

$$c = \frac{1}{2} f_c \times b ; f_c = \left(\frac{(f_y/n) \times x}{d - x} \right)$$

$$f_c = \frac{24 \text{ ksi}/8 \times x}{33 - x}$$

$$M = c \cdot (d - x/3)$$

$$450 \times 12 = \frac{1}{2} \left(\frac{24 \times x}{33 - x} \right) \cdot 18 \times (33 - \frac{x}{3})$$

$$9x^3 - 891x^2 - 5400x + 178,200 =$$

$$\text{solving by iteration, } \underline{\underline{x = 12.0283"}}$$

$$C = \frac{1}{2} \left(\frac{3 \times 12.0243}{33 - 12.0243} \right) \times 12.0243 \times 18$$

$$= 186.27 \text{ k}$$

$$\text{Req'd } A_s = \frac{186.27}{24 \text{ ksi}} = \underline{7.7611 \text{ in}^2}$$

$$\text{Check: } 7.76 < 12.70 \text{ (75}\rho_b\text{)}$$

$$7.76 > 1.98 \text{ (Min)}$$

OK - USE 5-#11

Check spacing

$$18" - 2 \left(2" + \frac{3}{4}" \right) - 1\frac{1}{2}" = 11"$$

clear stirrup bar

$$\frac{11"}{4 \text{ spa.}} - 1\frac{1}{2}" = 1.25" \text{ clear}$$

OK

SHEAR DESIGN

$$V_c = 1.1 \sqrt{4000} \times 18 \times 33 = 41.32 \text{ k}$$

@ Support

$$V_s = 60.0 - 41.32 = 18.68 \text{ k}$$

$$\text{for } A_v = 2 \times 4 = .40 \text{ in}^2,$$

$$\text{Spa.} = \frac{.40 \times 24 \text{ ksi} \times 33}{18.68 \text{ k}} = 16.96"$$

USE #4 @ 16" ENTIRE BEAM

ACI W.S.D. Beam Design

10
10

SF 04

Set flag 04
for ACI

XEQ "WSBM"

6:40:21 PM 12/09/88

TITLE:
TEST 2 RUN

ACI

F-S 20.24
24.0000 RUN

Fc' KSI
4.0000 RUN

N
8.0000 RUN

B"
18.0000 RUN

D"
33.0000 RUN

H"
36.0000 RUN

1.2Mc=184.4111 'K

200/FY=0.0033

A-S=1.9800 SQ"

.75Pb=0.0214

M, 'K
300.0000 RUN

M/BD2=183.6547

Y=10.1435"

RHO=0.0085

P/Pb=0.2991-OK

A-S=5.0643 SQ"

M, 'K
450.0000 RUN

M/BD2=275.4821

Y=12.0283"

RHO=0.0131

P/Pb=0.4583-OK

A-S=7.7611 SQ"

Design by ACI

Min. A_s figured
for $200/F_y$

Not over-
reinforced by
ACI due to
higher allowable
 f_c

M, 'K

100.0000 RUN

M/BD2=61.2182

Y=6.2049"

RHO=0.0027

P/Pb=0.0955-OK

A-S=1.6165 SQ"

4/3 A-S=2.1553 SQ"

M, 'K
RUN

SHEAR

Vc=41.3246 K

A-V
60.0000 RUN

V, K
RUN

A-V
.4000 RUN

V, K
60.0000 RUN

SPA=16.5000" (MAX)

V, K

$A_s < \text{Min. } A_s$ for
ACI so display
4/3 A_s

Bypass to access
shear design

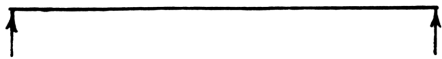
Oops - error

Bypass to return
to A_v prompt

Reinforced Concrete Beam Design

1/5

Design a beam for the following conditions:



Span = 20'-0", simple
DL = 2400 #/ft (heavy industrial)
LL = 3600 #/ft

$f'_c = 4000$ psi
 $f_y = 60$ ksi
 $n = 8$

Moment

$$DL = 2.40 \times 20^2 / 8 = 120.0 \text{ k}$$
$$LL = 3.60 \times 20^2 / 8 = 180.0 \text{ k}$$

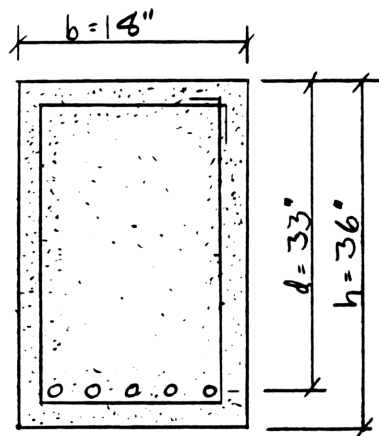
Shear

@ Reaction: $DL = 2.40 \times 20' / 2 = 24.0 \text{ k}$
 $LL = 3.60 \times 20' / 2 = 36.0 \text{ k}$

(Note: take @ reaction since beam depth for "d/2" is unknown)

@ 1/4 Pt.: $DL = 2.40 \times 5' = 12.0 \text{ k}$
 $LL = \frac{7.5'}{20'} \times (15' \times 3.60) = 20.25 \text{ k}$

Try this configuration:



Design by:

- 1) AASHTO W.S.D.
- 2) ACI W.S.D. ("Alternate")
- 3) AASHTO L.F.D.
- 4) ACI L.F.D. ("Strength")

1) and 2) use "WSBM"

3) and 4) use "LFBM"

Calculate the amount of reinforcing necessary for the same loads by AASHTO:

	<u>Serv.</u>	<u>Factored</u>
DL	120'·k	120'·k
LL	180'·k, 5/3	300'·k
		420'·k × 1.3 = <u>546'·k</u>

Note: undercapacity factor $\phi = .9$ not applied. ϕ is applied by LFBM.

$$f'_c = 4000 \text{ psi} \quad b = 18'$$

$$f_y = 60 \text{ ksi} \quad d = 33''$$

Calculate Min. and Max. Steel

Note: L.F.D. Min. and Max. steel are identical to W.S.D. limits. See preceding W.S.D. calculations.

$$1.2 M_{cr} = 184.42'·k$$

$$\text{Min. } A_s = \underline{1.266 \text{ in}^2}$$

$$.75 \rho_b = .0214$$

$$\text{Max. } A_s = \underline{12.70 \text{ in}^2}$$

Calculate Req'd Reinforcing

$$\rho = \frac{1}{m} \left(1 - \sqrt{1 - \frac{2m R_u}{f_y}} \right)$$

$$m = \frac{f_y}{.85 f'_c} = \frac{60}{.85 \times 4} = 17.647$$

$$R_u = \frac{M}{\phi b d^2} = \frac{546'·k \times 12}{.9 \times 18'' \times (33'')^2}$$

$$= .3714 \text{ ksi}$$

$$\rho = \frac{1}{17.647} \left(1 - \sqrt{1 - \frac{2 \times 17.647 \times .37}{60}} \right)$$

$$= .0066$$

$$\text{Req'd } A_s = .0066 \times 18' \times 33''$$

$$= \underline{3.9031 \text{ in}^2}$$

$$\text{check: } 3.90 < 12.70 \text{ in}^2 (.75 \rho_b)$$

$$3.90 > 1.266 \text{ in}^2 (\text{Min.})$$

OK - USE 4-#9

Check spacing

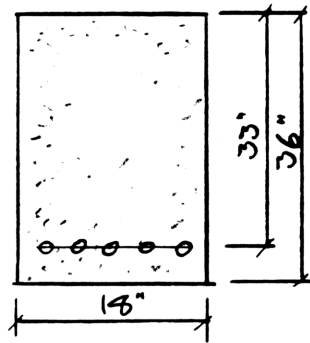
$$18'' - 2(2'' + \frac{3}{4}'') - \frac{\text{say}}{\text{clear stirrup}} \frac{1}{4}'' = 11.25''$$

$$\frac{11.25''}{3 \text{ spa.}} - \frac{1}{4}'' = 2.50'' \text{ clear bar}$$

OK

Note: LFBM does not check bar placement or clearance; fatigue; Z-cracking; or deflection. Program LFAN is somewhat more complete.

Design shear reinforcing
for this beam:



$$f'_c = 4000 \text{ psi}$$

$$f_y = 60 \text{ ksi}$$

$$V_c = 2 \sqrt{4000} \times 18 \times 33 = 75.14 \text{ k}$$

Use #4 stirrups - $A_v = 2 \times 20 = .40 \text{ in}^2$

Max. Spacing

$$A_{v-\min} = \frac{50 b \cdot \text{Spa}}{f_y} \quad (\text{AASHTO 8.19.1.2})$$

$$\therefore \text{Max spa.} = \frac{.40 \times 60,000}{50 \times 18}$$

$$\text{or } \frac{d}{2} = \frac{33}{2} = 16.5" \leftarrow \text{Governs}$$

$$\text{or Max} = 24"$$

@ Reaction

$$DL = 24.0$$

$$LL = \frac{5}{3} \times 36.0 = 60.0$$

$$84.0 \times 1.3 = 109.2 \text{ k}$$

$$V_s = 109.2 / (\phi = .85) - 75.14 = 53.33 \text{ k}$$

$$\text{Spa.} = \frac{A_v f_y d}{V_s} = \frac{.40 \times 60 \times 33}{53.33} = 14.85"$$

@ 1/4 Point

$$DL = 12.0$$

$$LL = \frac{5}{3} \times 20.25 = 33.75$$

$$45.75 \times 1.3 = 59.48$$

$$V_s = \frac{59.48}{(\phi = .85)} - 75.14 \text{ k} < 0$$

\Rightarrow Use Max. Spa. = 16.5"

XEQ "LFBM"

6:43:33 PM 12/09/88

TITLE:

TEST 3

RUN

AASHTO

F-Y 40,60

60.0000 RUN

Fc' KSI

4.0000 RUN

B"

18.0000 RUN

D"

33.0000 RUN

H"

36.0000 RUN

1.2Mc=184.4111 'K

A-S=1.2656 SQ"

200/FY=0.0033

.75Pb=0.0214

M, 'K

546.0000 RUN

M/BD2=371.3907

a=3.8265"

RHO=0.0066

P/Pb=0.2305-OK

A-S=3.9031 SQ"

M, 'K

RUN

SHEAR

Vc=75.1357 K

A-V

.4000 RUN

V, K

109.2000 RUN

SPA=14.8496"

V, K

59.4800 RUN

SPA=16.5000" (MAX)

DZ3

Calculate the amount of reinforcing necessary for the following loads:

	<u>Service</u>	<u>Factored</u>	
DL	$120 \text{ lb/ft} \times 1.4 = 168.0$		(includes beam)
LL	$140 \text{ lb/ft} \times 1.7 = 306.0$		
		<u><u>474.0 lb/ft</u></u>	

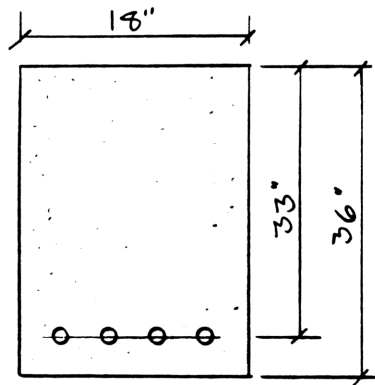
Note: under capacity factor $\phi = 0.9$ not applied. ϕ is applied by LFBM.

$$f'_c = 4000 \text{ psi}$$

$$f_y = 60 \text{ ksi}$$

$$b = 18"$$

$$d = 33"$$



Calculate Min. and Max. Steel

Note: L.F.D. Min. and Max. limits are identical to W.S.D. limits. See preceding W.S.D. calculations.

$$\frac{200}{f_y} = .0033$$

$$\text{Min. } A_s = \underline{1.98 \text{ in}^2}$$

$$.75 \rho_{bal} = .0214$$

$$\text{Max. } A_s = \underline{12.70 \text{ in}^2}$$

Calculate Req'd Reinforcing

$$\rho = \frac{1}{m} \left(1 - \sqrt{1 - \frac{2mR_u}{f_y}} \right)$$

$$m = \frac{f_y}{.85f'_c} = \frac{60}{.85 \times 4} = 17.647$$

$$R_u = \frac{M}{\phi b d^2} = \frac{474 \text{ lb/ft} \times 12}{.9 \times 18 \times 33^2}$$

$$= .3224 \text{ ksi}$$

$$\rho = \frac{1}{17.647} \left(1 - \sqrt{1 - \frac{2 \times 17.647 \times .3224}{60 \text{ ksi}}} \right)$$

$$= .0057$$

$$\text{Req'd } A_s = \rho b d = .0057 \times 18" \times 33" = \underline{3.3596 \text{ in}^2}$$

$$\text{check: } 3.36 \text{ in}^2 < 12.70 \text{ in}^2 \text{ (.75 } \rho_{bal})$$

$$3.36 > 1.98 \text{ in}^2 \text{ (Min)}$$

$$\underline{\text{OK - USE 4-}\#9\text{-}A_s = 4.00 \text{ in}^2}$$

Check Spacing

$$18" - 2 \left(2" + \frac{3}{4}" \right) - \frac{5}{16}" = 11.25" \text{ (say 11.25" clear)}$$

$$\frac{11.25"}{3 \text{ spa}} - 1\frac{1}{4}" = 2.50" \text{ clear}$$

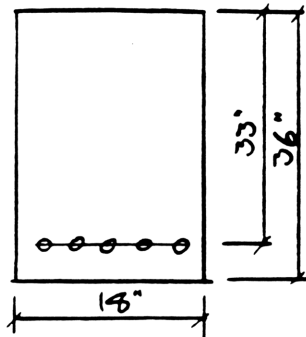
OK

Note: LFBM does not check bar placement or clearance as here. Neither does it check ϵ -cracking deflection. Program LFAN checks the latter two.

ACI L.F.D. Shear Design

5/5

Design shear reinforcing
for this beam:



$$f'_c = 4000 \text{ psi}$$

$$f_y = 60 \text{ ksi}$$

$$V_c = 2 \sqrt{4000} \times 18 \times 33 = 75.14 \text{ k}$$

Use #4 stirrups - $A_v = 2.20 = .40 \text{ in}^2$

Max. Spacing

$$A_{v-\min} = \frac{50 b \cdot s}{f_y} \quad (\text{ACI 11.5.5.3})$$

$$\therefore \text{Max spacing} = \frac{.40 \cdot 60,000}{50 \times 18}$$

$$\text{or } \frac{d}{2} = \frac{33}{2} = 16.5 \text{ " } \leftarrow \text{Controls}$$

$$\text{or Max} = 24"$$

@ Reaction

$$DL = 1.4 \times 24.0 = 33.6$$

$$LL = 1.7 \times 36.0 = 61.2$$

$$94.8 \text{ k}$$

$$V_s = 94.8 / (1.05) - 75.14 = 36.39 \text{ k}$$

$$Spa. = \frac{A_v f_y d}{V_s} = \frac{.40 \cdot 60 \cdot 33}{36.39} = 21.76"$$

Use 16" Spacing Entire Beam

SF 04

XEQ "LFBM"

6:47:04 PM 12/09/88

TITLE:

TEST 4 RUN

ACI

F-Y 40,60

60.0000 RUN

Fc' KSI

4.0000 RUN

B"

18.0000 RUN

D"

33.0000 RUN

H"

36.0000 RUN

1.2Mc=184.4111 'K

200/FY=0.0033

A-S=1.9800 SQ"

.75Pb=0.0214

M, 'K

474.0000 RUN

M/BD2=322.4161

a=3.2937"

RHO=0.0057

P/Pb=0.1984-OK

A-S=3.3596 SQ"

M, 'K

150.0000 RUN

M/BD2=102.0304

a=1.0056"

RHO=0.0017

P/Pb=0.0606-OK

A-S=1.0257 SQ"

4/3 A-S=1.3676 SQ"

M, 'K

RUN

SHEAR

Vc=75.1357 K

A-V

.4000 RUN

V, K

94.8000 RUN

SPA=16.5000" (MAX)

V, K

30.0000 RUN

.5Vc > V

V, K

D25

	X	Y	Z	T	L		X	Y	Z	T	L
01*LBL "WSBM"						53 XROM "BD"	D	E			
02 CF 06						54 X=0?					
03 14						55 GTO D					
04 GTO 03						56 RCL 07					
						57 "H"					
05*LBL "LFBM"						58 PROMPT	H				
06 SF 06						59 STO 07	d				
07 10						60 RCL 04	d ²				
						61 X↑2	b				
08*LBL 03						62 RCL 03					
09 CF 09						63 *	b d ²				$\phi b d^2 \text{ in R.05}$
10 XROM "S"						64 STO 10					
11 GTO 03						65 ST* 05					
						66 FS? 06					
12*LBL "CORBL"						67 GTO 00	.45f' _c				
13 SF 09						68 RCL 12	f' _{cy}				
14 17						69 RCL 13	N	f _{cy}	.45f' _c		$I_{deal} k = \frac{all \cdot f_c}{f'_{cy} + all \cdot f_c}$
15 XROM "W"						70 RCL 09	f _{cy} /n				
						71 /	.45f' _c				
16*LBL 03						72 RCL 12					
17 CF 10						73 +	k				
18 XEQ "W"						74 /	k	k			
19 STO 09	f' _c					75 ENTER↑	k				
20 FS? 06						76 CHS	3	-k	k		
21 GTO 03						77 3	1/3				
22 FC? 04						78 /	1				
23 .4						79 E	j	k			$\bar{j} = 1 - k/3$
24 FS? 04						80 +					
25 .45						81 *					
26 *						82 2	$\frac{1}{2}$				
27 STO 12	.45f' _c					83 /	.45f' _c				
28 8						84 RCL 12					
29 PROMPT	N					85 *	R _{ideal}				$I_{deal} R = \frac{1}{2}(all \cdot f_c) j k$
30 X<> 09	f' _c					86 STO 14					
						87*LBL 00	h				
31*LBL 03						88 RCL 07	d				
32 FC? 09						89 RCL 04	h-d				
33 GTO D						90 -					
34 5						91 FS? 09					
35 /	2f' _c					92 GTO 16	h				
36 .8	Abkai					93 RCL 07	h ²				
37 X>Y?						94 X↑2	b				
38 X<>Y	lower					95 RCL 03	b h ²				
39 STO 08						96 *	p _{min}				
40 .55						97 XROM "Mc"	b				
41 FC? 04						98 RCL 03	p _{min} b				
42 .45	USD factors					99 *	p _{min} b	p _{min} b			
43 FC? 06						100 RCL 04	d				
44 ST* 08						101 XROM "R"	A _{smm}				
45 "MU 1.1.4"						102 ARCL 02					
46 PROMPT						103 FC? 00					
47 STO 16	μ (friction factor)					104 AVIEW					
						105 STO 08	A _{smm}				
48*LBL D						106 SF 14					
49 .9											
50 FS? 09											
51 .85											
52 STO 05	φ										

	X	Y	Z	T	L		X	Y	Z	T	L
107*LBL E						162 R+	p	b			
108 CF 07						163 *					
109 XROM "LL"						164 RCL 04	d				
110 CLA						165 *	A _s				
111 CLX						166 CLA					
112 XROM "M"						167 XROM "AS"					
113 PROMPT	M ^{1-k}	12				168 XEQ "PV"					
114 *	M ^{1-k}					169 RCL 08	Min A _s	A _{6,red}			
115 X=0?						170 X<>Y	A _s	Min.			
116 GTO 37						171 X>Y?					
						172 GTO E					
117*LBL 07	M					173 BEEP					
118 "M/BD2="						174 4					
119 RCL 03	b					175 "4/3"					
120 .9	0					176 3					
121 FC? 06						177 /					
122 SIGN	1					178 *	4/3 A _s				
123 RCL 04	d					179 XROM "AS"					
124 X12	d ²	0	b	M		180 XEQ "PV"					
125 *						181 GTO E					
126 *											
127 /	$\frac{M^{1-k}}{0.6d^2}$	M	M	M		182*LBL 12					
128 E3	1000					183 "NG"					
129 *	$\frac{M^{1-k}}{0.6d^2}$					184 BEEP					
130 ARCL X						185 AVIEW					
131 FC? 00						186 GTO E					
132 FS? 09											
133 X=0?						187*LBL 05					
134 AVIEW	1000					188 RCL 14	R _{ideal}	R _{act}	M ^{1-k}		
135 LASTX	$\frac{M^{1-k}}{0.6d^2}$	M ^{1-k}	M ^{1-k}	M ^{1-k}		189 X>Y?					
136 /						190 GTO 10					
137 FC? 06						191 RCL 2	M				
138 GTO 05						192 RCL 04	d				
139 X<>Y	M ^{1-k}					193 2	Z				
140 XROM "U"						194 /	d/2				
141 STO a	A _s /b _d					195 CHS	-d/2				
142 RCL 04	d					196 RCL 12	.45d'				
143 *						197 RCL 03	b	.45d'	-d/2	M	
144 RCL 00	.45d'					198 *					
145 /	a					199 *	$\frac{A_{SC} \cdot b_d}{2}$				
146 X<> a	p					200 LASTX					
147 "a="						201 6	$\frac{.45d' \cdot b}{6}$	$\frac{-.45d' \cdot b}{2}$	M ^{1-k}		
						202 /					
148*LBL 19						203 XROM "Q"	Y				
149 FS? 09						204 X<>Y					
150 RTN						205 STO a	Y	Y			
151 ARCL a						206 ENTER↑	Y ²				
152 "t--"						207 X12	b				
153 FC? 00						208 RCL 03	n	b	Y ²	Y	
154 AVIEW						209 RCL 09					
155 .1	.1	P				210 /	Y ² b _n	Y	Y	Y	
156 X>Y?						211 *					
157 FS? 07						212 "OVER"					
158 GTO 12						213 RCL 04	d	d	Y ² b _n	Y	
159 RDN	p					214 R+	Y				
160 XEQ "RH"	P/P _{av}	.75	P			215 -	d-Y				
161 RCL 03	b			P		216 X<=0?					
						217 SF 07					

	X	Y	Z	T	L		X	Y	Z	T	L
218 /	$\frac{y^2 b/n}{d-y}$					274 ST+ 10					
219 2	$\frac{y^2 b/n}{d-y}$					275 RND					
220 FC? 09		$\frac{y^2 b/n}{d-y}$	Y	Y		276 RCL 10					
221 FS? 00						277 RND					
222 X=0?						278 X*Y?					
223 AVIEW						279 GTO 11					
224 GTO 08						280 ENTER↑	Y	Y			
						281 X(> a	M ¹⁰	Y			
225+LBL 10	R ₁ R ₂	R ₂	M ¹⁰			282 RCL 04	d	M	Y		
226 6						283 RCL 2	Y	d	M		
227 R↑	M ¹⁰	6				284 3					
228 STO a						285 /	Y/3	d	M	M	
229 *	6M					286 -	j d	M			
230 RCL 09	N					287 /	M/j d				
231 *	6 n M					288 RCL 13	f _{or}				
232 RCL 13	F _{or}										
233 /						289+LBL 08					
234 RCL 03	b					290 /	A _s				
235 /						291 RCL 03	b				
236 STO \	$\frac{6 n M}{b F_{or}}$					292 /					
237 RCL 04	d					293 RCL 04	d				
238 *						294 /	p				
239 STO]						295 "Y="					
240 LASTX	d					296 GTO 19					
241 3											
242 *	3d					297+LBL 16					
243 STO [298 STO 02	k-d				
244 9						299 RCL 01	F _y				
245 /	d/3					300 .85					
246 STO 10						301 *	.85 F _y				
247 E						302 FC? 06					
248 %						303 RCL 13	F _s				
249 STO ↑	d/300	d/3	$\frac{6 n M}{b F_{or}}$			304 STO 17	F _s				
250 CLST						305 "ARM"					
						306 PROMPT					
251+LBL 11						307 STO 15	Arm				
252 RCL 2						308 RCL 04	d				
253 STO 11						309 X*Y?					
254 RCL 10	Y?					310 SF 14					
255 ENTER↑						311 X*Y?					
256 ENTER↑						312 GTO 34					
257 ENTER↑						313 "ARM > D"					
258 RCL [A					314 BEEP					
259 -						315 PROMPT					
260 *						316 GTO D					
261 RCL \	B										
262 -						317+LBL 34					
263 *						318 XROM "L"					
264 RCL]	C					319 CLX					
265 +						320 "V, K"					
266 ST* ↑						321 PROMPT					
267 ST- 11						322 X=0?					
268 RCL ↑						323 GTO D					
269 RCL 11						324 STO 06	V				
270 X=0?						325 RCL 08	allow. f _y				
271 /						326 RCL 03	b				
272 STO ↑						327 *					
273 X(> 10						328 RCL 04					

$$A_s = \frac{M}{f_{or} j d}$$

$$x^2 - Ax^2 - Bx + C = 0$$

DZ9

	X	Y	Z	T	L		X	Y	Z	T	L
329 *	V _c	V									
330 "V>V _c "											
331 X<=Y?						387 X<=Y?					
332 BEEP						388 X<>Y	Greater	Lower			
333 X<=Y?						389 LASTX	A _n				
334 AVIEW						390 X<>Y	Greater	A _n			
335 X<=Y?						391 RCL 00	.856 _{c/f}				
336 GTO D						392 21.25	21.25	.856 _{c/f}	Greater	A _n	
337 CLX						393 /	.846 _{c/f}	Greater	A _n	A _n	
338 "TENS N _c , K"						394 "RHO-MIN="					
339 PROMPT						395 ARCL X					
340 STO 07	N _c					396 FC? 00					
341 "N _c >V"						397 AVIEW					
342 RCL 06	V	N _c				398 RCL 03	b				
343 X<=Y?						399 *					
344 BEEP						400 RCL 04	d				
345 X<=Y?						401 *	A _{omin}	A _s	A _n		
346 AVIEW						402 X>Y?					
347 X<=Y?						403 X<>Y	lower	greater			
348 GTO 34						404 RDN	greater	A _n			
349 5						405 ADV					
350 /	V/5	N _c				406 "PRIM -					
351 X>Y?						407 XROM "AS"					
352 STO 07						408 XEQ "PV"					
353 RCL 06	V					409 -	A ₁ -A ₂				
354 RCL 17	F _{or}					410 CHS	A ₂ -A ₁				
355 /						411 2					
356 RCL 16	u					412 /	1/2(A ₂ -A ₁)				
357 /						413 "SEC -					
358 "A-V="						414 XROM "AS"					
359 XEQ 01						415 XEQ "PV"					
360 2						416 GTO 34					
361 *						417 LBL 37					
362 3						418 RCL 02	.75P _b				
363 /	2/3A _v					419 STO 10					
364 X<> 06	V					420 RCL 03					
365 RCL 15	A _{rom}					421 XEQ "VS"					
366 *	M _u ard					422 RCL 10					
367 RCL 07	N _c					423 STO 02					
368 RCL 02	hi-d					424 GTO D					
369 *	M _u ard					425 LBL 01					
370 +	M					426 ARCL X					
371 XEQ 07	P _m					427 XROM "IS"					
372 RCL 03	b					428 FC? 00					
373 *						429 XEQ "PV"					
374 RCL 04	d					430 END					
375 *	A _f										
376 "A-F="											
377 XEQ 01											
378 RCL 07	N _c										
379 RCL 17	F _s										
380 /	A _n	A _f									
381 "A-N="											
382 XEQ 01											
383 +	A ₁ -A ₂										
384 RCL 06	2/3A _v										
385 LASTX	A _n										
386 +	2/3A _v A _n										

LOAD FACTOR (ULTIMATE STRENGTH) CONCRETE BEAM ANALYSIS

xeq LFAN (SIZE 029)

SOFTKEYS (USER mode ON)

A All new reinforcing input
C Correct previous layer of reinforcing input
D New beam dimensions
E New acting moment

USER FLAGS

Flag 00	clear set	Full output Only moment capacity and reinforcing limitation warnings
Flag 02	clear set	Warning BEEP disabled Warning BEEP enabled
Flag 04	clear set	AASHTO ACI

INPUT SUMMARY

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
AASHTO or ACI	A reminder, not a prompt. Press R/S to continue if correct.	
F-Y 40,60	Steel yield stress in ksi	60 ksi
Fc' KSI	Concrete <u>ultimate</u> strength	4.0 ksi
FL W"	Top flange width in inches	No default
WEB W"	Web width in inches. Bypass for a rectangular beam. WEB W" is greater than FL W" for an invert-T beam.	FL W"
FL TH"	Top flange thickness in inches. Only prompted if WEB W" is different than FL W".	No default
H"	Total depth of the member (full concrete depth, not depth to reinforcing), in inches	No default

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
Z 130,170 (AASHTO)	Allowable Z-cracking constant	130
Z 145,175 (ACI)	in kips per inch	
LF M,'K	Factored moment in foot-kips, <u>always positive</u> . Note: the program applies the under-capacity factor $\phi = 0.90$.	No default
SERV M,'K	Service moment in foot-kips, <u>always positive</u>	No default
MIN FATG M,'K (AASHTO only)	Minimum moment for fatigue considerations (<u>Service DL + LL</u>). If reversal takes place, this is negative. Bypass or input zero to skip fatigue calculations.	Zero
A-S1	Area of first layer of tension reinforcing in square inches. Bypass to access shear design.	Shear design
D"	Depth to reinforcing, in inches	No default
A-S2, A-S3, etc.	Area of second (third, etc.) layer of tension reinforcing. Bypass (or enter zero) to end reinforcing input. Layers need be in no particular order.	Zero
MAX BAR #	Size of the largest tension bar; e.g. #8 for a 1" diameter bar	No default
A-S'	Area of compression reinforcing.	Zero
D-TOP"	Depth to compression steel	No default

OUTPUT SUMMARY (Items in parenthesis are displayed only if flag
00 is clear)

(LF: EST A-S)	Estimated reinforcing necessary for the load factor moment (LF M) input
(WS: EST A-S)	Estimated reinforcing needed for the service moment (SERV M) input
(RHO)	The reinforcing ratio
P/Pb	Compares the actual reinforcing ratio to the "balanced" ratio.
AS<MIN	Warns that the area of steel is less than the calculated minimum. Design is still OK if the area of steel is greater than $\frac{4}{3} A_{s-req}$.

(Ie)	The "effective" moment of inertia based on service moment acting
(C)	The neutral axis location, measured from the top down
LF M	The "nominal" ultimate strength of the member, $O M_u$, in foot-kips.
SERV M	The allowable service moment on the member <u>based on Z-cracking considerations</u> , in ft-kips
(Z STRS)	The allowable service reinforcing stress in ksi, used to figure SERV M
(ACT STRS)	The acting service reinforcing stress in ksi
(Z)	The acting Z value
(ALL FAT STRS)	The allowable fatigue stress on the reinforcing, in ksi
(ACT FAT STRS)	The actual fatigue stress on the reinforcing
(OK or NG)	Approval or disapproval of the design, based on ultimate and service strength and fatigue. Does not look at maximum or minimum steel areas or constructibility of the design.

PROGRAM FLAGS

Flag 05	clear set	True T-beam, based on neutral axis depth Rectangular, or neutral axis in flange
Flag 06	clear set	Ic (WSD) LFAN (LFD)
Flag 07	clear set	OK NG
Flag 08	clear set	T-beam, based on configuration Rectangular beam
Flag 09	clear set	do Z-cracking calculation don't do Z-cracking calculation
Flag 10	clear	All concrete properties in Fc

STORAGE REGISTER USE

<u>Register</u>	<u>Value</u>
00	$.85 F_c' / F_y$
01	F_y , ksi

<u>Register</u>	<u>Value</u>
02	.75 P_{bal} for T-beam
03	Web width, inches
04	Maximum reinforcing depth
05	ϕbd^2
06	$\sqrt{F_c'}/253$ for 1.2 M_{cr} calculation
07	Flange width
08	Present A_{st} ; D' for compr. steel
09	N
10	Flange thickness
11	H (full beam depth)
12	Y (neutral axis position) from " I_c "
13	Z
14	H - 3" for reinforcing estimates; $\sum A_{st}$
15	"c" for .75 P_{bal}
16	S_{st} (reinforcing section modulus)
17	I_{cr}
18	B_1 (β_{a1})
19	Min A_{st}
20	counter; A_s'
21	$\sum A_{st} \times D$
22	M_{cr} , inch-kips
23	I_g (gross moment of inertia of the concrete alone)
24	.75 P_{bal} for rectangular beams
25	SERV M (inch-kips stored)
26	LF M (inch-kips stored)
27	MAX BAR #
28	M_{min} for fatigue

LOAD FACTOR (ULTIMATE STRENGTH) CONCRETE BEAM ANALYSIS

Program LFAN analyzes reinforced beams by ultimate strength methods, calculating the cracked-section moment capacity of a given beam configuration for different reinforcing options. It accepts rectangular, T, or invert-T beams, with or without compression steel (one layer only), and multiple layers of tension steel. It is not able to distinguish bars input as tension steel which are above the neutral axis, so don't include too many "side bars" as tension reinforcing. The program also calculates the gross moment of inertia (I_g) of the concrete section, and the "effective" moment of inertia (I_e) as defined by AASHTO 8.13.3 and ACI 9.5.2.3, using the service moment input.

AASHTO/ACI MINIMUM A_s

Like most of the structural programs, LFAN follows AASHTO if flag 04 is clear and ACI if it is set. The governing code is displayed at the beginning of each run. If it is the desired code, press R/S to continue. (If using a printer, you won't need to press R/S--it continues automatically.) If designing by the other code, change the status of flag 04 and start over. AASHTO calculates the minimum area of steel by means of a cracking-moment concept rather than with $200/F_y$ as ACI does. If designing by ACI, the $200/F_y$ ratio is based on average depth to reinforcing for multiple layers of reinforcing, and on the web width for T-beams. If working with AASHTO, on the other hand, the minimum reinforcing is that necessary for a "cracking moment" based on the concrete gross moment of inertia, assuming a homogenous material with a strength of $7.5\sqrt{f'_c}$ (the "modulus of rupture" of concrete per AASHTO 8.15.2.1). This reinforcing is calculated by ultimate strength methods, using an effective depth to reinforcing 3" less than the full concrete depth of the member, and is the same for both concrete flexural analysis programs, Ic and LFAN. AASHTO is not very clear in its present form that ultimate strength methods are to be used for all designs, WSD as well as LFD; however, the 1982 Interim Specifications, where it was introduced, spell it out a little better. Besides, it doesn't make sense to have a different minimum for WSD than for LFD. If the actual reinforcing is less than the minimum, the program displays AS<MIN; if the area of steel is greater than 4/3 of what is needed, it still meets code requirements.

MAXIMUM REINFORCING

LFAN also checks the reinforcing against a maximum of $0.75P_b$ (75% of the "balanced" reinforcing ratio based on ultimate-strength concepts) as required by both AASHTO and ACI. This check does take compression reinforcing into account, as well as the effect of T-beam flanges.

MOMENT CAPACITY

The moment capacity is found by calculating the moment causing full allowable stress in the tension steel only, as opposed to program Ic, which checks concrete stresses as well as tension and compression steel stresses. This is because, in ultimate strength design, concrete stresses will not exceed the allowable if the reinforcing ratio is held below $.75P_b$. If the reinforcing ratio goes above this maximum, the program displays a warning.

The output moment capacity (LF M=...) includes the undercapacity factor $\Phi = 0.90$.

At the end of a run, the program displays OK or NG (no good) in reference to the beam configuration just run. This reflects ultimate and fatigue capacity and service load (Z-cracking) capacity. It does not refer to maximum or minimum steel checks or deflection. In other words, a beam with too much reinforcing will effect a warning AS>MAX during the run but may well display OK at the end. This means it has adequate ultimate, cracking, and fatigue strengths; however, the beam is not acceptable.

REINFORCING ESTIMATES

After displaying the maximum and minimum steel calculations, LFAN prompts for the acting load factor and service moments, as well as minimum moment for fatigue. Based on the load factor and service moments, it estimates the amount of reinforcing required for each; these two quantities usually bracket the actual reinforcing required. If fatigue is a consideration, or if the Z-cracking requirement is strict ($Z = 130$ or 145), the actual steel required will be closer to the WSD estimate. If the Z required is higher (170 for AASHTO or 175 for ACI for "moderate" exposure), the area of steel required will be closer to the ultimate requirement. Using multiple layers of steel, of course, raises the required area since the reinforcing is being used less efficiently. The program bases these estimates on the following three assumptions:

- 1) One layer of reinforcing
- 2) Effective depth $D = H - 3"$

- 3) For a T-beam, the neutral axis is within the flange.

The LFD estimate is a true design, given these assumptions. The WSD estimate uses these rules of thumb (from the ACI Working Stress Handbook) where M is in footkips and D is in inches:

$$A_s = M / (1.76 D) \text{ for } f_s = 24 \text{ ksi (Grade 60)}$$
$$A_s = M / (1.24 D) \text{ for } f_s = 20 \text{ ksi (Grade 40)}$$

These two areas of reinforcing are, of course, only estimates, but are often useful as a place to start or to check if a given beam configuration is reasonable for the load.

FATIGUE AND CRACKING

LFAN calculates the elastic (cracked) section properties of the given beam configuration in order to evaluate service (Z-cracking) and, under AASHTO (flag 04 clear), fatigue properties. These calculations are done by the working portion of program Ic. The section properties are not displayed by LFAN; however, they are stored and may be recalled by the user. The program checks reinforcing fatigue only; concrete fatigue ($0.5 f_c'$) is not checked. According to ACI 343, no failures have ever been attributed to fatigue of concrete; and it never seems to govern. However, if felt necessary, this should be checked by the designer.

Fatigue calculations involving stress reversal are complicated in concrete analysis, since it is not correct to simply use the same cracked-section properties with signs reversed. Any stress calculations on tension steel under stress reversal usually involve assumptions concerning the opposite layer of reinforcing (formerly compression steel, now in tension). LFAN cavalierly makes a number of such assumptions, all based on concrete bridge slab design:

- 1) There is only a single layer of steel on each side.
- 2) Whether or not compression steel was input, the "top" steel (normally in compression, now in tension) is assumed to be twice the area of the steel being investigated for fatigue. This is predicated on the assumption of equal top and bottom mats (which is approximately true for continuous slab design) where the mat under investigation has had half its bars cut off (since most moment reversal problems happen at bar cutoffs).

3) Under moment reversal, the new "cracked" properties are not calculated precisely. The program assumes that $k' = 0.3$ and $j' = 0.9$ (where $j' = 1 - k'/3$).

Since the compressive force due to moment reversal is usually very low (on the order of 2 ksi or less) these assumptions are close enough. However, they really have no merit except in continuous slab design, so use negative MIN FATG M's with caution in other designs.

The program also assumes the reinforcing deformation ratio r/h is 0.8, meaning the maximum allowable fatigue stress is calculated as $21.0 + 3(r/h) - (f_{\min})/3$, or 23.4 ksi - $(f_{\min})/3$. Under moment reversal, f_{\min} is negative, thus raising the allowable fatigue stress.

USER FLAGS

The program displays full output (including limit checks, the neutral axis position, I_g , I_e , and fatigue and Z-cracking calculations) if flag 00 is clear. If flag 00 is set, the program displays only the ultimate and service moment capacity. The limit checks (minimum and maximum areas of steel) and their warnings are not disabled by setting flag 00; only the display of the actual minimums and maximums is suppressed. When flag 00 is set, some quantities (the moments) are not prompted to be input. Thus it works to set flag 00 in the middle of a run, but clearing it will cause problems.

All the programs have warning BEEPS for material property limits and some errors, which sound if flag 02 is set; however, LFAN does not have BEEPs for infractions concerning minimum steel areas, only displayed warnings. The program is unaffected by flag 01 or 03. As mentioned, flag 04 determines whether AASHTO (clear) or ACI (set) is followed.

SOFTKEYS

Softkeys are used extensively by LFAN for modification or correction of input. To access the beam dimension prompts, press softkey D (in USER mode); to input new moments, use softkey E (external loads, maybe?). Softkey C (correct) returns the program to the previous reinforcing prompt (for multiple layers of tension steel) while still in the tension steel input section. Softkey A, on the other hand, starts the reinforcing input from the beginning (A-S1, etc.) and may be used any time the program pointer is actually within LFAN; i.e., any time except while in the shear design subroutine (see next paragraph) or the initialization or material prompting subroutines at the beginning of the program.

SHEAR DESIGN

All four of the flexural programs access a shear reinforcing design subroutine, VS. This is a true design program, not an analysis program. LFAN is diverted to shear design by bypassing the A-S1 prompt. Unlike most other "defaults", the prompt must be bypassed; it doesn't work to input zero; and it must be the AS-1 prompt. Control returns to the flexural program by bypassing the A-V prompt in the shear design subroutine. To use the shear design subroutine, the designer must go through the flexural program at least once, since VS uses the beam width and reinforcing depth from the calling program. VS also uses the Flag 06 setting from the calling program to determine whether to design in WSD or LFD. In program LFAN, softkey A can be used to get to the A-S1 prompt, and thence to shear design, as long as at least one flexural analysis has been done; VS will use the beam configuration last run. Note that shear design is by the simplified equations of both codes and does not include the effect of moment or external tension or compression; this was done both to save programming space and because most designers usually use only the simplified methods anyway. For more information on this subroutine, see the documentation for VST, of which VS is a part.

T-BEAMS

The analysis of T-beams has a few peculiarities worth mentioning. First of all, the derivation of $.75P_{ba}$ is based on the actual beam configuration. The neutral axis position is calculated for $.75P_b$, and if it is found to be below the flange then everything is recalculated. This is shown in the T-beam example, to which reference should be made.

Second, rather than use the distance "c" to calculate the position of the neutral axis, LFAN uses the dimension "a", where "a", the compression block depth, is $B_1 \times c$. On most sections, this makes no difference. However, on those sections where "c" is slightly greater than the flange thickness and "a" is slightly less, the usual analysis has the flanges participating their full depth while the web is only effective to depth "a." In this situation, the program will have a different "a" somewhere between the two, and a slightly greater neutral-axis position "c." The difference is minimal practically but philosophically makes more sense. The "true" neutral axis position (i.e. that calculated by normal means) is still used to check compression steel yield and the value for $.75P_b$.

Invert T-beams are easily handled. In this case the FL W" (flange width) is less than the web thickness;

the flange is defined as always being in compression. The lesser of the two dimensions (flange or web width) is always used as the width for shear design.

ALLOWABLE STRESSES

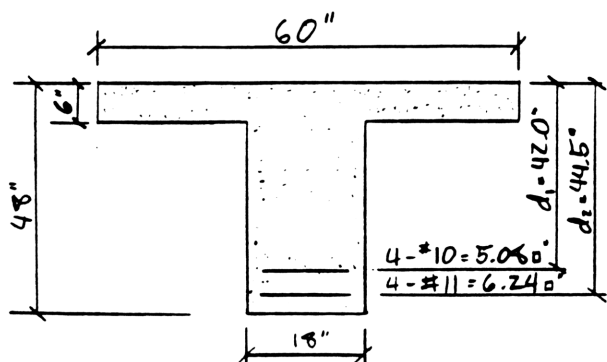
Program LFAN also has an idiosyncrasy related to the value of N , the modular ratio, which is needed by the working-stress portion of the program but which is not input. LFAN calculates the modular ratio as $N = 12 - f_c'$ (in ksi), which is correct for the normal range of concrete strengths used. Note that using a concrete strength, for instance, of 3500 psi, results in an N value of 8.5; normal hand calculations would use 9. This may result in some discrepancies when checking hand calculations.

REINFORCING STRESSES under ULTIMATE LOAD

Tension steel yield is never checked directly since, as mentioned, yield must occur if the reinforcing ratio is less than P_b . However, this indirect check applies at the average depth to reinforcing; in beams with more than one layer of reinforcing, bars above this which may not be yielding are not checked. Practically, the limitation of 75% of P_b ensures yielding in most reasonable designs. However, bars placed as "side bars" in deeper beams may not yield. If deemed desirable to include these bars in an analysis, the designer must check that they reach yield strain; the neutral-axis depth is displayed for this purpose. If found not to yield, they should be left out, or their area factored down by the ratio of their actual stress to the yield stress.

Compression steel yield, on the other hand, is checked. If this steel is found not to yield, the program recalculates using a factored-down area as described in the preceding paragraph. This, of course, is an approximation, but the error is typically less than 1% and is conservative. Concrete displaced by compression steel is accounted for. By way of comparison, neglecting to do this is non-conservative, by about 10%.

Check the T-beam for the loads given by AASHTO W.S.D. criteria.



$$f'_c = 4.0 \text{ ksi}$$

$$n = 8$$

$$f_s = 24 \text{ ksi (Grade 60)}$$

$$\text{Service DL} = 700 \text{ 'k}$$

$$\text{Service LL} = \frac{300 \text{ 'k}}{1000 \text{ 'k}}$$

Concrete

	A	d	Ad	\bar{x}	$A\bar{x}^2$	I_o
fl.	360	3"	1080	16.26	95,157	1080
web	756	27"	20,412	7.74"	45,313	111,132
	1116		21,492		140,470	112,212
					252,682 in ⁴	

$$\bar{x} = \frac{21,492}{1116} = 19.26"$$

$$S_{\text{bottom}} = \frac{252,682}{48 - 19.26} = 8791.4$$

$$f_r = 7.5 \sqrt{f'_c} = 7.5 \sqrt{4000} = 474.34 \text{ psi}$$

$$1.2 M_{cr} = 1.2 \times \frac{8791.4 \text{ in}^3 \times 474.34 \text{ psi}}{12,000} = 417.01 \text{ 'k}$$

$$= 5004.15 \text{ 'k}$$

Calculate min. reinf. by ultimate str. methods:

$$\rho = \frac{1}{m} (1 - \sqrt{1 - 2m R_u / f_y}); \quad m = \frac{f_y}{A_s f'_c} = 17.65$$

$$R_u = \frac{M}{\phi b d^2} = \frac{5,004.15 \text{ 'k}}{0.9 \times 60 \times 45} = .0458$$

Assumed

$$\text{Thus, } \rho_{\min} = \frac{1}{17.65} (1 - \sqrt{1 - 2m R_u / 60})$$

$$= .00077$$

$$A_{s,\min} = .00077 \times 60 \times 45$$

$$= 2.07 \text{ in}^2 \text{ - OK}$$

$$\text{Or, by ACI, } \rho_{\min} = \frac{200}{F_y} = .0033$$

$$A_{s,\min} = .0033 \times 18 \times 45$$

$$= 2.70 \text{ in}^2 \text{ - OK}$$

(much simpler)

$$\text{Check Max. } A_s = .75 \rho_{\text{bal}}$$

Assume N.A. in flange @ max.:

$$.75 \rho_b = .75 \left(\frac{.85 \beta_1 f'_c}{f_y} \times \frac{87}{87 + f_y} \right) = .0214$$

$$A_{s,\max?} = .0214 \times 60 \times 45 = 57.7 \text{ in}^2$$

$$T = 57.7 \text{ in}^2 \times 60 \text{ ksi} = 3463 \text{ k}$$

$$C = 60 \times 6 \times (.85 \times 4) \times \beta_1 = 1040.4 \text{ k}$$

\therefore N.A. is in web @ .75 ρ_{bal}

Then, flange \rightarrow equiv. A_s :

$$(60 - 18) \times 6 \times \frac{(.85 \times 40 \text{ ksi})}{60 \text{ ksi}} = 14.28 \text{ in}^2$$

$$\text{Web: } \rho_b = \frac{.0214}{.75}; 18 \times 45 = 23.09 \text{ in}^2$$

$$\Sigma = 37.37 \text{ in}^2$$

$$.75 A_{s-\text{bal}} = .75 \times 37.37 = 28.03 \text{ in}^2$$

$$.75 \rho_{\text{bal}} = \frac{28.03}{60 \times 45} = .0104$$

(relative to flange width)

$$A_s = (5.06 + 6.24) = 11.32 < 28.03$$

OK

(D41)

Section Properties

$$60" \times 6" \cdot (6/2) = 1080 \text{ in}^3$$

$$(n=8) \cdot (5.06" \cdot 36" + 6.24" \cdot 38.5")$$

$$= 3385 \text{ in}^3 > 1080$$

\Rightarrow N.A. in web @ service

$$\therefore (60-18) \cdot 6" \cdot (X - \frac{6"}{2}) + 18 \cdot X \cdot \frac{X}{2}$$

$$= (n=8) (5.06 \cdot (42-X) + 6.24 (44.5-X))$$

$$252.0 X - 756 + 9X^2 = (213.36 + 277.68 - 11.32X) \cdot 8$$

$$X = \underline{10.6785"} \quad (\text{Quadratic})$$

	Area	d	Ad ²	I _o
fl. (60-18) \cdot 6	= 252.0	7.68"	1934.98	756.0
web 18" \cdot 10.6785	= 192.21	5.34"	5479.59	1826.53

A _{s1} 5.06 \cdot 8	= 40.64	31.32"	39,869.22	—
A _{s2} 6.24 \cdot 8	= 49.92	33.82"	57,103.06	—
			117,309.77	2582.53

$$I_{cr} = 119,892.30 \text{ in}^4$$

$$\text{Concrete } S_e = \frac{I}{X} = 11,227.41 \text{ in}^3$$

$$M_c = \frac{(40.4 \text{ ksi}) \cdot 11,227.41}{12} = 1497.0 \text{ 'k}$$

$$\text{Steel } S_{st} = \frac{119,892.30}{(44.5 - X) \cdot (n=8)} = 443.11 \text{ in}^3$$

$$M_{st} = \frac{24 \text{ ksi} \cdot 443.11}{12} = 886.21 \text{ 'k} \quad \leftarrow \text{Governs}$$

$$886.21 < 1000.0 \Rightarrow \underline{\underline{NG}}$$

Section is inadequate by W.S.D.

XEQ "Ic"

8:30:24 AM 02/17/89

TITLE:

T-BEAM

RUN

AASHTO

F-S 20.24

24.0000

RUN

Fc' KSI

4.0000

RUN

N

8.0000

RUN

FL W"

60.0000

RUN

WEB W"

18.0000

RUN

FL TH"

6.0000

RUN

H"

48.0000

RUN

I-G=252,681.6777

1.2Mc=416.9831 'K

A-S=2.0732 SQ"

200/FY=0.0033

.75Pb=0.0104

SERV M. 'K

1,000.0000

RUN

MIN FATG M. 'K

RUN

MS: EST A-S=12.5549 SQ"

A-S1

5.0000

RUN

D"

42.0000

RUN

A-S2

6.2400

RUN

D"

44.5000

RUN

A-S3

RUN

A-S'

RUN

RHO=0.0043

Ic=119,892.2980

Y=10.6785"

Sc=11,227.4086

S-ST=443.1073

ST M=886.2146 'K

Ie=125,463.8348

NG

A-S1

Check the same design for adequacy by AASHTO L.F.D criteria.

$$f'_c = 4.0 \text{ ksi}$$

$$F_y = 60 \text{ ksi}$$

$$Z = 130 \frac{\text{in}^3}{\text{ft}} \text{ (severe exposure)}$$

$$DL = 700$$

$$LL = \frac{5}{8} \times 300' = \frac{500}{1200 \times 1.3} = 1560' \cdot \text{ft}$$

Strength

$$(5.04 + 6.24) \times 60 \text{ ksi} \leq 60'' \times (0.85 \times 4.0 \text{ ksi}) \times a$$

$$a = 3.33''$$

$$c = \frac{a}{\beta_1 = 0.85} = 3.92'' < 6'' \Rightarrow \text{N.A. in flange}$$

$$\begin{aligned} M &= (\phi = 0.9) (5.04 \times (42 - \frac{3.33}{2}) \\ &\quad + 6.24 \times (44.5 - \frac{3.33}{2})) \times \frac{60 \text{ ksi}}{12} \\ &= \underline{2124.88' \cdot \text{ft}} > 1560' \cdot \text{ft} \quad \underline{\text{OK}} \end{aligned}$$

Cracking - $Z = 130$; Max. bar = #11 = 1.56"

$$d = \frac{5.04 \times 42'' + 6.24 \times 44.5''}{11.32 \text{ in}^2} = 43.36''$$

$$\text{equiv. number of \#11} = \frac{11.32}{1.56} = 7.256 - \#11 \text{ bars}$$

$$A_c = Z \times (48 - 43.36'') \times 18'' / 7.256 \text{ bars} = 22.93 \text{ in}^2$$

$$d_c = 48'' - 44.5'' = 3.5''$$

$$\text{Allow. } f_s = \frac{Z = 130}{\sqrt{3.5'' \times 22.93}} = 30.14 \text{ ksi}$$

$$\text{Act. } f_s = \frac{1000.0' \cdot \text{ft} \times 12}{S = 443.11} = 27.08 \text{ ksi} < 30.14 \quad \underline{\text{OK}}$$

$$\text{Allow. serv. } M = \frac{30.14}{27.08} \times 1000' \cdot \text{ft} = \underline{1112.88' \cdot \text{ft}}$$

$$\text{Act. } Z = \frac{27.08}{30.14} \times 130 = 116.81 \text{ in}^3.$$

TITLE:
T-BEAM RUN

AASHTO
F-Y 40,60
60.0000 RUN
Fc' KSI 4.0000 RUN
FL M" 60.0000 RUN
WEB M" 18.0000 RUN
FL TH" 6.0000 RUN
H" 48.0000 RUN
I-G=252,681.6777
1.2Mc=416.9831 'K
A-S=2.0732 SQ"
200/FY=0.0033
.75Pb=0.0104

Z 130.170
130.0000 RUN
LF M, 'K 1,560.0000 RUN

SERV M, 'K 1,000.0000 RUN
MIN FATG M, 'K RUN
LF: EST A-S=7.9081 SQ"
WS: EST A-S=12.5549 SQ"

A-S1 5.0000 RUN
D" 42.0000 RUN
A-S2 6.2400 RUN
D" 44.5000 RUN
A-S3 RUN
MAX BAR # 11.0000 RUN
A-S' RUN
RHO=0.0043
Ie=125,463.8348
C=3.9170"

LF M=2,124.8799 'K
SERV M=1,112.8783 'K

Z STRS=30.1384 KSI
ACT STRS=27.0815 KSI
Z=116.8142

OK

043

WORKING STRESS CONCRETE BEAM ANALYSIS

xeq Ic (SIZE 029)

SOFTKEYS (USER mode ON)

- A All new reinforcing input
- C Correct previous layer of reinforcing input
- D New beam dimensions
- E New acting moment

USER FLAGS

Flag 00	clear set	Full output Only moment capacity and reinforcing limitation warnings
Flag 02	clear set	Warning BEEP disabled Warning BEEP enabled
Flag 04	clear set	AASHTO ACI

INPUT SUMMARY

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
AASHTO or ACI	A reminder, not a prompt. Press R/S to continue if not using a printer.	
F-S 20,24	Steel working stress in ksi	24 ksi
Fc' KSI	Concrete <u>ultimate</u> strength	4.0 ksi
N	Modular ratio, E_{st}/E_{conc}	8
FL W"	Top flange width in inches	No default
WEB W"	Web width in inches. Bypass for a rectangular beam. WEB W" is greater than FL W" for an invert-T beam.	FL W"
FL TH"	Top flange thickness in inches. Only prompted if WEB W" is different than FL W".	No default
H"	Total depth of the member (full concrete depth, not depth to reinforcing), in inches	No default

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
SERV M, 'K	Service moment in foot-kips, <u>always positive</u>	No default
MIN FATG M, 'K (AASHTO only)	Minimum moment for fatigue considerations (<u>Service DL + LL</u>). If reversal takes place, this is negative. Bypass or input zero to skip fatigue calculations.	Zero
A-S1	Area of first layer of tension reinforcing in square inches. Bypass to access shear design.	Shear design
D"	Depth to A-S1, in inches	No default
A-S2, A-S3, etc.	Area of second (third, etc.) layer of tension reinforcing. Bypass (or enter zero) to end reinforcing input. Layers need be in no particular order.	Zero
D"	Depth to A-S2, etc.	No default
A-S'	Area of compression reinforcing. Bypass for no compression reinf.	Zero
D-TOP"	Depth to compression steel	No default

OUTPUT SUMMARY (Items in parenthesis are displayed only if flag
00 is clear)

(WS: EST A-S)	Estimated reinforcing necessary for the design moment (SERV M) input
(RHO)	The reinforcing ratio
(Ic)	Cracked-section moment of inertia
(Y)	Neutral axis position measured from the compression side
(Sc)	Section modulus of the concrete
(S-ST)	Section modulus of the deepest layer of tension steel
ST M or CONC M	The service moment capacity, governed by steel or concrete, respectively
(Ie)	The "effective" moment of inertia
(ALL FAT STRS)	The allowable fatigue stress on the reinfor- cing, in ksi
(ACT FAT STRS)	The actual fatigue stress on the reinforcing

(OK or NG)

Approval or disapproval of the design, based on ultimate and service strength and fatigue. Does not look at maximum or minimum steel areas or constructibility of the design.

PROGRAM FLAGS

Flag 05	clear set	True T-beam Rectangular, or neutral axis in flange
Flag 06	clear set	Ic (WSD) LFAN (LFD)
Flag 07	clear set	OK NG
Flag 08	clear set	T-beam Rectangular beam
Flag 09	clear set	do Z-cracking calculation don't do Z-cracking calculation
Flag 10	clear	All concrete properties in Fc

STORAGE REGISTER USE

<u>Register</u>	<u>Value</u>
00	$.85 F_c' / F_y$
01	F_y , ksi
02	$.75 P_{bal}$ for T-beam
03	Web width, inches
04	Maximum reinforcing depth
05	ϕbd^2
06	$\sqrt{F_c' / 253}$ for $1.2 M_{cr}$ calculation
07	Flange width
08	Present A_{st} ; D' for compr. steel
09	N
10	Flange thickness
11	H (full beam depth)

<u>Register</u>	<u>Value</u>
12	Y (neutral axis position) from Ic
13	Steel working stress, f_{st}
14	H - 3" for reinforcing estimates; $\sum A_{st}$
15	"c" for $.75P_{bal}$
16	S_{st} (reinforcing section modulus)
17	I_{cr}
18	β_1 (β_{a1})
19	Min A_{st}
20	counter; A_s'
21	$\sum A_{st} \times D$
22	M_{cr} , inch-kips
23	I_g (gross moment of inertia of the concrete alone)
24	$.75 P_{bal}$ for rect. beams
25	SERV M (inch-kips stored)
26	not used
27	MAX BAR #
28	M_{min} for fatigue

WORKING STRESS CONCRETE BEAM ANALYSIS

Program Ic analyzes reinforced beams by working-stress methods, calculating the cracked-section properties and moment capacity of a given beam configuration for different reinforcing options. It accepts rectangular, T, or invert-T beams, with or without compression steel (one layer only), and multiple layers of tension steel. It is not able to handle bars input as tension steel which are above the neutral axis, so don't get greedy and include too many "side bars" as tension reinforcing. The program also calculates the gross moment of inertia (I_g) of the concrete section, and the "effective" moment of inertia (I_e) as defined by AASHTO 8.13.3 and ACI 9.5.2.3, using the service moment input.

AASHTO/ACI MINIMUM A_s

Like most of the structural programs, Ic follows AASHTO if flag 04 is clear and ACI if it is set. The governing code is displayed at the beginning of each run. If it is the desired code, press R/S to continue. (If using a printer, you won't need to press R/S--it continues automatically.) If designing by the other code, change the status of flag 04 and start over. AASHTO calculates the minimum area of steel by means of a cracking-moment concept rather than with $200/F_y$ as ACI does. If designing by ACI, the $200/F_y$ ratio is based on average depth to reinforcing for multiple layers of reinforcing, and on the web width for T-beams. If working with AASHTO, on the other hand, the minimum reinforcing is that necessary for a "cracking moment" based on the concrete gross moment of inertia, assuming a homogenous material with a strength of $7.5\sqrt{f'_c}$ (the "modulus of rupture" of concrete per AASHTO 8.15.2.1). This reinforcing is calculated by ultimate strength methods, using an effective depth to reinforcing 3" less than the full concrete depth of the member. AASHTO is not very clear in its present form that ultimate strength methods are to be used for all designs, WSD as well as LFD; however, the 1982 Interim Specifications, where it was introduced, spell it out a little better. Besides, it doesn't make sense to have a different minimum for WSD than for LFD. If the actual reinforcing is less than the minimum, the program displays AS<MIN; if the area of steel is greater than 4/3 of what is needed, it still meets code requirements.

MAXIMUM REINFORCING

Ic also checks the reinforcing against a maximum of $0.75P_b$ (75% of the "balanced" reinforcing ratio based

on ultimate-strength concepts) as a matter of good practice, although neither AASHTO nor ACI requires it. This check does take compression reinforcing into account, as well as the effect of T-beam flanges.

MOMENT CAPACITY

The moment capacity is found by calculating the moment causing full allowable stress in the concrete, tension steel, and compression steel, if any. The least of these is the output capacity. The governing capacity is displayed by the program as either CONC M=... or ST M=... for concrete or steel, respectively. If concrete governs, it's usually best to increase the size of the beam or add compression reinforcing. Note that the concrete section modulus S_c already has been adjusted by N , the modular ratio. See the examples.

At the end of a run, the program displays OK or NG (no good) in reference to the beam configuration just run. This reflects fatigue capacity and service load capacity. It does not refer to maximum or minimum steel checks or deflection. In other words, a beam with too much reinforcing will effect a warning AS>MAX during the run but may well display OK at the end. This means it has adequate service and fatigue strengths; however, the beam is not acceptable.

REINFORCING ESTIMATE

After displaying the maximum and minimum steel calculations, Ic prompts for the acting service moment. Based on this, it estimates the amount of reinforcing required. Using multiple layers of steel, of course, raises the required area since the reinforcing is being used less efficiently. The program bases this estimate on the following assumptions:

- 1) One layer of reinforcing
- 2) Effective depth $D = H - 3"$
- 3) For a T-beam, the neutral axis is within the flange.

The WSD estimate uses these rules of thumb (from the ACI Working Stress Handbook) where M is in foot-kips and D is in inches:

$$\begin{aligned} A_s &= M / (1.76 D) \text{ for } f_s = 24 \text{ ksi} \\ A_s &= M / (1.24 D) \text{ for } f_s = 20 \text{ ksi} \end{aligned}$$

This area of steel is, of course, only an estimate, but is often useful as a place to start or to check if a given beam configuration is reasonable for the load.

FATIGUE AND CRACKING

Ic calculates the elastic (cracked) section properties of the given beam configuration and, under AASHTO (flag 04 clear), fatigue properties. The program checks reinforcing fatigue only; concrete fatigue ($0.5 f_c'$) is not checked. According to ACI 343, no failures have ever been attributed to fatigue of concrete; and it never seems to govern, even in heavily reinforced members. However, if felt necessary, this should be checked by the designer.

Fatigue calculations involving stress reversal are complicated in concrete analysis, since it is not correct to simply use the same cracked-section properties with signs reversed. Any stress calculations on tension steel under stress reversal usually involve assumptions concerning the opposite layer of reinforcing (formerly compression steel, now in tension). Ic cavalierly makes a number of such assumptions, all based on concrete bridge slab design:

- 1) There is only a single layer of steel on each side.
- 2) Whether or not compression steel was input, the "top" steel (normally in compression, now in tension) is assumed to be twice the area of the steel being investigated for fatigue. This is predicated on the assumption of equal top and bottom mats (which is approximately true for continuous slab design) where the mat under investigation has had half its bars cut off (since most moment reversal problems happen at bar cutoffs).
- 3) Under moment reversal, the new "cracked" properties are not calculated precisely. The program assumes that $k' = 0.3$ and $j' = 0.9$ (where $j' = 1 - k'/3$).

Since the compressive force due to moment reversal is usually very low (on the order of 2 ksi or less) these assumptions are close enough. However, they really have no merit except in continuous slab design, so use negative MIN FATG M's with caution in other designs.

The program also assumes the reinforcing deformation ratio r/h is 0.8, meaning the maximum allowable fatigue stress is calculated as $21.0 + 3(r/h) - (f_{min})/3$, or $23.4 \text{ ksi} - (f_{min})/3$. Under moment reversal, f_{min} is negative, thus raising the allowable fatigue stress.

FLAGS

The program displays full output (including limit checks, cracked section properties, I_g , and I_e) if flag 00 is clear. If flag 00 is set, the program displays only the moment capacity. The limit checks (minimum and maximum areas of steel) and their warnings are not disabled by setting flag 00; only the display of the actual minimums and maximums is suppressed. All the programs have warning BEEPS for material property limits and some errors, which sound if flag 02 is set; however, Ic does not have BEEPs for infractions concerning minimum steel areas, only displayed warnings. The program is not affected by flags 01 and 03. As mentioned, flag 04 determines whether AASHTO (clear) or ACI (set) is followed.

SOFTKEYS

Softkeys are used extensively by Ic for modification or correction of input. To access the beam dimension prompts, press softkey D (in USER mode); to input new moments, use softkey E (external loads, maybe?). Softkey C (correct) returns the program to the previous reinforcing prompt (for multiple layers of tension steel) while still in the tension steel input section. Softkey A, on the other hand, starts the reinforcing input from the beginning (A-S1, etc.).

SHEAR DESIGN

All four of the flexural programs access a shear reinforcing design subroutine, VS. This is a true design program, not an analysis program. Ic is diverted to shear design by bypassing the A-S1 prompt. Unlike most other "defaults", the prompt must be bypassed; it doesn't work to input zero. And, it must be the A-S1 prompt. Control returns to the flexural program by bypassing the A-V prompt in the shear design subroutine. To use the shear design subroutine, the designer must go through the flexural program at least once, since VS uses the beam width and depth to reinforcing from the calling program. VS also uses the Flag 06 setting from the calling program to determine whether to design in WSD or LFD. In program Ic, softkey A can be used to get to the A-S1 prompt as long as at least one flexural analysis has been done; VS will use the beam configuration last run. Note that shear design is by the simplified equations of both codes and does not include the effect of moment or external tension or compression; this was done both to save programming space and because most designers usually use only the simplified methods anyway. For more information on this subroutine, see the documentation for VST, of which VS is a part.

T-BEAMS

The analysis of T-beams has a few peculiarities worth mentioning. First of all, the derivation of $.75P_{bal}$ is based on the actual beam configuration. The neutral axis position is calculated for $.75P_b$, and if it is found to be below the flange then everything is recalculated. This is shown in the T-beam example, to which reference should be made.

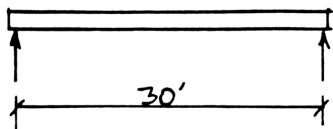
Second, rather than use the distance "c" to calculate the position of the neutral axis, LFAN uses the dimension "a", where "a", the compression block depth, is $B_1 \times c$. On most sections, this makes no difference. However, on those sections where "c" is slightly greater than the flange thickness and "a" is slightly less, the usual analysis has the flanges participating their full depth while the web is only effective to depth "a." In this situation, the program will have a different "a" somewhere between the two, and a slightly greater neutral-axis position "c." The difference is minimal practically but philosophically makes more sense. The "true" neutral axis position (i.e. that calculated by normal means) is still used to check compression steel yield and the value for $.75P_b$.

Invert T-beams are easily handled. In this case the FL W" (flange width) is less than the web thickness; the flange is defined as always being in compression. The lesser of the two dimensions (flange or web width) is always used as the width for shear design.

STRESSES

Program Ic (and the other working stress concrete programs, WSBM and VST) have a couple of idiosyncracies related to input of the allowable steel and concrete stresses. First, although the steel stress is a working stress (usually 20 or 24 ksi), the concrete stress to be input at the F_c' KSI prompt is the ultimate strength of the concrete, usually 3 to 4 ksi. The factor of .40 (AASHTO) or .45 (ACI) is applied by the program. Second, these programs (in subroutine N) calculate the reinforcing yield stress as $F_y = 5 \times (F_s - 12 \text{ ksi})$, giving 40 and 60 ksi from 20 and 24 ksi respectively. If the designer is using a different allowable stress (for example, some tank wall designs are done at 14 or 16 ksi to hold down cracking), the calculated yield stress will be incorrect, invalidating the maximum and minimum reinforcing checks (all of which involve ultimate strength concepts and yield stresses).

$$f'_c = 4000 \text{ psi}; f_y = 60 \text{ ksi}$$



Design a 30' slab for HS20. (24' Wide)

$$\text{AASHTO App. A: } M = 242.1 \text{ k/ft/lane}$$

$$\text{Impact} = 1 + \frac{50}{S_{\text{pan}} + 125} = 1.32 > 1.30$$

$$\therefore \text{Impact} = 1.30$$

Distribution Width (AASHTO 3.24.3.2)

$$E = (4 + .065) = 4 + .06 \cdot 30 = 5.8 \cdot 7.0'$$

$$\therefore M = \frac{242.1 \text{ k/ft/lane}}{2 \text{ wheel/lane}} \cdot 15.8' = 24.319 \text{ k/ft}$$

$$\text{check: } M \div 900 \text{ S} = .9 \cdot 30' = 27 \text{ k/ft OK}$$

$$\therefore \text{LL } M = 1.30 \cdot 24.319 = 31.61 \text{ k/ft}$$

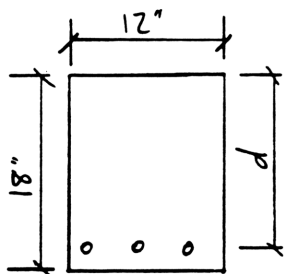
$$\text{AASHTO 4.9.2 recommends } h = 1.1 \cdot \frac{S + 10}{30}$$

$$h = 1.1 \cdot \frac{30 + 10}{30} \cdot 12 = 17.6"$$

USE 18" SLAB

$$\begin{aligned} \text{DL per foot} &= \text{Slab } 1' \times 1.5' \times .15 = .225 \text{ k/ft} \\ \text{Rail } &.44 \text{ k/ft} \div 13.67' = .032 \\ \text{Topping } &.02 \text{ k/ft} \cdot 12' \div 13.67' = .018 \\ &.275 \text{ k/ft} \end{aligned}$$

$$\begin{aligned} \text{DL } M &= .275 \cdot \frac{30^2}{8} = 30.94 \text{ k/ft} \\ \text{LL } M &= \frac{31.61}{62.55} \text{ k/ft} \end{aligned}$$



$$I_g = 12 \cdot \frac{18^3}{12} = 5432 \text{ in}^4$$

$$S_g = \frac{5432}{9} = 603.56 \text{ in}^3$$

Check cracking moment:

$$M_{cr} = f_r \cdot S_g$$

$$f_r = 7.5 \sqrt{f'_c} = 7.5 \cdot \sqrt{4000} = 474.34 \text{ psi}$$

$$1.2 M_{cr} = 1.2 \cdot 474.34 \cdot 603.56 \text{ in}^3$$

$$= 368.45 \text{ k/ft} = \underline{\underline{30.7373 \text{ k/ft}}}$$

By ultimate strength methods,

$$\rho = \frac{1}{m} (1 - \sqrt{1 - 2m R_u / f_y})$$

$$m = \frac{f_y}{.85 f'_c} = 17.6471$$

$$R_u = \frac{M}{b d^2} ; \text{ estimate } d = 18" - 3" = 15"$$

$$R_u = \frac{30.737 \cdot 12}{.9 \cdot 12" \cdot 15^2} = .1518 \text{ ksi}$$

$$\begin{aligned} \rho &= \frac{1}{17.6471} (1 - \sqrt{1 - 2 \cdot 17.65 \cdot .15 / 60}) \\ &= .0026 \end{aligned}$$

$$\text{Min } A_s = \rho b d = .0026 \cdot 12 \cdot 15$$

$$= \underline{\underline{.4660 \text{ in}^2/\text{ft}}}$$

Check by ACI: $\rho = 200 / F_y$

$$\text{Min. } A_s = \frac{200}{60,000} \cdot 12 \cdot 15 = .60 \text{ in}^2/\text{ft OK}$$

Estimate $A_{s, \text{req}}$

$$\begin{aligned} \text{Approx. } A_s &= \frac{M \text{ k/ft}}{1.76 \cdot d} = \frac{62.55 \text{ k/ft}}{1.76 \cdot 15"} \\ &= 2.37 \text{ in}^2/\text{ft} \end{aligned}$$

$$\begin{aligned} \text{Try } \#10 @ 6" ; A_s &= 1.27 \text{ in}^2/\text{ft} \cdot \frac{12'}{6"} \\ &= 2.54 \text{ in}^2/\text{ft} \end{aligned}$$

$$\text{Use } d = 18" - 1.5" - \frac{10}{8} \cdot \frac{1}{2} = 15.9" \text{ clear}$$

(055)

Check against min:

$$2.54 > .4660\% \quad \underline{\text{OK}}$$

Check against max:

$$\rho_{bal} = \frac{.45 \beta_1 f'_c}{f_y} \left(\frac{87}{87 + f_y} \right)$$

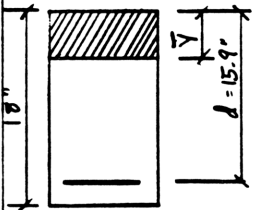
$$\text{for } f'_c = 4000, \beta_1 = .85$$

$$\therefore .75 \rho_{bal} = .75 \cdot \frac{.85 \cdot .45 \cdot 4}{60} \cdot \left(\frac{87}{87 + 60} \right)$$

$$= .0214$$

$$\text{Max } A_s = .0214 \cdot 12 \times 15.9 = 4.04 \text{ in}^2$$

$$2.54 \text{ in}^2 < 4.04 \text{ in}^2 \quad \underline{\text{OK}}$$

Calculate Section Properties (elastic)Use $n = 8$ for $f'_c = 4000$ 

$$\bar{Y} = \frac{n \cdot A_s}{b} \left(\sqrt{1 + \frac{2bd}{nA_s}} - 1 \right)$$

$$= \frac{8 \cdot 2.54}{12} \left(\sqrt{1 + \frac{2 \cdot 12 \cdot 15.9}{8 \cdot 2.54}} - 1 \right)$$

$$= 5.8376 \text{ in}$$

$$I_{cr} = 12 \cdot \frac{(5.8376)^3}{3} + (n \cdot A_s) \cdot 2.54 \cdot (15.9 - 5.8376)^2$$

$$= 2853.16 \text{ in}^4$$

$$S_{conc} = 2853.16 / 5.8376 = 488.76 \text{ in}^3$$

$$S_{st} = \frac{2853.16}{(n \cdot A_s) \cdot (15.9 - 5.8376)} = 35.443 \text{ in}^3$$

Either concrete or steel may govern:

$$M_{st} = 24 \text{ ksi} \cdot 35.443 / 12 = 70.89 \text{ k}$$

$$\text{check: } M_c = (.40 \cdot 4) \cdot 488.76 / 12 = 65.17 \text{ k} > 62.55 \text{ k}$$

OK

XEQ "Ic"

1:46:50 PM 02/07/88

TITLE:

BRIDGE SLAB RUN

AASHTO

N

8.0000 RUN

F-S 20,24

24.0000 RUN

Fc KSI

4.0000 RUN

FL W"

12.0000 RUN

WEB W"

RUN

H"

18.0000 RUN

I-G=5.832.0000

1.2Mc=30.7352 K

A-S=0.4660 SQ"

200/FY=0.0033

.75Pb=0.0214

A-S1

2.5400 RUN

D"

15.9000 RUN

A-S2

RUN

A-S'

RUN

RHO=0.0133

Ic=2.853.1635

Y=5.8376"

Sc=488.7539

S-ST=35.4435

CONC M=65.1672 K

A-S1

.4000 RUN

D"

16.0000 RUN

A-S2

RUN

A-S'

RUN

RHO=0.0021

AS<MIN

Ic=644.7407

Y=2.6667"

Sc=241.7778

S-ST=6.0444

ST M=12.0889 K

A-S1

Check fatigue (AASHTO)

Since this is a simple span, $M_{min} = M_{DL}$ and moment range = M_{LL} .

$$f_{min} = (30.94' \times 12) / 35.443 \text{ in}^2 \\ = 10.475 \text{ ksi}$$

$$\text{Allowable } f_f = 21 - \frac{f_{min}}{3} + 8(r/h)$$

$$\text{Use } (r/h) = 0.3 \quad (\text{deformation } \frac{\text{radius}}{\text{height}})$$

$$f_f = 21 - \frac{10.475}{3} + 8 \times 0.3 = \underline{19.9082 \text{ ksi}}$$

$$\text{Actual stress range} = \frac{(31.61 \times 12)}{35.443} = \underline{10.7023 \text{ ksi}}$$

$$\text{Calculate } I_{eff} = \left(\frac{M_{cr}}{M_{act}}\right)^3 \cdot I_g + \left(1 - \frac{M_{cr}}{M_{act}}\right)^3 \cdot I_{cr} \leq I_g$$

$$I_e = \left(\frac{30.73731 \times 12^3}{62.55}\right)^3 \cdot 5832.0 + (1 - \left(\frac{30.73731 \times 12^3}{62.55}\right)^3) \cdot 2453.16 = \underline{3057.72}$$

Check design for same loads by ACI

$$\text{Minimum } \rho = \frac{200}{F_y} = .0033$$

$$.0033 \times 12" \times 15" = .60 \text{ in}^2 < 2.54 \text{ in}^2 \text{ OK}$$

Section properties are the same

$$M_{ST} = 24 \text{ ksi} \times 35.443 \text{ in}^2 / 12 = 70.89 \text{ k}$$

$$M_{conc} = (.45 \times 4 \text{ ksi}) \times \frac{488.76}{12} = 73.31 \text{ k}$$

$$\text{Steel governs @ } \underline{70.89 \text{ k}} > 62.55 \text{ OK}$$

ACI has no fatigue requirements

$$.75p_b = .0214; \text{ act. } \rho = \frac{2.54}{12 \times 15.9} \cdot .0133 < .0214 \text{ OK}$$

SF 04

XEQ "IC"

7:42:48 PM 02/07/88

TITLE:

ACI WSD EXAMPLE

RUN

ACI

N

8.0000

RUN

F-S 20,24

24.0000

RUN

Fc KSI

4.0000

RUN

FL W"

12.0000

RUN

WEB W"

RUN

H"

18.0000

RUN

I-G=5,832.0000

1.2Mc=30.7352 K

200/FY=0.0033

A-S=0.6000 SQ"

.75Pb=0.0214

SERV M, K

62.5500

RUN

WS: EST A-S=2.3559 SQ"

A-S1

2.5400

RUN

D"

15.9000

RUN

A-S2

RUN

A-S'

RUN

RHO=0.0133

Ic=2,853.1635

Y=5.8376"

Sc=488.7539

S-ST=35.4435

ST M=70.8870 K

Ie=3,057.6799

OK

A-S1

Re-design bridge by LFD.

$$\begin{aligned} \text{AASTHO: } DL &= 30.94 \\ LL &= 5 \times 31.61 = 52.64 \\ &= 43.62 \\ &\times 1.3 \\ &= 104.71 \end{aligned}$$

Steel req'd for ultimate strength:

$$\rho = \frac{1}{m} \left(1 - \sqrt{1 - 2m R_u / f_y} \right)$$

$$m = \frac{f_y}{.45 f'_c} = \frac{60}{.45 \times 4} = 17.6471$$

$$R_u = \frac{M}{\phi b d^2} = \frac{104.71 \times 12}{.9 \times 12 \times 15.9^2}$$

$$= .4774$$

$$\begin{aligned} \therefore \rho &= \frac{1}{17.65} \left(1 - \sqrt{1 - 2 \times 17.65 \times .4774 / 60} \right) \\ &= .0086 \end{aligned}$$

$$A_s = .0086 \times 12 \times 15.9 = 1.644 \text{ in}^2$$

Positive moment reinf. - bottom of slab \Rightarrow Use $z = 170$ \therefore Expect A_s to be closer to LFD than WSD value ($z = 370$)

$$\text{Try } *10 @ 9"; A_s = 1.69 \text{ in}^2$$

Calculate elastic properties:

$$\bar{y} = \frac{n A_s}{b} \left(\sqrt{1 + \frac{z b d}{n A_s}} - 1 \right) = 4.9641$$

$$I_{cr} = 12 \times \frac{4.9641^3}{3} + (n-1) \times 1.69 \times (15.9 - 4.9641)^2$$

$$= 2106.22 \text{ in}^4$$

$$S_{st} = \frac{2106.22}{(15.9 - 4.9641) \times 4} = 24.0746 \text{ in}^3$$

Fatigue

$$f_{min} = \frac{M_{min}}{S_{st}} = \frac{30.94 \times 12}{24.0746}$$

$$= 15.42 \text{ ksi}$$

$$\text{allow. } f = 21.0 - \frac{f_{min}}{3} + 8(r/h)$$

$$\text{Use } (r/h) = 0.3$$

$$\therefore \text{allow. } f = 18.26 \text{ ksi}$$

$$\begin{aligned} \text{Act. Fatg. stress} &= \frac{\text{Mom Range}}{S_{st}} \\ &= \frac{31.61 \times 12}{24.0746} = 15.756 \text{ ksi OK} \end{aligned}$$

Ultimate strength

$$12 \times (.85 \times 4.0 \text{ ksi}) \times a = 1.69 \times 60 \text{ ksi}$$

$$a = 2.4853$$

$$c = \frac{a}{\beta_1} = \frac{2.4853}{0.85} = 2.9239$$

$$\begin{aligned} \text{UH. } M &= \phi \times (A_s f_y) \times (d - a/2) \\ &= .9 \times (1.69 \times 60) \times (15.9 - \frac{2.4853}{2}) \times 12 \\ &= 111.47 \text{ k} > 104.71 \text{ k OK} \end{aligned}$$

Z-Cracking

$$\begin{aligned} A_c &= 2 \times (14 - 15.9) \times 12 / \left(\frac{12}{9} \text{ bars per foot} \right) \\ &= 37.80 \text{ in}^2/\text{bar} \end{aligned}$$

$$d_c = 14 - 15.9 = 2.10$$

$$\text{Allow. } f_s = \frac{z = 170}{\sqrt[3]{37.80 \times 2.10}} = 39.556 \text{ ksi}$$

$$\Rightarrow \text{Use } f_s = 36.0 \text{ ksi}$$

$$\text{Act. } f_s = \frac{62.55 \times 12}{24.0746} = 31.1781 \text{ OK}$$

$$\text{Act. } z = \frac{31.1781}{36.0} \times 170 = 147.23$$

Service Moment

$$\text{Allowable } f_s = 36.00 \text{ ksi}$$

$$\therefore \text{Allowable serv. } M = f_s \cdot S_{sr}$$

$$= 36.0 \cdot 24.0746/12 = \underline{72.22} \text{ "k}$$

OK - Use #10 @ 9"

CF 04
XEQ "LFAN"
SIZE<28
SIZE 028
XEQ "LFAN"
3:13:18 PM 02/08/88

Size checked
by program
try again

TITLE:
BRIDGE SLAB--LFD RUN

AASHTO
F-Y 40,60
60.0000 RUN
F_c KSI
4.0000 RUN
FL W"
12.0000 RUN
WEB W"
RUN
H"
18.0000 RUN
I-G=5,832.0000
1.2Mc=30.7352 'K
A-S=0.4660 SQ"
200/FY=0.0033
.75Pb=0.0214

Z 130,170
170.0000 RUN
LF M, 'K
108.7100 RUN
SERV M, 'K
62.5500 RUN
MIN FATG M, 'K
30.9400 RUN
RANGE, 'K
31.6100 RUN

LF: EST A-S=1.7629 SQ"
WS: EST A-S=2.3559 SQ"

] estimates

A-S1
1.6900 RUN
D"
15.9000 RUN

A-S2
MAX BAR #
10.0000 RUN
A-S'
RUN
RHO=0.0089
Ie=2,362.0159
C=2.9239"

LF M=111.4692 'K
SERV M=72.2237 'K

Z STRS=36.0000 KSI
ACT STRS=31.1781 KSI
Z=147.2301

ALL FAT STRS=18.2593 KSI
ACT FAT STRS=15.7561 KSI

OK

A-S1
4.2000 RUN
D"
15.9000 RUN
A-S2
RUN
MAX BAR #
11.0000 RUN
A-S'
RUN

RHO=0.0220
P/Pb=0.7722

Too much steel

Ie=4,156.7368
C=7.2664"

LF M=242.1424 'K
SERV M=170.7604 'K

Z STRS=36.0000 KSI
ACT STRS=13.1869 KSI
Z=62.2715

ALL FAT STRS=21.2257 KSI
ACT FAT STRS=6.6641 KSI

NG

Check design for same loads by ACI

$$\begin{aligned} DL: 30.94 \times 1.4 &= 43.32 \\ LL: 31.61 \times 1.7 &= \frac{53.74}{97.06} \end{aligned}$$

Ultimate strength

Same: $M_{cap} = \underline{111.47} > 97.06^{kl}$ OK

Fatigue

ACI has no fatigue requirements

Z - Cracking

$$A_c = 37.80 \text{ in}^2/\text{bar}; d_c = 2.10''$$

Ext. exposure \Rightarrow use Z = 145

$$\text{Allow. } f_s = \frac{145}{\sqrt[3]{37.80 \times 2.10}} = \underline{33.74 \text{ ksi}}$$

$$\text{Act. } f_s = \frac{62.55 \times 12}{24.0746 \text{ in}^3} = \underline{31.18 \text{ ksi}} \quad \underline{OK}$$

$$\text{Act. } Z = \frac{31.18}{33.74} \times 145 = \underline{133.99}$$

$$\text{Allow. Serv. } M = \frac{33.74}{31.18} \times 62.55 = \underline{67.69^{kl}}$$

Area of steel could probably be reduced for ACI design

SF 04
XEQ "LFAN"
10:13:58 AM 02/09/88

TITLE:
ACI LFD EXAMPLE RUN

ACI
F-Y 40,60
60.0000 RUN
F_c' KSI 4.0000 RUN
FL W" 12.0000 RUN
WEB W" RUN
H" 18.0000 RUN
I-G=5.832.0000
1.2Mc=30.7352 'K
200/FY=0.0033
A-S=0.6000 SQ"
.75Pb=0.0214

Z 145,175
145.0000 RUN
LF M, 'K 97.0600 RUN

SERV M, 'K 62.5500 RUN

LF: EST A-S=1.5567 SQ"
WS: EST A-S=2.3559 SQ"

A-S1 1.6900 RUN
D" 15.9000 RUN
A-S2 RUN
MAX BAR # 10.0000 RUN
A-S' RUN

RHO=0.0089
Ie=2,362.0159
C=2.9239"

LF M=111.4692 'K
SERV M=67.6430 'K

Z STRS=33.7168 KSI
ACT STRS=31.1781 KSI
Z=134.0826

OK

A-S1

Check Max. A_s :

$$\rho_{bal} = \frac{.85 \beta_1 f_c'}{F_y} \cdot \frac{87}{87 + F_y}$$

for $f_c' = 3500$ psi, $\beta_1 = .85$

$$\rho_{bal} = \frac{.85^2 \cdot 3.5}{60} \cdot \frac{87}{87 + 60} = .0249$$

$$\text{Act. } \rho = (5.0 - 1.76) / 15 \cdot 21 = .0103$$

$$\frac{\rho}{\rho_{bal}} = \frac{.0103}{.0249} = .41 < .75 \text{ OK}$$

Calculate Elastic Section Properties

Neutral axis position, Y :

$$15 \cdot Y \cdot \frac{Y}{2} + [2 \cdot (n - 8.5) - 1] \cdot 1.76 \cdot (Y - 2.75) \\ = (n - 8.5) \cdot 5.0 \cdot (21 - Y)$$

$$7.5Y^2 + 28.16Y - 77.44 = 892.50 - 42.5Y$$

$$7.5Y^2 + 70.66Y - 969.94 = 0$$

$$Y = 7.5985"$$

	Area	d	Ad^2
Conc.: $15 \cdot 7.60$	$= 113.98$	3.80	1645.19
$A_s' : (2 \cdot 8.5 - 1) \cdot 1.76$	$= 28.16$	4.85	661.98
$A_s : (n - 4.5) \cdot 5.0$	$= 42.50$	13.40	7632.01
			9940.18
Conc. $I_o = 15 \cdot 7.5985^3 / 12$			$= 548.40$
			10,488.57

$$S_{ST} = \frac{10,488.57 / (n - 8.5)}{21 - Y} = 92.076 \text{ in}^3$$

Calculate Ultimate Section Properties

1st iteration - assume compr. $f_s = 60$ ksi:

$$(.85 \cdot 3.5 \text{ ksi}) \cdot 15 \cdot a$$

$$= (5.0 \cdot 60) - (1.76 \cdot (60 - (.85 \cdot 3.5)))$$

$$a = 4.47"$$

$$\text{Then compr. } f_s = 87 \cdot \frac{4.47 - 2.75 \cdot (\beta_1 \cdot .85)}{4.47}$$

$$= 41.54 \text{ ksi}$$

$$(87 = \text{conc. strain} \cdot .003 \cdot E_{st} = 29,000)$$

2nd iteration - use $f_s' = 41.54$ ksi

$$(.85 \cdot 3.5) \cdot 15 \cdot a = (5.0 \cdot 60)$$

$$- 1.76 \cdot (41.54 - .85 \cdot 3.5)$$

$$a = 5.20"$$

$$M = (\phi \cdot .9) \cdot [(5.20 \cdot 15 \cdot .85 \cdot 3.5) \cdot (21 - \frac{5.20}{2})$$

$$+ 1.76 \cdot (41.54 - .85 \cdot 3.5) \cdot (21 - 2.75)]$$

$$- (\phi \cdot .9) \cdot (4270.85 + 1238.71) = 4986.61 \text{ k} \cdot \text{ft}$$

$$= 413.22 \text{ k} > 355.84 \text{ k} \text{ OK}$$

Note: LFAN uses this two-iteration process. By way of comparison, an exact solution is given below:

$$(.85 \cdot 3.5) \cdot 15 \cdot a + 1.76 (87 \cdot \frac{a - (\beta_1 \cdot .85) \cdot 2.75}{a})$$

$$- .85 \cdot 3.5 = 5.0 \cdot 60$$

$$\text{Solving, } a = 5.01"; f_s' = 46.41 \text{ ksi}$$

$$M = 411.88 \text{ k} = 99.68\% \text{ of } 413.22$$

Neglecting displaced concrete: $a = 5.59"$;
 $f_s' = 50.61 \text{ ksi}$; $M = 445 \text{ k}$

ACI Doubly Reinforced Beam

3/4

Calculate allowable service moment

$$\text{Allowable } Z = 175$$

$$A_c = Z \cdot (24" - 21") \cdot 15/5 \text{ bars} = 18.0 \text{ in}^2/\text{bar}$$

$$d_c = 24" - 21" = 3"$$

$$\text{Allow. } f_s = \frac{Z=175}{\sqrt{3} \cdot 18.00} = 46.30 \text{ ksi} > 36.0$$

$$\Rightarrow \text{Use allow. } f_s = 36.0 \text{ ksi}$$

$$\text{Elastic } S_{xx} = 92.671 \text{ in}^3; \text{ act. } M = 233.60 \text{ k}$$

$$\text{Act. } f_{st} = \frac{233.60 \times 12}{92.076} = 30.44 \text{ ksi} < 36.0 \text{ OK}$$

$$\text{Acting } Z = \frac{30.44}{36.0} \times 175 = 146.0$$

$$\text{Allow. Serv. } M =$$

$$36.0 \text{ ksi} \times \frac{92.076}{12} = \underline{\underline{276.23 \text{ k}}}$$

\therefore Design is OK

Set flag 00 for Short

Form run

XEQ "LFAN"

9:39:09 AM 02/15/89

TITLE:
DOUBLE-REINF. BEAM RUN

ACI
F-Y 40,60
60.0000 RUN
Fc'KSI
3.5000 RUN
FL W"
15.0000 RUN
WEB W"
H"
24.0000 RUN
I-G=17,280.0000
1.2Mc=63.8892 'K
200/FY=0.0033
A-S=1.0500 SQ"
.75Pb=0.0187

Z 145,175
175.0000 RUN
LF M, 'K
355.8400 RUN
SERV M, 'K
233.6000 RUN

LF: EST A-S=4.3795 SQ"
WS: EST A-S=6.2846 SQ"

A-S1
5.0000 RUN
D"
21.0000 RUN
A-S2
MAX BAR #
9.0000 RUN
A-S'
1.7600 RUN
D-TOP"
2.7500 RUN

RHO=0.0103
P/Pb=0.4124 OK
Ie=10,568.9777
C=6.1195"

LF M=413.2223 'K
SERV M=276.2266 'K

Z STRS=36.0000 KSI
ACT STRS=30.4446 KSI
Z=147.9944

OK

0-01

D63

ACI Doubly Reinforced Beam - Working Stress

4/4

Calculate Working Stress Capacity

$$S_{st} = 92.076 \text{ in}^3$$

$$M_{st} = 92.076 \times 24 \text{ ksi} / 12 = \underline{184.15''\text{k}}$$

$$S_c = \frac{10,488.57}{7.5985} = 1380.35 \text{ in}^3$$

$$M_{conc} = 1380.35 \times (.45 - .35 \text{ ksi}) / 12 = \underline{181.17''\text{k}}$$

$$S'_{st} = \frac{10,488.57 / 8.5}{7.5985 - 2.75} = 254.50 \text{ in}^3$$

$$M'_{st} = 254.50 \times 24 \text{ ksi} / 12 = \underline{509.00''\text{k}}$$

Concrete governs: $M = 181.17''\text{k}$

\therefore Section is No Good by W.S.D.

XEQ "Ic"

11:28:41 AM 02/15/89

TITLE:

DOUBLY-REINF. BEAM RUN

ACI

F-S 20,24

24.0000 RUN

Fc' KSI

3.5000 RUN

N

8.5000 RUN

FL W"

15.0000 RUN

WEB W"

RUN

H"

24.0000 RUN

I-G=17,280.0000

1.2Mc=63.8892 'K

200/FY=0.0033

A-S=1.0500 SQ"

.75Pb=0.0187

SERV M, 'K

233.6000 RUN

WS: EST A-S=6.2846 SQ"

A-S1

5.0000 RUN

D"

21.0000 RUN

A-S2

RUN

A-S'

1.7600 RUN

D-TOP"

2.7500 RUN

RHO=0.0103

P/Pb=0.4124 OK

Ic=10,488.5732

Y=7.5985"

Sc=1,380.3469

S-ST=92.0755

CONC M=181.1705 'K

Ie=10,568.9777

NG

A-S1

01*LBL "LFAN"
02 SF 06
03 CF 09
04 GTO 00

05*LBL "Ic"
06 CF 06
07 SF 09

08*LBL 00
09 28
10 XROM "S"

11 CF 10
12 XEQ "N"

13 X<>Y
14 STO 18

15 12
16 RCL Z

17 -
18 FC? 06

19 PROMPT
20 STO 09

21 RCL 01
22 41

23 X>Y?
24 SF 09

25 RCL 02
26 STO 24

27*LBL D
28 CF 08

29 "FL W--"
30 PROMPT

31 STO 07
32 "WEB W--"

33 PROMPT
34 STO 03

35 RCL 07
36 X=Y?

37 SF 08
38 CLX

39 "FL TH--"
40 FC? 08

41 PROMPT
42 STO 10

43 "H--"
44 PROMPT

45 STO 11
46 3

47 -
48 STO 14

49 X12
50 RCL 07

51 *
52 .9

53 *
54 STO 05

55 CLA
56 RCL 11
57 RCL 10
58 -

59 RCL 03
60 RCL 07

61 LASTX
62 XROM "AY"

63 R↑
64 STO 23

65 "I-G="

66 ARCL X
67 FC? 00

68 AVIEW
69 R↑

70 6
71 *

72 XROM "Mc"
73 X<>Y

74 1.2
75 /

76 STO 22
77 X<>Y

78 RCL 07
79 *

80 RCL 14
81 XROM "R"

82 STO 19
83 RCL 24

84 RCL 14
85 RCL 18

86 RCL 00
87 *

88 /
89 *

90 STO 15
91 FS? 08

92 GTO 10
93 RCL 10

94 X>Y?
95 GTO 10

96 RCL 00
97 *

98 RCL 07
99 RCL 03

100 -
101 *

102 RCL 14
103 /

104 .75
105 *

106 RCL 24
107 RCL 03

108 *
109 +

110 RCL 07
111 /
112 RDN

	X	Y	Z	T	L		X	Y	Z	T	L
--	---	---	---	---	---	--	---	---	---	---	---

f_c' β_1
 ρ_1 ρ_1 f_c'
 f_c'

N
 F_y

.75p_w

Fl.W w_bw

Fl.Th

H

Est_d
Fl.W

ϕ
 $\phi b d^2$

H
Fl.Th
Web d
Web W
Fl.W
Fl.Th Fl.W Web d Web W
A Y S_c I

S_c

G S_b
P
M_{cr}

M_{cr}
P

Fl.W

d

M_{in}A_s

.75p_w
est. d

β_1
.95C₁K_y

β_1

C₁A_s

C

Fl.Th

.95C₁K_y

Fl.W
Web W

Width
Equiv A_s

est. d
Fl. d

.75p_w

.75p_w

Web W

R_ob_w

p_ob_w
p_ob
Fl. width
p_oequiv.

d

.75p_w

.75p_w

.75p_w

Fl.Th

Fl.W

C

A_s.+1.d = p.b

D65

	X	Y	Z	T	L		X	Y	Z	T	L
113*LBL 10						169 /	Est. A ₀				$A_s = \frac{M' \cdot b}{1.76 d''}$
114 R↑	.75A ₀					170 "WS"					$or = \frac{M' \cdot b}{1.43 d''}$
115 STO 02						171 XEQ 03					
116 ARCL X						172*LBL 23					
117 FC? 00						173*LBL A					
118 AVIEW						174 E	1				
119*LBL E						175 STO 20	counter				
120 CLX						176 CLX					
121 STO 25	Serv. M					177 STO 14					
122 STO 26	LF M					178 STO 21					
123 FS? 09						179 STO 17					
124 GTO 13						180 X<> 04	Max d				
125 XROM "L"						181 XROM "LL"					
126 130						182 CF 22					
127 "Z 130,170"						183*LBL 01					
128 FS? 04						184 "A-S"					
129 "Z 145,175"						185 RCL 20	counter				
130 PROMPT						186 XROM "AX"					
131 STO 13	Z					187 CLX					
132 FS? 00						188 PROMPT	A _s				
133 GTO 23						189 FC? 22	{shear				
134 "LF"						190 GTO 04	{end of				
135 XEQ 10						191 X=0?	S _{As}				
136 STO 26	LF. M					192 GTO 02	Input				
137 XROM "U"						193 ST+ 14					
138 STO 14	Est. P					194 STO 08	A _s				
139*LBL 13						195 CF 07					
140 FS? 00						196 ISG 20					
141 GTO 23						197 "					
142 XROM "L"						198 "D"	D				
143 "SERV"						199 PROMPT					
144 XEQ 10						200*LBL 12					
145 STO 25	Serv M					201 *	A _d				
146 "MIN FATG"						202 ST+ 21	± A _s d				
147 FC? 04						203 LASTX	d				
148 XEQ 10	Falg M					204 *	A _s d ²				
149 STO 28						205 ST+ 17	± A _s d ²				
150 ADV						206 SF 07					
151 RCL 14	Est. P					207 RCL 04	Max d				
152 RCL 07	W ₀₁					208 LASTX	d				
153 *						209 X>Y?					
154 RCL 11	H					210 STO 04	Max d				
155 3	3"					211 GTO 01					
156 -	Est d					212*LBL 04					
157 *	Est A _s					213 RDN	d _{max}				
158 "LF"						214 STO 04					
159 FS? 06						215 RCL 03	b _w				
160 XEQ 03						216 RCL 07	b _e				
161 RCL 25	Serv M					217 X>Y?					
162 LASTX	d					218 X<>Y	b _{min}				
163 /	m/d					219 XEQ "VS"					
164 RCL 01	f _y					220 GTO E					
165 5											
166 /											
167 9.24											
168 +	*										

166

* 1.76 * 12 for f_y = 60
1.43 * 12 for f_y = 40

	X	Y	Z	T	L		X	Y	Z	T	L
221 LBL C						278 LASTX	N				
222 DSE 20						279 ST* 17	$(A_0 d^2)_n$				
223 **						280 ST+ X	$2n$				
224 RCL 08	A_5					281 E	1				
225 ST- 14						282 -	$2n-1$				
226 CHS	$-A_5$					283 RCL 20	A_6'				
227 FC? 07						284 *					
228 .	$0 \text{ or } A_5$					285 STO 1	$(2n-1)A_6'$	nA_5			
229 LASTX	d	$-A_5$				286 +	ΣnA_5				
230 GTO 12						287 STO [
231 LBL 02						288 RCL 21	ΣAd				
232 *MAX BAR #						289 RCL 09	n				
233 FC? 09						290 *					
234 PROMPT	$\text{Max } \#$					291 RCL 00					
235 STO 27						290 *	nAd				
236 CF 07						291 RCL 08	d'				
237 CLX						292 RCL 1	$(2n-1)A_5'$				
238 *A-S**						293 *					
239 PROMPT						294 +	$\Sigma nA_5 d$				
240 STO 20	A_5'					295 STO \					
241 ENTER†						296 FS? 08					
242 *D-TOP**						297 GTO 04					
243 X>0?	d'					298 RCL 07	b_e				
244 PROMPT						299 RCL 10	$Fl.Th$				
245 STO 08						300 X†2	th^2				
246 RCL 15	"C"					301 *	Ad				
247 -	$d-c$					302 2					
248 LASTX	C					303 /	$Ad/2$				
249 /						304 RCL 10	$Fl.Th.$				
250 07						305 RCL 08	D'				
251 *						306 -					
252 RCL 01	f_y					307 RCL 1	$\Sigma n-DA_5'$				
253 /	$\frac{97(d-c)}{C-f_y}$					308 *	$A_6' d$				
254 E	-1					309 +	ΣAd				
255 CHS						310 RCL 21	ΣAd				
256 X>Y?						311 RCL 14	ΣA_5				
257 X<>Y						312 RCL 10	$Fl.Th$				
258 RDN	$\text{max } A_6'$					313 *					
259 *	$\text{eff. } A_5'$					314 -	$A_5(Fl-d)$				
260 RCL 14	ΣA_5					315 RCL 09	N				
261 +						316 *					
262 LASTX	$\frac{A_5-A_6}{A_6}$					317 X<=Y?					
263 *						318 GTO 04					
264 RCL 21	ΣAd					319 CF 05					
265 /	$(A_5-A_5')/d$					320 RCL 07	$Fl.W$				
266 RCL 07	b_{el}					321 RCL 03	$W_{el}W$				
267 /	$\text{eff. } p$					322 -	$Fl.W$				
268 XEQ "RH"						323 RCL 10	$Fl.Th$				
269 RCL 19	$\text{Min } A_5$					324 *	A_e				
270 RCL 14	ΣA_5					325 ST+ [(ΣA)				
271 *AS<MIN*						326 LASTX	Th				
272 X<=Y?						327 *	Ad				
273 AVIEW						328 2					
274 X<=Y?						329 /	$Ad/2$				
275 XROM "L"						330 ST+ \	(ΣAd)				
276 RCL 09	N					331 RCL 03	$W_{el}W$				
277 *	nA_5					332 GTO 05					

$X < Y ? \Rightarrow$
N.A. in flange

	X	Y	Z	T	L		X	Y	Z	T	L
333+LBL 04	fl.w.					387 RCL 09	n				
334 RCL 07						388 *	\bar{Y}				
335 SF 05						389 RCL 12	Y^2	nA_s			
						390 X+2	A_y^2				
336+LBL 05	b					391 *	I				
337 2						392 +	ΣA_d				
338 /	b/z					393 RCL 21	n				
339 RCL [ΣA					394 RCL 09	nA_{sd}				
340 RCL \	ΣAd					395 *	\bar{Y}				
341 CHS	$-2Ad$					396 RCL 12					
342 X< Z	b/z	ΣA	ΣAd			397 *	$2nA_d$				
343 XROM "Q"	\bar{Y}					398 ST+ X	I _{com}				
344 STO 12					Elastic N.A. position	399 -					
345 FS? 08						400 ST+ 17					
346 GTO 08						401 RCL 17	I _{cr}				
347 RCL 10	Fl.Th					402 RCL 09	n				
348 2						403 /					
349 /	Th/z	\bar{Y}				404 RCL 04	Max d				
350 X>Y?						405 RCL 12	\bar{Y}	I/n			
351 GTO 08	$\bar{Y}-I^2$					406 -	d				
352 -	d^2					407 /	S_{or}				
353 X+2	Th					408 STO 16					
354 RCL 10	Th^2					409 FS? 06					
355 X+2						410 GTO 06					
356 12	Th^2/z					411 .4					
357 /						412 FS? 04					
358 +						413 .45					
359 RCL 10	Fl.th.					414 RCL 00	$.45\bar{Y}/I$				
360 *						415 *					
361 RCL 07	Fl.W					416 .85	.45				
362 *	$I_o \cdot Ad^2$					417 /					
363 RCL 03	WebW					418 RCL 01	f_y				
364 RCL 12	\bar{Y}					419 *	$.45\bar{Y}'$				
365 RCL 10	Fl.th.					420 RCL 17	I _{cr}				
366 -	$\bar{Y}-Th.$	WebW	$I_o \cdot Ad^2$			421 RCL 12	\bar{Y}				
367 GTO 09						422 /	S_c	$.45\bar{Y}'$			
						423 *	M_c				
368+LBL 08						424 FS? 00					
369 .						425 GTO 11					
370 RCL 07	Fl.W					426 ADV					
371 RCL 12	\bar{Y}					427 "Ic="					
						428 ARCL 17					
372+LBL 09	\bar{Y}	b	$I_o \cdot Ad^2$			429 AVIEW					
373 3						430 "Y="					
374 Y+X	Y^3					431 ARCL 12					
375 LASTX						432 "t="					
376 /	$Y^3/3$	b	$I_o \cdot Ad^2$			433 AVIEW					
377 *	I_v					434 "Sc="					
378 +	I _{com}					435 ARCL L					
379 RCL 08	d'					436 AVIEW					
380 RCL 12	\bar{Y}					437 "S-ST="					
381 -						438 ARCL 16					
382 X+2	d^2					439 AVIEW					
383 RCL 1	$(n-1)A_s'$										
384 *	$A_s' d^2$					440+LBL 11	M_c				
385 +	I					441 ADV					
386 RCL 14	ΣA_s					442 RCL 13	f_{or}				
						443 RCL 16	S_{sr}	f_s	M_c		

	X	Y	Z	T	L		X	Y	Z	T	L
444 *	M _{sr}	M _c				500 LASTX	-C				
445 "CONC"						501 /	$\frac{C-d}{C}$	a	.	.	
446 X<=Y?						502 87					
447 "ST"						503 *	f _s		.	.	
448 X>Y?						504 RCL 01	f _y	f _s	a		
449 X<>Y	M _{min}					505 X<=Y?					
450 RCL 25	Serv M					506 X<>Y	Max	Min	a	Fi.Th	
451 XEQ 04	M _{min}					507 RDN	Min	a			
						508 STO a					
452+LBL 06						509 RDN	a	Fi.Th			
453 FS? 00						510 X<=Y?					
454 GTO 06						511 CLX					
455 RCL 23	I _{gross}					512 X<=Y?					
456 RCL 22	M _{sr}					513 GTO 07					
457 RCL 25	Serv M					514 RCL 10	Fi.Th				
458 X=0?						515 RCL 03	Vel.W				
459 GTO 06						516 STO \					
460 /	(M _{sr})	I _{gr}				517 RCL 07	Fi.W				
461 3	(M)					518 -	-Fi.V				
462 Y+X	(Y)					519 *	-Fi.A _{vec}				
463 *	I _{gr} (Y)					520 RCL 00	A _{vec} /f _y				
464 E	1					521 *					
465 LASTX	(Y)					522 STO [Fi.A _{vec} /f _y				
466 -	1-(Y)										
467 RCL 17	I _{cr}					523+LBL 07					
468 *	I _{cr} (1-(Y))	I _g (Y)				524 RCL 14	E _A s				
469 X<0?						525 +	E _A				
470 CLX						526 RCL 00	A _{vec} /f _y				
471 +	I _e					527 /	A _{vec} /f _y				
472 RCL 23	I _g					528 RCL 20	A _s				
473 X>Y?						529 RCL 00					
474 X<>Y	I _e					530 RCL 01	f _y				
475 "Ie="						531 *	A _s /f _y				
476 ARCL X						532 ST- a	(E _s -A _s)				
477 AVIEW						533 /	A _s /f _y				
						534 RCL a	f _s				
478+LBL 06						535 *	A _s /f _y	E _A f _y /f _s			
479 CLA						536 -	A _{vec} /f _y				
480 FC? 06						537 RCL \	b				
481 GTO 25						538 /	a				
482 RCL 10	Fi.Th					539 RCL 21	E _A .d				
483 RCL 14	E _A s	.				540 RCL 14	E _A s				
484 RCL 20	A _s		.			541 /	d				
485 -	A _s -A _s	.				542 STO]					
486 RCL 00	A _s /f _y	.				543 RCL 10	Fi.Th				
487 /	A _s /f _y	.				544 2	Z	Fi.Th	d	a	
488 RCL 20	A _s	.				545 /					
489 +		.				546 -	d-F _y /2	a	a	a	
490 RCL 07	b _e	.				547 RCL [Fi.A _{vec} /f _y				
491 STO \	a	Fi.Th				548 *	A _d f _y /f _y	a	a	q	
492 /						549 CHS					
493 ENTER↑	-a	a	Fi.Th			550 R↑	a				
494 CHS	β ₁			.		551 2					
495 RCL 18	C		.			552 /	a/2				
496 /	d'		.			553 RCL]	d	a/2	-A _s /f _y	a	
497 RCL 08				.		554 -	a/2-d				
498 X<>Y	-a/β ₁	d'	a			555 R↑	a				
499 +	A'-C		.			556 *	a.d				

Ultimate
N.A. in
Web

	X	Y	Z	T	L		X	Y	Z	T	L
557 RCL \	b					614 *	d Bar	Z			
558 *	A.d					615 RCL 04	Max d				
559 RCL 00	0.55 d					616 RCL 11	H				
560 *						617 -	d c				
561 -	-210 d					618 *					
562 RCL 01	f					619 RCL 03	b w				
563 *	-210 d					620 *	d. A. 2.5	Z			
564 RCL 1	d					621 RCL 14	2 A. 5				
565 RCL 08	d'					622 /	d. A				
566 -						623 3					
567 RCL 20	A. 5					624 1/X					
568 *	A. (0.8) 510					625 Y+X	R/A	Z			
569 RCL a	f. 5					626 /	All. str				
570 *						627 RCL 01	Ry				
571 +						628 .6	.6 fy				
572 .9	φ					629 *					
573 *	M					630 X>Y?					
574 X<>Y	q					631 X<>Y	All. str				
575 RCL 18	β.					632 ENTER†	All.	All.			
576 /	C	M				633 -SERV-					
577 "C="						634 RCL 16	S. 5				
578 ARCL X						635 *	All M				
579 "t="						636 RCL 25	Serv M				
580 FC? 00						637 XEQ 04					
581 AVIEW						638 FS? 00					
582 ADV						639 GTO 23					
583 X<>Y	M					640 ADV					
584 "LF"						641 CLX					
585 RCL 26	LF M					642 RCL 13	Z	All str			
586 XEQ 04						643 X<>Y	All str	Z			
587 FS? 09						644 -Z-					
588 GTO 25						645 XEQ 11					
589 RCL 13	Z					646 RCL 25	Serv M				
590 RCL d	d	Z				647 RCL 16	S. 5				
591 FIX 2						648 /	Act str	All str	Z		
592 RCL 27	Max. 2					649 "ACT"					
593 X†2	2					650 XEQ 11					
594 81	81	12	d	Z		651 /	All/Act	Z			
595 X=Y?						652 /	Act Z				
596 X>Y?						653 -Z="					
597 GTO 07						654 ARCL X					
598 CLX						655 AVIEW					
599 1.1											
600 Y†X											
601 125											
602+LBL 07						656+LBL 25					
603 /	Area					657 FS? 00					
604 RND						658 GTO 23	Z 3.4				
605 X<>Y	d	Area	Z			659 23.4	M. min				
606 STO d						660 RCL 28					
607 CLX						661 FC? 04					
608 RCL 21	2 A. 5 d					662 X=0?					
609 RCL 14	2 A. 5	A. 5 d	Area	Z		663 GTO 07	S. 5				
610 /	d	Area	Z			664 RCL 16	f. min	Z 3.4			
611 RCL 11	H					665 /					
612 -	d-H					666 X>0?					
613 ST+ X	d	Area				667 GTO 06					

070

Area of one bar

	X	Y	Z	T	L		X	Y	Z	T	L
668 RCL 11	H										
669 RCL 04	Max d					708 LBL 04	Min d	Max d			
670 -	H-d					709 X>Y?					
671 LASTX	d					710 SF 07					
672 /	H-d	f _{min}	23.4			711 RDN	Max d				
673 .3						712 XROM "M"					
674 -						713 XEQ "PV"					
675 *						714 RTN					
676 1.4											
677 /						715 LBL 10					
678 CHS	f _{min}					716 "t"					
						717 CLX					
679 LBL 06	f _{min}					718 XROM "M"					
680 STO 04						719 PROMPT	M ₁ -k	12			
681 3						720 *	M ₁ -k				
682 /	f/3	23.4				721 RTN					
683 -	All.f										
684 ADV						722 LBL 03					
685 "ALL"						723 "t": EST "					
686 XEQ 00						724 XROM "AS"					
687 RCL 25	Serv M					725 X=0?					
688 RCL 16	Sor					726 XEQ "PV"					
689 /	Actof					727 END					
690 RCL 04	f _{min}										
691 -											
692 "ACT"	Act.f	All.f									
693 XEQ 00											
694 X>Y?											
695 SF 07											
696 LBL 07											
697 ADV											
698 CLA											
699 XROM "OK"											
700 AVIEW											
701 GTO 23											
702 LBL 00											
703 "t FAT"											
704 LBL 11											
705 XROM "K"											
706 XEQ "PV"											
707 RTN											

CONCRETE SHEAR REINFORCING DESIGN

xeq VST (SIZE 015)

SOFTKEYS (USER mode ON)

- A New area of shear reinforcing
- D New dimensions

USER FLAGS

Flag 00	clear	Full output
	set	Title block not printed (output is not affected by flag 00)
Flag 02	clear	Warning BEEP disabled
	set	Warning BEEP enabled
Flag 04	clear	AASHTO
	set	ACI

INPUT SUMMARY

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
WSD?	Asking if by working stress design. Answer 1 (yes) for WSD, 0 (no) for LFD.	1 (WSD)
F-S 20,24 or F-Y 40,60	Reinforcing working or yield stress for WSD or LFD, respectively, in ksi.	24; 60
Fc'KSI	Concrete ultimate strength (for either WSD or LFD)	4.0 ksi
B"	Beam width (or web thickness), in inches	Previous B
D"	Beam depth to outermost layer of tension reinforcing, in inches	Previous D
A-V	Area of shear reinforcing, sq. inches; e.g. for one #5 stirrup input 0.62. Bypass (or input zero) to get new dimension prompts in VST or return to flexural program.	Zero
V, K	Design shear, in kips. For LFD, this must have the load factor included; however, <u>the program applies the undercapacity factor $\phi = 0.85$.</u>	Zero

OUTPUT SUMMARY (Items in parenthesis are displayed only if flag 00 is clear)

Vc	The portion of the total shear taken by the concrete, in kips.
SPA	The required stirrup spacing, in inches. Appends (MAX) if the spacing is governed by maximum spacing rules rather than shear requirements.
.5Vc>V	States that the design shear is less than half the allowable concrete shear; no shear reinforcing is necessary.
V-Vc NG	Design shear is too large for the given section. Increase section (or concrete strength).

PROGRAM FLAGS

Flag 06	clear	WSD
	set	LFD
Flag 10	set	Quick return from subroutine N

STORAGE REGISTER USE

<u>Register</u>	<u>Value</u>
00	Fc', ksi
01	Fy, ksi
02	Vc, kips (concrete shear)
03	B"
04	D"
05-07	Not used
08	$\sqrt{F_c'}$, ksi
09-12	Not used
13	F _{st} , ksi (WSD only)
14	B"

CONCRETE SHEAR REINFORCING DESIGN

Program VST designs the shear reinforcing for rectangular concrete beams. It can also be used for T-beams by simply inputting the web thickness at the B" (width, inches) prompt. VST will design by either working stress (WSD) or load factor (LFD) methods. It uses the ACI code if flag 04 is set; AASHTO is followed if flag 04 is clear. In LFD there is no difference between the two; in WSD, ACI allows a somewhat higher concrete shear than AASHTO. No matter which code is followed, VST uses the simplest shear design formula,

$$SPA = (F_{st} \text{ or } F_y) \times D'' \times A_v / (V - V_c).$$

It does not use the more complex moment-interaction formula.

After the dimensions and material properties are input, VST calculates and displays the concrete shear contribution, V_c , in kips. It then prompts for the area of shear reinforcing to be used, and then repeatedly calculates the stirrup spacing required for shear loads input. This allows the designer to quickly find the necessary spacings for varying shears along a beam. To input a new stirrup area (A-V), bypass the V, K (shear, in kips) prompt or press softkey A in USER mode. To input new dimensions, bypass the A-V prompt or press softkey D in USER mode.

MAXIMUM SPACING

The program follows all code provisions for maximum stirrup spacing ($d/2$ or 24") and maximum and minimum shear loads. In the case of small heavily loaded beams with large stirrup areas, the maximum spacing is cut in half (AASHTO 8.15.5.3.8/8.16.6.3.8 and ACI 11.5.4.3) to $d/4$ or 12". The recommended procedure if this happens is to decrease the stirrup size and run it again. The minimum area of shear reinforcing (per AASHTO 8.19.1.2 or ACI 11.5.5.3) is also checked. This sometimes governs the design of large, lightly loaded beams. See the examples for clarification.

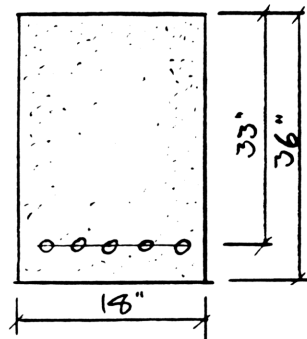
FOOTINGS and SLABS

Footing and slab designs are treated differently; they require no shear reinforcing for $V < V_c$. For these, any minimum shear reinforcing output (i.e. any SPA with (MAX) appended to it) is incorrect.

FLEXURAL SUBROUTINE

The main shear design program VST contains a subroutine, VS, which is called by the four concrete flexural programs (WSBM, LFBM, Ic, and LFAN). This subroutine is identical to VST except it uses the material properties and dimensions of the beam being investigated by the calling program. In the case of the analysis programs LFAN and Ic, which accept T-beams, the width used is the smaller of the web thickness or the flange width (the latter would be chosen for invert T-beams). The depth used is to the deepest layer of reinforcing (if there is more than one layer of tension reinforcing). To return to the calling program, bypass the V, K prompt and then the A-V prompt. The analysis programs return to the moment input sequence, while the design programs prompt for a new section.

Design shear reinforcing
for this beam:



$$f'_c = 4000 \text{ psi}$$

$$f_y = 60 \text{ ksi}$$

$$V_c = 2 \sqrt{4000} \times 18 \times 33 = 75.14 \text{ k}$$

Use #4 stirrups - $A_v = 2 \times .20 = .40 \text{ in}^2$

Max. Spacing

$$A_{v-\min} = \frac{50 b \cdot \text{Spa}}{f_y} \quad (\text{AASHTO 4.19.1.2})$$

$$\therefore \text{Max spa.} = \frac{.40 \times 60,000}{50 \times 18}$$

$$\text{or } \frac{d}{2} = \frac{33}{2} = 16.5" \leftarrow \text{Governs}$$

$$\text{or Max} = 24"$$

@ Reaction

$$DL = 24.0$$

$$LL = \frac{5}{3} \times 36.0 = 60.0$$

$$84.0 \times 1.3 = 109.2 \text{ k}$$

$$V_s = 109.2 / (\phi = .85) - 75.14 = 53.33 \text{ k}$$

$$\text{Spa.} = \frac{A_v f_y d}{V_s} = \frac{.40 \times 60 \times 33}{53.33} = 14.85"$$

@ 1/4 Point

$$DL = 12.0$$

$$LL = \frac{5}{3} \times 20.25 = 33.75$$

$$45.75 \times 1.3 = 59.48$$

$$V_s = \frac{59.48}{(\phi = .85)} - 75.14 \text{ k} < 0$$

\Rightarrow Use Max. Spa. = 16.5"

XEQ "VST"

3:52:40 PM 12/12/88

TITLE:

RECT. BEAM

RUN

WSD?

0.0000

RUN

0 -> "no" -> LFD

AASHTO

F-Y 40.60

60.0000

RUN

Fc' KSI

4.0000

RUN

B"

18.0000

RUN

D"

33.0000

RUN

Vc=75.1357 K

A-V

.4000

RUN

V, K

109.2000

RUN

SPA=14.8496"

V, K

59.4800

RUN

SPA=16.5000" (MAX)

V, K

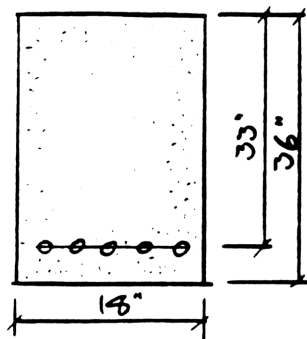
30.0000

RUN

.5Vc > V

V, K

Design shear reinforcing
for this beam:



$$f'_c = 4000 \text{ psi}$$

$$f_y = 60 \text{ ksi}$$

$$V_c = 2 \sqrt{4000} \times 18 \times 33 = 75.14 \text{ k}$$

Use #4 stirrups - $A_v = 2 \times 20 = .40 \text{ in}^2$

Max. Spacing

$$A_{v-\min} = \frac{50 b \cdot s_{pe}}{f_y} \quad (\text{ACI 11.5.5.3})$$

$$\therefore \text{Max } s_{pe} = \frac{.40 \times 60,000}{50 \times 18}$$

$$\text{or } \frac{d}{2} = \frac{33}{2} = 16.5" \leftarrow \text{Governs}$$

$$\text{or Max} = 24"$$

@ Reaction

$$DL = 1.4 \times 24.0 = 33.6$$

$$LL = 1.7 \times 36.0 = 61.2$$

$$94.8 \text{ k}$$

$$V_s = 94.8 / (1 - .95) - 75.14 = 36.39 \text{ k}$$

$$S_{pe} = \frac{A_v f_y d}{V_s} = \frac{.40 \times 60 \times 33}{36.39} = 21.76"$$

Use 16" Spacing Entire Beam

SF 04 ACI

XEQ "VST"

3:54:16 PM 12/12/88

TITLE:

RECT. BEAM 2 RUN

WSD?

0.0000 RUN

0 → "no" → LFD

ACI

F-Y 40.60

60.0000 RUN

Fc' KSI

4.0000 RUN

B"

18.0000 RUN

D"

33.0000 RUN

Vc=75.1357 K

A-V

.4000 RUN

V, K

94.8000 RUN

SPA=16.5000" (MAX)

V, K

RUN

A-V

.2200 RUN

V, K

94.8000 RUN

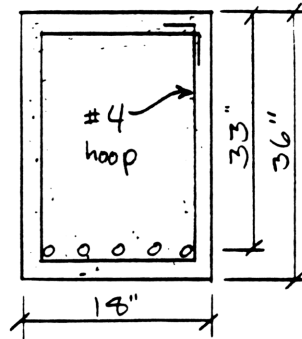
SPA=11.9691"

V, K

Bypass
prompt to
Try #3 hoop

From the preceding:

$$\begin{aligned} f'_c &= 4000 \text{ psi} \\ f_y &= 60 \text{ ksi} \\ f_{sr} &= 24 \text{ ksi} \end{aligned}$$



@ 1/4 Point

$$V_s = 32.25 - 35.69$$

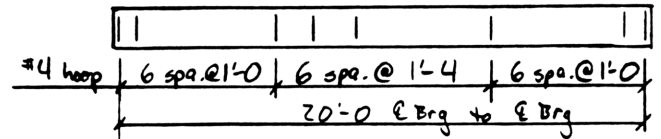
$$V_s < 0 \Rightarrow \text{Min. Steel}$$

$$\frac{1}{2} V_c = 17.84 \text{ k}$$

$$32.25 > 17.84 \Rightarrow \text{Need Stirrups}$$

$$\begin{aligned} \text{Max. Spa.} &: \\ d/2 &= 33/2 = 16.5'' \leftarrow \text{Governs} \\ \text{max} &= 24'' \end{aligned}$$

Use this layout:



@ Reaction:

$$\begin{aligned} \text{DL} &= 24.0 \text{ k (service)} \\ \text{LL} &= 36.0 \text{ k} \\ \Sigma &= 60.0 \text{ k} \end{aligned}$$

@ 1/4 Point:

$$\begin{aligned} \text{DL} &= 12.0 \text{ k} \\ \text{LL} &= 20.25 \text{ k} \\ \Sigma &= 32.25 \text{ k} \end{aligned}$$

Check Maximums

$$\begin{aligned} 2\sqrt{f'_c} b d &= 2\sqrt{4000} \times 18 \times 33 \\ &= 75.14 \text{ k} \end{aligned}$$

$$V_c = .95\sqrt{4000} \times 18 \times 33 = 35.69 \text{ k}$$

$$V_6 < 2\sqrt{f'_c} \Rightarrow \text{Spa. OK - AASHTO 4.15.5.3.6}$$

$$V_s < 4\sqrt{f'_c} \Rightarrow \text{OK - AASHTO 4.15.5.3.9}$$

@ Reaction.

$$V_s = 60.0 - 35.69 = 24.31 \text{ k}$$

Use stirrups - #4 hoop

$$A_v = 2 \times .20 = .40 \text{ in}^2$$

$$A_{v-\min} = \frac{50 b_v \cdot \text{Spa}}{f_y} \quad \text{AASHTO 4.19.1.2}$$

$$\therefore \text{Max spa.} = \frac{.40 \times 60,000}{50 \times 18}$$

$$\text{Spa.} = \frac{A_v f_s d}{V_s} = \frac{.40 \times 24 \text{ ksi} \times 33}{24.31 \text{ k}} = 13.03''$$

$$= 26.67'' \quad \underline{\text{OK}}$$

AASHTO W.S.D. Shear Design

4/4

CF 04 AASHTO

XEQ "VST"
3:56:10 PM 12/12/88

TITLE:
RECT. BEAM 3 RUN

WSD?

1.0000 RUN

W.S.D. → 1 = "yes"

AASHTO

F-S 20,24

24.0000 RUN

Fc KSI

4.0000 RUN

B-

18.0000 RUN

D-

33.0000 RUN

Vc=35.6895 K

.95 $\sqrt{f_c}$ · b · d

A-V

.4000 RUN

V, K

24.3100 RUN

Oops - wrong number

SPA=16.5000 (MAX)

V, K

60.0000 RUN

SPA=13.0314

V, K

32.2500 RUN

SPA=16.5000 (MAX)

V, K

42 381 20 SHEETS 5 SQUARE
42 382 100 SHEETS 5 SQUARE
42 383 200 SHEETS 5 SQUARE



040

	X	Y	Z	T	L		X	Y	Z	T	L
01*LBL "VST"						54*LBL 01					
02 14						55 RCL a					
03 XROM "W"						56 XROM "L"					
04 SF 10						57 STO a	A _v				
05 XEQ "N"						58 CLX					
06 STO 00	f _c '					59 "V, K"	V				
07*LBL D						60 PROMPT					
08 XROM "BD"	B'					61 X=0?					
09 RCL 03	B'					62 GTO A	Z4				
10 STO 14						63 24	d				
11 RCL 00	f _c '					64 RCL 04	Z	d	Z4	V	
12 XEQ 02						65 2	d/2	Z4			
13 GTO D						66 /					
						67 X>Y?					
						68 X<Y?	max spa				
14*LBL "VS"	B'					69 STO [
15 STO 14						70 RT	V				
16 XROM "L"						71 .85	Ø				
17 "SHEAR"						72 FC? 06					
18 AVIEW						73 SIGN	V				
19 RCL 00	f _c '/k _y					74 /	V _c				
20 RCL 01	f _y					75 RCL 02					
21 *						76 2	V _c /2	V			
22 .85	f _c '					77 /					
23 /						78 X>Y?					
						79 ".5V _c >V"					
24*LBL 02						80 X>Y?					
25 E3	1000					81 AVIEW					
26 /						82 X>Y?					
27 SQRT	f _c '/k _y					83 GTO 01					
28 STO 08						84 ST+ X	V _c	V			
29 .95						85 -	V _s				
30 FS? 04						86 RCL 08	f _c '				
31 1.1						87 FS? 06					
32 FS? 06						88 ST+ X					
33 2	factor					89 RCL 14	b				
34 RCL 08						90 RCL 04	d	b	f _c '	V _{sr}	
35 *	V _c					91 *	b d				
36 RCL 14	B					92 *	V _{limit}	V _{sr}			
37 *						93 /	V _{sr} /V _{lim}				
38 RCL 04	D					94 2	Z				
39 *						95 X>Y?					
40 STO 02	V _c /k					96 SIGN					
41 "V _c ="						97 ST/ [
42 ARCL X						98 CLX					
43 "K"						99 1.1					
44 FC? 00					FS? 06 or FC? 04	100 FC? 06					
45 AVIEW						101 FC? 04					
						102 SIGN	1 or 1.1				
46*LBL A						103 4					
47 XEQ "L"						104 *	4 or 4.4	V _{sr} /V _{lim}	V _{sr}	V _{sr}	
48 CLX						105 X>Y?					
49 "A-V"						106 GTO 00					
50 PROMPT						107 "V-V _c NG"					
51 X=0?						108 BEEP					
52 RTN						109 AVIEW					
53 STO a	A _v					110 GTO 01					

no shear
reinf.
needed

Max. Spa. in M.

1.1 only for ACI

(D81)

	X	Y	Z	T	L		X	Y	Z	T	L
111 LBL 00											
112 RCL a	A _v										
113 R↑	V _{or}										
114 X<0?											
115 CLX											
116 X=0?											
117 /											
118 FS? 06											
119 RCL 01	f _y										
120 FC? 06											
121 RCL 13	f _s										
122 *											
123 RCL 04	d										
124 *	Spa										
125 X=0?											
126 E3	100w _{sp}										
127 RCL a	A _v										
128 RCL 01	f _y										
129 RCL 14	b	f _y	A _v	Spa							
130 /											
131 *											
132 20											
133 *	Max' Spa	Spa	Spa	Spa							
134 X>Y?											
135 X<Y	Spa										
136 RCL [Max										
137 X>Y?											
138 X<Y	Spa?	Max	Max'	Spa							
139 R↑	Spa.										
140 X<Y	Spa?	Spa	Max	Max'							
141 "SPA="											
142 ARCL X											
143 "t--"											
144 X=Y?											
145 "t(MAX) "											
146 AVIEW	Spa.										
147 GTO 01											
148 END											

$$Spa. = \frac{A_v f_{st} d}{V_{st}}$$

$$Max. Spa. = \frac{A_v f_y \cdot 20}{b}$$

CONCRETE CORBEL DESIGN

xeg CORBL (SIZE 018)

SOFTKEYS (USER mode ON)

d	Estimate required depth given width and load
D	New corbel dimensions

USER FLAGS

Flag 00	clear set	Full output Only PRIM A-S and SEC A-S
Flag 02	clear set	Warning BEEP disabled Warning BEEP enabled
Flag 04	clear set	AASHTO ACI

INPUT SUMMARY

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
WSD?	Asking if by working stress design. Key in "1" for yes (WSD), "0" for no (LFD)	1 (WSD)
AASHTO or ACI	A reminder, not a prompt; push R/S	
F-Y 40,60 (LFD only)	Reinforcing yield stress, ksi	60 ksi
N (WSD only)	Modular ratio, E-st/E-conc	8
F-S 20,24 (WSD only)	Reinforcing working stress, ksi	24 ksi
Fc' KSI	Concrete strength in ksi (always ultimate strength)	4 ksi
MU 1,1.4	Friction coefficient. 1.4 for integrally cast concrete, 1.0 for "intentionally roughened" cold joint, 0.6 for cold joint	No default
B"	Corbel width, inches	Previous B
D"	Depth from top reinforcing to bottom of corbel, in inches	Previous D
H"	Total depth of corbel, in inches	Previous H

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
ARM"	Distance from face of supporting member to centroid of vertical load, in inches	No default
V, K	Vertical load, in kips. If bypassed or zero input, prompts for new dimensions	Zero
TENS Nc, K	Horizontal tension force due to temperature, shrinkage, etc.	V/5

OUTPUT SUMMARY (Items in parenthesis are displayed only if flag 00 is clear)

(A-V)	The area of steel for shear
(A-F)	The area of steel for bending
(A-N)	The area of steel for horizontal tension
(RHO-MIN)	Minimum reinforcing ratio, .04 Fc/Fy, for PRIM A-S
PRIM A-S	Primary (top) layer of reinforcing
SEC A-S	Secondary layers of reinforcing

PROGRAM FLAGS

Flag 06	clear set	WSD LFD
Flag 07	clear set	bending moment OK NG--too much bending moment in moment reinforcing subroutine
Flag 09	clear set	LFBM or WSBM CORBL
Flag 10	clear	subroutine N runs full length

STORAGE REGISTER USE

<u>Register</u>	<u>Value</u>
00	$.85 F_c' / F_y$
01	F_y , ksi
02	$(H'' - (D''))$, to calculate moment due to TENS N_c , K
03	B'' (width, inches)
04	D'' (depth from primary reinf. to bottom of corbel)
05	$\phi B D^2$ ($\phi = .85$ for shear and corbel moment)
06	V , K (vertical load, kips); $2/3 A_v$
07	H'' ; TENS N_c , K (horizontal tension force, kips)
08	Allowable concrete shear stress, ksi
09	N (modular ratio) (WSD only)
10	$D/3$ (WSD trial depth for neutral axis iteration)
11	Trial for iterative solution to neutral axis position (WSD only)
12	$.45 F_c'$ (ACI) or $.40 F_c'$ (AASHTO) (WSD only)
13	$F-S$ KSI (allowable steel stress) (WSD only)
14	R_{ideal} (see example for equations) (WSD only)
15	ARM'' (distance from center of load to column face)
16	μ (friction coefficient)
17	ϕF_y (LFD) or F_s (WSD) for horizontal tension

CONCRETE CORBEL DESIGN

Program CORBL designs the reinforcing for concrete corbels using either load factor (ultimate strength) or working stress (service) design. It follows AASHTO if flag 04 is clear, and ACI if flag 04 is set. AASHTO uses somewhat lower allowable stresses for WSD; for LFD the two codes are identical (except for factoring the loads, which is done by the user).

If flag 00 is set the display of A-F, A-N, A-V (the three component reinforcing areas), and RHO-MIN (the minimum primary reinforcing ratio) is suppressed; the program outputs only the necessary design quantities PRIM A-S and SEC A-S, the primary and secondary areas of steel.

WARNINGS

CORBL checks several items and displays the following warnings (with BEEPs if flag 02 is set) if they are violated:

- | | |
|---------|---|
| V > Vc | Shear is greater than allowable concrete shear stress; new dimensions are prompted |
| Nc > V | Horizontal tension is greater than shear; new loads are prompted |
| ARM > D | The position of the load with respect to the face of the supporting member is greater than the depth of the corbel; new dimensions are prompted |

The program also checks that the calculated primary (top) area of reinforcing is greater than the specified minimum.

LOAD and UNDER-CAPACITY FACTORS

The designer must be careful to input the correct loads to CORBL. If using LFD (Strength) design methods, the forces must have the load factor applied by the user. However, they must not include the under-capacity factor Phi (0.85), which is applied by the program. If using working stress design, this is not a problem.

HORIZONTAL TENSION

Both codes specify a minimum amount of horizontal tension on the corbel (due to thermal effects on the beam, concrete shrinkage, wind, etc.) of at least one-fifth the vertical load. If a "TENS Nc, K" force less than this is input, the program uses one-fifth V. CORBL also checks that this horizontal force is no greater than the vertical force, as mentioned above.

LISTING

CORBL is within the same program as WSBM and LFBM; the program listing has been placed with the documentation for these two programs.

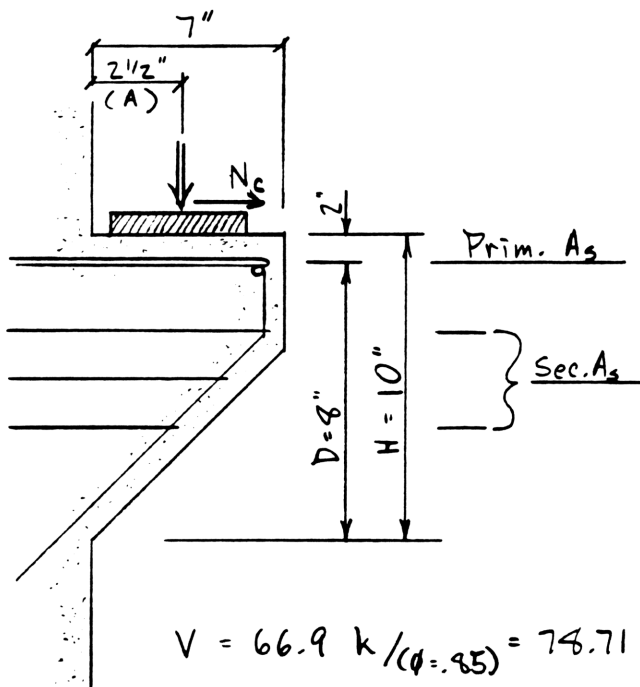
Design a corbel on a 14" wide column by LFD for the following loads, by AASHTO:

$$\begin{aligned} DL &= 30.3 \text{ k} \\ LL &= \frac{5}{3} \times 12.7 = 21.17 \\ &= 51.47 \times 1.3 = \underline{66.9 \text{ k}} \end{aligned}$$

$$\begin{aligned} f'_c &= 4.0 \text{ ksi} \\ f_y &= 60 \text{ ksi} \\ \phi &= 0.85 \end{aligned}$$

Cast monolithically $\Rightarrow \mu = 1.4$
(friction factor)

Try as follows:



Horizontal Tension $N_c = 0 \Rightarrow$ Use min.

$$N_{uc} = \frac{V_u}{5} = 15.74 \text{ k} \quad (\text{AASHTO 8.16.6.8.3 d})$$

Width $B = 14"$ (same as column)

Shear Friction Reinforcement

$$V_n : .2 f'_c = .2 \times 4000 = 800$$

$$800 b_w d = 800 \times 14" \times 8" = 89.6 \text{ k}$$

$$89.6 \text{ k} > 78.71 \text{ k} \quad \underline{OK}$$

$$\text{AASHTO 8.16.6.4: } A_{vf} = \frac{V}{\phi f_y \mu}$$

$$A_{vf} = \frac{78.71 \text{ k}}{60 \text{ ksi} \times 1.4} = \underline{.9370 \text{ in}^2}$$

Moment Reinforcement

$$M = V_u A + N_{uc} (h - d)$$

$$\begin{aligned} &= 78.71 \times 2.5" + 15.74 \times 2" \\ &= 228.26 \text{ in} \cdot \text{k} = 19.02 \text{ ft} \cdot \text{k} \end{aligned}$$

$$\frac{M}{\phi b d^2} = \frac{228.26 \times 1000}{0.85 \times 14" \times (8")^2} = 254.75$$

$$m = \frac{f_y}{.85 f'_c} = 17.6471$$

$$\begin{aligned} \rho &= \frac{1}{m} \left(1 - \sqrt{1 - \frac{2 m (M / \phi b d^2)}{f_y}} \right) \\ &= .0044 \end{aligned}$$

$$A_s = .0044 \times 8" \times 14" = \underline{.4948 \text{ in}^2}$$

Tension Reinforcement

$$A_n = \frac{N_{uc}}{\phi f_y} = \frac{15.74 \text{ k}}{60 \text{ ksi}} = \underline{.2623 \text{ in}^2}$$

Primary A_s

$$A_e + A_n = .4944 + .2623 = \underline{.7572 \text{ in}^2}$$

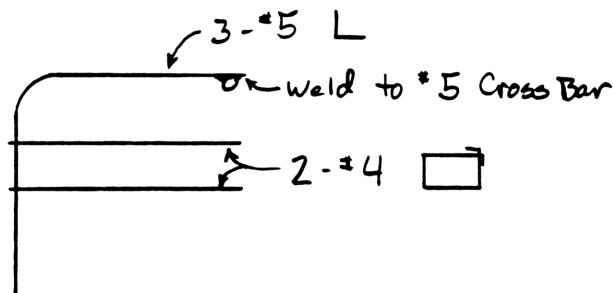
$$\begin{aligned} \frac{2}{3} A_{vf} + A_n &= \frac{2}{3} (.9370) + .2623 \\ &= \underline{.6870 \text{ in}^2} \leftarrow \text{Governs} \end{aligned}$$

Secondary A_s

$$A_h = \frac{1}{2} (.6870 - .2623) = \underline{.3123 \text{ in}^2}$$

Prim. A_s - provide 3-#5 ($A_s = .93 \text{ in}^2 > .687$)

Sec. A_s - provide 4-#4 ($A_s = .80 \text{ in}^2 > .3123$)

Check Minimum Reinforcing

AASHTO 8.16.6.8.5 :

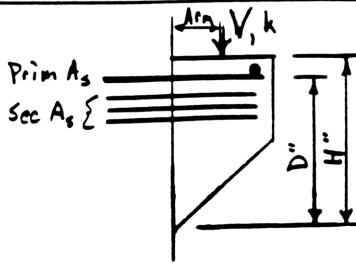
$$\rho_{min} = .04 \frac{f'_c}{f_y} = .04 \cdot \frac{4}{60} = \underline{.0027}$$

$$A_{s,min} = .0027 \times 8 \times 14 = \underline{.299 \text{ in}^2}$$

$$.6870 > .299 \quad \underline{OK}$$

"CORBL"-Load Factor Design

3/4



XEQ "CORBL"
7:27:07 AM 07/16/87

TITLE:
DEPTH EST. RUN

WSD?

0.0000 RUN

Load Factor
Design

AASHTO
F-Y 40.60
60.0000 RUN

Fc KSI
4.0000 RUN

MU 1.1.4
1.4000 RUN

friction
coeff.

B"
14.0000 RUN

D"
0.0000 RUN

depth unknown

V
66.9000 RUN

TRY H=9.9732"

Estimated
Depth

B"
14.0000 RUN

D"
8.0000 RUN

H"
10.0000 RUN

ARM"
2.5000 RUN

V, K

66.9000 RUN

TENS Mc, K
0.0000 RUN

A-V=0.9370 SQ"

A-F=0.4948 SQ"

A-N=0.2624 SQ"

RHO-MIN=0.0027

PRIM A-S=0.8870 SQ"

SEC A-S=0.3123 SQ"

V, K

90.0000 RUN

V>Vc

Concrete
Overstress for V=90.0 k

B"
14.0000 RUN

D"
10.0000 RUN

Try new
section

H"
12.0000 RUN

ARM"
2.5000 RUN

V, K

90.0000 RUN

TENS Mc, K

0.0000 RUN

A-V=1.2605 SQ"

A-F=0.5294 SQ"

A-N=0.3529 SQ"

RHO-MIN=0.0027

PRIM A-S=1.1933 SQ"

SEC A-S=0.4202 SQ"

V, K

66.9000 RUN

TENS Mc, K

0.0000 RUN

A-V=0.9370 SQ"

A-F=0.3900 SQ"

A-N=0.2624 SQ"

RHO-MIN=0.0027

PRIM A-S=0.8870 SQ"

SEC A-S=0.3123 SQ"

V, K

RUN

B"

Try new
section

"CORBL" - Working Stress Design

4/4

XEQ "CORBL"
7:33:55 AM 07/16/87

TITLE:
CORBEL DESIGN-PED RAMP

WSD? 1.0000 RUN

AASHTO
N 8.0000 RUN

F-S 20,24 24.0000 RUN

Fc KSI 4.0000 RUN

MU 1,1.4 1.4000 RUN

B- 14.0000 RUN

D- 8.5000 RUN

H- 10.5000 RUN

ARM- 2.5000 RUN

V, K 43.0000 RUN

V>Vc 4.0000 CLX

B- 14.0000 RUN

D- 10.5000 RUN

H- 12.5000 RUN

ARM- 2.5000 RUN

V, K 43.0000 RUN

TENS Mc, K 0.0000 RUN

A-V=1.2798 SQ"

A-F=0.5328 SQ"

A-N=0.3583 SQ"

RHO-MIN=0.0027

PRIM A-S=1.2115 SQ"

SEC A-S=0.4266 SQ"

Working Stress Design

Corbel too small
oops

V, K 43.0000 RUN

TENS Mc, K 44.0000 RUN

Mc>V -----

V, K 43.0000 RUN

TENS Mc, K 0.0000 RUN

A-V=1.2798 SQ"

A-F=0.5328 SQ"

A-N=0.3583 SQ"

RHO-MIN=0.0027

PRIM A-S=1.2115 SQ"

SEC A-S=0.4266 SQ"

V, K -----

B- -----

Tension greater than allowable

no more loads - new section

SHORT FORM, with flag 00 set:

SF 00
XEQ "CORBL"

WSD? 1.0000 RUN

N 8.0000 RUN

F-S 20,24 24.0000 RUN

Fc KSI 4.0000 RUN

MU 1,1.4 1.4000 RUN

B- 14.0000 RUN

D- 10.5000 RUN

H- 12.5000 RUN

ARM- 2.5000 RUN

V, K 43.0000 RUN

TENS Mc, K 0.0000 RUN

A-V=1.2798 SQ"

A-F=0.5328 SQ"

A-N=0.3583 SQ"

RHO-MIN=0.0027

PRIM A-S=1.2115 SQ"

SEC A-S=0.4266 SQ"

no title block

no breakdown of reinf. calcs - only final steel areas

REINFORCING BAR WEIGHT SUMMATION

xeq BWT (SIZE 004)

SOFTKEYS (USER mode ON)

- A Gives total weight of reinforcing (Σ), then returns to program. Can be checked at any time.
- B Common BAR size for succeeding bar sets. Bypass the prompt to return the program to the BAR # prompt at each bar set.
- C Correct an error. Subtracts the weight of the last bar set input from the sum of the weights.
- c Clears the sum (and the common BAR size, if any) and starts over. (The total is also zeroed every time BARWT is executed.)

USER FLAGS

Flag 00	clear	Full title block if printing.
	set	Title block is skipped if printing; output is not affected by flag 00.

INPUT SUMMARY

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
BAR #	Standard CRSI bar sizes 3-11, 14, and 18. Also recognizes 1/4" pencil rod (#2).	No default
QUANT	The quantity of bars in a given bar set	No default
L FT.IN	Length of one bar in feet and <u>inches</u> . Thus 12'-7 is input 12.07. For a variable bar length, key in the shortest length, enter/, then the longest length.	No default
BAR	(From softkey B) The bar size common to succeeding bar sets. Bypass to get BAR# prompt back for each bar set.	Zero

OUTPUT SUMMARY (Flag 00 has no effect on program BWT)

WT Weight of a given set of bars.

Σ (From softkey A) Total weight thus far.

$\Sigma - \text{Weight} = \text{New } \Sigma$ (From softkey C) The corrected total after subtracting an incorrect entry.

PROGRAM FLAGS

Flag 21 set Controls display/printing

Flag 27 set Turns USER mode on

STORAGE REGISTER USE

<u>Register</u>	<u>Value</u>
00	Bar weight for a given size bar; Weight x Quantity
01	Current bar set weight (for softkey C)
02	Summation of weights
03	Common bar size

REINFORCING BAR WEIGHT SUMMATION

Program BWT is used for calculating and summing the weights of reinforcing bars. Upon execution, BWT prompts for the bar size (BAR #), the number of bars of that type (QUANT), and the length of each bar (L FT.IN) in feet and inches. Six feet eleven inches is input as 6.11; 12'-3 is 12.03. See the notes concerning length and variable bar sets on the next page.

BAR SIZE

For regular reinforcing bar sizes, answer the BAR # prompt with the bar size (3-11, 14, 18); for 1/4" pencil rod, respond with a 2. Responding with other numbers will give strange results because the program uses a weight calculation algorithm based on the bar number to figure a given bar's weight. For instance, the program calculates the weight of a #12 bar as 6.428 pounds per foot, between that of a #11 and #14 bar.

If the bar list has many bars of the same size (diameter), press softkey B and key in that size in response to the BAR prompt. The BAR# prompt will be skipped for succeeding bar sets, and the program will use the BAR size input for softkey B. This feature saves a lot of repetitive input. To get the BAR# prompt back (e.g. if the bar size changes), press softkey B again and bypass the BAR prompt (press R/S without keying anything in).

WEIGHT SUMMATION

As mentioned, the program adds all weights to register 02, which is zeroed at the beginning of the program. This can also be accomplished by pressing softkey c (shift, SQRT). The program gives no evidence that it has cleared anything; the first time through, the user may wish to use softkey A to check it. See the final paragraph, regarding accidental clearing.

BWT actually sums the rounded-off weights, as determined by the display format. Since bar lists usually state the weights to the nearest pound or tenth of a pound, it is suggested to set the calculator to FIX 0 or FIX 1 format, respectively, before starting (or soon thereafter). Then the sum figured by the program will be the same as the total of the weights on the bar list; there is no round-off difference.

In order to get the total of the weights (which is, after all, why we're doing this) press softkey A. This can be done at any time, as many times as desired. After viewing the sum, pressing R/S gets you back into the input sequence, carrying your total. As mentioned

above, press softkey c (lowercase; shifted) to clear the total and start over.

LENGTHS

As mentioned, lengths are input in feet and inches, which is usually how bar lists state them. If a length is mistakenly input in decimal feet (e.g. 6.17 for 6'-2"), the program will catch it and shunt you back to the L FT.IN prompt for another go. It does this by checking that the fractional part of the length is less than .12; the only thing it can't catch is an input of 6'-1", for instance, as 6.08, which it sees as 6'-8".

VARIABLE BAR SETS

For a group of variable bars, key in the length of the longest bar, enter/ , then key in the length of the shortest bar and press R/S. Again, input must be in feet and inches. The program takes the average of these lengths, times the number of bars (QUANT) input. It's good to try this a few times to get comfortable with it.

ERROR CORRECTION

There are several ways to make mistakes with this program. The first and easiest, of course, is to simply key in the wrong number in response to a prompt. This is fixed by pressing softkey C (for "correct") after the incorrect weight has been displayed. The program will remove this weight from the total and return you to the input sequence. This can be done any time between when the incorrect weight is displayed and when the next weight is calculated and displayed (i.e. any time before you key in the next L FT.IN and press R/S). If you realize too late and finish inputting another bar, re-input the incorrect bar (incorrectly, as before) and then press softkey C twice, thus subtracting the incorrect weight twice. All of this, of course, hinges on realizing that a mistake has been made in the first place, which is the really tricky part.

Another mistake that happens when using the softkey B feature is to get used to bypassing the bar size, and then forget to input it when you're supposed to. Then, if you're paying more attention to the bar list than the calculator (the usual case), the BAR # prompt gets the quantity input and the QUANT prompt gets the length input. If the length happened to be an even number of feet, you'll be surprised by a L FT.IN prompt instead of a weight. The easiest way to take care of this is to press softkey B. If you try to input a non-integer number for the QUANT prompt (i.e. the length had inches) the program will realize what happened and do

this for you. This mistake happens the other way around just as often--inputting the BAR# at the QUANT prompt, and the QUANT at the Length prompt. Use softkey C to fix this one.

Potentially the worst mistake is to press softkey c (shifted, to clear) instead of C (unshifted, to Correct). The shifted key was chosen for "clear" because it takes more effort to do, but in spite of this innovative protection scheme the mistake still happens. When it does, you'll be surprised by a BAR# prompt instead of the reassuring " Σ - Weight = New Σ " display. However, before the program zeroes the total weight, it recalls it. To continue working, press RDN (roll down) to bring the total into the X-register, store it in register 02 (STO 02), then press softkey B and take it from there.

BILL OF BARS - SLAB			
BAR	NO.	LENGTH	WEIGHT
S501	54	2'-7	145.5
S602	10	20'-10	312.9
S503	12	Varies 2'-0 / 7'-10	61.5
S1004	32	46'-0	6334.0

$$501: 54 \times 2.543' \times 1.043 = 145.5 \text{ lbs.}$$

$$602: 10 \times 20.833' \times 1.502 = 312.9$$

$$503: 12 \times \frac{1}{2}(2.0 + 7.83') \times 1.043 = 61.5$$

$$1004: 32 \times 46.0' \times 4.303 = 6334.0$$

$$6853.9$$

FIX 1
FROM "BWT"
12:45 PM 04/10

TITLE:
SLAB RUN

BAR #

5.0 RUN

QUANT

54.0 RUN

L FT. IN

145.5 RUN

L FT. IN

2.07 RUN

WT=145.5

BAR #

6.0 RUN

QUANT

10.0 RUN

L FT. IN

20.10 RUN

WT=312.9

BAR #

5.0 RUN

QUANT

12.0 RUN

L FT. IN

2.0 ENTER

7.10 RUN

WT=61.5

BAR #

10.0 RUN

QUANT

32.0 RUN

L FT. IN

46.0 RUN

WT=6334.0

BAR #

XEQ A

S=6853.9

BAR #

	X	Y	Z	T	L		X	Y	Z	T	L
01*LBL "BWT"						50 CLX	0				
02 3						51 "QUANT"					
03 XROM "S"						52 PROMPT	Quant.				
04 SF 27	USER	make				53 ST* 00					
05*LBL c						54 X=0?					
06 SF 21						55 GTO 02	Frc=0?				
07 CLX						56 FRC					
08 STO 03	0					57 X=0?					
09 X<> 02	Old Σ					58 GTO 02					
10*LBL 02						59*LBL 07	0				
11 "BAR #"						60 .					
12 RCL 03	Common					61 "L FT.IN"	L Ft.IN				
13 X=0?						62 PROMPT	L ₁	L ₂ ?			
14 PROMPT	Bar #					63 XEQ 00	L				
15 8	8					64 ENTER†	L ₁	L ₁	L ₂ ?	L ₂ ?	
16 X<Y?						65 R†	L ₂ ?	L ₁	L ₁	L ₂ ?	
17 GTO 01						66 X>0?					
18 /	Dia."					67 XEQ 00	L ₂ ?				
19 2						68 X=0?					
20 /	Radius					69 RDN	L ₁ or L ₂	L ₁			
21 X†2	r ²					70 +					
22 PI	π					71 2					
23 *	Area					72 /	L ₁ or Any				
24 GTO 00						73 RCL 00	Wt. Quant				
25*LBL 01						74 *	Wt				
26 X<>Y	Bar #	8				75 RND	Rounded				
27 13	13					76 ST+ 02	(Σ)				
28 X<Y?						77 STO 01	Weight				
29 GTO 01						78 "WT"					
30 SIGN	1	Bar #	8			79 GTO 04					
31 GTO 02						80*LBL B					
32*LBL 01	13	Bar #	8			81 .					
33 RDN						82 "BAR"					
34 2	Z	Bar #	8			83 PROMPT	Common Bar				
35*LBL 02						84 STO 03					
36 -	Bar #	8				85 GTO 02					
37 X<>Y	8					86*LBL A					
38 /	Dim					87 "Σ"					
39 X†2	Area					88 RCL 02					
40*LBL 00						89 GTO 04					
41 3.4	Area					90*LBL C					
42 *	Wt.					91 "Σ"					
43 RCL d	Flags					92 ARCL 01	Σ				
44 X<>Y	Wt	Flags				93 RCL 02	Last Wt.				
45 FIX 3						94 RCL 01	Prev. Σ				
46 RND	Weight	Flags				95 -	Σ				
47 STO 00	Wt.					96 STO 02					
48 RDN	Flags										

Check if
QUANT prompt
was bypassed.
Check if L FT.IN
accidentally input.
Quantity must
be an integer

Dim = Side of equiv.
square bar

$$\frac{490 \text{ pcf}}{144 \text{ in}^3/\text{ft}} = 3.4$$

Rea.d the fla register.

D99

	X	Y	Z	T	L		X	Y	Z	T	L
97*LBL 04											
98 "t="											
99 ARCL X											
100 RVIEW											
101 GTO 02											
102*LBL 00	3.07	Y				Use example length 3.07 = 3'-7"					
103 INT	3										
104 LASTX	3.07										
105 FRC	.07	3	Y								
106 .12	.12	.07	3	Y							
107 X<=Y?											
108 GTO 07											
109 /	.563	3	Y	Y							
110 +	3.563	Y	Y	Y							
111 END											

RECTANGULAR CONCRETE COLUMN ANALYSIS

Program CC is a subroutine which calculates points on the load-moment interaction diagram for a rectangular reinforced concrete column. The points calculated represent levels of force in the tension-side reinforcing, and are ultimate (factored) strengths only. The following points are calculated:

MAX	The maximum usable axial load and the corresponding moment
OT	The eccentricity producing zero force in the tension-side reinforcing
25%T	The eccentricity producing a stress of 15 ksi in the tension reinforcing
50%T	The eccentricity producing a stress of 30 ksi in the tension reinforcing
BAL	The eccentricity producing a "balanced" condition, with concrete and steel reaching their ultimate strengths simultaneously

Output consists of axial loads in kips and moments in foot-kips, stored in numbered storage registers. CC has no displayed output; it is solely a computational subroutine requiring its input in the expected storage registers. Lack of space in the ROM prevented CC and its calling program CCOL from being included. However, a listing of both programs is included for use or as an example, to be keyed in and run in main memory (RAM). CCOL is a very simple input/output shell program for subroutine CC; however, its results are useful and the program can be expanded by the user.

Program CC uses modified versions of the CRSI Universal Strength Formulas, and includes all the limitations and assumptions of these formulas.

As mentioned, CC calculates only ultimate strengths; all values have been reduced by a phi-factor of 0.70, appropriate for strength design of tied columns. However, program CCOL is written to analyze by either strength or allowable stress (LFD or WSD) methods. In working stress design, CCOL uses $\phi = 1.0$ but factors the allowable loads and moments down to 0.35 (AASHTO) or 0.40 (ACI) of their ultimate levels.

RECTANGULAR CONCRETE COLUMN ANALYSIS

Note: this document serves as the Summary Sheet for both sub-routine "CC" and main program "CCOL", which uses CC. Both CCOL and CC must be keyed into memory by the user; space constraints prevented them from being included in the ROM. The only sections which pertain directly to CC are the SIZE requirement and storage register information.

(SIZE 019)

SOFTKEYS (USER mode ON)

- A New area of steel
- D New concrete dimensions

USER FLAGS

- | | | |
|---------|-------|---------------------|
| Flag 00 | clear | Full output |
| | set | Title block skipped |
| Flag 04 | clear | AASHTO |
| | set | ACI |

INPUT SUMMARY

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
WSD?	Asking if working stress or load factor design. Answer 1 (yes) or bypass for working stress design, 0 (no) for load factor (strength).	1 (WSD)
AASHTO or ACI	A reminder, not a prompt. Press to continue if not using a printer.	
F-S 20,24 or F-Y 40,60	Reinforcing working or yield stress in ksi	60 ksi
Fc' KSI	<u>Ultimate</u> concrete strength	4.0 ksi
B"	Column width in inches	Previous B
D"	Column depth (full dimension, not depth to reinforcing) in direction of bending.	Previous D
D" TO CL BAR	Distance from concrete surface to center of reinforcing bar (cover + tie + 1/2 bar diam.)	No default

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
#BARS	Total number of bars in column (all bars must be the same size)	No default
BAR A SQ"	The area of one bar in square in.	No default
#SIDE BARS	The number of bars <u>not</u> in either the tension or compression face	No default

OUTPUT SUMMARY (Items in parenthesis are displayed only if flag 00 is clear)

Output for CCOL consists of the following five points on the column's load-moment interaction diagram, each having an allowable axial load P in kips, and a corresponding moment M in foot-kips. The reinforcing ratio is also calculated, displayed, and checked to be between .01 and .08; and the value of $0.1 f_c' \times A_g$ is displayed.

BAL	The point of balanced stress between concrete and tension reinforcing.
MAX	The point of maximum usable axial load ($0.80 P_0$)
0T	The point producing zero stress in the tension reinforcing
25%T	The point producing one quarter of the allowable stress in the tension reinforcing
50%T	The point producing half the allowable stress in the tension reinforcing

PROGRAM FLAGS

Flag 06	clear	Working stress design (.35 or .40 times the ultimate strength)
	set	Ultimate strength (load factor) design

STORAGE REGISTER USE

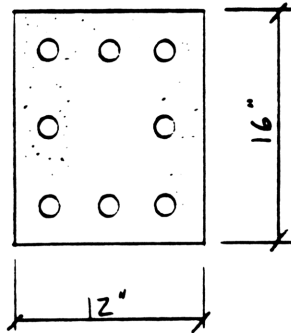
<u>Register</u>	<u>CC Value</u>	<u>Add'l CCOL Value</u>
00	F_c' , ksi	
01	B_1	
02	F-Y KSI	
03	B"	
04	D"	
05	# Side Bars	Counter

<u>Register</u>	<u>CC Value</u>	<u>Add'l CCOL Value</u>
06	"g" (bar separation/depth)	
07	Non-side-bar A_s/bd	
08	Side-bar A_s/bd	(.35 or .40)/ ϕ or 1.0
09	#BARS; BAL P	
10	1 bar area; BAL M	
11	MAX P	(.80 $f_c' A_g + A_s F_y$ whether by WSD or LFD)
12	MAX M	
13	OT P	
14	OT M	
15	25%T P	
16	25%T M	
17	50%T P	
18	50%T M	

Rect. Column Analysis

1/1

Analyze this column by
"strength" design and
compare to tabular values
in CRSI Handbook.



8 - #8 bars

$$f_y = 60 \text{ ksi} \quad \phi = .70$$

$$f'_c = 5.0 \text{ ksi}$$

$$\begin{array}{rcl} \text{Cover} & 1\frac{1}{2}'' & \\ \text{tie} & \frac{3}{8}'' & \\ \hline \frac{1}{2} \text{ bar} & \frac{1}{2}'' & \\ \hline 2.38'' = d' & & \end{array}$$

* Side bars = 2

CRSI (Major Axis):

Point	P, k	e (in)	M''k
Balanced	224 k	9.73"	141.6''k
Max (.80 P _b)	654 k	1.43"	77.9''k
Zero Tension (Interp.)	516.2	2.93"	126.0''k

XEQ "CCOL"

5:01:07 PM 01/24/89

TITLE:
CRSI TRIAL RUN

WSD?

0.0000 RUN

AASHTO

F-Y 40,60

60.0000 RUN

Fc'KSI

5.0000 RUN

B"

12.0000 RUN

D"

16.0000 RUN

D" TO CL BAR

2.3800 RUN

BARS

8.0000 RUN

BAR A SQ"

.7900 RUN

SIDE BARS

2.0000 RUN

RHO=0.0329 OK

.1Fc'A=96.0000 K

BAL: P=225.9900 K

M=181.7334 'K

MAX: P=654.2704 K

M=69.3719 'K

0T: P=518.1019 K

M=127.1233 'K

25%T: P=428.1258 K

M=144.7068 'K

50%T: P=352.9934 K

M=160.7045 'K

BARS

0105

	X	Y	Z	T	L			X	Y	Z	T	L	
01*LBL "CCOL"							58 RCL 07						
02 18							59 +						
03 XROM "W"							60 CF 07						
04 CF 10							61 .01						
05 XROM "N"							62 X>Y?						
06 STO 00							63 SF 07						
07 RDN							64 8						
08 X<> 01							65 *						
09 STO 02							66 X<Y?						
							67 SF 07						
10*LBL D							68 RDN						
11 XROM "BD"							69 ARCL X						
12 "D" TO CL BAR"							70 XROM "OK"						
13 PROMPT							71 AVIEW						
14 ST+ X							72 ".1Fc'A="						
15 RCL 04							73 RCL 00						
16 -							74 RCL 03						
17 LASTX							75 *						
18 /							76 RCL 04						
19 CHS							77 *						
20 STO 06							78 10						
							79 /						
21*LBL A							80 ARCL X						
22 XROM "L"							81 "I K"						
23 "I BARS"							82 AVIEW						
24 PROMPT							83 "BAL"						
25 STO 09							84 XEQ 01						
26 "BAR A "							85 "MAX"						
27 XROM "IS"							86 XEQ 01						
28 PROMPT							87 "0T"						
29 STO 10							88 XEQ 01						
30 "I SIDE BARS"							89 "25"						
31 PROMPT							90 XEQ 02						
32 STO 05							91 "50"						
33 STO 08							92 XEQ 02						
34 CHS							93 SF 14						
35 RCL 09							94 GTO A						
36 +													
37 STO 07							95*LBL 02						
38 RCL 10							96 "I%T"						
39 RCL 03													
40 /							97*LBL 01						
41 RCL 04							98 ADV						
42 /							99 "I: P="						
43 ST* 07							100 RCL IND 05						
44 ST* 08							101 RCL 08						
45 XEQ "CC"							102 *						
46 XROM "LL"							103 ARCL X						
47 9.1							104 "I K"						
48 STO 05							105 AVIEW						
49 4							106 ISG 05						
50 "RHO="							107 CLA						
51 7							108 RCL IND 05						
52 /							109 RCL 08						
53 FC? 04							110 *						
54 .5							111 XROM "M="						
55 FS? 06							112 AVIEW						
56 SIGN							113 ISG 05						
57 X<> 08							114 END						

	X	Y	Z	T	L		X	Y	Z	T	L
01 *LBL "CC"						57 5					
02 E						58 /					
03 RCL 07						59 *					
04 RCL 08						60 STO 12					
05 +						61 RCL 06					
06 -						62 .5					
07 LASTX						63 +					
08 RCL 02						64 XEQ 05					
09 *						65 16					
10 X<>Y						66 /					
11 .85						67 6					
12 *						68 *					
13 RCL 00						69 STO 15					
14 *						70 1.5					
15 +						71 LASTX					
16 .56						72 7					
17 *						73 /					
18 RCL 03						74 XEQ 09					
19 RCL 04						75 ST+ 15					
20 *						76 .22					
21 STO I						77 XEQ 07					
22 *						78 STO 16					
23 STO 11						79 4					
24 1.8						80 1/X					
25 ENTER↑						81 XEQ 08					
26 SIGN						82 3.2					
27 XEQ 09						83 /					
28 STO 13						84 ST+ 16					
29 4						85 RCL 06					
30 1/X						86 1.8					
31 XEQ 07						87 *					
32 STO 14						88 XEQ 05					
33 .28						89 4					
34 XEQ 08						90 /					
35 4						91 STO 17					
36 /						92 1.3					
37 ST+ 14						93 .75					
38 2.7						94 XEQ 09					
39 RCL 06						95 ST+ 17					
40 1.4						96 .19					
41 *						97 XEQ 07					
42 -						98 STO 18					
43 RCL 06						99 .24					
44 *						100 XEQ 08					
45 XEQ 05						101 3					
46 2						102 *					
47 /						103 8					
48 ST+ 13						104 /					
49 RCL 11						105 ST+ 18					
50 .8						106 RCL 06					
51 /						107 3					
52 STO Y						108 *					
53 RCL 13						109 2					
54 -						110 -					
55 /						111 .					
56 RCL 14						112 XEQ 06					
						113 LASTX					

	X	Y	Z	T	L		X	Y	Z	T	L
114 /						169 1/X					
115 .15											
116 *						170*LBL 00					
117 STO 09						171 RCL \					
118 SIGN						172 *					
119 .6						173 RCL 08					
120 XEQ 09						174 *					
121 ST+ 09						175 RCL 07					
122 .15						176 +					
123 XEQ 07						177 RCL 06					
124 STO 10						178 *					
125 .2						179 RCL 04					
126 XEQ 08						180 *					
127 2											
128 /						181*LBL 10					
129 ST+ 10						182 RCL 02					
130 RTN						183 *					
131*LBL 05						184*LBL 12					
132 RCL 07						185 RCL [
						186 *					
133*LBL 06						187 .7					
134 X<>Y						188 *					
135 RCL 08						189 RTN					
136 *											
137 X<0?						190*LBL 09					
138 CLX						191 RCL 06					
139 +						192 E					
140 GTO 10						193 +					
						194 *					
141*LBL 08						195 X<>Y					
142 RCL 06						196 RCL 08					
143 *						197 *					
144 .19						198 -					
145 +						199 RCL 01					
146 .37						200 *					
147 X>Y?						201 RCL 07					
148 X<>Y						202 -					
149 STO \						203 RCL 00					
150 RCL 05						204 *					
151 4						205 .425					
152 X>Y?						206 *					
153 CLX						207 GTO 12					
154 X=0?											
155 GTO 00						208*LBL 07					
156 RDN						209 RCL 01					
157 6						210 *					
158 X<=Y?						211 RCL 06					
159 SIGN						212 E					
160 E						213 +					
161 X=Y?						214 *					
162 GTO 00						215 .5					
163 R↑						216 -					
164 R↑						217 CHS					
165 12						218 RCL 04					
166 /						219 *					
167 1/X						220 *					
168 TNT						221 END					

STEEL DESIGN PROGRAMS

This module contains three programs concerned with the design of steel structures:

COMP	Composite steel/concrete beam analysis
PLG	Slender built-up plate girder analysis
SCOL	Steel column analysis (axial loads only)

In addition, another program, BM, which calculates elastic section properties of I- and T-shapes, is also frequently useful for steel design problems.

FLAGS

All the programs recognize flag 00; they are shorter and run faster when it is set. They also recognize flag 04. When flag 04 is set, the AISC Code is followed; when it is clear, AASHTO is used. AASHTO and AISC have much greater differences than do AASHTO and ACI (concrete design) or NDS (timber design). It is therefore very important that the status of flag 04 be checked before using these programs.

Flag 02 is also used by PLG and COMP. These programs have a BEEP at errors and warnings; when flag 02 is clear these BEEPs are disabled. Flags 01 and 03 have no effect on these programs.

STRESSES

All the programs work with stresses in KSI (kips per square inch); this changes the form of some AASHTO equations, making the program listings a little more difficult to trace through.

SUBROUTINES

As mentioned, SCOL deals only with axially loaded columns. However, the working calculation portion of the program has been given a global label, SFa, so that it may be called from a user-written program. In addition, program PLG contains a subroutine dealing with bending stresses, SFb. These are called in a program analyzing steel beam-columns named SBCOL, a listing of which follows. This may be keyed into main memory by the user. Unfortunately, space constraints prevented this from being included in the ROM.

I-BEAM SECTION PROPERTIES

Program BM is used to calculate elastic section properties for I- and T-shaped homogenous beams. It prompts for web depth and thickness, and the width and thickness of each flange, then calculates the moment of inertia, neutral axis position, section moduli of the top and bottom flanges, and the cross-sectional area. In addition, if flag 06 is set, the plastic modulus Z is calculated. If flag 00 is clear, all of these are displayed; if it is set, none of them are.

BM is intended for use as a subroutine. It may be used as a stand-alone program without any problems; however, it differs from most programs in this package on three points:

- 1) It does not use subroutine S; there is no SIZE check or title block. The user must verify a SIZE of at least 015.
- 2) Once finished, the program simply stops. It does not return for another section to be input.
- 3) If flag 00 is set there isn't just minimal output; there is no output at all.

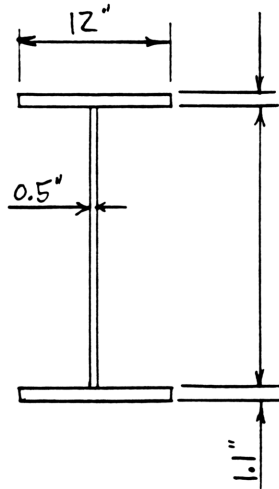
USE as SUBROUTINE

Program BM is used as a subroutine by COMP; all but the web input portion is used by PLG through subroutine SBM. Both BM and SBM in turn use subroutine AY (so called because the output is in the stack as A, Y, I, S_p) to actually calculate the section properties from the dimensions input. AY is also used by PIER and LFAN/Ic. AY is one of the three routines in this module that cannot be single-stepped through, due to its use of synthetic register P. See "Synthetic Programming and Other Loose Ends" for more details.

SOFTKEYS

This program uses no softkeys itself. However, it is a part of program COMP and thus will recognize softkeys from that program. While within COMP (or BM), softkey B will execute BM from the beginning. However, it will then return to COMP. The listing, and storage register use, of program BM are within the documentation for COMP.

"BM": Elastic Section Properties



XEO "BM"		
WEB D"	34.0000	RUN
TH"	.5000	RUN
COMP FL W"	12.0000	RUN
TH"	1.1000	RUN
TENS FL W"	12.0000	RUN
TH"	1.1000	RUN

$I = 9,771.5947$
 $Y = 18.1000"$
 $S-C = 539.8671$
 $S-Te = 539.8671$

AREA = 43.4000 SQ"

	Area	d	Ad	d	Ad^2	I_o
Top Fl.	13.2	.55	7.26	17.55	4065.633	1.331
Web	17.0	18.10	307.70	0	—	1637.667
Bottom Fl.	13.2	35.65	470.58	17.55	4065.633	1.331
	<u>43.4</u>	<u>18.10</u>	<u>785.54</u>		<u>8131.266</u>	<u>1640.329</u>
					9771.595	

$$S_{comp} = 9771.595 / 18.10" = 539.867$$

STEEL COLUMN ANALYSIS

xeq SCOL (SIZE 007)

SOFTKEYS

D New effective lengths

USER FLAGS

Flag 00	clear	Outputs allowable stress and maximum axial load
	set	Outputs only allowable axial load
Flag 02	clear	Warning BEEP disabled
	set	Warning BEEP enabled
Flag 04	clear	AASHTO
	set	AISC

INPUT SUMMARY

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
WSD?	Asking if the analysis is by working stress (ASD) methods. Answer "0" (no) to analyze by AASHTO LFD or AISC LRFD rules.	1 (WSD)
F-Y KSI	Steel yield stress, ksi	36 ksi
KL-X"	Major axis effective length, in.	No default
KL-Y"	Minor axis effective length, in.	KL-X
AREA	Trial member area, sq. inches. Bypass to re-use the previous section (e.g. after softkey D).	Previous Section
R-X	Major axis rad. of gyration, in.	No default
R-Y	Minor axis radius of gyration	R-X

OUTPUT SUMMARY (Items in parenthesis are displayed only if flag 00 is clear)

(Fa) Allowable axial stress, ksi

ALL P Allowable axial load, kips

In addition, the Euler stress is held in the Y-register.

PROGRAM FLAGS

Flag 06	clear	WSD (ASD)
	set	LFD (LRFD)

STORAGE REGISTER USE

<u>Register</u>	<u>Value</u>
00	not used
01	KL-X (inches stored)
02	KL-Y (inches stored)
03	R-X
04	AREA, square inches
05	R-Y
06	F-Y KSI

STEEL COLUMN ANALYSIS

Program SCOL analyzes steel columns of any shape by the AISC or AASHTO codes for allowable service or ultimate (factored) stresses, using program SFa as a subroutine. SCOL prompts for steel yield stress, axial load, and effective lengths in each direction. It then asks for the area and radii of gyration of a trial section, calculates the allowable compressive stress and corresponding load, and returns for a new trial section.

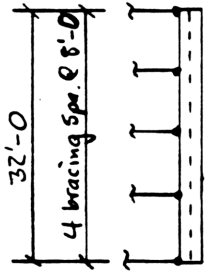
The program displays a warning and BEEPs if the KL/r value is too high (above 120 per AASHTO 10.7.1; and 200 per AISC ASD B7. (1.8.4) or LRFD B.7). It does still output an allowable stress by the appropriate formulae; it is up to the judgement of the designer if the design is acceptable.

SCOL is not written to analyze beam-columns; however, the allowable-stress subroutine, SFa, was given a global label in order to allow its inclusion in user-written programs for interaction analysis. SFa needs only the effective length KL/r in the X-register, and F-Y KSI in register 06. It outputs the allowable stress in ksi in the X-register and the Euler stress in the Y-register. The WSD versions of the program for both AASHTO and AISC incorporate the proscribed safety factor (2.12 and 23/12, respectively) into the Euler stress value, while the AASHTO LFD and AISC LRFD versions do not. This follows the way the Euler values are used in the interaction equations. See the examples.

The program is written to analyze prismatic members only; it is unable to account for tapered columns. Built-up members can be analyzed if the separate parts can be considered integrally connected. The user must calculate the radii of gyration. Program BM may be of help for this.

SCOL/SFa assume the flanges and web to be compact, or at least more so than the overall length. The only stress/strength reduction taken is for unbraced length.

Calculate the allowable load on the column shown below.



$$KL_y = 8'-0'' = 96''; KL_x = 32'-0'' = 384''$$

$$W6 \times 20: A = 5.87 \text{ in}^2$$

$$r_x = 2.66''$$

$$r_y = 1.50''$$

$$F_y = 36 \text{ ksi}$$

Effective Length

$$(KL/r)_x = \frac{384}{2.66} = 144.36 \leftarrow \text{Governs}$$

$$(KL/r)_y = \frac{96}{1.50} = 64.0$$

$$C_c = \sqrt{\frac{2\pi^2 E}{F_y}} = \sqrt{\frac{2\pi^2 (29,000)}{36}} = 126.10$$

$$144.36 > 126.10 \Rightarrow \text{"long"}$$

$$\therefore F_a = \frac{12\pi^2 E}{23(KL/r)^2} = 7.166 \text{ ksi}$$

$$\text{Allowable } P = 7.166 \times 5.87 \text{ in}^2 = \underline{42.06 \text{ k}}$$

Add a brace at midlength - $KL = 192''$

$$(KL/r)_x = 72.18 < C_c \Rightarrow \text{"short"}$$

$$\therefore F_a = \left(1 - \frac{(KL/r)^2}{2C_c^2}\right) \times \frac{F_y}{F.S.}$$

$$F.S. = \frac{5}{3} + \frac{3}{8} \left(\frac{KL/r}{C_c}\right) - \frac{1}{8} \left(\frac{KL/r}{C_c}\right)^3$$

$$= 1.458$$

$$F_a = \left(1 - \frac{1}{2} \left(\frac{KL/r}{C_c}\right)^2\right) \times \frac{36}{1.458}$$

$$= 16.20 \text{ ksi}$$

$$\text{Allowable } P = 16.20 \times 5.87 \text{ in}^2$$

$$= \underline{95.11 \text{ k}}$$

For interaction equation,

$$\text{Euler stress } F_e = \frac{\pi^2 E}{23(KL/r)^2} = 28.66 \text{ ksi}$$

SF 04
XEQ "SCOL"
3:14:49 AM 01/01/00

Set flag 04 to
use AISC

TITLE:

ASD

RUN

AISC

WSD?

1.0000

RUN

1 = "yes" \Rightarrow A.S.D.

F-Y KSI

36.0000

RUN

KL-X"

384.0000

RUN

32'-0" = 384"

KL-Y"

96.0000

RUN

8'-0" = 96"

AREA

5.8700

RUN

W6x20

R-X

2.6600

RUN

R-Y

1.5000

RUN

Fa=7.1655 KSI

Pa=42.0613 K

AREA

XEQ D

KL-X"

192.0000

RUN

KL-Y"

96.0000

RUN

AREA

RUN

Fa=16.2025 KSI

Pa=95.1087 K

AREA

Press softkey "D"
for new length
(Dimension) prompts

Bypass to use
same section as
previous trial

(D117)

Same column - W6x20

$$A = 5.57 \text{ in}^2$$

$$r_x = 2.66 \text{ in}$$

$$r_y = 1.50 \text{ in}$$

First Case

$$KL_x = 384 \text{ in}; KL/r = 144.36$$

$$KL_y = 96 \text{ in}; KL/r = 64.0$$

$$\text{Check flange } b/t: \frac{6.02 \text{ in}}{2 \times .365 \text{ in}} = 8.25$$

$$\lambda_r = \frac{95}{\sqrt{F_y}} = \frac{95}{\sqrt{36}} = 15.8 > 8.25 \text{ OK}$$

∴ Can use full allowable stresses

Note: The program does not check flange slenderness (λ_r). The user must check this to determine if SCOL/SFa may be used.

$$\lambda_c = \frac{KL}{\pi r} \sqrt{\frac{F_y}{E}} = 1.619 > 1.5$$

$$\Rightarrow F_{cr} = \frac{.877}{\lambda_c^2} \times F_y = 12.04 \text{ ksi}$$

$$F_a = (\phi = .85) \times F_{cr} = \underline{10.24 \text{ ksi}}$$

$$\text{Allowable } P = 10.24 \times 5.57 = \underline{60.10 \text{ k}}$$

(Factored Load)

$$\text{Euler stress } F_e = \frac{F_y}{\lambda_c^2} = 13.734 \text{ ksi}$$

for interaction equation

Second Case - Brace @ Midpoint

$$KL_x = 16' = 192 \text{ in}$$

$$\lambda_c = \frac{KL}{\pi r} \sqrt{\frac{F_y}{E}} = .8095 < 1.5$$

$$\Rightarrow F_{cr} = (.658^{\lambda_c^2}) F_y$$

$$= .658^{.6583} \times F_y$$

$$= 27.364 \text{ ksi}$$

$$F_a = (\phi = .85) \times 27.364 = \underline{23.26 \text{ ksi}}$$

$$\text{Allow. } P = 23.26 \times 5.57 \text{ in}^2 = \underline{136.53 \text{ k}}$$

$$\text{Euler stress} = F_y / \lambda_c^2 = 54.94 \text{ ksi}$$

3:16:36 AM XEQ SF 04
SCOL- 01/01/00

Set flag 04
to use AISC

TITLE:

LRFD RUN

AISC

MSD?

0.0000 RUN

0 = "no" → LRFD

F-Y KSI

36.0000 RUN

KL-X"

384.0000 RUN

KL-Y"

96.0000 RUN

AREA

5.8700 RUN

R-X

2.6600 RUN

R-Y

1.5000 RUN

Fa=10.2380 KSI

Pa=60.0973 K

AREA

XEQ D

Safety "D"

KL-X"

192.0000 RUN

KL-Y"

96.0000 RUN

AREA

RUN

Bypass to use
Same section

Fa=23.2597 KSI

Pa=136.5343 K

D116

Calculate the allowable load on the same column (W6x20) by AASHTO working stress criteria.

$$W6 \times 20: A = 5.47 \text{ in}^2$$

$$r_x = 2.66 \text{ in}$$

$$r_y = 1.50 \text{ in}$$

First Case

$$KL_x = 384 \text{ in}; KL/r = 144.36 \xrightarrow{\text{Use}}$$

$$KL_y = 96 \text{ in}; KL/r = 64.0$$

$$C_c = \sqrt{\frac{2\pi^2 E}{F_y}} = 126.10$$

$$144.36 > 126.10 \Rightarrow \text{"Long"}$$

$$\therefore F_a = \frac{\pi^2 E}{2.12 (KL/r)^2} = 6.478 \text{ ksi}$$

$$\text{Allowable } P = 5.47 \times 6.48 = \underline{35.03 \text{ k}}$$

$$\text{Euler stress } F_e = 6.478 \text{ ksi}$$

(Note: this column violates AASHTO 10.7.1)

Second Case - Brace @ Mid height

$$KL_x = 192 \text{ in}; KL/r_x = 72.18 > 64.0$$

$$\therefore KL/r = 72.18 < (C_c = 126.10)$$

\Rightarrow "Short"

$$F_a = \frac{F_y}{2.12} \times \left(1 - \frac{(KL/r)^2 F_y}{4\pi^2 E}\right) = 14.20 \text{ ksi}$$

$$\text{Allowable } P = 5.47 \times 14.20 = \underline{43.35 \text{ k}}$$

$$\text{Euler stress} = \frac{\pi^2 E}{2.12 (KL/r)^2} = 25.91 \text{ ksi}$$

XEQ "SCOL"

3:13:09 AM 01/01/00

TITLE:

WSD RUN

AASHTO

WSD?

1.0000 RUN

1 = "yes" \Rightarrow WSD

F-Y KSI

36.0000 RUN

KL-X"

384.0000 RUN

KL-Y"

96.0000 RUN

AREA

5.8700 RUN

R-X

2.6600 RUN

R-Y

1.5000 RUN

KL/R > 120.0000

Fa = 6.4783 KSI

Pa = 38.0278 K

Warning: AASHTO 10.7.1

AREA

XEQ D

Sortkey "D"

KL-X"

192.0000 RUN

KL-Y"

96.0000 RUN

AREA

RUN

Bypass for same section

Fa = 14.1992 KSI

Pa = 83.3492 K

AREA

Calculate the allowable load
on the same column (W6x20)
by AASHTO load factor criteria.

$$W6 \times 20 : A = 5.87 \text{ in}^2$$

$$r_x = 2.66 \text{ in}$$

$$r_y = 1.50 \text{ in}$$

First Case:

$$KL_x = 364 \text{ in} ; KL/r = 144.36 \leftarrow \text{Governs}$$

$$KL_y = 96 \text{ in} ; KL/r = 64.0$$

Note: $KL/r = 144.36 > 120$; this
column violates AASHTO 10.7.1.
It is up to the discretion of
the engineer whether the
results are acceptable.

$$C_c = 126.10 < 144.36 \Rightarrow \text{"Long"}$$

$$F_{cr} = \frac{\pi^2 E}{(KL/r)^2} = 13.734 \text{ ksi}$$

$$F_a = (\phi = .85) \times 13.734 = 11.674 \text{ ksi}$$

$$\text{Allowable } P = 11.674 \times 5.87 = \underline{68.53 \text{ k}}$$

Second Case - Brace @ Midheight

$$KL_x = 192 \text{ in} ; KL/r_x = 72.18 \leftarrow \text{Governs}$$

$$KL/r_y = 64.0$$

$$72.18 > C_c (=126.10) \Rightarrow \text{"short"}$$

$$F_{cr} = \left(1 - \frac{(KL/r)^2 F_y}{4\pi^2 E}\right) \times F_y = 30.10 \text{ ksi}$$

$$F_a = (\phi = .85) \times F_{cr} = 25.59 \text{ ksi}$$

$$\text{Allowable } P = 25.59 \times 5.87 = \underline{150.20 \text{ k}}$$

(D120)

$$\text{Euler stress} = \frac{\pi^2 E}{(KL/r)^2} = 54.94 \text{ ksi}$$

XEQ "SCOL"
3:11:30 AM 01/01/00

TITLE:

LFD

RUN

AASHTO

WSD?

0.0000

RUN

0 \Rightarrow "no" \Rightarrow L.F.D.

F-Y KSI

36.0000

RUN

KL-X"

384.0000

RUN

KL-Y"

96.0000

RUN

AREA

5.8700

RUN

R-X

2.6600

RUN

R-Y

1.5000

RUN

KL/R > 120.0000

Fa=11.6739 KSI

Pa=68.5260 K

AREA

XEQ D

Softkey "D"

KL-X"

192.0000

RUN

KL-Y"

96.0000

RUN

AREA

RUN

Bypass to use
existing
section

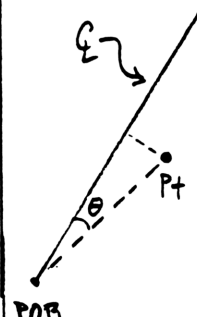
Fa=25.5869 KSI

Pa=150.1952 K

AREA

	X	Y	Z	T	L		X	Y	Z	T	L	
01*LBL "SCOL"						54*LBL "SFa"	Kl/r					$\lambda_c = \frac{Kl}{\pi r} \sqrt{\frac{F_y}{E}} \geq 1.5$
02 6						55 8						$\rightarrow \frac{Kl}{r} > \sqrt{\frac{2.25 \pi^2 E}{F_y}}$
03 "AISC"						56 "KL/R)"						LRFD: $C_c = \sqrt{\frac{2.25 \pi^2 E}{F_y}}$
04 XROM "AA"						57 9						$2.25 = \frac{9}{4} \cdot 2$
05 CF 06						58 /	b/q					
06 E	1					59 FS? 04						
07 "WSD?"						60 FC? 06	1?					
08 PROMPT						61 SIGN						
09 X=0?						62 2						
10 SF 06						63 /	$1/2$					
11 XROM "FY"	F_y					64 X<Y	Kl/r	$1/2$				
						65 FC? 04	limit					
12*LBL D						66 120						AASHTO 10.7.1
13 "KL-X"						67 FS? 04						AISC 1.6.4
14 PROMPT						68 200						LRFD B.7
15 STO 01	Kl_x					69 ARCL X						
16 "KL-Y"						70 X<Y?						
17 PROMPT						71 BEEP						
18 STO 02	Kl_y					72 X<Y?						
						73 AVIEW						
19*LBL 10						74 RDN	Kl/r	$1/2$				
20 XROM "L"						75 PI	π					
21 CF 22						76 /	$(\frac{Kl}{r})^2$					
22 "AREA"						77 X12	E					
23 PROMPT						78 29 E3						
24 FC? 22						79 /	F_y					
25 GTO 09						80 RCL 06	$\frac{Kl/r^2}{\pi^2 E F_y}$	$1/2$				
26 STO 04	A_{req}					81 *	$\frac{Kl/r^2}{C_c^2}$	$\frac{Kl/r^2}{\pi^2 E F_y}$				
27 "R-X"						82 STO Z	$\frac{Kl/r^2}{C_c^2}$	$\frac{Kl/r^2}{\pi^2 E F_y}$				
28 PROMPT						83 *	1					
29 STO 03	r_x					84 CF 19						
30 "R-Y"						85 E						
31 PROMPT						86 X<Y?	$\Rightarrow \ln g$	$\frac{Kl/r^2}{C_c^2}$				
32 STO 05	r_y					87 SF 19	$\frac{Kl/r^2}{\pi^2 E F_y}$	$\frac{Kl/r^2}{C_c^2}$				
						88 X<Z	F_c	λ_c^2	$\frac{Kl/r^2}{C_c^2}$			
33*LBL 09	Kl_x					89 ENTER↑						
34 RCL 02	r_y					90 1/X						
35 RCL 05						91 STO [
36 /						92 FS? 06						
37 RCL 01	Kl_x					93 GTO 11						
38 RCL 03	r_x					94 FS? 04						
39 /						95 1.9167	F.S.	F_c	λ_c^2	$\frac{Kl/r^2}{C_c^2}$		
40 X<Y?						96 FC? 04						
41 X<Y	Kl/r					97 2.12						
42 XEQ "SFa"	F_a	F_c				98 /	F_{c-req}	λ_c^2				
43 "Fa"						99 STO [
44 XROM "KS"						100 FS? 19						
45 FC? 00						101 GTO 00						
46 XEQ "PV"						102 X<L	2.12	λ_c^2				
47 RCL 04	A_{req}					103 R↑	$\frac{Kl/r^2}{C_c^2}$					
48 *	P_k	F_c				104 SQRT	$Kl/r/C_c$					
49 "Pa="						105 ST* L						
50 ARCL X						106 3	3	()	2.12	λ_c^2	() ³	
51 "t K"						107 ST* Y		3()				
52 AVIEW						108 X<L	()	3()	2.12	λ_c^2	() ³	
53 GTO 10						109 -	3()-()	2.12	λ_c^2	λ_c^2		(DIZ)

	X	Y	Z	T	L		X	Y	Z	T	L
110 8	8										
111 /											
112 .6						160 CF 22					
113 1/X	5/3					161 "PT ON CL"	Pt	POB			
114 +	115 F.S. 2.12	λ_c^2	λ_c^2			162 PROMPT					
115 FC? 04						163 FS? 22					
116 X<>Y	F.S.	λ_c^2	λ_c^2			164 XROM "XY"	ΔX	ΔY			
117 1/X						165 X<>Y	Y	X			
118 Rt	$\frac{(X/L)^2}{\pi^2 E/F_y}$	F.S.				166 R-P	d	Az			
119 LBL 01						167 X<>Y	Az	d			
120 4						168 "AZ"					
121 /						169 FC? 22					
122 SIGN	1					170 XROM "Pd"					
123 LASTX	$\frac{(X/L)^2}{4\pi^2 E/F_y}$	1	F.S.			171 STO 01	Az				
124 -	1-()	F.S.									
125 *	Fa					172 LBL 04					
126 GTO 00						173 XROM "L"					
127 LBL 11	Fc	λ_c^2				174 "PT"					
128 .877	.877					175 PROMPT	Pt				
129 FC? 04						176 XEQ 02	sta	offset			
130 SIGN	1?	Fc	λ_c^2			177 XROM "PV"					
131 *						178 X<>Y	offset				
132 .85	ϕ					179 RND	offset				
133 *	Fe					180 CHS					
134 FS? 19						181 "OS"					
135 GTO 00						182 XROM "V"					
136 X<>L	$\phi = .45$	λ_c^2				183 GTO 04					
137 X<>Y	$\frac{(X/L)^2}{\pi^2 E/F_y}$.45									
138 FC? 04						184 LBL 02	POB				
139 GTO 01						185 RCL 00	Pt	POB			
140 .658	.658	λ_c^2	.45			186 X<>Y	ΔX	ΔY			
141 X<>Y	λ_c^2	.658	.45			187 XROM "XY"	Y	X			
142 Y+X						188 X<>Y	d	Az			
143 *	$\phi = .458$					189 R-P	Az	d			
144 LBL 00	Fa					190 X<>Y	ΔAz				
145 RCL [Fe					191 RCL 01	θ				
146 X<>Y	Fa	Fe				192 -	d	θ			
147 RCL 06	Fy					193 X<>Y	sta	offset			
148 ST* Z						194 P-R					
149 *	Fa	Fe									
150 RTN						195 LBL 03					
151 LBL "POB"						196 ENTER	sta	sta	offset		
152 SF 21						197 "STA="					
153 RDN						198 E2	100	sta	sta	OS	
154 FS?C 11						199 /					
155 GTO IND T						200 INT	$\frac{sta}{100}$	sta	OS	OS	
156 SF 14						201 XROM "AX"					
157 "POB"						202 "I+"	$\frac{sta}{100}$				
158 PROMPT						203 X<>L	2nd Part				
159 STO 00	POB					204 FRC	100				
						205 E2	2nd Part				
						206 *					
						207 ARCL X	sta	offset	offset		
						208 RDN					
						209 END					



STEEL BEAM-COLUMN ANALYSIS

xeq SBCOL (SIZE 032)

Note: SBCOL is not resident in ROM and must be keyed into main memory by the user. A listing follows. The program requires bytes (registers) of memory in RAM.

SOFTKEYS

D New effective lengths

USER FLAGS

Flag 00	clear	Outputs allowable stress and maximum axial load
	set	Outputs only allowable axial load
Flag 02	clear	Warning BEEP disabled
	set	Warning BEEP enabled
Flag 04	clear	AASHTO
	set	AISC

INPUT SUMMARY

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
WSD?	Asking if the analysis is by working stress (ASD) methods. Answer "0" (no) to analyze by AASHTO LFD or AISC LRFD rules.	1 (WSD)
F-Y KSI	Steel yield stress, ksi	36 ksi
AX P, K	Axial load on the member in kips	No default
M, 'K	Moment on the member in footkips	No default
KL-X"	Major axis effective length, in.	No default
KL-Y"	Minor axis effective length, in.	KL-X
AREA	Trial member area, sq. inches. Bypass to re-use the previous section (e.g. after softkey D).	Previous Section
R-X	Major axis rad. of gyration, in.	No default
R-Y	Minor axis radius of gyration	R-X

OUTPUT SUMMARY (Items in parenthesis are displayed only if flag 00 is clear)

INTRX Interaction in the X or Y direction. Should be less than unity.

PROGRAM FLAGS

Flag 06	clear	WSD (ASD)
	set	LFD (LRFD)

Flag 18	clear	WSD (ASD)
	set	LFD (LRFD)

STORAGE REGISTER USE

<u>Register</u>	<u>Value</u>
00	not used
01	KL-X (inches stored)
02	KL-Y (inches stored)
03	R-X
04	AREA, square inches
05	R-Y
06	F-Y KSI
07	
08	
09	
10	
11	
12	
13	
14	
15	
16	
17	



SBCOL FR -3

D 125

SBCCL 1
D 126

SBCOL-2

7127

SBCOL ex.

DISE

SBC02 ex

D129

SBCOL List 1

D 130

SBCOL List 2

D131

SBC02 List 3

D132

COMPOSITE BEAM ANALYSIS

xeq COMP (SIZE 028)

SOFTKEYS (USER mode ON)

B	New beam input
D	New slab dimension input
E	New moment input

Note: if designing by AASHTO, do not use these softkeys during "N=3N" calculations; allow the program to return to a new beam prompt first.

USER FLAGS

Flag 00	clear	Full output
	set	Only live load capacity, or sums of stresses on flanges
Flag 02	clear	Warning BEEP disabled
	set	Warning BEEP enabled
Flag 04	clear	AASHTO
	set	AISC/ACI

INPUT SUMMARY

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
WSD?	Asking if designing by working stress or load factor design. Respond 1 (yes) for WSD, 0 (no) for LFD/LRFD.	1 (WSD)
PLAST? (LFD/LRFD only)	Asking if section is compact enough to be analyzed by plastic rather than elastic methods. Respond 1 for "yes", 0 for "no." Elastic section properties will still be computed but capacity will be based on plastic properties.	0 (no)
N	Modular ratio, E_{st}/E_{conc}	8
Fc'KSI	Ultimate concrete cylinder strength in kips per square inch	4.0 ksi
F-Y KSI	Steel yield stress, for shear and plastic calculations	36 ksi
Fb KSI (Elastic analysis only)	Allowable steel bending stress. If by LFD, will be near or equal to yield stress.	F_y

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
(STUD D")	Shear stud diameter in inches. Bypass to input capacity directly.	CAP prompt
(STUD CAP, K)	Shear capacity of one shear stud from AISC table 1.11.4, or as calculated by the user for special connectors (e.g. channels) or for reduced capacity due to formed metal deck ribs	No default
SL TH"	Slab thickness, inches	No default
BM SPA'	Beam spacing in feet	No default
SL Be=..."	Displays calculated effective slab width. If OK, press R/S to continue. If not, key in the correct width and press R/S. Does <u>not</u> include slab thickness ($B_t + 16 \times Th$) for AISC ASD.	Displayed
H"	Haunch or stool between the <u>top</u> of the top flange and bottom of slab, or the rib dimension for transverse metal deck, in inches	Zero
NCDL M, 'K	Non-composite dead load moment (load applied before slab has achieved its strength), in foot-kips	Zero
CDL M, 'K	Composite dead load moment (e.g. from guard rail, future wearing surface, sidewalks, etc.)	Zero
LL M, 'K	Live load moment in foot-kips. Bypass (or enter zero) to find remaining live load capacity.	Zero
WEB D"	Web depth (inside of flanges) in inches. <u>Bypass for rolled beam input.</u>	Zero

If inputting a built-up plate member:

WEB TH"	Web thickness in inches.	No default
COM FL W"	Compression flange width, inches. Bypass for rolled beam input.	Zero
TH"	Compression flange thickness, in.	No default
TENS FL W"	Tension flange width in inches	No default
TH"	Tension flange thickness, inches	No default

If inputting a rolled beam:

BM A SQ"	Beam cross-sectional area in square inches	No default
BM D"	Beam depth in inches (out to out)	No default
Y"	Neutral axis position, from the top flange. Used only for cover-plated or rivetted beams; bypass for symmetrical (rolled) beams.	(BM D")/2
I	Moment of inertia of the beam, in inches ⁴	No default

All beams:

#STUDS	Number of shear studs to be used between points of inflection. For a simple span, this is the number of studs on the entire beam. If input is greater than #STUDS REQD or prompt is bypassed, full composite action is assumed.	#STUDS REQD
--------	---	-------------

OUTPUT SUMMARY (Items in parenthesis are displayed only if flag 00 is clear)

(AASHTO or AISC)	Displays code being followed. If not right, change flag 04 and start again. If OK, press R/S to continue.
SL Be	Effective slab width in inches. If OK, press R/S to continue. If not, key in the correct width and press R/S.
(HOR V)	Horizontal shear, in kips (H/2 for AISC ASD)
(#STUDS REQD)	The minimum number of shear studs required to resist the HOR V, taken between points of inflection. See softkey note for HOR V.

If a built-up beam: (if WEB D" prompt was answered)

(I)	Moment of inertia of the steel beam
(Y)	Neutral axis position of the bare beam, in inches, measured from the compression flange
(COM S)	Section modulus of the compression flange, in inches ³
(TEN S)	Section modulus of the tension flange
(AREA)	Area of the bare beam, in square inches
(BM Z)	The plastic section modulus of the bare beam (only if by LFD/LRFD)

All beams:

- (a) The effective slab depth used in calculating section properties, as limited by the number of shear studs used.
- (Z) The composite plastic section modulus, in inches³, if plastic analysis was chosen
- PLAST M The full plastic moment capacity of the composite beam, multiplied by $\phi = 0.85$ if by AISC LRFD (if plastic analysis was chosen)
- LL M The live load moment capacity of the composite beam (if plastic analysis was chosen)
- (N) The modular ratio (n) used to calculate the (elastic) section properties which follow
- (I-COMP) Composite moment of inertia, inches⁴
- (Y) Neutral axis position, measured from the top of the slab, in inches
- (Sc) Section modulus of the concrete. Do not divide by N to use. See the examples.
- (TF S) Section modulus of the top (compression) flange, in inches³
- (BF S) Section modulus of the bottom (tension) flange
- (EFF S)
(AISC ASD only) The effective section modulus of the bottom flange, reduced due to insufficient shear connectors, by AISC ASD equation I2-1.
- (Q) (AASHTO only) Statical moment of the slab about the neutral axis, in cubic inches

Note: if designing by AASHTO, the above elastic section properties are repeated for a modular ratio of 3N per AASHTO 10.38.1.4. The values for Q, HOR V, and MIN # STUDS are not repeated.

If a live load moment was input:

- (B FL) The stresses which follow are on the bottom
(T FL) flange (or coverplate, if there is one), top
(SLAB) flange, or slab, respectively.

(NCDL STRS)	Stress due to non-composite dead load, compo-
(CDL STRS)	site dead load, live load, and the summation
(LL STRS)	of these stresses, respectively, in ksi, on
Σ BF STRS	the part indicated (top or bottom flange or
Σ TF STRS	slab).
Σ SL STRS	

If a live load moment of zero was input:

WS or LF LL M	The remaining allowable live load moment, by working stress or load factor design methods, governed by stresses on the bottom flange, top flange, or slab; in foot-kips. For example, WS LL M=107.95 'K means the remaining moment, calculated by working stress design, is 107.95 foot-kips. This will not be displayed if a plastic section is being used under LFD/LRFD.
---------------	---

PROGRAM FLAGS

Flag 05	clear set	Have a live load moment Solve for live load moment capacity
Flag 06	clear set	WSD/ASD LFD/LRFD
Flag 07	clear set	Modular ratio equals N Modular ratio equals 3N (AASHTO only)
Flag 08	clear set	Elastic analysis only Plastic analysis permitted <u>by user</u> (PLG)
Flag 09	clear set	Composite Non-composite (from PLG)
Flag 10	clear set	AISC LRFD, or AASHTO AISC ASD
Flag 11	set	Subroutine return
Flag 18	clear set	Elastic analysis only Plastic analysis permitted <u>by code</u> (PLG)

STORAGE REGISTER USE

<u>Register</u>	<u>COMP</u>	<u>BM</u>
00	Beam Depth	Beam depth
01	Beam Area; Allowable LL Moment	Beam area
02	Beam I (moment of inertia)	Beam I
03	Compr. Flange Width; Beam S_{bf} ; "TF" or "BF"	Comp. Fl. W.

<u>Register</u>	<u>COMP</u>	<u>BM</u>
04	Compr. Flange Th.; S_{tf} ($N = N$)	Comp. Fl. Th.
05	Y (neutral axis) of bare beam	Y
06	F_y	
07	Web Depth (built-up beams)	Web Depth
08	Web Thickness (built-up beams and some plastic sections)	Web Thickness
09	Tens. Fl. Width; Composite Y (compos. neutral axis position)	Tens. Fl. W.
10	Tens. Fl. Th.; Slab Compr. Force; S_{conc} ($N = N$)	Tens. Fl. Th.
11	(Slab + Haunch + Beam Y); $S_{bottom\ flange}$ ($N = 3N$)	Beam S_{tf}
12	Effective slab thickness (reduced for inadequate number of shear connectors)	Beam Z
13	Slab thickness	
14	Effective slab area/N	Beam S_{bf}
15	H + S (Haunch or stool + Slab thickness)	
16	$F_c' \times (.40, .45, \text{ or } .85)$	
17	NCDL Moment (inch-kips stored)	
18	F_b (allowable bending stress) or OF_y , ksi	
19	$S_{top\ flange}$ ($N = 3N$)	
20	N (modular ratio, E_{st}/E_{conc})	
21	CDL Moment (inch-kips stored)	
22	Number of studs required/provided	
23	$S_{bottom\ flange}$ ($N = N$)	
24	$.85 F_c'$	
25	One half of shear stud capacity, kips	
26	Live load moment (inch-kips stored)	
27	Slab Area	

COMPOSITE BEAM ANALYSIS

Program COMP analyzes composite concrete slab/steel beam sections. It will follow AISC (ASD or LRFD) or AASHTO (WSD or LFD) depending on the status of flag 04, and the user's response to the "WSD?" prompt. It will accept rolled beams with known section properties or built-up sections. If a built-up (welded plate) beam is used, COMP calculates and displays the section properties of the steel beam before calculating the composite properties. Only elastic section properties are calculated for WSD designs; however, under LFD/LRFD designs, the program will calculate plastic properties if desired.

COMP does not check compactness, slenderness, or strength (allowable stress) criteria; these must be checked or calculated by hand or by some other means (for instance, program PLG in this package). For plastic design, COMP is thus more useful for rolled beams, which are usually compact, while PLG is usually used for welded beams.

Code references are given throughout this documentation. For portions dealing with AISC ASD, the specification number given is for the Ninth Edition (green book), with the Eighth Edition (red book) specification in parenthesis following. This should help ease the transition to the new format followed in the Ninth Edition.

BEAM CAPACITY

COMP can output beam strength in two different ways. If the live load moment prompt LL M 'K is bypassed (or answered zero) the program calculates the composite section properties, subtracts the dead load stresses from the allowable stresses (yield stress for compact beams), and then finds the allowable live load moment which can be applied. Method of calculation of dead load stresses is addressed in the following section.

If the actual live load moment is input, on the other hand, the program calculates the stresses due to NCDL (non-composite dead load), CDL (composite dead load) and LL (live load), and checks the sum of these (on the bottom flange, top flange, and slab--see following paragraphs) against the allowable stress input. The former mode is useful for rating of existing structures and preliminary rough-out work; the latter mode is intended for detailed final design of beams.

Elastic Design: AASHTO/AISC

AASHTO and AISC treat live load capacity somewhat differently under WSD rules. For design by AISC ASD, the program considers the entire composite section under the combination of dead and live loads, and CDL stress is figured on the same section as the LL stress. The output stresses for NCDL are those on the bare beam, but the stresses for combined dead and live load (BF and _ TF STRS) are calculated with the full composite section acting. Thus, the individual stresses on the beam due to NCDL, CDL, and LL will not add up to the summation given. Although this seems rather awkward at first, it is useful because it gives both the NCDL stress on the bare beam (to help the designer check the beam adequacy under NCDL) and the stress produced by all loads, acting on the composite section, for checking the stresses under full load (or checking the remaining live load capacity). COMP does not check Eighth Edition AISC equation 1.11-2; if the ratio of live to dead load is too high, the results will be incorrect and unconservative. The user must verify that this ratio is within the limits prescribed. The Ninth Edition ASD specification has dropped this rule and requires only that the superimposed stresses be less than $.9 F_y$. The user must also check this, by adding up the output stresses.

For design by AASHTO, the procedure is a little more logical, if somewhat more involved. NCDL moment is applied to the bare beam. CDL moment is applied to a composite section figured with a modular ratio of $3N$ for the top and bottom flanges, and N for the slab (per AASHTO 10.38.1.4, to account for creep). The live load moment is applied to, or figured from, the composite section calculated using a modular ratio of N . The sum of the stresses on any part (top flange, bottom flange, or slab) is calculated directly by adding these. Thus, design by AASHTO involves calculating two sets of composite properties; for AISC, only one set is needed. For an AASHTO design, the further refinement of considering two different slabs, one including a topping thickness (for live load, $N = N$) and one not including this extra thickness (for CDL, $N = 3N$), was not instrumented. This was left out both for simplicity, and because it's usually slightly conservative to consider only the final slab dimensions. Since contractors are usually left the option of constructing a slab as a single pour, it would be slightly unconservative to consider two different sets of slab dimensions.

For LFD/LRFD, the program also uses elastic properties if the PLAST? prompt was answered 0 (no) or bypassed (i.e. for non-compact beams). The same methodology is used as described above with one exception: under AISC LRFD rules, the summation of

stresses is figured from the superposition of elastic stresses. That is, the dead load (NCDL) is taken on the bare beam instead of being combined with live load and composite dead load on the composite section. An undercapacity (ϕ) factor of 0.90 is applied to the resulting LL capacity if the LL M input prompt was bypassed. (Actually, the factor of 0.90 is applied by the program only to the allowable steel stress; if the concrete strength governs, the results will be incorrect by this amount. This happens only rarely.)

For AASHTO LFD analysis for non-compact beams, the methodology described above is followed exactly; the strength is taken as the allowable steel stress (usually close to yield) or the concrete strength ($0.85 f_c'$) times the appropriate elastic section modulus. No reduction (ϕ) factor is applied.

Plastic Design: AASHTO/AISC

If the "PLAST?" prompt is answered 1 (yes), COMP figures the strength based on a fully plastic stress distribution over the composite section. All loads are taken on this composite section. AISC LRFD uses an undercapacity (ϕ) factor of 0.85; AASHTO uses 1.0. As mentioned, COMP leaves it up to the user to determine that the section meets all compactness requirements. (When portions of COMP are called as subroutines by program PLG, the calling program determines this.)

For built-up beams, the program also calculates the plastic section modulus, BM Z, of the bare beam, for possible use by the designer. However, program COMP makes no use of this value (again, PLG does). Also, the program assumes that the plastic neutral axis falls within the web. For very unbalanced beams where one flange contains over half the area, this assumption is invalid, and COMP will not output a BM Z value.

For those following the examples in the AISC LRFD manual, the refinement of using the beam's fillet area (e.g. as shown on page 4-4) is not included in this program. However, for the condition of the plastic neutral axis (P.N.A.) falling within the web or flange of a rolled beam, the program will prompt for the flange and web dimensions in order to calculate the P.N.A. position. For built-up plate beams, this is not done since the program already has these dimensions.

INPUT SEQUENCE

The program is written to check several different beams in succession with a given slab configuration. To enter a built-up (welded) beam, answer the web and flange dimension prompts. To enter a rolled beam,

bypass the WEB D" prompt (or enter zero) and the program will prompt for the tabulated elastic section properties. COMP is not able to directly figure the properties of riveted beams built up with angle (L) shapes in the flanges, or rolled beams with coverplates; for these beams, the designer must calculate the section properties by hand and enter them as for a rolled beam. (The program's neutral axis prompt, Y", is expressly for this purpose. For symmetric rolled beams, it can be bypassed.)

COMP prompts for slab thickness, span length, and beam spacing. Then it calculates the effective slab width (for an interior beam) from these parameters and displays it, using the minimum of:

Span/4
Beam spacing
12 x Slab thickness (AASHTO only)

Note that the Eighth Edition AISC ASD criteria (Flange width plus 16 x Slab thickness) is not used, at the recommendation of LRFD Commentary section I3.1 and the Ninth Edition ASD.

If the effective width is correct, press R/S to continue; if not (for instance, for an edge beam) key in the correct value and press R/S. For a given beam, if the calculated neutral axis is below the bottom of the slab, the entire slab is considered effective in compression; if the neutral axis is within the slab, only that portion above the axis is considered. The number of shear studs used also can affect the neutral axis position; see the following paragraphs.

If the neutral axis is above the top flange of the beam (i.e. in the slab or haunch), the composite section modulus of the top flange calculates out as a negative number, and the stress on the top flange due to live load is subtracted from the dead load stress. Slab reinforcing is not taken into account, nor is any concrete in the haunch between the beam and slab considered (although the separation between beam and slab due to the haunch is taken into account). Thus, transverse formed metal deck may be accommodated by entering the height of the ribs for the H" prompt. The slab thickness to be input would then be that amount above the ribs. Formed metal deck with the ribs running along the beam is a little more difficult. The simplest procedure is probably to ignore the concrete in the ribs and treat them as a haunch, the same as for transverse ribs. This is conservative.

If the beam is slightly embedded in the slab (i.e. a negative haunch), the "negative area" of concrete in the haunch is still neglected. That is, the program

considers there to be concrete and a beam flange in the same place. This, of course, is slightly unconservative if the neutral axis falls beneath the top of the top flange. For embedded beams with the N.A. in the beam, the user should reduce the input slab thickness to obtain zero embedment.

SHEAR CONNECTORS

COMP also calculates shear connector requirements using ultimate strengths of the slab and beam. If designing by AASHTO or AISC LRFD, the stud diameter is input, and the capacity and required number of studs is calculated by AASHTO equation 10-66 or LRFD equation I5-1. Research has shown that concrete strengths above 4000 psi do not increase connector strength. However, the program does not take this into account. If using concrete stronger than 4000 psi, see the next paragraph.

If designing by AISC ASD, the shear connector capacity is input directly from Table I4.1 (1.11.4) at the STUD CAP (connector capacity) prompt. When designing by LRFD or AASHTO, if using something other than standard shear studs at full strength (e.g. channel connectors, or studs at reduced strength due to formed metal deck ribs); or if the input slab concrete strength is greater than 4000 psi; bypass the STUD D" (diameter) prompt in order to get the STUD CAP prompt and key in a strength value directly. (That is, the program does not calculate a reduction per LRFD I3-1.)

The program does not check AASHTO fatigue requirements; however, the statical inertia Q is displayed if flag 04 is clear. For partial connection, this value of Q is still figured on the full slab area if the slab governs. If flag 00 is set, this shear output is suppressed.

The requirements of AISC ASD equation I4-5 (1.11-7) regarding moments due to concentrated loads are not checked, nor are those of AASHTO 10.38.5.1.3 regarding negative moment regions. The designer must check vertical shear on the steel beam web; COMP does not do this.

Partial shear connection is handled differently depending on which code is being followed. If designing by AISC LRFD or AASHTO, the program simply reduces the effective slab area by the ratio of actual to required shear connectors. This is done by reducing the effective thickness of the slab rather than the width; the haunch dimension is increased accordingly. The difference between the two methods is small.

If designing by AISC ASD, formula I4-4 (1.11-1) is

used, interpolating the strength between that of the fully composite section and the bare beam. However, the slab used to figure the "fully composite section" is first reduced as explained in the preceding paragraph. This minor error will cause a small discrepancy when checking hand calculations. The error is conservative and typically is very small, on the order of two percent.

ALLOWABLE STRESSES

As with the other structural programs in this module, the concrete strength prompted for, F_c' KSI, is the ultimate strength, whether designing by WSD or LFD. The steel stress F_b KSI is the working stress to be used; for load factor designs it will be close to or equal to the yield stress. F_y KSI (yield) is used only for the shear connector design and plastic design.

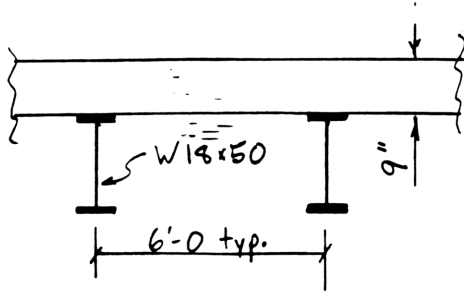
OTHER CONSIDERATIONS

Shored construction can be analyzed by simply considering the dead loads (NCDL and CDL) as part of the live load, and inputting them as zero.

COMP does not check span/depth ratios per AASHTO 10.5; the designer should do this. It also will not handle negative-moment areas (i.e. a beam acting compositely with slab reinforcing).

When using program PLG to analyze a welded composite beam, the flange width to thickness requirement is assumed to be met. The AASHTO requirements for $b/2t_f$ related to dead load stress (10.34.2.1.5 and 10.50c) are not checked in this situation by either program COMP or PLG and should be reviewed by the designer. AASHTO 10.50(d), related to dead load moment, also must be checked by the designer. Program PLG calculates the constant R_b for this purpose.

COMP does not check vertical shear on the web. Program PLG may be used for this. Neither program checks AASHTO 10.50(e), concerning tension field action in the web prior to hardening of the slab.

Composite Section Properties

	A	d	Ad
Slab $72" \times 9" / n=8$	81.00	4.5"	364.5
Beam $9" \times 17.99" / 2$	14.7	17.995	264.53
	95.7	6.57"	629.03

Trial N.A. = 6.57" < 9" \Rightarrow Cracked section

$$\frac{72}{n=8} \times Y \cdot \left(\frac{1}{2} Y\right) = 14.7 \times \left(\frac{17.99}{2} + 9 - Y\right)$$

$$4.5 Y^2 = -14.7 Y + 264.53$$

$$Y = \frac{-14.7 + \sqrt{14.7^2 - 4 \times 4.5 (-264.53)}}{2 \times 4.5}$$

$$= \underline{6.2058"}$$

$$F_y = 36 \text{ ksi}; f_y = 24 \text{ ksi}$$

$$f_c' = 4.0 \text{ ksi}; n = 8$$

Solve for Live Load capacity
by AISC, Span = 26'-0".

$$\begin{aligned} W18 \times 50: d &= 17.99" \\ A &= 14.7 \text{ in}^2 \\ I &= 800 \text{ in}^4 \end{aligned}$$

Effective slab width:

$$\begin{aligned} \text{Span}/4 &= 26'/4 = 7'8" \\ \text{Beam Spa.} &= 7'2" \leftarrow \text{Governs} \\ \text{Slab } 16 \times 9" &+ 7\frac{1}{2}" = 151.5" \end{aligned}$$

$$\begin{aligned} \text{DL: } 6' \times 7'5" \times .15 &= .675 \text{ k/ft} \\ \text{Beam } &\underline{.05} \\ &= .725 \end{aligned}$$

$$\text{DL M} = .725 \text{ k/ft} \times \frac{(26')^2}{8} = 61.26 \text{ k-ft}$$

Check erection stress:

$$(61.26 \times 12) \times \left(\frac{17.99}{2}\right) / 800 \text{ in}^4 = 4.27 \text{ ksi OK}$$

Re-Calc. Composite Section Properties

	A	d	Ad^2	I_o
Slab	55.65	3.10	537.73	179.25
Beam	14.7	11.79"	2043.10	800.0
			2580.83	979.25
			3560.08	

$$S_{\text{slab}} = \frac{3560.08}{6.2058} \times (n=8) = 4589.34 \text{ in}^3$$

$$S_{\text{T-Fl.}} = \frac{3560.08}{6.2058 - 9"} = -1274.08 \text{ in}^3$$

(minus sign indicates top fl. in tension)

$$S_{\text{B-Fl.}} = \frac{3560.08}{17.99" + 9" - 6.2058"} = 171.29 \text{ in}^3$$

Live Load Capacity

$$\left(\frac{24 \text{ ksi} \times 171.29}{12} \right) - 61.26 \text{ k-ft}$$

$$= \underline{281.31 \text{ k-ft}}$$

Design Shear Studs

$$\text{Slab: } (.45 \times 4 \text{ ksi}) \frac{72''}{4} \times 9''/2 = 137.70 \text{ k}$$

$$\text{Beam: } 14.7 \square'' \times 36 \text{ ksi} / 2 = 264.60 \text{ k} \leftarrow \text{Governs}$$

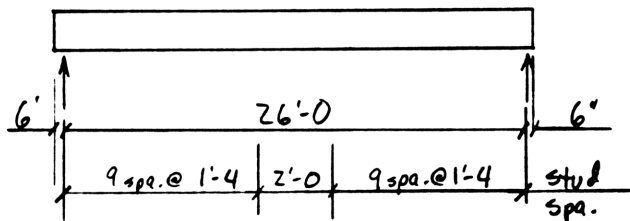
Use $\frac{3}{4}'' \phi$ studs; capacity = 13.3 k/stud
(Table 1.11.4)

$$\text{Min. \# studs} = 19.89 \text{ each side of } \angle$$

USE TWO ROWS OF 20 - $\frac{3}{4}'' \phi$ STUDS

Check diameter: flange th. = .57"

$$.57'' \times 2 \frac{1}{2} = 1.425'' < \frac{3}{4}'' \phi \quad \underline{\text{OK}}$$



Check w/ 2 rows of 15 studs

$$S_{\text{eff}} = 88.94 \text{ in}^3 + \sqrt{\frac{15}{19.99}} (171.29 - 88.94)$$

$$= \underline{160.45 \text{ in}^3}$$

$$\text{Live Load: } 160.45 \times 24 \text{ ksi} / 12 = 320.90$$

$$\quad \quad \quad \underline{- 61.26}$$

$$\quad \quad \quad 259.64 \text{ k}$$

42 381 50 SHEETS 3 SQUARE
42 382 100 SHEETS 3 SQUARE
42 383 200 SHEETS 3 SQUARE



XEQ "COMP"
8:55:58 AM 05/16/89

TITLE:

RUN

AISC

MSD?

RUN

N

8.0000

RUN

Fc'KSI

4.0000

RUN

STUD CAP, K

13.3000

RUN

F-Y KSI

36.0000

RUN

Fb KSI

24.0000

RUN

SL TH"

9.0000

RUN

BM SPA'

6.0000

RUN

SPAN'

26.0000

RUN

SL Be=72.0000"

RUN

H"

0.0000

RUN

NCDL M, 'K

61.2600

RUN

CDL M, 'K

0.0000

RUN

LL M, 'K

RUN

WEB D"

RUN

BM A SQ"

14.7000

RUN

BM D"

17.9900

RUN

Y"

RUN

I

800.0000

RUN

HOR V=264.6000 K

#STUDS REQD=39.7895

#STUDS

RUN

N=8

I-COMP=3,560.0754

Y=6.2058"

Sc=4,589.3792

TF S=-1,274.0782

BF S=171.2873

B FL

NCDL STRS=8.2655 KSI

T FL

NCDL STRS=8.2655 KSI

MS LL M=281.3146 'K

WEB D"

RUN

BM A SQ"

14.7000

RUN

BM D"

17.9900

RUN

Y"

RUN

I

800.0000

RUN

HOR V=264.6000 K

#STUDS REQD=39.7895

#STUDS

30.0000

RUN

N=8

I-COMP=3,560.0754

Y=6.2058"

Sc=4,589.3792

TF S=-1,274.0782

BF S=171.2873

EFF S=160.4430

B FL

NCDL STRS=8.2655 KSI

T FL

NCDL STRS=8.2655 KSI

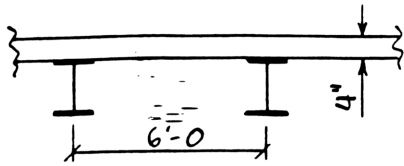
MS LL M=259.6261 'K

WEB D"

Use less than
full shear
connection

Effective S
for reduced
number of
studs

D147



$$\text{Span} = 30'-0$$

$$\text{Bm Spa.} = 6'-0$$

$$\text{Slab Th.} = 4"$$

$$\text{Live Load} = 100 \text{ psf}$$

$$\text{Partition + Ceiling Load} \\ = 30 \text{ psf}$$

$$f_c' = 3.0 \text{ ksi}$$

$$n = 9$$

$$f_y = 36 \text{ ksi}$$

$$f_{\text{pr}} = 24 \text{ ksi}$$

Check Shear

$$\frac{16.40 \text{ k}}{.275 \times 15.48} = 3.85 \text{ ksi} < .40 f_y \text{ OK}$$

Note: COMP does not check shear.

Check DL on bare beam

$$\frac{38.25''^4 \times 12}{47.2 \text{ in}^3} = 9.72 \text{ ksi} < 24 \text{ ksi} \text{ OK}$$

Calculate Composite Properties

	A	d	Ad	\bar{x}	$A\bar{x}^2$	I.
Slab	31.11	2"	62.22	2.25"	157.96	41.48
Beam	9.12	11.94"	108.89	7.69"	538.86	375
	40.23		171.11		696.82	416.4
					1113.30	

$$\frac{171.11}{40.23} = 4.25'' \Rightarrow \text{N.A. below slab}$$

$$I_{\text{comp}} = 1113.30 \text{ in}^4$$

$$S_{\text{conc}} = \frac{1113.30}{4.25} \cdot (n-9) = 355.75$$

$$S_{\text{TF}} = \frac{1113.30}{4.25 - 4"} = 4395.18 \text{ in}^3$$

$$S_{\text{BF}} = \frac{1113.30}{15.48 + 4" - 4.25"} = 71.21 \text{ in}^3$$

Effective Slab Width

$$\text{Span}/4 = 30' \times 12/4 = 90"$$

$$\text{Bm Spa.} = 6' \times 12 = 72"$$

$$16 t_{\text{sl}} + \text{Flange} = 16 \times 4" + 6" = 70" \leftarrow$$

$$b_{\text{eff}} = 70"; \text{ effective Area} = \frac{70"}{n=9} \cdot 4" \\ = 31.11 \text{ in}^2$$

TRY W 16x31

$$A = 9.12 \text{ in}^2$$

$$d = 15.48"$$

$$I_x = 375 \text{ in}^4$$

$$S_x = 47.2 \text{ in}^3$$

$$t_w = .275"$$

$$b_f = 5.53" \approx 6" \text{ OK}$$

$$\begin{aligned} \leq M &= 38.25 \text{ NCDL} \\ &20.25 \text{ CDL} \\ &67.50 \text{ LL} \\ &126.00''^k \end{aligned}$$

$$\text{Slab: } \frac{(20.25 + 67.50) \times 12}{2355.75} = .447 \text{ ksi} < .45 f_c' \text{ OK}$$

$$\text{Bottom Fl: } \frac{126.0''^k \times 12}{71.24} = 21.22 \text{ ksi} < 24 \text{ OK}$$

$$\text{Top Fl: OK by inspection}$$

$$\text{Check: } S_{\text{max}} = (1.35 + .35 \frac{(67.50 + 20.25)}{38.25}) \times 47.2'' \\ = 101.62 > 71.24 \text{ OK}$$

Design Shear ConnectorsUse $\frac{3}{4} \times 3$ Studs (2 lines)

$$\text{Max stud } \phi = 2.5 \cdot t_{cl} = 2.5 \cdot .440$$

$$= 1.1 \text{ " OK}$$

Horiz. Spacing (do not use part. comp. action)

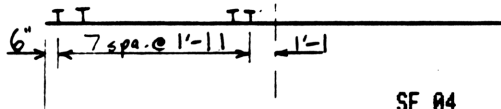
$$(.85 \cdot 3.06 \cdot 70) \cdot 4 \cdot \frac{1}{2} = 357.0 \text{ k}$$

$$36 \text{ ksi} \cdot 9.12 \text{ in} / 2 = 164.16 \text{ k} \leftarrow$$

$$\frac{3}{4} \times 3 \text{ studs} \rightarrow 11.5 \text{ kips/stud}$$

$$\frac{164.16 \text{ k}}{11.5} = 14.27 \text{ studs per half}$$

\therefore Use two rows of 8 studs
per half beam
Sym. about Beam E



SF 04

XEQ "COMP"

8:41:50 AM 05/16/89

TITLE:
OFFICE BLDG FLOOR DESIGN
RUN

AISC
WSD?

1.0000 RUN

N

9.0000 RUN

Fc KSI

3.0000 RUN

STUD CAP, K

11.5000 RUN

F-Y KSI

36.0000 RUN

Fb KSI

24.0000 RUN

SL TH

4.0000 RUN

BM SPA

6.0000 RUN

SPAN' 30.0000 RUN

SL Be=72.0000"

70.0000 RUN

H" 0.0000 RUN

NCDL M, 'K

38.2500 RUN

CDL M, 'K

20.2500 RUN

LL M, 'K

67.5000 RUN

WEB D"

RUN

BM A SQ"

9.1200 RUN

BM D"

15.8800 RUN

Y"

RUN

I

375.0000 RUN

HOR V=164.1600 K

#STUDS REQD=28.5496

#STUDS

RUN

N=9

I-COMP=1,113.3023

Y=4.2533"

Sc=2,355.7516

TF S=4,395.1765

BF S=71.2436

B FL

NCDL STRS=9.7186 KSI

CDL STRS=3.4108 KSI

LL STRS=11.3694 KSI

 Σ B FL STRS=21.2230 KSI

T FL

NCDL STRS=9.7186 KSI

CDL STRS=0.0553 KSI

LL STRS=0.1843 KSI

 Σ T FL STRS=0.3440 KSI

SLAB

CDL STRS=0.1032 KSI

LL STRS=0.3438 KSI

 Σ SLAB STRS=0.4470 KSI

WEB D"

Non-composite DL

Composite DL

Live load

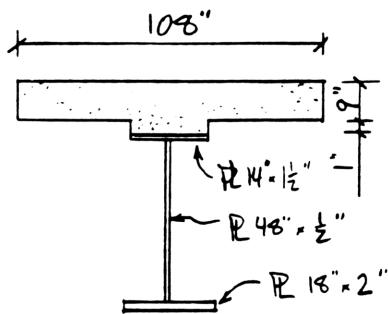
by pass for
rolled beamcan bypass for
symm. beamNCDL stress on
bare beam Σ on composite
section

6150

Calculate plastic section properties of this beam. Find allowable load by AASHTO and AISC LRFD.

$$F_y = 36 \text{ ksi}$$

$$f_c' = 4.0 \text{ ksi}$$



Check Compactness - $\frac{D}{t_w} = \frac{48}{.5} = 96.0$

If by AASHTO, $\frac{D}{t_w} < \frac{19,230}{\sqrt{F_y}} = 101.35$ - OK

If by AISC, $\frac{H_c}{t_w} < \frac{640}{\sqrt{F_y}} = 106.67$ - OK

Composite section is considered compact by either code. Non-composite section would require additional compression flange checks.

Note: Program COMP does not check compactness; the user must do this. Program PLG does check compactness.

Calculate Z of bare beam:

$$14 \times 1.5 = 21.0$$

$$48 \times 0.5 = 24.0$$

$$18 \times 2.0 = 36.0$$

$$\text{Area} = 81.0 \text{ in}^2$$

$$\frac{81.0}{2} - 21.0 \text{ in} = 19.5 \text{ in}$$

$$\frac{19.5}{0.5} = 39.0 \text{ in}, + 1\frac{1}{2} \text{ in} = 40.5 \text{ in to N.A.}$$

Bare Beam (cont'd)

$$\text{Top half C.G.} : 21.0 \times .75 = 15.75$$

$$\frac{19.5 \times 21.0}{40.5 \times 10.5} = \frac{409.50}{425.25}$$

$$\text{Bottom half C.G.} : 36.0 \times 1.0 = 36.0$$

$$\frac{4.5 \times 6.5}{40.5 \times 1.611} = \frac{29.25}{65.25}$$

$$Z = 40.5 \text{ in} \times (51.5 \text{ in} - 1.611 \text{ in} - 10.5 \text{ in})$$

$$= \underline{\underline{1595.25 \text{ in}^3}}$$

Composite Beam Z:

$$C = 108 \text{ in} \times 9 \text{ in} \times (.85 \times 4.0 \text{ ksi}) = 3304.8 \text{ k}$$

$$T = 81.0 \text{ in}^2 \times 36 \text{ ksi} = 2916.0 \text{ k}$$

\therefore N.A. is in slab

$$a = \frac{2916.0}{3304.8} \times 9 \text{ in} = \underline{\underline{7.9412 \text{ in}}}$$

$$\text{Slab } \frac{1}{2} \times (7.9412 \text{ in}) \times 108 \text{ in} \times \frac{(.85 \times 4)}{36 \text{ ksi}} = 321.62$$

$$\text{Top Fl. } 21 \text{ in} \times (1.0588 \text{ in} + 1 \text{ in} + 3/4 \text{ in}) = 58.99$$

$$\text{Web } 24 \text{ in} \times (1.0588 \text{ in} + 1 \text{ in} + 1/2 \text{ in} + 24 \text{ in}) = 661.41$$

$$\text{Bot. Fl. } 36 \text{ in} \times (1.0588 \text{ in} + 50.5 \text{ in} + 1 \text{ in}) = 1892.12$$

$$Z = \underline{\underline{2934.13 \text{ in}^3}}$$

$$\text{AASHTO Capacity} = F_y Z = 36.0 \times 2934.13 \times \frac{1}{2}$$

$$= \underline{\underline{5281.44 \text{ k}}}$$

$$\text{Capacity by AISC LRFD} = \phi F_y Z$$

$$= .85 \times 5281.44 = \underline{\underline{4489.22 \text{ k}}}$$

42 381 20 SHEETS 5 SQUARE
42 382 100 SHEETS 5 SQUARE
42 383 200 SHEETS 5 SQUARE



SF 04
XEQ "COMP"
8:12:15 AM 05/16/89

TITLE:
PLASTIC SECTION TRIALS
RUN

AISC
MSD?

0.0000 RUN

PLAST?

1.0000 RUN

N

8.0000 RUN

Fc' KSI

4.0000 RUN

STUD D"

1.0000 RUN

F-Y KSI

36.0000 RUN

SL TH"

9.0000 RUN

BM SPA'

9.0000 RUN

SPAN'

100.0000 RUN

SL Be=100.0000"

RUN

H"

1.0000 RUN

NCDL M, 'K

1,610.0000 RUN

CDL M, 'K

800.0000 RUN

LL M, 'K

RUN

WEB D"

48.0000 RUN

WEB TH"

.5000 RUN

COM FL W" 14.0000 RUN

TH" 1.5000 RUN

TEN FL W" 18.0000 RUN

TH" 2.0000 RUN

I=38,202.6875

Y=30.1944"

COM S=1,265.2224

TEN S=1,793.0857

BM Z=1,595.2500

AREA=81.0000 SQ"

HOR V=2,916.0000 K

#STUDS REQD=123.7154

#STUDS RUN

Z=2,934.1324

PLAST M=7,482.0375 'K

LL M=5,072.0375 'K

N=8

I-COMP=100,943.7500

Y=18.7778"

Sc=43,005.6213

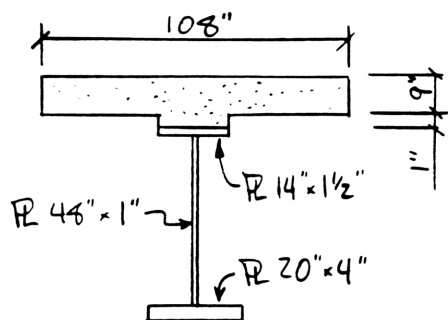
TF S=11,499.9209

BF S=2,362.7926

Calculate composite plastic section properties of this beam, and capacities by AASHTO and AISC LRFD.

$$F_y = 36 \text{ ksi}$$

$$f_c' = 4.0 \text{ ksi}$$



$$\begin{aligned} \text{Bm Area: } 14 \times 1\frac{1}{2} &= 21.0 \\ 48 \times 1 &= 48.0 \\ 20 \times 4 &= 80.0 \\ \hline &149.0 \text{ in}^2 \end{aligned}$$

Note: COMP will not figure a BM Z for this beam since one flange contains over half the beam area.

$$C = 108 \times 9 \times (.45 \times 4.0) = 3304.8$$

$$T = 149.0 \text{ in}^2 \times 36 = 5634.0$$

$$C' = (5634.0 - 3304.8) / 2 = 1029.6 \text{ k}$$

$$\frac{1029.6}{36 \text{ ksi}} = 28.6 \text{ in} > 21.0 \text{ in}$$

\Rightarrow N.A. in Web

$$\frac{28.6 - 21.0}{1} = 7.60"$$

$$\begin{aligned} \text{Slab } \left(\frac{3304.8}{36 \text{ ksi}} \right) \times (7.6 \times 1.5 + 1 + 4.5) &= 1340.28 \\ \text{Top Fl. } 21.0 \times (7.60 + 7.5) &= 175.35 \\ \text{Web } 7.60 \times 1 \times (7.60 / 2) &= 28.88 \\ \text{Web } 40.4 \times 1 \times (40.4 / 2) &= 816.08 \\ \text{B.Fl. } 80 \text{ in}^2 \times (40.4 + 2) &= 3392.00 \\ \hline \text{Comp. Z} &= 5752.59 \text{ in}^3 \end{aligned}$$

$$\begin{aligned} \text{AASHTO Capacity} &= F_y Z \\ &= 36.0 \times 5752.59 / 12 \\ &= \underline{\underline{17,257.8 \text{ k}}} \end{aligned}$$

$$\begin{aligned} \text{AISC Capacity} &= \phi F_y Z \\ &= .95 \times 17,257.8 = \underline{\underline{14,669.1 \text{ k}}} \end{aligned}$$

WEB D"	48.0000	RUN
WEB TH"	1.0000	RUN
COM FL W"	14.0000	RUN
TH"	1.5000	RUN
TEN FL W"	20.0000	RUN
TH"	4.0000	RUN
I=59,932.2955		
Y=35.9715"		
COM S=1,666.1061		
TEN S=3,419.1297		

AREA=149.0000 SQ"

HOR V=3,304.8000 K
#STUDS REQD=140.2108

#STUDS

RUN

Z=5,752.5900
PLAST M=14,669.1045 K
LL M=12,259.1045 K

N=8

I-COMP=175,857.4426
Y=27.3438"
Sc=51,458.7545
TF S=10,139.4945
BF S=4,863.8264

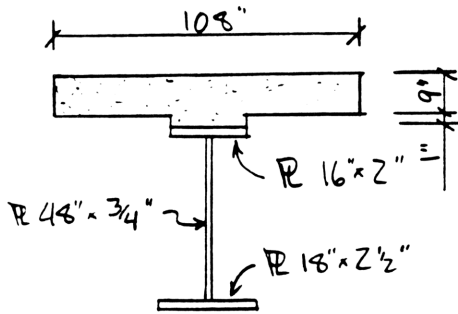
WEB D"

(153)

Calculate composite plastic section properties of this beam, and capacities by AISC LRFD and AASHTO.

$$F_y = 36 \text{ ksi}$$

$$f'_c = 4.0 \text{ ksi}$$



$$C = 9" \times 108" \times (.85 \times 4) = 3304.8$$

$$\begin{aligned} \text{Bm Area} &= 16 \times 2 = 32.0 \\ 48 \times 3/4 &= 36.0 \\ 18 \times 2\frac{1}{2} &= 45.0 \\ \hline &113.0 \text{ in}^2 \end{aligned}$$

$$T = 113.0 \times 36 \text{ ksi} = 4068.0 \text{ k}$$

$T > C \Rightarrow$ N.A. in beam

$$C' = \frac{4068 - 3304.8}{2} = 381.6 \text{ k}$$

$$\frac{381.6 \text{ k}}{36 \text{ ksi} \times 16"} = .6625" < 2"$$

\therefore N.A. is in flange

$$\begin{aligned} \text{Slab} & \frac{3304.8 \text{ k}}{36 \text{ ksi}} = 91.8 \text{ in} \times .6625" = 565.72 \text{ in}^2 \\ \text{Fl.} & .6625 \times 16 = 10.6 \text{ in} \times .3312" = 3.51 \text{ in}^2 \\ \text{Fl.} & 1.3375 \times 16 = 21.4 \text{ in} \times .6648" = 14.31 \text{ in}^2 \\ \text{Web} & 48 \times 3/4 = 36.0 \text{ in} \times .253375 = 912.15 \text{ in}^2 \\ \text{B.Fl.} & 18 \times 2\frac{1}{2} = 45.0 \text{ in} \times .505475 = 2276.44 \text{ in}^2 \end{aligned}$$

$$\text{Composite } Z = 3772.13$$

By AASHTO, capacity = $F_y Z$

$$= 36 \times 3772.13 / 12$$

$$= \underline{\underline{11,316.4 \text{ k}}}$$

By AISC LRFD, capacity = $\phi F_y Z$

$$= .85 \times 11,316.4 = \underline{\underline{9618.9 \text{ k}}}$$

WEB D"	48.0000	RUN
WEB TH"	.7500	RUN
COM FL W"	16.0000	RUN
TH"	2.0000	RUN
TEN FL W"	18.0000	RUN
TH"	2.5000	RUN
I=54,635.8497		
Y=28.9757"		
COM S=1,885.5772		
TEN S=2,322.5246		

$$\text{BM } Z=2,311.9167$$

$$\text{AREA}=113.0000 \text{ SQ"} \\ \text{-----}$$

$$\text{HOR V}=3,304.8000 \text{ K}$$

$$\# \text{STUDS REQD}=140.2108$$

$$\# \text{STUDS} \quad \text{RUN}$$

$$\text{-----}$$

$$Z=3,772.1275$$

$$\text{PLAST M}=9,618.9251 \text{ 'K}$$

$$\text{LL M}=7,208.9251 \text{ 'K}$$

$$\text{-----}$$

$$N=8$$

$$\text{I-COMP}=125,044.4220$$

$$Y=21.1130"$$

$$S_c=47,381.0009$$

$$\text{TF S}=11,252.0787$$

$$\text{BF S}=3,021.3459$$

01+LBL "COMP"

02 CF 09

03 CF 11

04 27

05 XEQ 31

06 "Fb KSI"

07 FC? 18

08 PROMPT

09 .9

10 FS? 04

11 FC? 06

12 SIGN

13 *

14 STO 18

15+LBL D

16 XROM "L"

17 "SL TH"

18 PROMPT

19 STO 13

20 "BM SPA"

21 PROMPT

22 "SPAN"

23 PROMPT

24 4

25 /

26 X>Y?

27 X<>Y

28 ENTER↑

29 FC? 04

30 RCL 13

31 X>Y?

32 X<>Y

33 12

34 *

35 "SL Be"

36 ARCL X

37 XEQ "AI"

38 PROMPT

39 RCL 13

40 *

41 STO 27

42 CLX

43 "H"

44 PROMPT

45 RCL 13

46 +

47 STO 15

48+LBL E

49 XROM "L"

50 "H"

51 XEQ 13

52 STO 17

53 CLA

54 XEQ 13

55 STO 21

56 CF 05

57 "LL"

58 XEQ 14

59 X=0?

60 SF 05

61 STO 26

62 FS? 11

63 RTN

64+LBL B

65 XROM "L"

66 CF 07

67 XEQ "BM"

68 X=0?

69 GTO 39

70 "BM A"

71 XROM "IS"

72 PROMPT

73 STO 01

74 "BM D"

75 PROMPT

76 STO 00

77 2

78 /

79 "Y"

80 PROMPT

81 STO 05

82 "I"

83 PROMPT

84 STO 02

85+LBL 39

86 RCL 24

87 RCL 27

88 *

89 STO 10

90 LASTX

91 RCL 20

92 /

93 STO 14

94 X<>Y

95 RCL 06

96 RCL 01

97 *

98 X>Y?

99 X<>Y

100 2

101 FC? 10

102 SIGN

103 /

104 "HOR V"

105 ARCL X

106 "T K"

107 FC? 00

108 XEQ "PV"

CDL

LLM

Bm A

d

Y

Y

I

.45f_c

sl. area

sl. force

area

n

a/n

force

F_y

A_{bm}

Tens. Compt.

force

hor. Shear

AISC ASD
uses H=
A·F_y/2

(155)

	X	Y	Z	T	-		X	Y	Z	T	L
109 RCL 25	Stud cap						164*LBL 14				
110 X#0?							165 CLX				
111 /	#studs						166 XROM "M"				
112 *STUDS REQD							167 PROMPT				
113 X#0?							168 *				
114 XEQ 02							169 RTN				
115 ADV											
116 ENTER↑	#studs #studs					"#STUDS"	170*LBL 01	sl.A Bm A			Neutral axis
117 ASTO 23							171 RCL 04	T.F.Th?			is in beam
118 CLA							172 X#0?				
119 ARCL 23							173 GTO 01				
120 X#0?							174 XEQ 11				
121 PROMPT	#avail #req'd						175 STO 03	T.F.W.			If there is a
122 X#0?							176 PROMPT				value for top
123 /	Req/avail						177 STO 04	T.F.Th.			flange thickness,
124 SIGN	1										STO 01. Otherwise
125 LASTX	Req/avail 1						178*LBL 01	T.F.Th.			(ie rolled beam)
126 X<Y?	#required					If more studs	179 RCL 03	T.F.W.			prompt for it.
127 SIGN						are furnished	180 *	Fl Area			
128 STO 22	*					than are req'd,	181 RCL 01	Bm A			
129 ST/ 10	*					consider only	182 RCL 10	sl.A			
130 ST/ 14	sl.th.					the number	183 -	ΔA			
131 RCL 13						req'd.	184 2				
132 X<Y							185 /	ΔA/2 Fl.A.			
133 /						*Reduces slab	186 X<Y?				
134 STO 12	eff sl.th.					area and compr.	187 GTO 01				
135 "a="						force if less	188 -	Δ			Neutral axis
136 FC? 10						studs are	189 STO Y	Δ	Δ		in web
137 X=Y?						richer than	190 RCL 08	Webth.			
138 X#0?						are required.	191 /	N.A.			
139 XEQ 23							192 RCL 05	Bm N.A.			
140 XROM "L"							193 RCL 04	T.F.Th Bm NA Pl. NA	Δ		
141 FC? 18							194 -				
142 GTO 29							195 ST+ X				
143 RCL 15	sl+h						196 +	Ze''	Δ		
144 RCL 01	Bm A						197 *	3rd term			
145 RCL 10	sl.compr.						198 CHS				
146 RCL 06	Fy						199 RCL 05	Bm N.A.			
147 /							200 ST+ X				
148 STO 10	*	Bm A				*Slab area	201 RCL 04	T.F.Th.			
149 X<Y?						transformed	202 -	Ze'			
150 GTO 01						to equiv. steel	203 RCL 03	Fl.W.			
151 /	Bm A	sl+h	sl+h	sl+h		area	204 LASTX	T.F.Th. Fl.W. Ze' 3rd term			
152 RCL 12	sl.th.						205 *	Fl. Ze' 3rd term			
153 *	eff sl.th.					Neutral axis	206 *	2nd term 3rd term			
154 2						is in flange	207 +	Steel term			
155 /							208 GTO 03				
156 -	arm										
157 RCL 05	Bm P						209*LBL 01				Neutral axis
158 +	arm						210 STO Y	ΔA/2 ΔA/2			in flange
159 RCL 01	Bm A						211 RCL 03	Fl.W.			
160 *	Moment						212 /				
161 GTO 09							213 CHS				
							214 RCL 05	Bm N.A.			
162*LBL 13							215 ST+ X				
163 *FCDL -							216 +	Ze' ΔA/2			
							217 *	Steel term			

	X	Y	Z	T	-		X	Y	Z	T	L
218*LBL 03	Steel term					272 *	$A \cdot d_{bm}$				
219 RCL 15	sl+h					273 CHS	$-A \cdot d$				
220 RCL 12	eff.sl.th.					274 LASTX	$B_m A$				
221 2	Z	th.	sl+h	steel term		275 RCL 14	$sl.A/n$	$B_m A$	$B_m A \cdot d$	sl.th.	
222 /	e_c					276 R+	$Z \cdot sl.th$	$sl.A/n$	$B_m A$	$B_m A \cdot d$	
223 -	\bar{Y}					277 ST+ X	$\frac{eff.W}{Z}$	$B_m A$	$-A \cdot d$		
224 RCL 05	e					278 /	\bar{Y}				
225 +	sl. force					279 XROM "Q"					
226 RCL 10	conc. term	steel term				280 STO 09					
227 *	Z					281*LBL 06	*				*
228 +	Z					282 ENTER↑	*	*			\bar{Y} or sl.thickness
229*LBL 09	Z					283 X↑2	* ²				(uncracked
230 "Z"						284 12					portion
231 XEQ 02						285 /	* ² / ₁₂	*			of slab)
232 RCL 06	F_y					286 RCL 14	sl.A/n		$\bar{Y}^2/12$	\bar{Y}	
233 *	M					287 RCL 12	sl.th.	A/n	$\bar{Y}^2/12$	\bar{Y}	
234 .85	ϕ					288 /	eff.W		$\bar{Y}^2/12$	\bar{Y}	
235 FC? 04						289 R+	\bar{Y}	eff.W	$\bar{Y}^2/12$	*	
236 SIGN						290 *	eff.sl.A	$\bar{Y}^2/12$	*	*	
237 *	ϕM					291 R+	*	Z	*	eff.sl.A	$\bar{Y}^2/12$
238 "PLAST"						292 2	Z	*			
239 XROM "M"						293 /	* ² / ₁₂				
240 XEQ "PV"						294 RCL 09	\bar{Y}				
241 LASTX	12					295 -	d				
242 *	$\phi M \cdot k$					296 X↑2	d^2	eff.A	$\bar{Y}^2/12$	$\bar{Y}^2/12$	
243 RCL 21	NCOL					297 R+	$\bar{Y}^2/12$	d^2	eff.A		
244 -						298 +					
245 RCL 17	COL					299 *	$I_{sl.}$				
246 -	LLM					300 RCL 09	\bar{Y}				
247 CLA						301 RCL 11	sl.th, but				
248 XEQ 05						302 -	$b_m d$				
						303 X↑2	d^2				
249*LBL 29						304 RCL 01	$B_m A$				
250 RCL 15	sl.h					305 *	$A \cdot d^2$	sl. I			
251 RCL 05	\bar{Y}					306 +	$Z \cdot I$				
252 +						307 RCL 02	$B_m I$				
253 STO 11	sl.h, \bar{Y}					308 +	I				
254 RCL 01	$B_m A$					309 ENTER↑	I	I			
255 *	$A \cdot d_{bm}$					310 "N"					
256 RCL 14	$sl.A/n$					311 RCL 20	n	I	I		
257 RCL 12	eff.sl.th.					312 XROM "AX"					
258 2						313 FC? 00					
259 /	$d_{sl.}$	$A_{sl.}$	$A \cdot d_{bm}$			314 AVIEW					
260 *	$A \cdot d_{sl.}$					315 ADV					
261 +	$\Sigma A \cdot d$					316 X<>Y	I	n	I		
262 RCL 01	$B_m A$					317 "I-COMP"					
263 RCL 14	$sl.A/n$					318 XEQ 02					
264 +	ΣA					319 RCL 09	\bar{Y}	I	n	I	
265 /	trial \bar{Y}					320 "Y"					
266 STO 09	\bar{Y}					321 XEQ 23					
267 RCL 12	eff.sl.th.					322 /	S_c	n	I	I	
268 X<=Y?						323 *	$S_c \cdot h$	I	I	I	
269 GTO 06						324 "Sc"					
270 RCL 11	sl.h, \bar{Y}					325 FC? 07					
271 RCL 01	$B_m A$					326 STO 10					
						327 XEQ 02					

Neutral Axis
in slab

	X	Y	Z	T	L		X	Y	Z	T	L
328 RDN	I	I	I	Se		383+LBL 01					
329 RCL 09	Y					384 FC? 07					
330 RCL 15	slab					385 FS? 04					
331 -	T.F.I.d	I	I	I		386 X<0?					
332 /	T.F.S					387 GTO 16					
333 STO 19						388 FS? 18					
334 FC? 07						389 GTO 16					
335 STO 04						390 XROM "LL"					
336 "TF"						391 RCL 17	NCDL M				
337 XEQ 22						392 RCL 03	S _{BF}				Bottom Flange
338 RDN	I					393 "B"					
339 RCL 00	Bm d					394 RCL 11	S _{3n} *				* Both are
340 LASTX	T.F.I.d					395 RCL 23	S _n *	S _{3n}	S _{6m}	NCDL	S _{BF} for
341 -	B.F.I.d	I				396 XEQ 20	Z f _b				AISC
342 /	B.F.I.S	I	I	I		397 RCL 18	allow F _b				
343 STO 11	(comp)					398 XEQ 21	act. f _b	allow F _b			
344 "BF"						399 -	Δ F _b				
345 XEQ 22						400 RCL 23	S _n				
346 FS? 07						401 *	LL M				
347 GTO 01						402 STO 01	allow LL M				
348 STO 23						403 RCL 17	NCDL M				
349 RCL 02	Bm I					404 RCL 02	Bm I				
350 RCL 00	Bm d					405 RCL 05	Bm Y				
351 RCL 05	Bm Y					406 /	Bm T.F.S				
352 -	B.F.I. dom					407 "T"					
353 /	B.F.I.S					408 RCL 19	S _{3n}				
354 STO 03	(care beam)					409 RCL 04	S _n	S _{3n}	S _{6m}	NCDL M	Top Flange
355 FS? 04						410 XEQ 20	act. f _b				
356 GTO 06						411 RCL 18	allow F _b				
357 ADV						412 XEQ 21	act. f _b	F _b			
358 RCL 09	Comp Y					413 RCL 04	S _{TF(n)}				If compos. S _{TF}
359 RCL 13	sl. th *					414 SIGN	± 1				is negative
360 2						415 *	± act f _b				(i.e. N.A. in slab)
361 /						416 -	Δ F _b				then DL stress
362 -	Slab d					417 RCL 04	S _{TF}				counteracts LL stress
363 RCL 14	sl. Aln					418 *	allow LL M				
364 *	Q					419 ABS					
365 "Q"						420 RCL 01	BF LL M				
366 XEQ 02						421 X<Y?	min.				
367 GTO 01						422 X<Y	allow LL M				
368+LBL 06	S _{6m}	S _{comp}				423 STO 01	S _{comp}				Slab
369 -	Δ S					424 RCL 10	O *				* S _c for n=∞
370 RCL 22	*					425 .	S _{c-n}				(NCDL)
371 SORT	1/*	S _{TF} -S _{6m}				426 RCL 10					
372 /						427 "SLAB"					
373 RCL 03	S _{6m}					428 RCL 21	CDL M				Slab stress
374 +	S _{eff.}					429 ASTO 03	"SLAB"				calculations
375 STO 11						430 FS? 00	CDL M	S _c	0	S _c	do not use n:3n
376 X<Y? 23	S _{comp}					431 GTO 01					modulus
377 RCL 11	S _{eff}					432 FS? 05					
378 X<Y?						433 X<0?					
379 ADV						434 AVIEW					
380 "EFF"						435+LBL 01					
381 X<Y?						436 X<Y	S _c	CDL M	0	S _c	
382 XEQ 22	S _{eff.}	S _{comp}				437 XEQ 08	act. f _b				

	X	Y	Z	T	L		X	Y	Z	T	L
438 RCL 16	all. f_c					490 X=0?					
439 XEQ 21	act. f_c	all. f_c				491 XEQ 15					
440 FC? 05						492 +	$\Sigma D f_b$				
441 GTO 16						493 FS? 05					
442 -	Δf_c					494 RTN					
443 RCL 10	S_c					495 RCL 26	LL M	$\Sigma D f_b$	S_n	S_n	
444 *	all. LL M					496 R+	S_n				
445 -WS -						497 /	LL f_b				
446 FS? 06						498 -LL-					
447 -LF -						499 XEQ 15					
448 RCL 01	steel LL M					500 +	Σf_b				
449 X>Y?						501 FC? 00					
450 X<Y	LL M					502 ADV					
451 XEQ 05						503 RTN					
452+LBL 16						504+LBL 15					
453 FS? 04						505 FS? 00					
454 GTO 12						506 RTN					
455 3	3					507 XROM -K-					
456 FS? 07						508 XEQ -PV-					
457 1/X	3 or 3					509 RTN					
458 ST* 20	(N)										
459 ST/ 14	(sl. A/n)					510+LBL 21	allow F_b	act. f_b			
460 FS?C 07						511 X<Y	act. f_b	allow			
461 GTO 12						512 FS? 05					
462 SF 07						513 GTO 08					
463 GTO 29						514 -Z -					
464+LBL 12						515 ARCL 03					
465 FS? 11	*					516 XROM -K-					
466 RTN						517 GTO 01					
467 GTO 8											
468+LBL 20	S_n	S_{3n}	S_{0m}	M_{NCDL}		518+LBL 05					
469 -t FL-						519 -tLL-					
470 ASTO 03						520 XROM -M=-					
471 FC? 00						521 RCL 26					
472 AVIEW											
473 -NCDL-						522+LBL 01					
474 R+						523 XEQ -PV-					
475 R+	S_{0m}	M	S_n	S_{3n}		524 FC? 11					
476 /	$f_{b.1}$					525 X=0?					
477 XEQ 15						526 GTO 08					
478 FC? 10						527 -NG-					
479 GTO 04						528 X>Y?	act. f_b	allow F_b			
480 LASTX	S_{0m}					529 BEEP					
481 *	M_{NCDL}	S_{0m}	S_{0m}			530 X>Y?					
482 R+	S_{0m}					531 AVIEW					
483 /	$f_{b.1}$										
484+LBL 04	$f_{b.1}$	S_n	S_{3n}			532+LBL 08					
485 RCL 21	CDL M	$f_{b.1}$	S_n	S_{3n}		533 FC? 00					
486 R+	S_{3-n}	CDL M	$f_{b.1}$	S_n		534 XROM -L-					
487+LBL 08						535 RTN					
488 -CDL-											
489 /	$f_{b.2}$					536+LBL -BM-					
						537 CLX	0				
						538 STO 04					
						539 -WEB D--					
						540 PROMPT					

see line
171

(159)

	X	Y	Z	T	L		X	Y	Z	T	L
541 X=0?						592 STO 05	Y				
542 RTN						593 -Y=					
543 STO 07	Web d					594 XEQ 23	S _T	A	S _b	S _b	
544 XEQ 11	Compr.F.W	WebTh				595 /					
545 GTO 05						596 STO 11					
						597 -COM-	Scarf				
546+LBL -SBM-					From PLG	598 XEQ 22					
547 SF 11						599 -TEN-	Scarf				
548 CF 27						600 R↑	Scarf				
549 GTO IND X						601 STO 14					
						602 XEQ 22					
550+LBL 01						603 FC? 06					
551 GTO 39						604 GTO 03					
						605 CLA					
552+LBL 03						606 RCL 07	Web d				
553 GTO D						607 RCL 09	Tens W				
						608 RCL 10	Tens Th				
554+LBL 04						609 *	Tens A				
555 GTO E						610 RCL 03	Compr W				
						611 RCL 04	Compr Th	Compr W	Tens A	Web d	
556+LBL 00						612 *	Compr A	Tens A	Web d		
557 STO 12						613 -	Δ A				
558 XEQ 09						614 RCL 08	Web th				
						615 /	Δ Web d				
559+LBL 05						616 -					
560 DEG						617 2					
561 X=0?						618 /	N.A. pos	Web d			
562 RTN						619 X<Y?					
563 STO 03	Compr.F.W				X > Y? or X < 0?	620 X<0?					
564 PROMPT					(i.e. N.A. in flange?)	621 GTO 03					
565 STO 04	Compr.F.Th.					622 ADV					
566 -TEN-						623 STO 12	N.A.				
567 XEQ 10						624 RCL 08	Web th				
568 STO 09	Tens. W					625 RCL 09	Tens W				
569 PROMPT						626 RCL 10	Tens Th	Tens W	Web th	d	
570 STO 10	Tens. Th.					627 RAD					
571 STO I						628 XEQ -AY-	A d				
572 RCL 04	Compr Th					629 RCL \	Σ A d				
573 +						630 RCL I	Σ A				
574 RCL 07	Web d					631 /	d				
575 +	Bm d					632 X<> 12	N.A.				
576 STO 00						633 RCL 07	Web d				
577 RCL 09	Tens. W					634 -	new N.A.				
578 STO ↑						635 CHS					
579 RCL 07	Web d					636 RCL 08	Web th				
580 RCL 08	Web th					637 RCL 03	Compr W				
581 RCL 03	Compr.F.W					638 RCL 04	Compr Th	W	Web th	d	
582 RCL 04	Compr.F.Th.					639 XEQ -AY-					
583 XEQ -AY-	A	Y	S _b	I		640 DEG					
584 STO 01	A					641 RCL \	Σ A d				
585 R↑						642 RCL I	Σ A				
586 SF 21						643 /	d				
587 -I-	I					644 RCL 12	d				
588 STO 02						645 +					
589 XEQ 02	S _b	I	A	Y		646 2					
590 R↑	Y	I	A	S _b		647 /	d/2				
591 X<> T						648 CHS					

D160

	X	Y	Z	T	L		X	Y	Z	T	L
750 LASTX	Y	Y ²	I			805 FS? 04					
751 X(>) I	ZA	Y ¹	I			806 FS? 06					
752 *	SA ²	I				807 CF 10					
753 LASTX	A					808 CF 22					
754 RDN	AY ²	I		A		809 FC? 09					
755 +	I					810 FS? 00					
756 RCL I	Y	I		A		811 GTO 05					
757 R↑	A	Y	I			812 "STUD"					
758 X(>) \	ZA	Y				813 FS? 10					
759 *	SA ²	I				814 GTO 07	f _c				
760 -	I					815 .75					
761 ENTER↑						816 Y↑X	(f _c) ^{3/4}				
762 ENTER↑	I	I	I			817 7.25	7.25(f _c) ^{3/4}				
763 RCL [d					818 *					
764 RCL]	Y _{1,2,3}	I	I	I		819 "FD"					
765 -	S _b					820 PROMPT	Stud det.				
766 /	Y					821 X↑2	d ²				
767 RCL I	A	Y	S _b	I		822 *	Stud cap.				
768 RCL \						823 .87					
769 RTN						824 FC? 04					
770+LBL 31						825 SIGN					
771 "AISC"						826 /	Stud cap.				
772 XROM "AA"						827 FS? 22					
773 CF 06						828 GTO 05					
774 .4						829 CLA					
775 FS? 04						830+LBL 07					
776 .45	.45					831 "FCAP, K"					
777 STO 16						832 PROMPT	Stud Cap.				
778 "MSD?"						833 2					
779 PROMPT						834 /					(for AISC ASD)
780 X=0?						835+LBL 05					
781 SF 06						836 FC? 22					
782 CF 08						837 CLX					
783 CLX						838 STO 25	Stud Cap.				
784 "PLAST?"						839+LBL "FY"					
785 FS? 06						840 36					
786 PROMPT						841 "F-Y KSI"					
787 X>0?						842 PROMPT	F _y				
788 SF 08						843 STO 06					
789 .85	.85					844 66 *					* AISC
790 STO 24						845 X(>)Y	F _y				LRFD
791 FS? 06						846 X>Y?					A5.1
792 STO 16	.45?					847 BEEP					
793 8						848 X>Y?					
794 "N"						849 CF 08					
795 FC? 09						850 CF 18					
796 PROMPT						851 FS? 08					
797 STO 20	n					852 SF 18					
798 SF 10						853 END	F _y				
799 CLA											
800 FC? 09											
801 XEQ "Fc"	f _c										
802 ST* 24	(.45+.85)										
803 ST* 16	(.45+.85)										
804 SF 10											

PLATE GIRDER ANALYSIS

xeq PLG (SIZE 032)

SOFTKEYS (USER mode ON)

- a Input intermediate stiffener spacing
- b Input new flanges and find moment capacity
- c Input new moments and lateral bracing spacing
- d Input new web section
- e Input new shear within a panel (e.g. for a new stiffener location)

Note: do not use softkey "b" before having begun flange input once already, since web and stiffener information must be processed before the beam's section and strength properties can be calculated.

USER FLAGS

Flag 00	clear	Full output
	set	Reduced output-many intermediate calculations not displayed
Flag 02	clear	Warning BEEP disabled
	set	Warning BEEP enabled
Flag 04	clear	AASHTO
	set	AISC

INPUT SUMMARY Note: The following summary is for a non-composite plate girder. If analyzing a composite plate girder, many of the prompts encountered are summarized in the documentation for program COMP and are not repeated here. The user should become familiar with COMP before using PLG on a composite beam.

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
AASHTO or AISC	A reminder, not a prompt. Push R/S to continue.	
WSD?	Asking if by working stress method. Respond 1 for yes (WSD/ASD), 0 for no (LFD/LRFD).	1 (WSD)
F-Y KSI	Steel yield stress, ksi.	No default
M1, 'K	Greatest moment within panel being analyzed, in foot-kips. For composite designs, the sum of the NCDL, CDL, and LL moments input is used for M1 and it is not prompted.	No default

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
M2, 'K	Smallest moment within the panel in question. If moment reversal occurs, M2 is negative. <u>For an end panel, input M2 as zero.</u>	M1, 'K
V, K	Shear at the location of interest within the panel.	No default
LAT BR SPA'	Spacing of the lateral bracing, or unbraced length of the compression flange, in <u>feet</u> . Assumed zero and not prompted for composite beams.	No default
WEB D"	Depth of the web plate, inches. Bypass (or enter zero) to accept last web configuration entered and begin stiffener calculations (if stiffeners are needed) or flange trials (if stiffeners aren't needed).	Zero
TH"	Web thickness, inches.	No default
END or INT PANEL:	A message, not a prompt. Push R/S.	
ST SPA"	Stiffener spacing, inches. Bypass (or enter zero) to accept last spacing and begin flange calculations. Press softkey e (lowercase) to input a new shear value (i.e. to look at another location within the panel).	Zero
COMP FL W"	Compression flange width, inches. If satisfied with previous flange, bypass (or key in zero) to enter new loads and begin analyzing a new panel.	Zero
TH"	Compression flange thickness.	No default
TENS FL W"	Tension flange width, inches.	No default
TH"	Tension flange thickness, inches.	No default

OUTPUT SUMMARY

(Items in parenthesis are displayed only if flag 00 is clear)

(MAX D/TW)

Maximum allowable web slenderness

D/TW

Actual web slenderness



ACT FV	Acting shear stress on the trial web
STIFF REQD or NO STIFF REQD	Telling whether or not intermediate stiffeners are needed for shear
MAX ST SPA"	Maximum allowable stiffener spacing, determined by either spacing (for interior panels) or shear requirements (for end panels). If an interior panel, shear requirements may dictate a spacing considerably less than this calculated maximum.
TENS FIELD or NO TENS FIELD	Telling whether or not tension field action is being utilized. The program only uses t.f.a. when allowed by code; and for AISC designs, only when needed.
CV	Shear constant.
ALL FV	Allowable shear stress, ksi. If greater than ACT FV, displays OK; otherwise NG (no good).
(I, Y, S-C, S-Te, A)	Elastic section properties of the beam: moment of inertia, neutral axis depth, section moduli of the compression and tension flanges, and area of the section.
(Z)	Plastic section modulus of the beam, in inches cubed (only if by LFD/LRFD)
b/T	Compression flange width to thickness ratio.
Cb	Bending constant, to account for moment gradient on allowable bending stress.
ALL L/b	Allowable compression flange unbraced length to width ratio. See "FLANGES" section.
L/b OK or NG (AASHTO WSD only)	Checks compression flange L/b.
(L/r _T)	Unbraced length property for AISC designs.
(K _C) (AISC ASD)	Web/flange slenderness interaction constant
(R _b) (AASHTO LFD)	Composite dead load factor
TENS and COMP Fb	Allowable bending stresses for tension and compression flanges, in ksi (WSD/ASD only).
M	Allowable moment capacity.

PROGRAM FLAGS

Flag 05	clear	Interior web panel
	set	End panel-no tension field action
	clear	Have live load moment in COMP
Flag 06	clear	WSD (working stress design)
	set	LFD (load factor design)
Flag 07	clear	OK
	set	NG (no good)
	clear	No stiffener required
	set	Stiffener is required
	clear	L/b_{f1} is under limit (OK) (AASHTO WSD)
	set	L/b_{f1} is over limit (NG)
	clear	h/t_w is under limit (beam) (AISC LRFD)
	set	h/t_w is over limit (plate girder)
Flag 08	clear	Elastic design only
	set	Plastic design permitted <u>by designer</u>
Flag 09	clear	Composite beam
	set	Non-composite beam
Flag 10	clear	AISC LRFD or AASHTO
	set	AISC ASD
Flag 11	set	Within a subroutine
Flag 18	clear	Elastic design only
	set	Plastic design allowed <u>by code</u> (compact)
Flag 19	clear	Symmetrical beam (identical flanges)
	set	Asymmetrical beam
Flag 20	clear	Tension field action allowed
	set	No tension field action allowed

STORAGE REGISTER USE

<u>Register</u>	<u>Value</u>
00	Total beam depth (in SBM)
01	C or Cv; Beam area (in SBM); Allow. LL Moment
02	Beam I (in SBM);
03	Max. D/t_w ; Compr. flange width; Beam S_{bf} ; "B FL"
04	$(D/t_w)^2$; Compression flange thickness; S_{tf} (n=n)
05	$(1 + (D/d_o)^2)/(D/t_w)^2 F_y$; Beam N.A.

<u>Register</u>	<u>Value</u>
06	Fy, ksi
07	Web depth
08	Web thickness
09	Allowable Fv; Tension flange width (SBM); $1/r_T$; Composite N.A. \bar{r}_T
10	Actual D/tw; Tension flange thickness; M_R ; Concrete compr. force; S_c (n=n)
11	D/t _w ; Top flange S; (Slab + Haunch + Beam N.A.); S_{bf} (n=3n)
12	Beam Z; Allowable capacity M_u (in SFb); Effective slab thickness
13	Slab Thickness
14	Beam S_{bf} ; Slab area/N
15	Slab thickness + Haunch
16	.40, .45, or .85 x F_c'
17	NCDL M ("K stored)
18	Allowable tension flange stress
19	M_p (in SFb); S_{tf} (n=3n)
20	N
21	CDL Moment
22	Actual F_v ; Number of studs required/available
23	r_y ; $r_y X_1$; "#STUDS"; S_{bf} (n=n)
24	.85 F_c'
25	Stud capacity/2
26	Live load moment
27	Slab area
28	M_1/M_2
29	Shear interaction reduction factor
30	Lateral bracing spacing (inches stored)
31	M_2

PLATE GIRDER ANALYSIS

Program PLG analyzes a plate girder design for compliance with code requirements, and determines allowable stresses. It considers a beam one section at a time, considering shear, intermediate stiffener spacing, moment gradient, and section properties within each bracing bay or section of interest. The program can analyze a girder by one of four codes:

AISC ASD (Ninth Edition)
AISC LRFD (First Edition)
AASHTO WSD (1989)
AASHTO LFD (1989)

AISC is followed if flag 04 is set. If flag 04 is clear, AASHTO is followed. At the beginning of a run, the designer is prompted "WSD?" Answering this with a 1 (yes) sets the program for ASD/ WSD; responding with a 0 (no) results in the use of LFD/LRFD rules.

For each bracing bay or stiffener panel analyzed, the program steps through the following sequence:

- 1) Prompt maximum shear in panel, moments at each end, and spacing of lateral bracing.
- 2) Try web configurations; check D/t_w and acting shear stress; see if intermediate stiffeners are needed.
- 3) Using last web configuration input, calculate minimum stiffener spacing; try different spacings, calculating allowable shear stress for each.
- 4) Using last web and stiffener input, try different flange sizes; calculate section properties; check B/t_f and bracing requirements.
- 5) Calculate allowable stresses (ASD/WSD) or ultimate moments (LFD/LRFD) for each flange trial.
- 6) Start sequence for new panel.

Each step of the sequence repeats itself for new trials until the user bypasses the first prompt of that step, thus accepting the last parameter input. See the examples for clarification.

As mentioned, PLG analyzes a girder one section or bracing bay at a time; each section can be different. It prompts for moments M_1 and M_2 at the ends of the bay or section, the maximum shear V within the section, and the unbraced length if non-composite (the length of the

bracing bay for noncomposite beams). If a uniform section is desired, it's still not a good idea to input the maximum moment and maximum shear as occurring at one point if they don't. For one thing, the lower shear coincident with the maximum moment may give a higher allowable bending stress. Also, the web stiffener analysis will change with the stiffener spacing, which presumably will change.

The value of the shear is a design variable affecting the intermediate stiffener area requirements and spacing and the allowable flange stress calculations, among other things. As such it must be calculated and input carefully. The moment values M_1 and M_2 are a little less critical. They are used for three things. First, if M_2 is input as zero, the program assumes an end panel and the shear and stiffener calculations are done accordingly.

Second, the program uses the moments to figure C_b (for AISC) or to check for the 20% increase in bending stress allowed by AASHTO LFD. If designing by AISC, and the panel has a higher moment in the middle than at the ends, M_1 and M_2 must be input the same to force $C_b=1.0$.

Third, for AASHTO LFD, the maximum unbraced length for full plastic moment capacity is based on the moment ratio.

The sign convention used by PLG is that moments inducing single curvature are both positive; if they induce reverse curvature, the signs are opposite (either M_1 or M_2 is negative). Note that this is opposite the convention used by AISC.

Some of the things PLG does not do are:

- 1) Analyze hybrid girders. The web and flanges must be the same yield strength. Program BM may be used to calculate the section properties of a hybrid girder.
- 2) Design or analyze bearing or intermediate stiffeners, unless RAM program "STIF" is keyed into main memory; a listing follows. Space constraints prevented this from being included with PLG in the ROM.
- 3) Consider fatigue. The designer must do this.
- 4) Check stresses (yield on gross section against ultimate on net section) at splices or other bolt holes, or check that bolt hole area is less than 15% of gross area.

- 5) Check deflection or camber (although it does calculate moment of inertia).
- 6) Analyze beam-columns. Direct axial stress P/A is assumed to be negligible. See program SBCOL.

The following information concerning what the program does do, and how it does it, is broken down into seven sections, concerned respectively with:

- softkeys
- plastic analysis
- the web
- intermediate stiffener spacing
- the flanges
- allowable stresses and moment capacity
- composite design
- stiffener analysis

SOFTKEYS

Softkeys are used in PLG mainly for jumping from one area of analysis (for instance, stiffener spacing), to another (e.g. flanges). Softkey operation is complicated by the fact that PLG uses many subroutines outside the main program. However, most of these are either general-purpose subroutines such as "S", which have no softkeys of their own; or portions of program COMP (for input and composite beam design). All of the softkeys in COMP are uppercase (unshifted); all of PLG's softkeys are lowercase (shifted). Thus, to use a softkey in PLG, hold it down so that you can read its function in the display. If the softkey calls its native function (X/2, Y/X, etc.) instead of XEQ a or XEQ c, it means that you aren't within PLG and the softkey is inaccessible at the moment; continue holding it until it reads NULL. Then continue running the program until you're back in the main body, and try again.

Note that when analyzing composite beams, the program spends a lot of time within program COMP; but when running non-composite beams, most of the external subroutines are simply short output and display routines which can be gotten through quickly with a couple of SST's. As a quick check, USER mode is turned off while within COMP and turned back on upon re-entry into PLG.

PLASTIC DESIGN

Another potentially confusing item is the PLAST? prompt near the end of the initial input sequence. For design by LFD/LRFD, this prompt gives the user the option of disallowing plastic section analysis (the prompt does not appear for WSD/ASD designs). In other words, the maximum moment capacity is computed using

elastic section properties at first yield. This "veto" ability was initiated for several reasons:

- 1) AASHTO eq. 10-94 involving flange-web slenderness interaction is not checked. For right-to-the-wire designs where this might be a problem, the user must check it and disallow plastic properties if necessary.
- 2) AASHTO 10.48.3 involving web thickness transitions is ignored, i.e. the allowable moment capacity is interpolated between elastic and plastic values. If the user is not comfortable with this, PLAST? can be disallowed for these "in-between" designs.
- 3) PLG automatically disallows plastic design for steel yield strengths above 65 ksi per AISC LRFD A5.1. This is assumed to meet the requirements of AASHTO 10.48.1.2.
- 4) For composite beams, flange b/t ratios are not checked since the flange is assumed restrained from local buckling by the shear connectors. For thin flanges with wide connector spacings (or nervous designers) it may be desirable to override this assumption.
- 5) Inevitably, some agencies will refuse to allow full utilization of plastic strength properties on their structures.

Answering 1 (yes) to the PLAST? prompt does not mandate plastic section analysis; it merely allows it. Applicable code provisions are still checked.

The above discussion is concerned with plastic section analysis; PLG assumes that loads are derived from an elastic structure analysis. AASHTO 10.44.1 is assumed to be met, with the 9/10 exception allowed by 10.48.13; for AISC LRFD designs, equation F1-4 is checked, not equation F1-1. If the user verifies that a section meets the requirements of F1-1, then PLG may be used for a plastically analyzed structural member by LRFD (i.e. the loads may be from a plastic structural analysis). Doing this generally results in a significant decrease in the design loading and thus in the level of conservatism.

Note that the status initially set by the PLAST? prompt can be changed manually at any time by changing the setting of flag 08. This should be done before the program begins calculating the allowable capacity; before or during input of the flange dimensions is a good time.

WEB

The effective depth of the web used by PLG is the distance between the flanges rather than the full member depth, both for web slenderness and shear stress calculations. This is in keeping with AASHTO criteria and is in the spirit of AISC ASD rules (assuming web depth is 95% of full depth) for web slenderness; it is contrary to the liberalization allowed by AISC LRFD F2.1.

The value of h_c is correctly taken as twice the neutral axis depth for asymmetrical non-composite beams; however, for composite beams, the value used is that of the bare beam. This is conservative compared to the codes, and seems prudent given the uncertainty of construction loads prior to hardening of the slab. These limits are concerned primarily with slender web buckling. Putting a composite slab on a beam with its web already undergoing flexural buckling does not reverse the buckling.

AISC rules require calculation of a bending stress reduction factor, R_{PG} , for slender webs. PLG uses web depth per ASD section G2 rather than h_c as called for by LRFD A-G2-3.

The shear constant C_v is the ratio of a panel's buckling strength to its yield strength in shear. C_v may calculate and be displayed as greater than 1.0; however, for AASHTO and LRFD designs, a maximum of 1.0 is used since the web will yield in shear before it buckles. AISC ASD effectively liberalizes this limit to (0.4×2.89) or 1.156; i.e. F_v is never greater than $0.40 F_y$.

The maximum web depth:thickness ratio for AISC is figured assuming stiffeners spaced at 1.5 times the depth; this ratio is only slightly greater than that for an unstiffened girder. However, maximum stiffener spacing/depth is calculated as the greater of $(260t_w/h)^2$ or 3.0, not 1.5. This will very seldom cause conflict, since for the large h/t ratios typically involved, either the $(260t_w/h)^2$ requirement or the shear stress requirements will dictate spacings less than $1.5h$.

Maximum D/t_w requirements for AASHTO are based on AASHTO 10.34.3.1 and 10.48.5.1, and similarly assume maximum stiffener spacing of $3.0 D$, or $1.5 D$ for end panels (panels where one of the moments is input as zero). The increase allowed by 10.34.3.1.1 for bending stress less than allowable is not used, since at this point in the program the bending stress is not known. Stiffeners are required for D/t_w greater than 150, and to meet shear requirements.

The web slenderness is checked against the AASHTO value of $608/\sqrt{F_y}$ over the web depth instead of $640/\sqrt{F_y}$ over the full depth, assuming the flanges comprise 5% of the depth ($608 = .95 \times 640$). This is within the spirit of ASD rules and is conservative for LRFD. Another minor adjustment is made to the AISC ASD shear strength constant of 190 used for the calculation of C_v . The program uses $\sqrt{36000}$, which is 189.74, to match the AASHTO wording. For the same reason, PLG uses 186.95 instead of 187 in this portion of the LRFD code. These differences are obviously insignificant but may cause very minor differences between computed and hand-calculated versions of the same design.

PLG does not check web crippling; this must be investigated by the designer. Neither does it does check that the web is at least 5/16" thick per AASHTO 10.8.1 if designing by AASHTO. No thickness minimum is required for AISC designs.

INTERMEDIATE STIFFENERS

Stiffener spacing is checked and a warning displayed if it is greater than 3.0 D. If the limit is violated, the program still continues; however, the user may wish to use softkey "a" to input a new stiffener spacing. If an end panel is being analyzed by AASHTO, the limit used is 1.5 rather than 3.

Thicker webs may not require any intermediate stiffeners. However, if the actual/allowable shear stress is greater than 0.6 they may be desirable to eliminate flange stress reduction. Thus, even if the program finds NO STIFF REQD, the stiffener input sequence will still be entered. Or, press softkey (shift) a (in USER mode) to get to the stiffener input sequence and input a new stiffener spacing from elsewhere in the program. If no stiffeners are desired, simply bypass the STIFF SPA prompt (press R/S with no numeric entry) to get on to the flange design portion.

The portion of the program related to the analysis of the actual stiffeners had to be left out of this version of BRIDGEROM due to severe space constraints (largely due to the addition of AISC LRFD provisions, absent in the previous version). However, a listing of RAM program "STIF" is included. If this is present in main memory, it will be accessed as a subroutine by PLG and will analyze both intermediate and bearing stiffeners. NOTE: if you have any programs presently in main memory with the Label STIF, please change the title if intending to use PLG. For more information, see page D178.

FLANGES

For composite designs, PLG assumes an unbraced length of zero and L_b' (lateral bracing spacing) is never prompted.

Compression flange b/t ratios are checked using full flange width. For composite beams before hardening of the slab, the requirements of AASHTO 10.34.2.1.5 or 10.50(c), and AISC LRFD I3.4, regarding width to thickness ratios as a function of dead load stress, are not checked by the program and should be verified by the designer.

AASHTO WSD has upper limits on compression flange L/b , while AASHTO LFD and AISC simply keep reducing the allowable (compressive) bending stress. If an AASHTO WSD run finds "L/B NG", it is a Code violation, and the program returns for a new beam section. For situations where neither the unbraced length nor the flange width can be changed (i.e. rating of existing structures), the user is left with the choice of ignoring the limit and extrapolating the parabolic reduction equation beyond that point (as the other codes do) or pronouncing the structure impossible to rate or possessing no strength. PLG will be of no help in any of these situations, since it does acknowledge the code cutoff point.

For very wide thin flanges, AISC LRFD equation A-B5-3 is identical to the Eighth Edition ASD equation C2-3. This has been changed in the Ninth Edition ASD spec., appearing as ASD equation A-B5-3; the equations are similar except for the introduction of a web slenderness interaction constant K_c . PLG incorporates K_c into ASD equation F1-4; however the infrequently used Appendix B equation is the older Eighth Edition ASD C2-3 (LRFD A-B5-3). For designers who like to stretch the bounds of present knowledge concerning very wide thin flanges, the newer Ninth Edition equation should be checked by hand.

The program makes a few minor adjustments to code constants in the interest of efficient use of code. The LRFD constant of 11,200 in the $b/2t$ check is taken as 11,250. Also, for AISC ASD designs, the program assumes $b/2t < 195//F_y K_c$; it doesn't check equation A-B5-4 in appendix B.

ALLOWABLE STRESSES and MOMENT CAPACITY

The AISC LRFD code explicitly considers the strength of a beam as the lowest of three values based on analysis of web slenderness (WLB), flange slenderness (FLB), and overall (unbraced length) buckling (LTB). Program PLG uses this approach for all four of the

codes accepted (AASHTO WSD/LFD and AISC ASD/LRFD); the moment capacity of the beam is calculated based on each of the three criteria (WLB is not considered for LRFD "plate girders" (very slender-webbed members) and for AASHTO WSD analysis, which has no such provisions). The final moment capacity output is the least of these two or three values. If by LRFD, this final capacity is multiplied by the undercapacity factor ϕ (which is why it is not the same as any of the preliminary values).

Except for AASHTO WSD, the codes recognize three levels of "compactness" for each of the three beam criteria: compact

intermediate between compact and non-compact
non-compact

"Compact" beams have a moment capacity (allowable stress) equal to the plastic moment of the beam ($0.66 F_y$ for AISC ASD). "Non-compact" beams have a capacity (allowable stress) less than or equal to the elastic strength at first yield ($0.60 F_y$ for ASD). And, "intermediate" beams have a strength (stress) interpolated between these levels.

ALLOWABLE STRESSES: WSD/ASD

For WSD/ASD, the program uses the actual value of 0.55, .60, or .66 F_y rather than the rounded-off values often used; i.e. for compact AISC designs it outputs 21.6 ksi instead of the familiar 22 ksi (Grade 36 steel). These values are then reduced for lateral-bracing, flange-slenderness, and shear-interaction requirements; AISC designs also are reduced for web slenderness.

For reduction in the allowable bending stress due to high shear stress in the web (shear interaction), AISC reduces only the allowable tension web stress, while AASHTO simply reduces all stresses. PLG reduces both tension and compression flange stress, which is correct for AASHTO and makes no difference for most AISC designs. In the unlikely event that AISC would allow a higher compression than tension stress, the output allowable compression stress will thus be slightly conservative.

AISC only makes this reduction if tension field action is being utilized; AASHTO considers it for all high shear areas. Thus, when designing by AISC, the REDX=... display only appears if tension field action was necessary to carry a shear load with a given web and stiffener spacing. If the web has sufficient shear capacity without relying on the tension field, no reduction is applied. AASHTO takes a simpler view of things; tension field action is always assumed used and

the reduction is always investigated. For simplicity, PLG takes the constants in AASHTO WSD equation 10-29 in 10.34.4.4 as $.55 \times (1.375 \text{ and } .625)$ instead of $.754$ and $.34$; the difference is negligible.

This reduction is not additive with reductions in compressive flange stress due to lateral bracing requirements for WSD/ASD designs. However, due to the way the equations are written, the reduction is additive with other reductions for LFD/LRFD. Program PLG takes this shear-interaction reduction as additive in all cases and is thus slightly conservative for ASD/WSD designs. While the AASHTO WSD code is a generally conservative code, AISC ASD often results in members which are lighter than the other codes would allow. Thus this small degree of conservatism is not inappropriate.

MOMENT CAPACITY: LFD/LRFD

As mentioned, moment capacities for beams with compactness properties in the "intermediate" range are figured by interpolating between the elastic moment at first yield and the plastic moment. For WLB (slender web) calculations, the elastic moment is figured with the section modulus taken with respect to the compression flange, and when this flange is smaller than or equal to the tension flange this calculation is straightforward. However, when the compression flange is distinctly larger than the tension flange, its section modulus may be greater than the plastic modulus Z of the beam, which, of course, represents a physically impossible situation. In this case, neither AASHTO nor AISC offers much advice. Whether the intent is to use the moment corresponding to first yield of the compression flange with the tension zone partially yielded (a difficult computation) or the much lower moment at tension flange yield, or some other value, is unknown. In this instance, PLG uses the full compression section modulus, but with an upper limit of the plastic modulus. In other words, the program may interpolate between Z and Z . Please note that this applies only to WLB calculations.

For the case of flange slenderness (FLB) calculations, the elastic moment is figured differently for AASHTO LFD and AISC LRFD designs. For thin-webbed "plate girder" design, LRFD appendix G2 is quite straightforward, and the program follows it directly. For stout-webbed "beam" design, LRFD reduces the yield stress by F_r , the assumed residual stress; PLG uses the compression flange S_x for this case. AASHTO also specifies the full compression flange S_x , but with the full yield stress. In this case, as for web slenderness, PLG uses this moment up to a limiting value of the plastic moment.

COMPOSITE BEAMS

Note: this section mostly reiterates information concerning composite built-up beams from the preceding material. The documentation for program COMP concerns itself with all composite beams and should be consulted along with this documentation for composite plate girders.

The stability of a composite beam during construction is not checked directly by the program. To do this a separate run must be made with the COMP? prompt answered 0 (no). The designer should input the weight of the wet slab and beam as dead load, and whatever construction live load is likely to be applied. Depending on the degree of restraint afforded by the slab falsework, the unbraced length of the compression flange will probably be the distance between cross bracing. AASHTO 10.50(e), forbidding tension field action in a composite beam before hardening of the slab, is not checked by the program and must be checked by the designer. AASHTO 10.50(d) is not checked, but the constant R_p is calculated for use by the designer.

Non-composite beams are checked on the basis of three criteria: web slenderness (WLB), flange slenderness (FLB), and unbraced length (LTB). For composite beams, only web slenderness is checked. For plastic design, if the web is compact, a fully plastic composite section is used. However, if the web falls between the compact and non-compact limits, PLG does not interpolate between the plastic and elastic (first yield) levels of strength; the moment capacity output is that of the elastic section at first yield, as directed by both AASHTO 10.50.1.2 and AISC LRFD I3.2.

The value of h_c is correctly taken as twice the neutral axis depth for asymmetrical non-composite beams; however, for composite beams, the value used is that of the bare beam. This is conservative compared to the codes, and seems prudent given the uncertainty of construction loads prior to hardening of the slab.

For composite beams, flange b/t ratios are not checked since the flange is assumed restrained from local buckling by the shear connectors. For thin flanges with wide connector spacings, the designer should check by hand that FLB does not govern the design. For composite beams before hardening of the slab, the requirements of AASHTO 10.34.2.1.5 or 10.50(c), and AISC LRFD I3.4, regarding flange width to thickness ratios as a function of dead load stress, are not checked by the program and should be verified by the designer.

The flange is also assumed restrained against LTB by the slab and shear connectors. For composite designs, PLG assumes an unbraced length of zero and L_b' (lateral bracing spacing) is never prompted.

STIFFENER ANALYSIS

Program PLG does not analyze stiffeners by itself. However, it will access a subroutine in Main Memory named STIF if such a program exists. (Note: If by chance you have any programs in memory named "STIF", change the name if using program PLG.) A listing of this program follows. It uses the same flag conventions and storage registers as PLG, and requires 393 bytes (57 registers) of memory in RAM.

STIF analyzes both bearing and intermediate stiffeners. Bearing stiffeners are assumed to consist of one plate on each side of the web; intermediate stiffeners may be one or two plates. All applicable provisions of AASHTO or AISC are checked, with these exceptions:

- 1) AASHTO LFD bearing stiffener analysis gives an allowable factored load instead of following WSD criteria as specified.
- 2) AASHTO WSD intermediate stiffener analysis uses the minimum area requirement from the LFD spec.
- 3) Shear transfer (i.e. weld size) is not addressed. Neither is intermediate stiffener length.
- 4) All bearing stiffeners are analyzed as columns with effective length equal to web depth. This is contrary to the 75% criterion of AISC.
- 5) Single intermediate stiffener moment of inertia is calculated from center of web instead of edge of stiffener plate.

01+LBL "STIF"
 02 CF 25
 03 XROM "LL"
 04 CLX
 05 CLA
 06 XROM "NO"
 07 "I" INT ST 1,2"
 08 PROMPT
 09 X=0?
 10 RTN
 11 1
 12 X=Y?
 13 2.4
 14 STO 00
 15 RCL 11
 16 1/X
 17 ENTER↑
 18 X↑2
 19 ENTER↑
 20 SIGN
 21 LASTX
 22 +
 23 SQRT
 24 /
 25 -
 26 RCL 07
 27 RCL 08
 28 *
 29 RCL 00
 30 *
 31 2
 32 /
 33 1
 34 RCL 01
 35 -
 36 *
 37 RCL 09
 38 /
 39 RCL 22
 40 *
 41 FC? 10
 42 GTO 00
 43 *
 44 RCL 07
 45 50
 46 /
 47 X↑2
 48 X↑2
 49 GTO 03

 50+LBL 00
 51 .3
 52 *
 53 RCL 08
 54 X↑2
 55 18
 56 *
 57 -

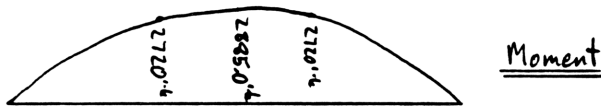
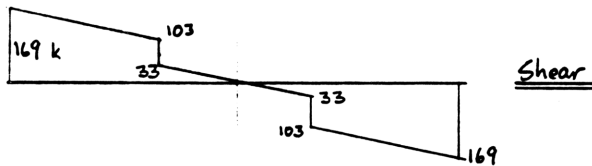
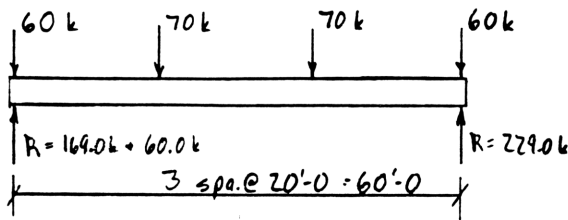
X	Y	Z	T	L		X	Y	Z	T	L
					58 RCL 11	a/d				
					59 X↑2					
					60 2.5					
					61 *					
					62 2					
					63 -					
					64 .5					
					65 X>Y?					
					66 X<>Y					
					67 RDN					
					68 "J"					
					69 XROM "V"					
					70 RCL 07					
					71 RCL 11					
					72 *					
					73 *					
					74 RCL 08					
					75 3					
					76 Y↑X					
					77 *					
					78+LBL 03					
					79 STO 10					
					80 "MIN I"					
					81 XROM "V"					
					82 "MIN A"					
					83 X<>Y					
					84 X<0?					
					85 "I<0 "					
					86 STO 12					
					87 XROM "V"					
					88 256					
					89 RCL 06					
					90 *					
					91 FS? 06					
					92 6760					
					93 CLA					
					94 XEQ 14					
					95 SF 14					
					96+LBL 07					
					97 XROM "L"					
					98 CF 07					
					99 "INT"					
					100 XEQ 11					
					101 X=0?					
					102 GTO 10					
					103 RCL 10					
					104 R↑					
					105 RCL 08					
					106 INT					
					107 ST/ 02					
					108 /					
					109 *					
					110 3					
					111 /					
					112 STO 05					

221 *LBL 14
222 SQRT
223 FS? 04
224 95
225 RCL 06
226 SQRT
227 /
228 STO 09
229 *-MAX b/T-
230 XROM *-V-
231 R↑
232 END

Design a Welded Plate Girder in A36 steel to serve as a transfer beam.
Design by AISC ASD.

$$F_y = 36 \text{ ksi}$$

Uniform load = 3.1 k/ft + Selfwt.
(Assume 200 lbs/ft selfwt.)
Consider braced only at 1/3 pts.



Select web depth

Try $h/t = 72$; try $1/4$ " web th.

$h/t = 288$; $\frac{760}{\sqrt{22}} = 162 < 288 \Rightarrow$ Plate Girder
check web h/t by AISC G1-1

$$\frac{14,000}{\sqrt{36(36+16.5)}} = 322 > 288 \text{ OK}$$

Intermediate stiffeners

$h/t > 260 \Rightarrow$ Int. stiff. req'd (AISC F5.)

$$\text{Shear stress @ end: } f_v = \frac{169 \text{ k}}{72" \cdot 1/4"} = 9.39 \text{ ksi}$$

$$\text{Max. Stiffener spacing: } \frac{a}{h} < \left(\frac{260}{h/t} \right)^2 \leq 3.0$$

$$\text{Max. } a = 72" \times \left(\frac{260}{72/1/4} \right)^2 = 58.68" \leq 216.0"$$

In end panel, spacing will be governed by shear stress considerations. (AISC 1.10.5.2)

Try first space @ 24" :

$$a/h = \frac{24}{72} = .333$$

$$a/h < 1 \Rightarrow k = 4.0 + \frac{5.34}{(a/h)^2} = 52.06$$

$$\text{Try } C_v = \frac{190}{h/t} \sqrt{k/F_y} = \frac{190}{288} \sqrt{52.06/36} = .7933$$

$C_v < 0.8 \Rightarrow$ re-calculate

$$C_v = \frac{45,000 \cdot k}{F_y \cdot (h/t)^2} = \frac{45,000 \cdot 52.06}{36 \cdot 288^2} = .7846$$

$$\text{Allow. } F_v = \frac{F_y}{2.89} (C_v) \leq .40 F_y$$

$$\frac{36}{2.89} \cdot .7846 = 9.7731 \text{ ksi} \leftarrow \text{Governs}$$

$$.40 \cdot 36 = 14.40 \text{ ksi}$$

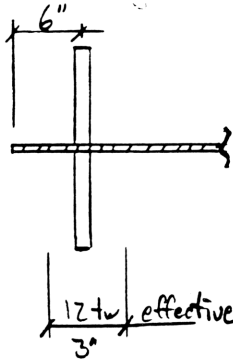
$$9.77 \text{ ksi} < 9.39 \text{ actual OK}$$

Locate first int. stiff. @ 24"

Thereafter, use tension field action (AISC eq. G3-1).
Spacing probably governed by minimum 58.68".

Between first int. stiff. and bearing stiffeners under 70 k loads, have:

$$\begin{aligned} 20' \cdot 12 &= 240" \\ 1^{st} \text{ spa} &= \frac{24"}{216"} \end{aligned}$$

End Bearing Stiffness

12tw effective web (AISC K1.8)

$$\text{Allow. } P = 20.45 \text{ ksi} \times 12.75 \text{ in}^2 = 265.9 \text{ k}$$

$$265.9 \text{ k} > 229.0 \text{ k} \quad \underline{\text{OK}}$$

Since third pt. stiffeners have greater effective web width and smaller load, and since a thinner stiffener will impinge on width/th limit,

USE SAME STIFF. @ 1/3 PTS.

$$\text{Reaction} = 229.0 \text{ k}$$

Estimate 20" flanges \Rightarrow Use 8" plates

Size for ~ 20 ksi allowable.

$$\frac{229.0 \text{ k}}{20 \text{ ksi}} = 11.45 \text{ in}^2$$

$$11.45 - 3 \times \frac{1}{4} = 10.70 \text{ in}^2$$

$$10.70 / 2 \times 8 = .67$$

\Rightarrow width/thickness shouldn't govern

$$(\text{AISC B5.1}) \frac{b}{t} < \frac{95}{\sqrt{F_y}} = \frac{95}{\sqrt{36}} = 15.83$$

\therefore Use ZR 8" \times 3/4"

$$b/t = 10.7 < 15.83 \quad \underline{\text{OK}}$$

$$I = 16.25^3 \times .75 / 12 = 268.19 \text{ in}^4$$

$$\text{Area} = 2 \times 8 \times .75 + 3 \times \frac{1}{4} = 12.75 \text{ in}^2$$

$$r = \sqrt{I/A} = 4.586 \text{ in}$$

Use eff. length = $d = 72$ " (AISC K1.8 allows 54")

From AISC App. A, Table 3-36 for $\frac{KL}{r} = 15.7, F_a = 20.45 \text{ ksi}$

FLANGE DESIGN

Check allowable comp. stress (AISC G2)

$$\text{Web } \frac{h}{t} = \frac{72}{1/4} = 288$$

$$\text{Max. } \frac{h}{t} = \frac{760}{\sqrt{F_b}} = \frac{760}{\sqrt{21.6}} = 163.5$$

$$288 \gg 163.5 \Rightarrow \text{reduce allow. } F_b$$

Assume $F_b = 20$ ksi for trial Flange

$$\text{@ Center, } M = 2445.0 \text{ k-in}$$

$$A_f \approx \frac{M}{F_b h} - \frac{A_w}{6} = \frac{2445.0 \times 12}{20 \times 72} - \frac{72 \times 1/4}{6}$$

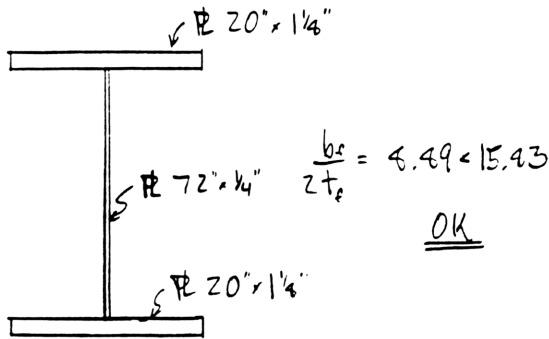
$$= 21.04 \text{ in}^2$$

Long unbraced length \Rightarrow desire wide flanges.

$$(\text{AISC B5.1}) \frac{b_f}{t_f} < \frac{95}{\sqrt{F_y}} = 15.83$$

$$b_f < 31.66 \text{ in}$$

TRY 20" \times 1 1/8" FLANGES



$$\begin{aligned} \text{Web: } 72^3 \times \frac{1}{4} / 12 &= 7776.0 \\ \text{Fl. } 2 \times 1.125^3 \times 20 / 12 &= 4.75 \\ 2 \times (36.5625)^3 \times 20 \times 1.125 &= 60,156.74 \\ &= 67,937.49 \text{ in}^4 \end{aligned}$$

$$S = \frac{67,937.49}{37.125} = 1829.97$$

$$\text{Act. } f_b = \frac{2445.0 \times 12}{1829.97} = 16.92 \text{ ksi}$$

Flange stress reduction in center panel

$$I_{fl} = \frac{20^3 \times 1.125}{12} = 750.0$$

$$A_f + \frac{A_w}{6} = 20 \times 1.125 + \frac{1}{6} \times 72 \times \frac{1}{4} = 25.50 \text{ in}^2$$

$$r_T = \sqrt{\frac{I}{A}} = 5.4233"$$

$$\text{Center Panel: } M_1 = M_2 \Rightarrow C_b = 1.0$$

$$\sqrt{\frac{102 \times 10^3 \times C_b}{F_y}} = 53.23 \quad (\text{AISC 1.5.1.4.5(2)})$$

$$\frac{l}{r_T} = \frac{20' \times 12}{5.42} = 44.25$$

$$\frac{l}{r_T} < 53.23 \Rightarrow F_b = .60 F_y = \underline{21.6 \text{ ksi}}$$

Reduction due to web slenderness

$$F'_b = F_b \left[1 - 0.0005 \frac{A_w}{A_f} \left(\frac{h}{t} - \frac{760}{\sqrt{F_b}} \right) \right] \quad (\text{AISC G2.})$$

$$\begin{aligned} F'_b &= 21.6 \left(1 - 0.0005 \frac{14.0}{22.5} \left(244 - \frac{760}{\sqrt{22}} \right) \right) \\ &= \underline{20.511 \text{ ksi}} \quad (\text{Comp. Flange}) \end{aligned}$$

$$\text{Max. Tens. Flange } F_b = \underline{21.6 \text{ ksi}}$$

$$\begin{aligned} \text{Max. } M &= 20.511 \text{ ksi} \times 1829.97 \text{ in}^3 / 12 \\ &= 3127.90 \text{ ft-k} > 2445 \text{ OK} \end{aligned}$$

Flange stress in End Panels

$$M_1 = 0; M_2 = 2720 \text{ ft-k}$$

$$\frac{M_1}{M_2} = 0$$

$$C_b = 1.75 + 1.05 \frac{M_1}{M_2} + 0.3 \left(\frac{M_1}{M_2} \right)^2 = 1.75$$

$$\sqrt{\frac{102 \times 10^3 \times 1.75}{36}} = 70.42$$

Reduce flanges to 20" x 3/4"

at approx. 10' from supports

$$\frac{b_s}{2t_f} = 13.33 < 15.43 \text{ OK}$$

$$\begin{aligned} \text{I: web} &= 7776.0 \\ \text{fl } 2 \times .75^3 \times 20 / 12 &= 1.41 \\ 2 \times (36.375)^3 \times 20 \times .75 &= 39,694.22 \\ \text{I} &= 47,471.63 \end{aligned}$$

$$S = \frac{I}{36.75} = 1291.74 \text{ in}^3$$

Consider small fl. entire 20' for l/r_T

$$I_{fl} = \frac{20^3 \times .75}{12} = 500 \text{ in}^4$$

$$A_f + \frac{A_w}{6} = 20 \times .75 + 3.0 = 18.0$$

$$r_T = \sqrt{\frac{I}{A}} = 5.2705"$$

Reduced Flange

$$\frac{l}{r_T} = \frac{20' \times 12}{5.27} = 45.54$$

$$45.54 < 70.42 \quad \underline{\text{OK}}$$

$$\text{Tens. } F_b = .60 F_y = 22 \text{ ksi}$$

$$\begin{aligned} \text{Comp. } F_b &= 21.6 \left(1 - 0.0005 \frac{16.0}{15.0} \left(244 - \frac{760}{\sqrt{22}} \right) \right) \\ &= 19.967 \text{ ksi} \end{aligned}$$

Max. M @ flange reduction

$$= 19.967 \times 1291.74 \text{ in}^3 / 12$$

$$= 2149.37 \text{ ft-k}$$

$$\text{Approximately, } \frac{2149}{2720} = \frac{x}{20'}$$

$$x = 15.8'$$

Locate Flange Reduction

15'-0 from Q Reaction

Check shear interaction @ Conc. Load

$$f_v = \frac{103 \text{ k}}{72 \times \frac{1}{4}} = 5.722 \text{ ksi}$$

$$F_v = 9.44 \text{ ksi (see sh. 2)}$$

$$\left(0.825 - 0.375 \frac{F_v}{F_y} \right) \cdot F_y = .598 \cdot F_y \approx .60 F_y$$

Use F_b as above

(AISC G5-1)

Check Web Crippling

Under 3.1 k/ft uniform load,
with flange restrained
against rotation

$$\frac{3.1 \text{ k/ft}}{12" \times \frac{1}{4}"} = \underline{1.03 \text{ ksi}}$$

Allowable comp. stress ^{Eighth Edition} (AISC 1.10-10)

$$\left(5.5 + \frac{4}{(a/h)^2} \right) \times \frac{10,000}{(h/t)^2} = \left(5.5 + \frac{4}{(.7447)^2} \right) \times \frac{10,000}{244^2}$$

$$= 1.45 \text{ ksi} \quad \underline{\text{OK}}$$

Note: Program PLG does not
check web crippling. This
must be done by the engineer.
Ninth Edition AISC is not particularly
clear on this requirement.

Check Weight

$$\text{Heavy section} = 214.2 \text{ lb/ft} \times 30' = 6426 \text{ lb}$$

$$\text{Light section} = 163.2 \text{ lb/ft} \times 30' = 4896 \text{ lb}$$

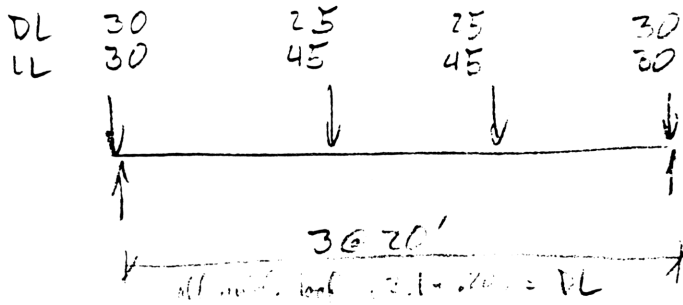
$$11,322$$

$$\frac{11,322}{60'} = 188.7 \text{ lb/ft avg.}$$

\Rightarrow 200 lb/ft assumed is OK

ASL cr. 6/6

Same design by LRFD



$$DL: M_{1/3} = 1.2 (25 \times 20' + 3.3 \times 20' \times 40' / 2)$$

$$= 1.2 \times 1420 = 2184 \text{ k}$$

$$LL: M_{1/3} = 1.6 (45 \times 20') = 1440 \text{ k}$$

$$3624.0 \text{ k} @ 1/3 \text{ pt}$$

$$DL: M_{mid} = 1.2 \times (25 \times 20' + 3.3 \times \frac{60^2}{4})$$

$$= 1.2 \times 1965 = 2362.0$$

$$LL: \frac{1440}{3622.0 \text{ k} @ mid}$$

Shear

$$1.2 (3.3 \times 30' + 25 \text{ k}) = 146.4$$

$$1.6 \times 45 \text{ k} = 72.0$$

$$1.7 \times 1.6 \times 30 = 81.6$$

$$220.8$$

$$\frac{44.0}{304.8}$$

$$1.2 \times 3.3 \times 10' = 39.6 \text{ k}$$

$$1.2 \times 25 \text{ k} = 30.0$$

$$1.6 \times 45 = 72.0$$

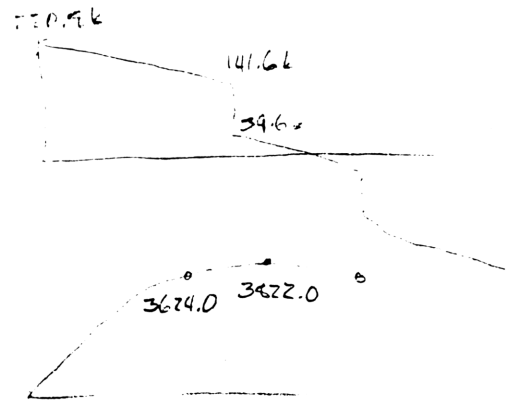
$$141.6 \text{ k}$$

$$(1.2 \times 1.6) \times 30 \text{ k} = 57.6 \text{ k}$$

$$225.6$$

Max. end spa.

$$= 72'' \sqrt{\frac{5}{\frac{f_v / 1.6}{44,000 / (W \cdot t)^2} - 5}} = 26.18''$$



Use same web. $t_w = 1/4''$

end shear stress:

$$f_v = \frac{220.8}{72 \times 1/4} = 12.27 \text{ ksi}$$

$$Try @ 24'' : \phi_n = \frac{24}{72} = .333$$

$$k = 5 + \frac{5}{(a/h)^2} = 50.0$$

OK

$$234 \sqrt{\frac{k}{F_y}} = 275.8$$

$$\frac{b}{t_w} = \frac{72}{1/4} = 288 > 275$$

$$\therefore C_v = \frac{44,000 \text{ k}}{(W/t_w)^2 F_y} = .7368$$

$$all. F_y \phi_n = (4.9) \phi_n = .6 \phi_n C_v F_y$$

$$= .6 \times (.9) \times 220.8 \times .7368 = 36$$

$$257.81 \text{ k} > 220.8 \text{ k}$$

$$= 14.32 \text{ ksi} > 12.27 \text{ ksi}$$

Note: PLG rounds slightly to

$$C_v = .7316$$

$$V = 14.2218 \times 72 \times 1/4 = 256.0 \text{ k}$$

LRFD ex 2/6

LRFD ex. 3/6

CRFD ex 4/6

LRFD ex. 5/6

LRFD ex. 6/6

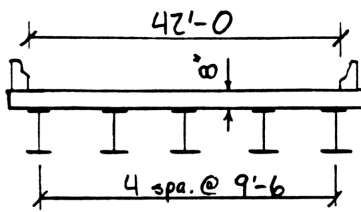
Design the interior girders for a 90'-0' simple span bridge

Design Load: HS20

$$F_y = 50 \text{ ksi (A588)}$$

$$f_c = 4.0 \text{ ksi}$$

Use AASHTO Working Stress Design



LOADS

$$\text{NCDL: Beam est. } .20 \text{ k/ft}$$

$$\text{Deck } .67' \times 9.5' \times .15 = .95 \text{ k/ft}$$

$$1.15 \text{ k/ft}$$

$$\text{NCDL } M = \frac{1.15 \times 90^2}{8} = 1164.38 \text{ k-ft}$$

CDL (Composite Dead Load)

$$\text{Rails } 2 \times .44 \text{ k/ft} / 5 \text{ beams} = .18 \text{ k/ft}$$

$$\text{FWS } 42' \times .02 \text{ ksf} / 5 = .17 \text{ k/ft}$$

$$.35 \text{ k/ft}$$

$$\text{CDL } M = \frac{.35 \times 90^2}{8} = 354.38 \text{ k-ft}$$

Live Load

From AASHTO App. A for 90' span:

$$M = 1344.4 \text{ k-ft/lane} ; V = 64.5 \text{ k/lane}$$

$$\text{Impact} = \frac{50}{L+125} = \frac{50}{90'+125} = .2326$$

$$\text{Distribution} = \frac{B_m \text{ Spa}}{5.5} = \frac{9.5}{5.5} = 1.7273 \frac{\text{wheel}}{\text{beam}}$$

$$\therefore M = 1344.4 \text{ k-ft/lane} \times (1.2326 \times \frac{1 \text{ lane}}{2 \text{ wheel}} \times 1.7273)$$

$$LL M = 1344.4 \times 1.0645 = 1431.09 \text{ k-ft}$$

$$LL V = 64.5 \times 1.0645 = 68.66 \text{ k}$$

$$\text{CDL } V = .35 \text{ k/ft} \times 45' = 15.75$$

$$\text{NCDL } V = 1.15 \text{ k/ft} \times 45' = 51.75$$

$$136.16 \text{ k}$$

AASHTO 10.5.2 recommends $L/d < 30$ for comp. constr.

Simple span \Rightarrow deflection problems

Use 48" web ; $L/d = 22.5 \text{ OK}$

Max. $D/t_w = 140$ for A588 (10.34.3.1.2)

AASHTO 10.6.1 limits min. thick. = $5/16"$

Try $3/8"$ thick web

$$f_v = \frac{136.16 \text{ k}}{48" \times 3/8"} = 7.56 \text{ ksi}$$

$$D/t_w = \frac{48'}{3/8"} = 128 < 150 \text{ OK}$$

(10.34.4.1)

$$F_v = \frac{5.625 \times 10^4}{(D/t_w)^2} = 3.43 \text{ ksi} < 7.56 \text{ ksi} \text{ NG}$$

\therefore Int. Stiffeners Req'd

First Stiffener Spa. $\leq D/2$

Try 24"

$$F_v = \frac{7 \times 10^4 [1 + (D/20)^2]}{(D/t_w)^2} \leq \frac{F_y}{3}$$

$$= \frac{7 \times 10^4 \cdot [1 + (48/20)^2]}{128^2} = 21.36 \text{ ksi}$$

$$F_y/3 = 16.67 \text{ ksi}$$

$$16.67 > 7.56 \text{ ksi} \Rightarrow \text{24" OK}$$

Locate remaining stiffeners

$$\text{Max. spa.} = 1.5D = 72" \quad (10.34.4.2)$$

$$\text{Try spa. @ max: } 72"$$

$$C = \frac{2.2 \times 10^5 [1 + (D/d)^2]}{F_y (D/t_w)^2} \leq 1$$

$$C = \frac{2.2 \times 10^5 [1 + (4^2/72^2)]}{50(174)^2} = .3879$$

$$\text{Allow. } F_v = \frac{F_y}{3} \left[C + \frac{.87(1-C)}{\sqrt{1 + (d_o/D)^2}} \right] = \underline{11.38 \text{ ksi}}$$

$$11.38 > 7.56 \text{ ksi} \quad \underline{\text{OK}}$$

$$\text{Check } 10.34.4.4 : \frac{7.56}{11.38} = .66 > .60$$

⇒ reduce bending stress. (10.34.4.4)

$$\text{Max. } F_b = (.754 - .34 \cdot \frac{7.56}{11.38}) \cdot 50 = \underline{26.42 \text{ ksi}}$$

Design Int. Stiffeners

Use 1-side (single) stiffeners

$$\text{Min. } I = d_o t_w^3 J$$

$$J = 2.5 \left(\frac{D}{d_o} \right)^2 - 2 \geq 0.5$$

$$\text{First stiff. : } J = 2.5 \times \left(\frac{48}{24} \right)^2 - 2 = 8.0$$

$$\text{Min. } I = 24" \cdot (.375)^3 \cdot 8.0 = 10.125 \text{ in}^4$$

$$\text{Rest of stiff. : } J = 2.5 \times \left(\frac{48}{72} \right)^2 - 2 = -.89$$

$$\Rightarrow J = .5$$

$$\text{Min. } I = 72" \cdot (.375)^3 \cdot .5 = 1.896 \text{ in}^4$$

$$\therefore \text{Min } I = 10.125 \text{ in}^4 \quad (10.34.4.7)$$

$$\text{Min. stiff. width} = 2" + \frac{D}{30}$$

$$= 2" + \frac{48}{30} = 3.6" \quad (10.34.4.10)$$

Try 5" x 5/16" Stiffeners

$$I = \frac{5^3 \times 5/16}{3} = 13.02 \text{ in}^4 \quad \underline{\text{OK}}$$

$$\frac{b}{t} = \frac{5}{5/16} = 16 \quad \underline{\text{OK}}$$

USE 5" x 5/16" SINGLE STIFF.

BEARING STIFFENERS

$$\text{Design for reaction} = \underline{136.16 \text{ k}}$$

Assume 14" Wide Top Flange

⇒ try 6 1/2" stiffeners

(Use A588 for brg. stiff.)

$$\text{Min. th.} = \frac{6\frac{1}{2}}{12} \cdot \sqrt{\frac{50 \text{ ksi}}{33}} = .67"$$

Try 3/4" x 6 1/2" stiff.

$$(10.34.6.1) \quad A_{web} = (18 \times \frac{3}{8}) \times \frac{3}{8}$$

$$= 2.53 \text{ in}^2$$

$$A_{stiff} = 2 \cdot 6\frac{1}{2} \times \frac{3}{4} = \frac{9.75}{12.28 \text{ in}^2}$$

$$I = (2 \cdot 6\frac{1}{2} \times \frac{3}{8})^3 \times \frac{3}{12} = 149.54 \text{ in}^4$$

$$r = \sqrt{\frac{I}{A}} = 3.49"$$

Use L = 48"; assume K = 1

$$\frac{KL}{r} = 13.755$$

$$C_c = \sqrt{\frac{2\pi^2 E}{F_y}} = 107.0$$

$$\frac{KL}{r} < C_c \Rightarrow F_a = 23.54 - 0.00103 \left(\frac{KL}{r} \right)^2$$

$$= 23.385 \text{ ksi}$$

$$\text{Act } f_a = \frac{136.16 \text{ k}}{12.28 \text{ ft}^2} = 11.09 \text{ ksi OK}$$

Note: PLG does not check bearing stiff. design

FLANGES

Desire $\approx 14"$ min. width of top flange for shear stud placement, erection stability.

$$\text{Min. th.} = \frac{b}{20} = \frac{14}{20} = .70"$$

(10.34.2.1.4)

Try 14" \times 3/4" top flange

Approximately, bottom flange has moment arms of:

$$50" \pm \text{ for NCDL}$$

$$57" \pm \text{ for CDL + LL + I}$$

Assume $f_b \approx 26 \text{ ksi}$

$$\therefore A_{bf} = \frac{12}{26 \text{ ksi}} \left(\frac{1164.38 \text{ ft}^2}{50"} + \frac{354.38 + 1431.09 \text{ ft}^2}{57"} \right)$$

$$= 25.2 \text{ ft}^2$$

Try 14" \times 1 1/2" bottom flange

Beam Properties:

	A	d	Ad	A d ²	I _o
Top Fl	10.5	.375	3.94	10.620.73	0.419
Web	14.0	24.75	445.5	993.44	3456.0
Bot Fl	27.0	49.5	1336.5	8100.41	5.06
	55.5		1745.94	19714.58	3461.55
				23,176.13	

$$\bar{y} = \frac{1745.94}{55.5} = 32.18"$$

$$S_{top} = \frac{23,176.13}{32.18} = 720.22 \text{ in}^3$$

$$S_{bot} = \frac{23,176.13}{50.25 - 32.18} = 1282.51 \text{ in}^3$$

Check erection stresses

Bracing @ 1/4 pts. = 22.5'

$$L_b < 25' \text{ OK (10.)}$$

Top flange governs by inspection

$$\text{Table 10.32.1A: } F_a = 27,000 - 14.4 \left(\frac{L}{b} \right)^2$$

$$= 27,000 - 14.4 \left(\frac{22.5' \times 12}{14"} \right)^2$$

$$= 21,644 \text{ psi (*Note: PLG solves as below)}$$

Check NCDL on bare beam;
increase by 10% for constr. loads

$$f_b = \frac{(1.10 \times 1164.38 \times 12)}{720.22} = 21.34 \text{ ksi}$$

OK

Note: decrease in F_b due to web slenderness (sheet 2) does not apply here due to lower shear.

* PLG solves for F_b as follows:

$$F_b = 27,000 \cdot \left[1 - \frac{12 \left(\frac{L}{b} \right)^2 \cdot F_y}{4\pi^2 E} \right]$$

$$= 21,737.04 \text{ psi}$$

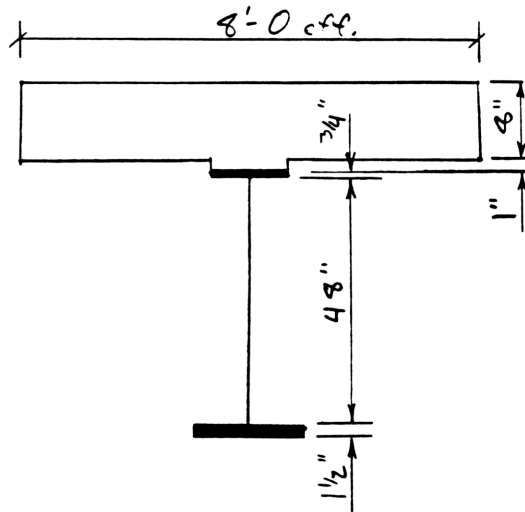
Calculate composite properties

Effective slab width:

$$\begin{aligned} \text{Bm Sp.} &= 9.5' \\ \text{Span}/4 &= 9.5/4 = 2.375 \\ 12 \times t_h &= 12 \times 8" = 9.0' \leftarrow \text{Governs} \end{aligned}$$

$$\therefore \text{For } n=8, \text{ eff. width} = \frac{12 \times 8"}{n=8} = 12"$$

Assume 1" stool



$$n=24:$$

$$S_{\text{slab}} = \frac{51,403.24}{27.54} \times (n=24) = 44,727 \text{ in}^3$$

$$S_{\text{TF}} = \frac{51,403.24}{27.54 - 9} = 2766.27 \text{ in}^3$$

$$S_{\text{BF}} = \frac{51,403.24}{(59.25 - 27.54)} = 1623.20 \text{ in}^3$$

Actual Stresses:

Slab Use S_{s1} for $n=8$ (conserv.)

$$\begin{aligned} & (354.38 + 1431.09) \times \frac{12}{32,826.57} \\ & = .65 \text{ ksi} \end{aligned}$$

(Note: PLG does not calculate composite section properties or stresses thereon. Program COMP can be used for this.)

For $n=8$

	A	d	Ad	$A d^2$	I_o
Slab $12" \times 4"$	96.0	4"	384.0	17,404.60	512.0
Beam	55.5	41.1791"	2285.44	30,404.05	23,176.13
	151.50	17.6201"	2669.44	48,612.65	23,688.13
				72,300.74	

$$\bar{y} = 17.62" > 4" \rightarrow \text{entire slab effective}$$

$$S_{\text{slab}} = \frac{72,300.74}{17.62} \times (n=8) = 32,826.57 \text{ in}^3$$

$$S_{\text{TF}} = \frac{72,300.74}{(17.62 - 4)} = 4347.50 \text{ in}^3$$

$$S_{\text{BF}} = \frac{72,300.74}{(59.25 - 17.62)} = 1736.75 \text{ in}^3$$

For $n=24$ (Per 10.38.1.4)

	A	d	Ad	$A d^2$	I_o
Slab $4" \times 8"$	32.0	4"	128.0	17,795.40	170.67
Beam	55.5	41.1791"	2285.44	10,260.64	23,176.13
	47.5	27.5422"	2413.44	28,056.45	23,346.80
				51,403.24	

Stresses on comp. section

Top Flange Since the neutral axis is below the top flange, for both $n = 9$ and $n = 24$, stresses are additive.

$$\begin{aligned} \text{NCDL} & 1164.39 \times 12 / 720.22 = 19.40 \text{ ksi} \\ \text{CDL} & 354.36 \times 12 / 2766.77 = 1.54 \\ \text{LL+I} & 1431.09 \times 12 / 4327.50 = \underline{2.05} \\ & 22.99 \text{ ksi} \end{aligned}$$

Bottom Flange

$$\begin{aligned} \text{NCDL} & 1164.39 \times 12 / 1242.51 = 10.89 \text{ ksi} \\ \text{CDL} & 354.39 \times 12 / 1623.20 = 2.62 \\ \text{LL+I} & 1431.09 \times 12 / 1736.75 = \underline{9.89} \\ & 23.40 \text{ ksi} \end{aligned}$$

Allowable stresses on Comp. section

From Table 10.32.1-A, $F_{\text{tens}} = 27 \text{ ksi}$

and for $L/b \approx 0$ (due to slab), $F_{\text{comp}} = 27 \text{ ksi}$ (assuming no splice)

$\therefore \text{Max } F_b = \underline{26.42 \text{ ksi}}$ (see sh. 2) OK

(Actually, this reduction is only true at high shear areas, i.e. the ends. $F_b = 27.0 \text{ ksi}$ where shear drops below .6 V_u :

$$.6 \times 11.34 = 6.43 \text{ ksi}$$

$$\text{or } 6.43 \times 44 \times \frac{3}{4} = 122.90 \text{ k.}$$

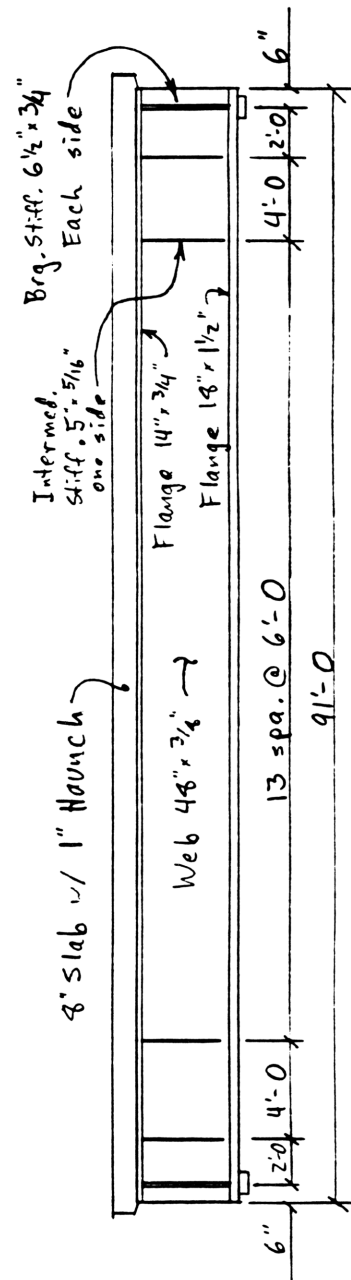
However, erection stresses govern this design so advantage cannot be taken of the full allowable.

$\therefore \text{USE SECTION AS DESIGNED}$

Note: Ordinarily the bottom flange would be reduced for the end 20' or so. This example does not consider this reduction.

This example also excludes design of shear connectors; consideration of field splices; and deflection and camber calculations.

LAYOUT



XEQ "PLG"
9:12:59 AM 03/19/88

TITLE:
90 FT SIMPLE SPAN BRIDGE
RUN

AASHTO

WSD? 1.0000 RUN

F-Y KSI 50.0000 RUN

M1, 'K 2,000.0000 RUN

M2, 'K 0.0000 RUN

V, K 136.1600 RUN

LAT BR SPA' 0.0000 RUN

MAX D/TW=140.0000

WEB D" 48.0000 RUN

TH" .3750 RUN

D/TW=128.0000 OK

ACT FV=7.5644 KSI

STIFF REQD

WEB D" RUN

END PANEL:

MAX ST SPA=24.0000"

ST SPA" 24.0000 RUN

NO TENS FIELD

ALL FV=16.6667 KSI OK

ST SPA" RUN

#STIFF 1.0000 RUN

J=8.0000

MIN I=10.1250

INT PANEL:

MAX ST SPA=72.0000"

Working stress

M1 is arbitrary
due to composite
action
O check end
panel

fully braced by slab

New web trial?
Bypass

First stiffener
space

New spacing trial?
Bypass

single stiffener

ST SPA" 72.0000 RUN

CV=0.3879

TENS FIELD

ALL FV=11.3883 KSI OK

ST SPA" RUN

#STIFF 1.0000 RUN

J=0.5000

MIN I=1.8984

STIFF W" 4.0000 RUN

TH" .2500 RUN

STIFF NG

STIFF W" 6.0000 RUN

TH" .3125 RUN

STIFF NG

STIFF W" 5.0000 RUN

TH" .3125 RUN

STIFF OK

STIFF W" RUN

COMP FL W" 14.0000 RUN

TH" .7500 RUN

TENS FL W" 18.0000 RUN

TH" 1.5000 RUN

I=23,176.1267

Y=32.1791"

S-C=720.2240

S-Te=1,282.5077

A=55.5000

b/T=18.6667 OK

ALL L/b=30.0000

L/b OK

I < Min

$\frac{b}{T} > 16$

OK

Go on

Check flange $\frac{b}{T}$
and slenderness

TENS Fb=26.3961 KSI
COMP Fb=26.3961 KSI

COMP FL W"

RUN

M1, 'K

1,280.8200

RUN

M2, 'K

RUN

V, K

56.9300

RUN

LAT BR SPA'

22.5000

RUN

MAX D/TW=140.0000

WEB D"

XEQ B

COMP FL W"

14.0000

RUN

TH"

.7500

RUN

TENS FL W"

18.0000

RUN

TH"

1.5000

RUN

I=23,176.1267

Y=32.1791"

S-C=720.2240

S-Te=1,282.5077

A=55.5000

b/T=18.6667 OK

ALL L/b=30.0000

L/b OK

TENS Fb=27.0000 KSI

COMP Fb=21.7370 KSI

COMP FL W"

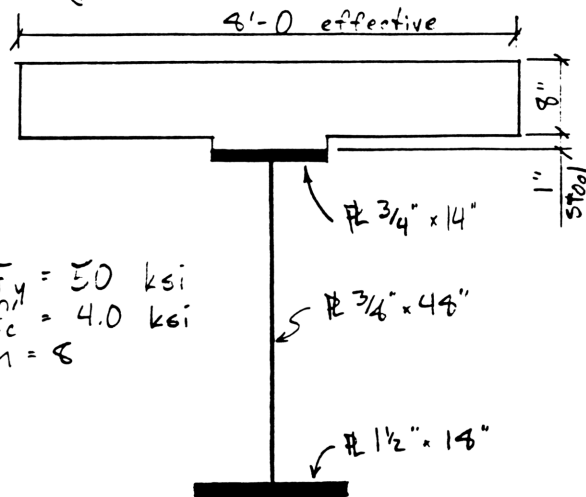
Allowables for
fully supported
flange-governed
by shear reduction

Bypass flange input
for new load and
support conditions-
check erection
stresses

Web dimensions and
shear properties stay
the same - check
new flange allowables

New allowable stresses
for unbraced length
of 22.5'

Re-analyze this design
using LOAD FACTOR DESIGN



$$F_y = 50 \text{ ksi}$$

$$f_c = 4.0 \text{ ksi}$$

$$n = 8$$

Span = 90'-0
Braced @ 1/4 Pts. $\Rightarrow L_b = 22'-6$

LOADS

Moments:	Serv.	LF
NCDL	1164.38 $\times 1.3$	= 1513.69 ft-k
CDL	354.38 $\times 1.3$	= 460.69 ft-k
LL+I	1431.09 $\times 1.3 \cdot \frac{5}{3}$	= 3100.70 ft-k

Shear	Serv.	LF
NCDL	51.75 $\times 1.3$	= 67.28 k
CDL	15.75 $\times 1.3$	= 20.48
LL+I	64.66 $\times 1.3 \cdot \frac{5}{3}$	= 148.76
		236.52 k

For construction, consider 10%
increase in NCDL.

$$\text{LF } M = 1.1 \times 1513.69 = 1665.06 \text{ ft-k}$$

$$\text{LF } V = 1.1 \times 67.28 = 74.01 \text{ k}$$

Check compactness requirements (10.4.2.1)

$$a) \frac{b'}{t} = \frac{7''}{.75''} = 9.33$$

$$\frac{2200}{\sqrt{50,000}} = 9.84 > \frac{b'}{t} \quad \underline{\text{OK}}$$

$$b) \frac{D}{t_w} = \frac{48''}{3/4''} = 128 < 150 \quad \underline{\text{OK}}$$

$$c) \frac{20,000 \cdot (14'' \cdot .75'')}{50 \text{ ksi} \times 50.25''} = 43.6''$$

$$L_b = 22.5' \times 12 = 270'' > 43.6''$$

NG during erection

$$L_b = 0' < 43.6'' \quad \underline{\text{OK}} \text{ for service}$$

$$d) \text{Max. axial comp.} = 0 \quad \underline{\text{OK}}$$

$$e) 1.015 \times 10^5 \frac{t_w^3}{D} = 1.015 \times 10^5 \frac{(3/4'')^3}{48}$$

$$= 111.5 \text{ k}$$

$$\text{Shear} = 236.52 \text{ k} > 111.5 \quad \underline{\text{NG}}$$

\Rightarrow transverse stiff. req'd

Check stiffened web (10.44.5.1)

$$\frac{D}{F_w} = \frac{48''}{\frac{3}{4}} = 128 < 163 \quad \underline{\underline{OK}}$$

Space first stiffener (10.44.5.5)

$$M_{\max} = \frac{P}{2} = 24''; \text{ check allowable @ } 24''$$

$$v_p = .54 F_y = 29.0 \text{ ksi}$$

$$\text{Allow. } \gamma = 1.2 \times 10^5 \left[1 + \left(\frac{\gamma}{\delta} \right)^2 \right] \cdot \frac{t^2}{D^2} = 36.62 \text{ ksi}$$

$$36.62 > 29.0 \Rightarrow \text{allow. } v = \underline{29.0 \text{ ksi}}$$

$$\text{Act. } v = \frac{236.52 \text{ k}}{48,375} = \underline{13.14 \text{ ksi}} < \underline{29.0 \text{ OK}}$$

Remaining stiffeners

$$\text{Max. Spo.} = 1.5 D = 72" \quad (10.49.5.5)$$

Check v @ 72" ; $D/d_o = 2/3$

$$C = 18,000 \left(\frac{t_w}{D} \right) \sqrt{\frac{1 + (D/d_s)^2}{F_y}} - 0.3 < 1.0$$

$$= 14,000 \left(\frac{.375}{48} \right) \sqrt{\frac{1 + (.667)^2}{50,000}} - 0.3$$

$$= .45\bar{3}4$$

$$v_u = v_p \left(C + \frac{.47(1-C)}{\sqrt{1 + (.6/v)^2}} \right) = 29.0 + .7184$$

$$20.83 > 13.14 \text{ kN}; \quad \underline{OK}$$

Use 1 spa. @ 24" then spa. @ 72"

Check shear interaction (10.4.5.4)

$$\frac{V}{V_u} = \frac{13.14}{20.43} = .63 > .60$$

\therefore limit bending stress

Note: AISI limits only the tension bending stress, while AASHTO does not make the distinction.

As a simplification, PLG applies this reduction to both tensile and comp. stresses no matter which code is being followed.

$$\frac{F_b}{F_y} = 1.375 - .625 \frac{V}{V_u}$$

$$F_b = 50(1.375 - .625 \cdot \frac{13.14}{20.45})$$

$$= \underline{49.04 \text{ ksi}}$$

Check bending stresses

Bottom Flange:

$NCDL \quad 1513.69^{14} / 12 / 1242.51_{10} = 14.163 \text{ ksi}$
 $CDL \quad 460.69^{14} / 12 / 1623.20 = 3.406$
 $LL+I \quad 3100.70 / 12 / 1736.75 = 21.424$
 $\underline{\hspace{1.5cm}}$
 39.993 ksi

$$34.99 < 49.04 \quad \underline{OK}$$

Top Flange :

$$\begin{array}{rcl} \text{NCDL } 1513.69 \cdot 12 / 720.72 & = & 25.22 \\ \text{CDL } 460.69 \cdot 12 / 2766.77 & = & 2.00 \\ \text{LL+I } 3100.70 \cdot 12 / 4387.50 & = & 4.44 \\ & & \hline & & 31.66 \text{ ksi} \end{array}$$

$$31.66 < 49.04 \quad \text{OK}$$

Slab:

$$CDL \quad 460.69 \times 12 / 32,426.57 = .164$$

$$LL+I \quad 3100.70 \times 12 / 32,426.57 = \frac{1.134}{1.302}$$

$$\text{Allow. } f_c = .45 f_c' = 3.40 \text{ ksi} > 1.302 \quad \underline{\text{OK}}$$

Check erection stresses

$$L_b = 22.5' \times 12 = 270" > 43.6"$$

$$\therefore F_b = F_y \left[1 - \frac{3F_y}{4\pi^2 E} \left(\frac{L_b}{r_b} \right)^2 \right]$$

$$= 50 \cdot \left[1 - \frac{3 \cdot 50}{4\pi^2 \cdot 29,000} \left(\frac{270}{9.77} \right)^2 \right]$$

$$= 37.968 \text{ ksi} \quad (10.48.4.1)$$

@ Midspan, $\frac{M_1}{M_2} \approx 1 \Rightarrow$ no increase

$$\text{Act. } f_b = \frac{1665.06 \text{ in}^4 \cdot 12}{720.22 \text{ in}^3} = 27.74 \text{ ksi}$$

$$27.74 < 37.97 \quad \underline{\text{OK}}$$

Note: Under LFD this section could have been considerably smaller than under WSD and would normally be re-designed. Deflection could become an important consideration.

Bearing Stiffener Design

Use WSD design (10.48.7)

Intermediate Stiff. Design (10.48.5E)

$$\text{Min. } A = (.15 B \cdot D \cdot t_w (1-C) \cdot \frac{Y}{\sqrt{A}} - K_1 t_w) \times Y$$

$$Y = 1.0 \quad (\text{non-hybrid})$$

$$B = 2.4 \quad (\text{single plate})$$

$$C = .4559$$

$$\text{Min } A = (.15 \cdot 2.4 \cdot 46 \cdot \frac{3}{4} (1-.4559) \cdot \frac{13.14}{20.43}$$

$$- 18 \cdot \frac{3}{4} \cdot \frac{1}{2})$$

$$= -.3071 \quad (\text{for internal panel})$$

For end panel, $C = 1.0$

$$\Rightarrow \text{Min. } A = -18 \left(\frac{3}{4} \right)^2 = -2.5313$$

Both are $< 0 \Rightarrow$ no min. area requirement (stiffened portion of web alone is adequate)

Min I is calculated as for WSD:

$$\text{End Panel Min. } I = 10.125 \text{ in}^4$$

$$\text{Int. Panel Min } I = 1.498 \text{ in}^4$$

$$\frac{b'}{t_{st}} < \frac{2600}{\sqrt{50,000}} = 11.63$$

Try $5" \cdot \frac{5}{16}"$:

$$\frac{b'}{t} = \frac{5}{5/16} = 16 > 11.63 \quad \underline{\text{NG}}$$

Try $4" \times \frac{1}{2}"$: $\frac{b'}{t} = 9.14 < 11.63 \quad \underline{\text{OK}}$

$$I = \frac{4^3 \times .5}{3} = 10.67 > 10.125 \quad \underline{\text{OK}}$$

USE $4" \times \frac{1}{2}"$ INT. STIFF.

LAYOUT will be the same ^{as WSD} except for size of int. stiffeners. As mentioned, the flanges (esp. the tension flange) would most likely be downsized.

AASHTO LFD

11/12

AASHITO LTD

12/12

	X	Y	Z	T	L		X	Y	Z	T	L
109 FC? 07						162 X<0?					
110 "NO "						163 CLX					
111 "STIFF READ"						164 SQRT	min-spa				
112 AVIEW						165 X=0?					
113 GTO 06						166 3					
						167 ENTER↑	min.	min.			
114+LBL 12						168 FC? 04					
115 CF 07						169 1.5	1.5 ?	min			
116 "V, K"	V					170+LBL 03					
117 PROMPT	d					171 X>Y?					
118 RCL 07						172 X<>Y	min.				
119 /						173 RCL 07	d				
120 RCL 08	+w					174 *	min-spa				
121 /	f v					175 "ST SPA="					
122 "ACT FV"						176 STO 00	min.				
123 STO 22						177 ARCL X					
124 XROM "KS"						178 "T"					
125 XEQ "PV"						179 FC? 00					
126 RTN						180 XEQ "PV"					
						181 CF 22	1				
127+LBL e						182 E	redux				
128 XROM "LL"						183 STO 29					
129 XEQ 12											
						184+LBL 21					
130+LBL a						185 CF 07					
131 XEQ 01						186 XROM "L"					
132 X<>Y						187 CLX					
133 1/X						188 "STIFF SPA"					
134 FS? 10						189 PROMPT					
135 2.89						190 FS? 22					
136 STO 04	all. Fv/f					191 X=0?					
137 XROM "L"						192 GTO 17					
138 "MAX INT"						193 STO 11	stiff-spa				
139 FS? 05						194 CF 20					
140 "MAX END"						195 FS? 05					
141 260						196 SF 20					
142 RCL 07	d					197 RCL 00					
143 RCL 08	+w					198 X<>Y	spa.	min			
144 /						199 X>Y?					
145 STO 10	d/tw					200 XEQ 05	d				
146 /						201 RCL 07					
147 X↑2	(260/10)²					202 ST/ 11					
148 3	3	C 3				203 5	5				
149 FC? 05						204 ENTER↑	5				
150 GTO 03						205 FS? 10					
151 /						206 5.34	5.34				
152 ST+ X	45000 (20/10)²					207 FS? 10					
153 5						208 4	4	5.34			
154 RCL 04	2.89					209 RCL 11	a/d				
155 RCL 22	act f v	2.89	5		45000 (20/10)²	210 E	1	a/d	4	5.34	
156 R↑						211 X>Y?					
157 /						212 XEQ "ZT"	1	a/d	5.34	4	
158 *						213 RDH					
159 5						214 X↑2	(a/d)²	5.34	4		
160 -						215 /					
161 /	min. spa.					216 +	K				
						217 "K"					

	X	Y	Z	T	L		X	Y	Z	T	L
218 XROM -V-						276 X<0?					
219 .8						277 SF 20					
220 X<>Y	.4	K				278 RDN					
221 RCL 06	$\frac{4}{F_1}$	K				279 FC? 20					
222 /						280 +					
223 36 E3						281 CLA					
224 *						282 FS? 20					
225 1.03						283 -NO -					
226 FS? 04						284 -TENS FIELD-					
227 FC? 06						285 FC? 00					
228 SIGN						286 AVIEW					
229 /						287 XEQ 08					
230 SQRT						288 GTO 21					
231 RCL 10											
232 /						LIST 122					
233 X>Y?											
234 ENTER†						289*LBL 01					
235 X†2						290 3					
236 X<>Y						291 1/X					
237 /						292 733 E2					
238 SIGN						293 FS? 06					
239 LASTX						294 .58					
240 X>Y?						295 FS? 06					
241 SF 20						296 127 E3					
242 -CV-						297 FS? 04					
243 XEQ -V-						298 .54					
244 SIGN						299 FS? 04					
245 FS? 10						300 1188 E2					
246 1.156						301 RTN					
247 LASTX											
248 X>Y?						302*LBL 08					
249 X<>Y						303 STO 09	all. fv				
250 STO 01						304 -ALL FV-					
251 ENTER†						305 XROM -KS-					
252 SIGN						306 RCL 22	act fv				
253 LASTX						307 XEQ 05					
254 -						308 X>Y?					
255 .87						309 RTN					
256 *						310 FS? 04					
257 RCL 11						311 FC? 20					
258 X†2						312 X<0?					
259 E						313 RTN	act fv	act fv			
260 +						314 11	11				
261 SQRT						315 -REDX-					
262 /						316 5	5	11	act	all fv	
263 X<0?						317 R†	Fv				
264 CLX						318 /	5/Fv	11	act	act	
265 X=0?						319 R†	act fv				
266 SF 20						320 *	5G/Fv	11			
267 X<>Y						321 -					
268 RCL 06						322 8					
269 RCL 04						323 /	*				
270 /						324 E	1				
271 ST* Z						325 X>Y?					
272 *						326 X<>Y	reduction				
273 RCL 22						327 STO 29					
274 X<Y?											
275 FC? 04											

FS?04
and
FS?20

*1.375-.625 $\frac{fv}{Fv}$

	X	Y	Z	T	L		X	Y	Z	T	L
328*LBL -V-						381 +					
329 FS? 00						382 12					
330 RTN						383 /					
331 -t=-						384 RCL 01					
332 ARCL X						385 /					
						386 SQRT					
333*LBL -PV-						387 STO 23					
334 SF 25						388 -RY-					
335 PRA						389 XEQ -V-					
336 FC?C 25						390 FC? 04					
337 AVIEW						391 GTO 13					
338 RTN						392 RCL 08					
						393 RCL 07					
339*LBL 05						394 *					
340 X>Y?						395 RCL 03					
341 SF 07						396 /					
342 XROM -OK-						397 RCL 04					
343 XEQ -PV-						398 /					
344 RTN						399 6					
						400 +					
345*LBL 17						401 ST+ X					
346 FS?C 05						402 SQRT					
347 GTO a						403 RCL 03					
348 SF 25						404 /					
349 FS? 22						405 1/X					
350 XEQ -STIF-						406 STO 09					
351 CF 25						407 -RT-					
						408 XROM -V-					
352*LBL b											
353 XROM -L*-						409*LBL 13					
354 CLX						410 ADV					
355 -Lb*-						411 XEQ 11					
356 FS? 09						412 RCL 07					
357 PROMPT						413 2 E3					
358 12						414 RCL 12					
359 *						415 FS? 06					
360 STO 30						416 GTO 02					
361 CLX						417 760					
362 XEQ -SBM-						418 XEQ 00					
363 X=0?						419 FS? 04					
364 GTO c						420 AVIEW					
365 RCL 00						421 FS? 04					
366 CF 19						422 SIGN					
367 RCL 05						423 *					
368 ST+ X						424 *					
369 X*Y?						425 -COMP-					
370 SF 19						426 XEQ 09					
371 RCL 03						427 -TENS-					
372 3						428 RCL 18					
373 Y+X						429 RCL 29					
374 RCL 04						430 *					
375 *						431 XEQ 09					
376 RCL 10						432 CF 18					
377 RCL 09						433 FC? 09					
378 3						434 GTO 13					
379 Y+X						435 RCL 14					
380 *						436 *					

	X	Y	Z	T	L		X	Y	Z	T	L
437 X<>Y						489*LBL 00					
438 RCL 11						490 X<>Y					
439 *						491 SQRT					
440 X>Y?						492 /					
441 X<>Y						493 RCL 08					
442 GTO 07						494 *					
						495 R↑					
443*LBL 02						496 -					
444 RCL 11						497 LASTX					
445 /						498 *					
446 970						499 RCL 03					
447 XEQ 00						500 /					
448 FS? 04						501 RCL 04					
449 AVIEW						502 /					
450 *						503 X>0?					
451 *						504 CLX					
452 .9						505 X<>Y					
453 FC? 04						506 /					
454 SIGN						507 E					
455 FC? 09						508 +					
456 GTO 02						509 *R-PG=-					
457 *						510 ARCL X					
						511 FC? 04					
458*LBL 07						512 SIGN					
459 CLA						513 RCL 12					
460 XROM *M=-						514 RCL 29					
461 AVIEW						515 RTH					
462 GTO b											
						516*LBL 11					
463*LBL 02						517 CLX					
464 RCL 06						518 SF 18					
465 *						519 FS? 04					
466 STO 12						520 FS? 20					
467 RCL 05						521 X>0?					
468 RCL 04						522 CF 18					
469 -						523 CF 07					
470 5 E2						524 FC? 04					
471 RCL 06						525 FS? 06					
472 395						526 X>0?					
473 XEQ 00						527 GTO 10					
474 LASTX						528 .66					
475 *Rb-						529 FC? 10					
476 FC? 04						530 RCL 12					
477 XEQ *V-						531 RCL 06					
478 *						532 *					
479 *						533 STO 12					
480 *ALL-						534 STO 18					
481 FC? 18						535 STO 19					
482 XEQ 09						536 *PL-					
						537 XROM *M=-					
483*LBL 13						538 FS? 06					
484 CF 05						539 FC? 18					
485 XROM *L-						540 X=0?					
486 E						541 AVIEW					
487 XEQ *SBM-						542 ADV					
488 GTO b						543 RCL 11					

Z
F_y
M_p

S_T

	X	Y	Z	T	L		X	Y	Z	T	L	
544 FS? 10						601 /						
545 .6						602 RCL 06						
546 RCL 06	F_y					603 SQRT						
547 *	M_R					604 *						
548 RCL 12						605 X>Y?						
549 X>Y?						606 CF 18						
550 X<Y						607 FC? 07						
551 STO 10	M_R					608 X>Y?						
552 FC? 20						609 X=0?						
553 STO 12						610 GTO 02						
554 608	$.95 \sim 640$					611 RCL 07						
555 RCL 05	\bar{Y}					612 RCL 08						
556 RCL 04	$T \text{ FLSH}$					613 /						
557 -						614 70						
558 ST+ X	h_c					615 X>Y?						
559 RCL 08	t_w					616 GTO 11						
560 /	W/t					617 RDN						
561 RCL 06						618 .46						
562 SQRT						619 CHS						
563 *						620 Y+X						
564 X>Y?						621 4.05						
565 CF 18						622 *						
566 150						623 "Kc"						
567 LASTX						624 FS? 10						
568 *						625 XEQ "V"						
569 970						626 FC? 10						
570 FS? 04						627 SIGN						
571 X<Y						628 SQRT						
572 RDN						629 *						
573 FS? 04						630 R+						
574 X>Y?						631 R+						
575 X=0?												
576 SF 07						632*LBL 11						
577 FC? 06						633 R+						
578 CF 07						634 R+						
579 XEQ 07						635 FC? 06						
580 RCL 10						636 95						
581 X<Y?						637 FS? 06						
582 X<Y						638 70						
583 FC? 08						639 FS? 06						
584 FC? 06						640 FC? 04						
585 X<0?						641 GTO 03						
586 CF 18						642 RDN						
587 FC? 18						643 FC? 07						
588 STO 12						644 GTO 04						
589 FC? 09						645 RCL 10						
590 RTN						646 STO 19						
591 "D/TW"						647 2						
592 RCL 12						648 /						
593 FC? 07						649 STO 10						
594 XEQ 14						650 RDN						
595 SF 18						651 150						
596 65						652 GTO 03						
597 RCL 03												
598 RCL 04						653*LBL 04						
599 /						654 RCL 06						
600 2						655 16.5						

If tens field
action, no
plastic moment

	X	Y	Z	T	L		X	Y	Z	T	L
656 -						709 FS? 18					
657 RCL 11						710 GTO 02					
658 X<>Y						711 FC? 04					
659 *						712 GTO 03					
660 STO 10						713 RCL 09	γ_T				
661 RDN						714 FC? 07					
662 RCL 06						715 RCL 23	γ_Y				
663 LASTX						716 3 E2	γ_{300}				
664 /						717 RCL 06					
665 SQRT						718 SQRT	$\gamma_{F_{100}}$	γ_{300}	$\gamma_{T_{100}}$		
666 106						719 /					
667 *						720 *					
668*LBL 03						721 STO a					
669 X<=Y?						722 RCL 30	L_b''				
670 GTO 03						723 X<=Y?					
671 XEQ 07						724 GTO 05					
672 GTO 02						725 CF 18					
						726 "X1"					
673*LBL 03						727 FC? 07					
674 FC? 06						728 PROMPT					
675 GTO 11						729 RCL 23					
676 X<>Y						730 *					
677 /						731 STO 23					
678 X+2						732 "X2"					
679 RCL 10						733 FC? 07					
680 *						734 PROMPT					
681 GTO 02						735 STO \					
						736 RCL 06	F_{100}				
682*LBL 11						737 16.5					
683 1.415						738 -	γ_{100}				
684 R+						739 ST/ Z					
685 229						740 X+2					
686 /						741 *					
687 -						742 E					
688 *						743 +					
						744 SQRT					
689*LBL 02						745 E					
690 "b/2T"						746 +					
691 XEQ 14						747 SQRT					
						748 *					
692*LBL "SFb"						749 756					
693 SF 18						750 RCL 06					
694 RCL 10	M_R					751 SQRT					
695 RCL 14	S_{bp}					752 /					
696 RCL 06	F_Y					753 RCL 09					
697 *						754 /					
698 X<=Y?						755 FC? 07					
699 STO 10						756 X<>Y	λ_R				
700 2 E4	γ_{100}					757 RCL 30	L_b				
701 RCL 03						758 X>Y?					
702 *						759 GTO 04					
703 RCL 04						760 RCL a	λ_P				
704 *						761 X<> Z	γ_r	L_b	γ_r	λ_F	
705 RCL 00	d					762 XEQ 07					
706 /						763 XEQ 08					
707 RCL 06						764 X>Y?					
708 /	γ_{100}					765 X<>Y					
	F_{100}					766 GTO 05					

	X	Y	Z	T	L			X	Y	Z	T	L
767*LBL 04						823 /						
768 RCL 23						824 2						
769 X<>Y						825 -						
770 /						826 CHS						
771 ENTER↑												
772 X↑2						827*LBL 02						
773 RCL \						828 RCL 06						
774 *						829 *						
775 2						830 3						
776 +						831 /						
777 SQRT						832 RCL 10						
778 *	M ₂					833 X>Y?						
779 286 E3						834 X<>Y						
780 RCL 30	L ₀					835 GTO 05						
781 RCL 09	r _T											
782 /						836*LBL 03						
783 X↑2						837 FC? 06						
784 /						838 GTO 10						
785 FC? 07						839 36						
786 X<>Y						840 RCL 31	M ₂					
787 RCL 23						841 RCL 19	P					
788 *						842 -						
789 XEQ 08						843 22						
790 GTO 05						844 *						
						845 -						
791*LBL 02	2000 A ₁					846 E2						
792 RCL 03	b ₁					847 *						
793 76						848 RCL 06						
794 *						849 /						
795 RCL 06	F ₁					850 RCL 23						
796 SQRT						851 *	λ _F					
797 /						852 RCL 30	L					
798 X>Y?						853 X<=Y?						
799 X<>Y	λ _P					854 GTO 05						
800 RCL 30	L					855 CF 18						
801 X<=Y?						856 R↑	2000 A ₁ dF ₁	L				
802 GTO 05						857 X>Y?						
803 CF 18						858 GTO 04	λ _r	λ	λ _P			
804 RCL 10	.6F ₄					859 XEQ 07						
805 RCL 12	.66F ₅					860 GTO 05						
806 X>Y?												
807 X<>Y						861*LBL 07	λ _r	λ	λ _P			
808 STO 18	f ₁ b ₁ c ₁ d ₁ e ₁ f ₁ g ₁ h ₁ i ₁ j ₁ k ₁ l ₁ m ₁ n ₁ o ₁ p ₁ q ₁ r ₁ s ₁ t ₁ u ₁ v ₁ w ₁ x ₁ y ₁ z ₁			L	λ _P	862 RCL 19	P _P	λ _r	λ	λ _P		
809 RCL 09						863 X<> T	λ _P	λ _r	λ	λ _P		
810 R↑	L					864 ST- Z						
811 /						865 -	λ _r λ _P	λ _r	λ _P			
812 51 E4						866 /						
813 XEQ 08						867 X<>Y	λ _P	λ _r	λ _P			
814 RCL 06						868 RCL 10						
815 /						869 -						
816 X<>Y						870 *						
817 X↑2						871 -						
818 *						872 RTH	M ₁					
819 SIGN												
820 LASTX						873*LBL 04						
821 X<=Y?						874 XEQ 03						
822 GTO 02						875 *						
						876 RCL 28						

TIMBER BEAM DESIGN

xeq WBM for wood beam analysis (SIZE 012)

SOFTKEYS (USER mode ON)

D New dimensions
E New LDF
F New material properties (skips dimension and load/LDF prompts to re-evaluate same trial with new allowable stresses)

USER FLAGS

Flag 00	clear	full output
	set	short format
Flag 04	clear	AASHTO
	set	NDS/UBC

INPUT SUMMARY

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
F-V PSI	Allowable horizontal shear stress	No default
Fb PSI	Allowable bending stress	No default
E PSI	Modulus of elasticity	No default
LDF	Load Duration Factor	1.0
Le-b"	Effective unbraced length in the "B" direction, in inches	Previous Le-b
B"	B dimension, inches. Enter zero for round.	Previous B
D"	D dimension, inches. Enter zero for round members.	Previous D
DIA" (round only)	Diameter, inches	No default

OUTPUT SUMMARY (Items in parenthesis are displayed only if flag 00 is clear)

(C-K)	Slenderness limit, dependent on E and F_b values
(S)	Section modulus in inches cubed
(C-S)	Effective length (slenderness) constant, $L_e d/b^2$
(F_b')	Allowable bending stress in psi
ALL M	Allowable moment on the section
ALL V	Allowable shear on the section

PROGRAM FLAGS

Flag 06	clear	check slenderness
	set	no slenderness check necessary
Flag 07	clear	rectangular
	set	round
Flag 08	clear	new material properties input via softkey "E"; use same loads and dim.
	set	normal input sequence
Flag 09	clear	column analysis (WCOL)
	set	beam analysis (WBM)

STORAGE REGISTER USE

<u>Register</u>	<u>WBM</u>
00	S (in ³)
01	F_b
02	L_e (inches stored)
03	B or Diameter, in.
04	D, inches
05	LDF
06	Area
07	E, psi
08	C-F (form)
09	F-V, psi
10	C-K
11	not used (column effective length)

TIMBER BEAM ANALYSIS

Program WBM analyzes the strength of timber beams. It calculates and displays the section properties and allowable stresses, as well as all factors and coefficients used to figure these properties.

MEMBER SECTION

The program is intended for use with solid or glue-laminated (glulam) members only; the analysis of built-up members (e.g. nail-laminated or spaced members) or any solid shape other than rectangular or round, is not covered.

The program is not able to make allowance for bolt holes or notches; a uniform section is assumed. However, in many cases (e.g. slender members with bolt holes at the ends) the allowable stress may still be used. This is a matter requiring engineering judgment. Deflection is not calculated or checked.

FLAGS

WBM will analyze by AASHTO (1988) if flag 04 is clear, and by NDS (National Design Specification for Wood Construction, 1986, upon which AITC and UBC base their specs) if flag 04 is set. As with other structural programs, the code being followed is displayed at the beginning of each run. The differences between the two codes are minor; C_k and the slender beam stress formula for bending are slightly different between the codes. Also, of course, some of the allowable stresses are different for different species and grades of timber. Actually, since AASHTO bases its recommendations on the NDS, any differences merely reflect AASHTO's slowness in changing.

As with other programs in this package, setting flag 00 disables the output of intermediate calculations and constants (as well as the title block for printed output), shortening the output.

DIMENSIONS

The dimension prompts for B" and D" (B and D, in inches) assume D is the dimension in the direction of applied side load (or eccentricity for column design). Thus if a member is being used in weak-axis bending, B will be greater than D. For columns under pure axial load, B and D are interchangeable as long as the effective unbraced length and the member dimension correlate with each other in each direction. After being input once, the prompts for $Le(b)$, $Le(d)$, B, and D all

default to their previous values. See the "Input Summary" section on the front sheet of this documentation.

ALLOWABLE STRESSES

Engineers using the NDS should note that tabulated allowable stresses have decreased in recent editions; an attempt should be made to keep up to date. The higher equation constants allowed for machine-stress-rated (MSR or E-rated) timber are not implemented in these programs.

Factors affecting allowable stresses are input and used in different ways. The following factors are the most common.

CUF	Condition-of-use or wet-use factor
C-FR	0.9 for fire-retardant treated wood
C-FU	Flat-use factor

The above three factors must be chosen by the designer and applied to the tabulated allowable bending stress (F_b) values before input; CUF and C-FR must be applied to the tabulated F_v and E values as well. (CUF for these is 1.0 for sawn timber but not for glulam.)

C-F	Form factor (1.0 for rect. and 1.18 for round members)
C-f	Size factor for deep beams

C-F and C-f are calculated by the program and applied to the allowable bending stress input. C-f is calculated by the basic exponential formula; no adjustment is made for different span/depth or loading conditions. A note of caution: tabulated stresses for poles or piles (NDS 6.2, UBC ch. 25) already have the form factor (1.18) included. If these values are used they must be reduced by 1.18 before input.

LDF	Load duration factor
-----	----------------------

The LDF is input by the user. It is applied where appropriate by the program (e.g. beams or columns which have a "short" or "intermediate" effective length) and must not be included in the allowable stresses input.

C-S	Slenderness factor
-----	--------------------

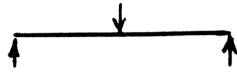
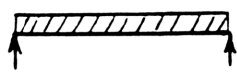
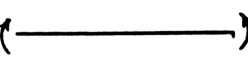
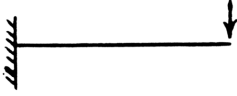
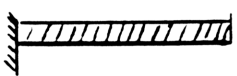
The slenderness factor is calculated and applied by the program. It is used in conjunction with the CUF, C-FR, and C-FU factors (obviously, since the allowable stresses are multiplied by them before input); C-F and C-f are also used in conjunction with these factors. However, C-S is not used with C-F and C-f. The allowable bending stress is calculated based on both

slenderness (C-S) and size (C-F x C-f). The lesser of these, multiplied by the LDF, is the allowable stress which is output. This will hopefully be made more clear in the examples: Fb_0 is the (user-modified) tabulated allowable stress to be input. Fb_1 is the calculated allowable stress involving size and form. Fb_2 is the calculated allowable stress involving slenderness. And finally, Fb is the least of these two, multiplied by the LDF, which is output.

WBM is applicable to round, sawn, or (rectangular) glulam beams. However, for heavily cambered or curved glulam beams the user must modify the allowable stress by the curvature factor.

EFFECTIVE LENGTH

For rectangular beams which are as wide or wider than they are deep, and for round members used as beams, no reduction for slenderness is made; i.e. C-S = 1.0. For deep rectangular beams which are not fully braced, C-S depends on the effective length, which in turn is a multiple of the unbraced length depending on the loading condition. For those cases where the unbraced length is not zero, the following multipliers apply:

$1.37 L_u + 3D$	
$1.63 L_u + 3D$	
$1.84 L_u$	
$1.44 L_u + 3D$	
$0.90 L_u + 3D$	

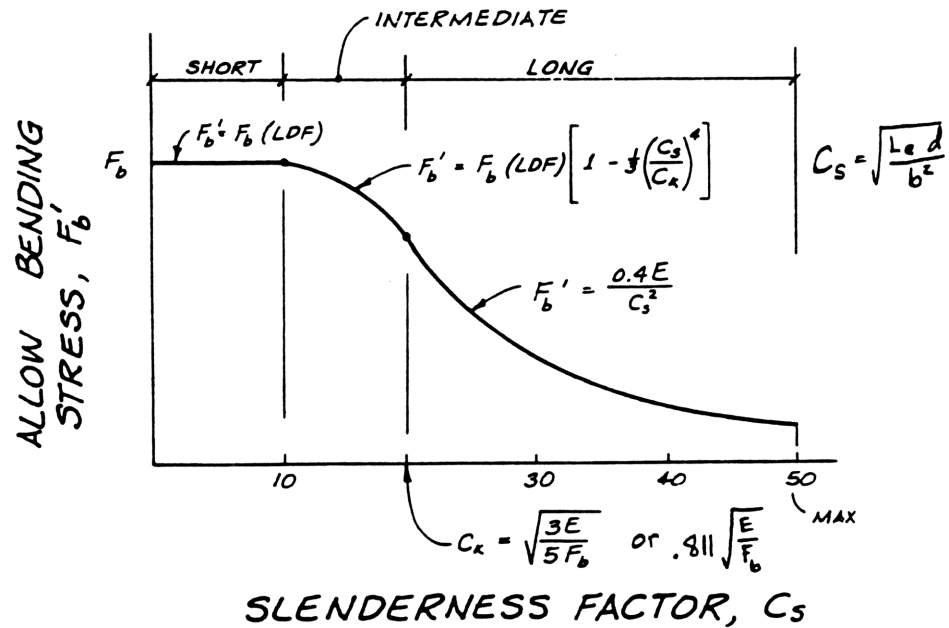
For C-S greater than 50 the program calculates an allowable stress of zero. See the next page for a graph showing the effect of unbraced length on allowable bending stresses.

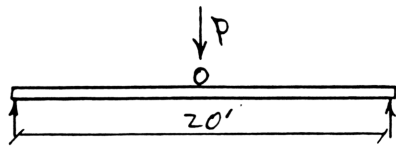
SHEAR

Shear analysis is a little simpler: the only factors which pertain are the CUF and C-FR, which the user must factor into the allowable stress input; and the LDF, which the user inputs directly. The actual shear stress compared is VQ/Ib , which amounts to the average shear stress times $3/2$ for rectangular members and $4/3$ for round beams.

ALLOWABLE BENDING STRESS considering EFFECTIVE LENGTH

The effect of unbraced length is shown graphically by the following illustration, taken by permission from Design of Wood Structures by Donald E. Breyer (McGraw-Hill, 1980)





Design a sawn S4S beam to support a movable equipment hoist.

$$\begin{aligned} \text{Crane DL} &= 2.90 \text{ k} \\ \text{Max. Expected LL} &= \frac{1.20 \text{ k}}{4.10 \text{ k}} \end{aligned}$$

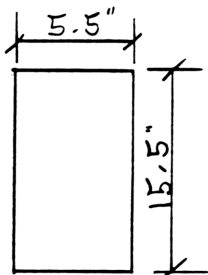
$$M_p = 4.10 \text{ k} \times \frac{20'}{4} = 20.50' \cdot \text{k}$$

Beam requires fire-retardant treatment.

$$\therefore C_{FR} = 0.90$$

$$CUF = 1.0 \text{ (Indoor use)}$$

TRY 6" x 16" - #1 Hem-Fir ("B+S")



$$\begin{aligned} \text{NDS: } F_b &= 1050 \text{ psi} \times C_{FR} \times CUF \\ F_v &= 70 \text{ psi} \times C_{FR} \\ E &= 1,300,000 \text{ psi} \times C_{FR} \end{aligned}$$

$$\begin{aligned} S &= \frac{5.5 \cdot (15.5)^2}{6} \\ &= 220.229 \text{ in}^3 \end{aligned}$$

Beam DL (use 50 pcf.):

$$\begin{aligned} \frac{5.5 \times 15.5}{144} \times 50 \text{ pcf} \times \frac{(20')^2}{8} / 1000 \\ = 1.48' \cdot \text{k} \end{aligned}$$

Check loading combinations

$$\begin{aligned} \text{DL: } (1.48' \cdot \text{k} + \frac{2.90 \times 20'}{4}) / (LDF = 1.0) \\ = 15.98' \cdot \text{k} \end{aligned}$$

$$\text{LL: } (1.48' \cdot \text{k} + 20.50' \cdot \text{k}) / 1.33 = 16.53' \cdot \text{k}$$

\therefore Governing load is LL:

$$\underline{\underline{DL + LL = 21.98' \cdot \text{k}}}$$

$$\begin{aligned} \text{Shear} &= (21.98' \times 10') + \frac{(20' - 15.5')}{20'} \times 4.10 \text{ k} \\ &= 4.13 \text{ k} \end{aligned}$$

$$F_{b.0} = 1050 \times 0.9 \times 1.0 = 945 \text{ psi}$$

$$F_{v.0} = 70 \times 0.9 \times 1.0 = 63 \text{ psi}$$

$$E_0 = 1,300,000 \times 0.9 \times 1.0 = 1,170,000$$

rect. beam \Rightarrow Form Factor $C_F = 1.0$

$$\begin{aligned} \text{Size Factor } C_s &= \left(\frac{12}{d}\right)^{1/9} = \left(\frac{12}{15.5}\right)^{1/9} \\ &= .9720 \end{aligned}$$

$$F_{b.1} = 945 \times 1.33 \times .9720 = \underline{\underline{1221.7 \text{ psi}}}$$

$$L_u = 20.0' \text{ (unbraced)}; L_{eff} = 1.61 \times 20' = 32.20'$$

$$\begin{aligned} C_s &= \sqrt{\frac{L_e d}{b^2}} = \sqrt{\frac{32.20' \times 12 \times 15.5}{(5.5)^2}} \\ &= 14.071 > 10 \Rightarrow \text{not "short"} \end{aligned}$$

$$C_K = .411 \sqrt{\frac{E}{F_b}} = .411 \sqrt{\frac{1,170,000}{945}} = 24.54$$

$$14.071 < 24.54 \Rightarrow \text{"Intermediate"}$$

$$\begin{aligned} \therefore F_{b.2} &= 945 \times 1.33 \times \left[1 - \frac{1}{3} \left(\frac{C_s}{C_K}\right)^4\right] \\ &= \underline{\underline{1232.1 \text{ psi}}} \end{aligned}$$

$$\underline{\underline{F_{b.1} \text{ Governs @ } 1221.7 \text{ psi}}}$$

Sawn Beam Design

5/2

Moment Capacity

$$M = F_b \cdot S = \frac{1221.7 \text{ psi} \times 220.229 \text{ in}^3}{12,000}$$

$$= \underline{22.42 \text{ k}} > 21.94 \text{ k} \quad \underline{\text{OK}}$$

Shear Capacity

$$\text{Shear stress} = \frac{3}{2} \times \frac{4.13 \text{ k}}{5.5'' \times 15.5''} = .0727 \text{ ksi}$$

$$\text{Allowable stress} = .063 \text{ ksi} \times \text{LDF}$$

$$= .0438 \text{ ksi} \quad \underline{\text{OK}}$$

$$\text{Allowable shear} = \frac{.0438}{.0727} \times 4.13 = \underline{4.76 \text{ k}}$$

Short Form - Set Flag 00

no title
block

	SF 00 XROM "WBM"	
F-V PSI	63.0000	RUN
Fb PSI	945.0000	RUN
E PSI	1,170,000.000	RUN
LDF	1.3300	RUN
Le-b"	386.4000	RUN
B"	5.5000	RUN
D"	15.5000	RUN

ALL M=22.4196 'K
ALL V=4.7621 K

Le-b"

no inter-
mediate
calculations

Long Form

SF 00
XEQ "WBM"

← NDS

SIZE<12

SIZE 012

XEQ "WBM"

6:11:14 PM 11/02/88

TITLE:
CRANE RAIL RUN

NDS
F-V PSI

63.0000 RUN

Fb PSI
945.0000 RUN

E PSI
1,170,000.000 RUN

LDF
1.3300 RUN

C-K=28.5363

Le-b"
386.4000 RUN

B"
5.5000 RUN

D"
15.5000 RUN

S=220.2292
C-S=14.0709
Fb'=1,221.6123 PSI

ALL M=22.4196 'K
ALL V=4.7621 K

Le-b"

Use Adjusted
values

TIMBER COLUMN and BEAM-COLUMN DESIGN

xeq WCOL for wood column and beam-column analysis (SIZE 019)

SOFTKEYS (USER mode ON)

- D New dimensions
- E New loads and LDF
- F New material properties (skips dimension and load/LDF prompts to re-evaluate same trial with new allowable stresses)

USER FLAGS

- | | | |
|---------|--------------|-------------------------------|
| Flag 00 | clear
set | full output
short format |
| Flag 02 | clear
set | BEEP disabled
BEEP enabled |
| Flag 04 | clear
set | AASHTO
NDS |

INPUT SUMMARY

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
Fc PSI	Allowable compressive stress	No default
Fb PSI	Allow. bending stress	No default
E PSI	Modulus of elasticity	No default
P, K	Axial load, kips	No default
ECCd"	Eccentricity of axial load in the "D" direction, in inches	0
LATd M, 'K	Moment due to lateral load, in the "D" direction, in foot-kips	No default
LDF	Load Duration Factor	1.0
Le-b"	Effective unbraced length in the "B" direction, in inches	Previous Le-b

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
Le-d"	Effective unbraced length in the "D" direction, in inches	Previous Le-d
B"	B dimension, inches. Zero for round members.	Previous B
D"	D dimension, inches. Zero for round members.	Previous D
DIA" (round only)	Diameter, inches	No default

OUTPUT SUMMARY (Items in parenthesis are displayed only if flag 00 is clear. Asterisked items are displayed only if the member is in bending.)

(NDS or AASHTO)	A reminder of the code used. Press R/S to continue.
(C-K) *	Bending slenderness limit, based on E and F_b
(K)	Slenderness limit, based on E, F_c , and LDF
(S) *	Section modulus
(C-S) *	Effective length (slenderness) constant
(F_b') *	Allowable bending stress in psi
(LONG, INTRM, or SHORT)	The length category of the column in each direction
($F_c'd$ and $F_c'b$)	Allowable axial compressive stresses in the D and B axes, respectively, in psi
ALL P	The allowable axial load in kips, if no bending stress is present
(J) *	Interaction (P-delta) constant
INTRXN-b or -d (*)	The governing interaction value. Must be less than unity to be acceptable.

PROGRAM FLAGS

Flag 06	clear set	check slenderness no slenderness check necessary
Flag 07	clear set	rectangular round

PROGRAM FLAGS

Flag 08	clear	new material properties input via softkey "E"; use same loads and dim. normal input sequence
	set	
Flag 09	clear	column analysis (WCOL)
	set	beam analysis (WBM)
Flag 10	clear	beam-column
	set	pure axial load

STORAGE REGISTER USE

<u>Register</u>	<u>WCOL</u>	<u>WFC</u>
00	S (in3)	
01	Fb	
02	Lb (inches stored)	
03	B or Diameter, in.	
04	D, inches	
05	LDF	LDF
06	Area	
07	E, psi	E, psi
08	C-F (form)	
09	Fc, psi	Fc, psi
10	C-K	
11	Ld (inches stored);	
12	P, kips	
13	M (inch-kips stored)	
14	ECC(d), inches	
15	Axial stress due to P	Length Category (SHORT, INTRM, LONG)
16	Fc'(b), psi	
17	Fb', psi	
18	K	K

TIMBER COLUMN ANALYSIS

Program WCOL analyzes the properties and strength of timber columns and beam-columns. It calculates and displays the section properties and allowable stresses, as well as all factors and coefficients calculated to arrive at these properties.

MEMBER SECTION

The program is intended for use with solid or glue-laminated (glulam) members only; the analysis of built-up members (e.g. nail-laminated or spaced members) or any solid shape other than rectangular or round, is not covered.

The program is not able to make allowance for bolt holes or notches; a uniform section is assumed. However, in many cases (e.g. slender members with bolt holes at the ends) the allowable stress may still be used. This is a matter requiring engineering judgment.

FLAGS

The program will analyze by AASHTO (1988) if flag 04 is clear, and by NDS (National Design Specification for Wood Construction, 1986, upon which AITC and UBC base their specs) if flag 04 is set. As with other structural programs, the code being followed is displayed at the beginning of each run. The differences between the two codes are minor. Also, of course, some of the allowable stresses are different for different species and grades of timber. Actually, since AASHTO bases its recommendations on the NDS, any differences merely reflect AASHTO's slowness in changing.

Beam-column interaction analysis is based entirely on the NDS no matter what the status of flag 04. However, the allowable stresses in bending and compression are figured using the appropriate code.

As with other programs in this package, setting flag 00 disables the output of intermediate calculations and constants (as well as the title block for printed output), shortening the output.

DIMENSIONS

The dimension prompts for B" and D" (B and D, in inches) assume D is the dimension in the direction of applied side load or eccentricity for beam-column design. Thus if a member is being used in weak-axis bending, B will be greater than D. For columns under

pure axial load, B and D are interchangeable as long as the effective unbraced length and the member dimension correlate with each other in each direction. After being input once, the prompts for $L_e(b)$, $L_e(d)$, B, and D all default to their previous values. See the "Input Summary" section on the front sheet of this documentation.

MATERIAL PROPERTIES

The program prompts for material properties F_c , F_b , and E. F_b is not needed for axial column analysis; E is not needed for "short" columns. Thus if the design fits one of these situations, the unnecessary input prompt can be bypassed (hit R/S without answering the prompt). Constants calculated using these values (e.g. C-K and K) will be displayed, but if the estimate of length category was correct or the column is axially loaded, these incorrect constants will not enter into the calculation. If the length category estimate was not correct, however, garbage results. It's usually best to key in the correct values.

See the WBM documentation for information on factors affecting allowable stress input for beam-columns.

ALLOWABLE STRESSES

Engineers using the NDS should note that tabulated allowable stresses have decreased in recent editions; an attempt should be made to keep up to date. The higher equation constants allowed for machine-stress-rated (MSR or E-rated) timber are not implemented in these programs.

The program only checks column buckling on the gross section. It is left to the user to check crushing at the net section (i.e. at bolt holes). Shear is not checked in beam-columns. For truss chords the user must calculate C_t and apply it to E before input.

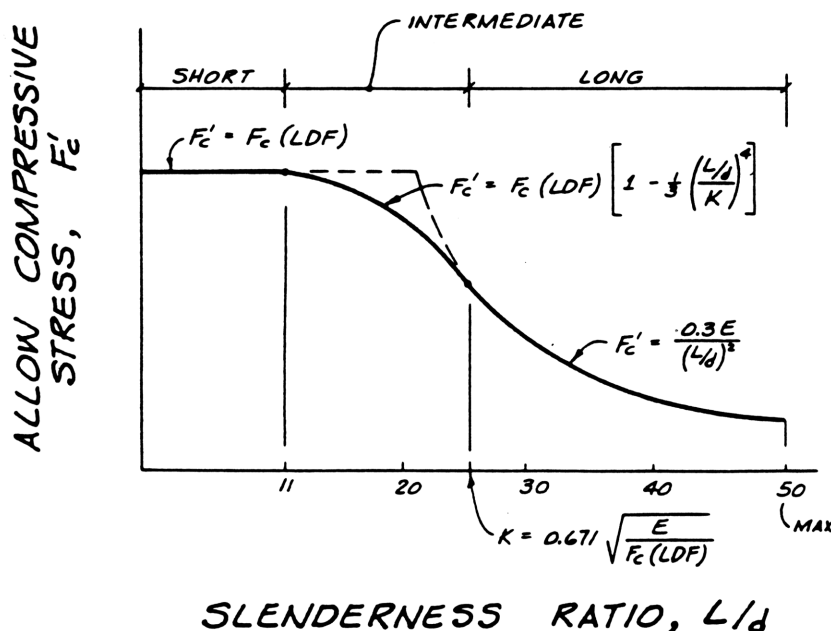
WCOL calculates and displays the allowable compressive stress about each axis, as well as the length category. For axially loaded columns it then picks the governing (least) of these and calculates the allowable axial load. For beam-columns the program calculates and displays the allowable compressive and bending stress, and the P-delta constant J. Then it calculates the interaction formula for each axis (which for the B axis is simply the actual/allowable compressive stress) and displays the governing (largest) one. If this is less than one the design is OK.

EFFECTIVE LENGTH

The effective length for columns depends on the end fixity and the presence or absence of sidesway; such conditions are a matter of engineering judgement, but for conditions where sidesway is prevented, $L_e < L_u$. For unbraced beams, on the other hand, $L_e > L_u$. For the design of beam-columns, then, the choice of effective length probably should be based on whether the member is more beam or column. If doing the calculations by hand, both lengths could be used; the program, however, accepts only one effective length in each direction. Most members which carry side loading are braced laterally by the member delivering the load so this isn't a problem. For those infrequent cases where beam slenderness is a factor, it is up to the judgement of the user whether the program's results are acceptable.

If L_e/D is greater than 50, the allowable stress output is zero.

All this is shown graphically by the following illustration, taken by permission from Design of Wood Structures by Donald E. Breyer (McGraw-Hill, 1980)



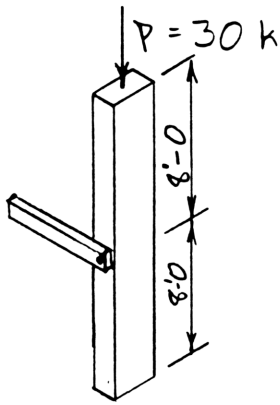
SUBROUTINE "WFC"

As mentioned, spaced column analysis is not covered. However, the allowable column stress subroutine of the program has been given a global label, "WFC," so that

it may be called from user-written programs. The user's program must store the values of K, E, Fc, and LDF in the proper registers and D and Le must be in the stack. See the program listing for WFc (within WBM/WCOL, line 345) and the portion of WCOL which calls WFc (lines 238-243) for clarification.

BIAXIAL BENDING

WCOL does not analyze columns under biaxial bending; only uni-axial bending is considered. The moment prompted for is only that due to side loading, not axial eccentricity. The moments due to side loading and eccentricity are assumed to be additive. If they are of opposite sign, tending to cancel each other, the interaction is evaluated in a mathematically correct way; however, the intent of the Specification is rather vague on this point. Normally it's not a problem since the axial load and any side load are usually from different sources (i.e. side loads are usually from wind) so one of them would simply be neglected.



Design column for axial load = 30 k (including snow)

Sidesway prevented } Use $L_e = L_u$
Free ("Simple") Ends

Snow Load \Rightarrow LDF = 1.15

Design for wet conditions:

$$\begin{aligned} CUF &= .91 \text{ for } F_c \\ &= 1.0 \text{ for } E \\ &= 1.0 \text{ for } F_b \end{aligned}$$

Use #1 Doug Fir, full cut : $F_c = 925$ psi
 $E = 1,600,000$ psi
(Assume "B+S" category due to bracing) $F_b = 1300$ psi

$$F_{c,0} = .91 \cdot 925 = \underline{841.75 \text{ psi}}$$

$$\text{Try } 5" \times 8" : f_c = \frac{30,000}{5 \times 8} = 750 \text{ psi}$$

$$K = .671 \sqrt{\frac{E}{F_c \cdot LDF}} = .671 \sqrt{\frac{1,600,000}{841.75 \cdot 1.15}} = 27.28$$

$$\frac{L_b}{D} = \frac{8' \cdot 12}{5} = 19.2 ; \frac{L_d}{D} = \frac{16' \cdot 12}{8} = \underline{24.0}$$

$11 < 24.0 < 27.28 \Rightarrow$ Intermediate

$$\therefore F'_c = F_{c,0} \cdot \left[1 - \frac{1}{3} \left(\frac{L_d}{K} \right)^4 \right] \times 1.15 = 774.7 \text{ psi}$$

$$774.7 > 750 \quad \underline{OK}$$

$$\text{Allowable } P = 774.7 \cdot 5" \times 8" = \underline{30.988 \text{ k}}$$

As an example, try 10" x 5" oriented other way

$$\frac{L_b}{B} = \frac{8' \cdot 12}{10} = 9.6 < 10 \Rightarrow \text{Short} ; F_{c,b} = F_{c,0} \cdot LDF = 968.01 \text{ psi}$$

$$\frac{L_d}{D} = \frac{16' \cdot 12}{5} = 38.4 > 27.28 \Rightarrow \text{Long}$$

$$F_{c,d} = \frac{0.30 E}{(L/d)^2} = \frac{.30 \cdot 1,600,000}{(38.4)^2} = 325.52 \text{ psi} \leftarrow \text{Governs}$$

$$\text{Allow. } P = 325.52 \cdot 5" \times 10" = \underline{16.276 \text{ k}} \quad \underline{NG}$$

XEQ "WCOL"

6:17:39 PM 11/02/88

TITLE:

ROOF SUPPORT RUN

NDS

Fc PSI

841.7500 RUN

Fb PSI

RUN

E PSI

1,600,000.000 RUN

P, K

30.0000 RUN

ECCd"

0.0000 RUN

LATd M, 'K

0.0000 RUN

LDF

1.1500 RUN

K=27.2799

Le-b"

96.0000 RUN

Le-d"

192.0000 RUN

B"

5.0000 RUN

D"

8.0000 RUN

INTRM

Fc'b=888.8360 PSI

INTRM

Fc'd=774.7104 PSI

ALL P=30.9884 K=OK

Le-b"

RUN

Le-d"

RUN

B"

10.0000 RUN

D"

5.0000 RUN

SHORT

Fc'b=968.0125 PSI

LONG

Fc'd=325.5208 PSI

ALL P=16.2760 K=NG

Le-b"

F_b not needed - can bypass

input same as before - can bypass

Round Column

2/4

Redesign for round column. Remove brace.

$$L_b = L_d = 16.0'$$

Same material: $F_{c.o} = 841.75 \text{ psi}$

$$E = 1,600,000$$

Use "equiv. D" from rad. of gyration:

$$D = r \times \sqrt{12}$$

$$r = \frac{\text{Diam}}{4} \Rightarrow D = .466 \cdot \text{Diam}$$

Try 8" ϕ member

$$\frac{L}{D} = \frac{16.0' \times 12}{.466 \times 8"} = 27.71 > 27.28$$

\Rightarrow Long

$$\therefore F'_c = \frac{0.30 E}{(L/d)^2} = \frac{.30 \cdot 1,600,000}{(27.71)^2} = 624.96 \text{ psi}$$

$$\text{Area} = \frac{\pi}{4} (8")^2 = 50.26 \text{ in}^2$$

$$\text{Allowable } P = 50.26 \cdot 624.96 = \underline{\underline{31.414 \text{ k}}}$$

$$31.414 > 30 \quad \underline{\underline{OK}}$$

Or, analyze as "equiv. square" member:

$$D = \sqrt{50.26} = 7.09"$$

$$\frac{L}{D} = \frac{16' \cdot 12}{7.09} = 27.09 < 27.28$$

\Rightarrow Intermediate (barely)

$$\therefore F'_c = F_{c.o} \cdot LDF \cdot \left[1 - \frac{1}{4} \left(\frac{L/d}{K} \right)^4 \right]$$

$$= 841.75 \cdot 1.15 \cdot \left[1 - \frac{1}{4} \left(\frac{27.09}{27.28} \right)^4 \right] = 654.65 \text{ psi}$$

$$\text{Allowable } P = 50.26 \times 654.65 = \underline{\underline{32.906 \text{ k}}}$$

0222 Results are about the same

XEQ "WCOL"
6:21:25 PM 11/02/88

TITLE:
RE-DESIGN: ROUND COLUMN
RUN

NDS
Fc PSI

841.7500 RUN

Fb PSI
RUN

E PSI
1,600,000.000 RUN

P, K
30.0000 RUN

ECCd"
RUN

LATd M, 'K
RUN

LDF
1.1500 RUN

K=27.2799

Le-b"
192.0000 RUN

Le-d"
192.0000 RUN

B"
0.0000 RUN

D"
0.0000 RUN

DIA"
8.0000 RUN

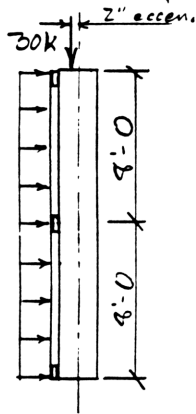
LONG
Fc'b=624.9633 PSI

LONG
Fc'd=624.9633 PSI

ALL P=31.4141 K-OK

Le-b"

Design a wall beam-column for the same structure. Assume wind @ 40 psf. Columns @ 10' spa.



Use same wood:

$$F_{c.o} = 441.75 \text{ psi}$$

$$E_o = 1,600,000 \text{ psi}$$

$$F_{b.o} = 1300 \text{ psi}$$

$$L_{e.b} = 8.0' = 96''$$

$$L_{e.d} = 16.0' = 192''$$

$$\text{With wind, LDF} = 1.33$$

$$\text{Without wind, LDF} = 1.15 \text{ (Snow)}$$

$$\begin{aligned} \text{Wind Moment} &= (40 \text{ psf} \times 10' \times 8') \times \frac{16'}{4} \\ &= 12.40 \text{ k} \end{aligned}$$

Try 6" x 10" $f_c = \frac{30,000}{6 \times 10} = 500 \text{ psi}$

Check minor-axis interaction:
(being the easiest to check)

$$K = .671 \sqrt{\frac{E}{f_{c.o}} \cdot \text{LDF}} = .671 \sqrt{\frac{1,600,000}{441.75 (1.15 \text{ or } 1.33)}}$$

$$K_{\text{wind}} = 25.367$$

$$K_{\text{snow}} = 27.240$$

∴ Check without wind

$$\frac{L_b}{B} = \frac{96''}{6} = 16.0$$

$$11 < 16.0 < 27.240 \Rightarrow \text{Intermed.}$$

$$F'_{c.b} = 441.75 \cdot 1.15 \left[1 - \frac{1}{3} \left(\frac{16}{27.240} \right)^4 \right] = 929.83 \text{ psi}$$

$$\text{Interaction}_b = \frac{500 \text{ psi}}{929.83} = .5377 < 1.0$$

Minor Axis OK

Check major axis, with wind:

$$S = \frac{6 \times 10^2}{6} = 100.0 \text{ in}^3 ; C_F \cdot C_F = 1.0$$

$$C_K = .811 \sqrt{\frac{E}{F_{b.o}}} = .811 \sqrt{\frac{1,600,000}{1300}} = 24.45$$

$$C_S = \sqrt{\frac{L_{e.b} \cdot d}{b^2}} = \sqrt{\frac{8' \cdot 12 \cdot 10''}{6^2}} = 5.16$$

$$5.16 < 10 \Rightarrow \text{"Short" for bending}$$

$$\therefore F'_{b.1} = F'_{b.2} = 1300 \cdot 1.33 = 1729 \text{ psi}$$

$$\text{In compression, } \frac{L_d}{D} = \frac{16' \cdot 12}{10'} = 19.20$$

$$11 < 19.2 < 25.367 \Rightarrow \text{Intermed. for col.}$$

$$F'_{c.d} = 441.75 \cdot 1.33 \cdot \left[1 - \frac{1}{3} \left(\frac{19.2}{25.367} \right)^4 \right] = 997.05$$

$$J = \frac{L_d - 11}{K - 11} = \frac{19.2 - 11}{25.367 - 11} = .5708$$

$$f_b = \frac{12.40 \text{ k} \times 12}{100 \text{ in}^3} = 1.536 \text{ ksi} \text{ this doesn't look good}$$

$$\text{Interaction} = \frac{f_c}{F'_{c.1}} + \frac{f_b + f_c (6 + 1.5J) (L'/D)}{F_{b.o} - J \cdot f_c}$$

$$= \frac{500}{997.05} + \frac{1536 + 500 (6 + 1.5 \cdot .5708) (2' / 10')}{1729 - .5708 (500)}$$

$$= 2.0404 > 1.0 \quad \underline{\underline{NG}}$$

To see which governs, check without wind:

$$J = \frac{19.2 - 11}{27.24 - 11} = .5037 ; f_b = 0$$

$$F'_{b.o} = 1300 \cdot 1.15 = 1495 \text{ psi}$$

$$F'_{c.d} = 441.75 \cdot 1.15 \cdot \left[1 - \frac{1}{3} \left(\frac{19.2}{27.240} \right)^4 \right] = 844.9365 \text{ psi}$$

$$\text{Interaction} = \frac{500}{844.94} + \frac{500 (6 + 1.5 \cdot .5037) (2' / 10')}{1495 - .5037 \cdot 500}$$

$$= 1.1060 > 1.0$$

Clearly, Loading with wind governs

(D223)

Wall Column

4/4

Try 6" x 14"

$$f_c = \frac{30,000}{6 \times 14} = 357.143 \text{ psi}$$

$$S = \frac{14^2 \times 6}{6} = 196.0 \text{ in}^3$$

$$f_b = \frac{12,401.12}{196.0} = 743.673 \text{ psi}$$

$$K = 25.367; C_k = 24.45 \text{ (previous page)}$$

$$C_s = \sqrt{\frac{L_e \cdot b \cdot D}{B^2}} = \sqrt{\frac{8' \times 12' \times 14'}{6^2}} = 6.11$$

$$6.11 < 10.0 \Rightarrow \text{"Short" for bending}$$

$$\therefore F_{b2} = 1300 \times 1.33 = 1729.0 \text{ psi}$$

$$C_F = 1.0 \text{ (rect.)}; C_F = \left(\frac{12}{14}\right)^{1/4} = .9430$$

$$\therefore F_{b1} = 1300 \times 1.33 \times .9430 = 1699.64$$

F_{b1} governs @ 1699.64 psi

$$\frac{L_e \cdot d}{D} = \frac{16' \times 12}{14} = 13.714$$

$$11 < 13.714 < K = \text{"Intermediate"}$$

$$\therefore F'_c = 441.75 \times 1.33 \left[1 - \frac{1}{3} \left(\frac{13.714}{25.367} \right)^4 \right]$$

$$= \underline{1047.647 \text{ psi}}$$

$$J = \frac{L/D - 11}{K - 11} = \frac{13.714 - 11}{25.367 - 11} = .1449$$

$$\text{Interaction} = \frac{f_c}{F'_c} + \frac{f_b + f_c(6 + 1.5J)(e/d)}{F'_{b1} - Jf_c}$$

$$= 1.0049 \approx 1.0 \quad \underline{\text{say OK}}$$

(less than 1% over)

D224

XERO "WOOD"
4:45:47 PM 01/25/88

TITLE:
WALL BEAM-COLUMN RUN

NDS
Fc PSI 841.7500 RUN
Fb PSI 1,300.0000 RUN
E PSI 1,600.000.000 RUN

P, K 30.0000 RUN
ECCd" 2.0000 RUN
LATd M, 'K 12.0000 RUN
LDF 1.3300 RUN
C-K=28.4518
K=25.3668

Le-b" 96.0000 RUN
Le-d" 192.0000 RUN
B" 6.0000 RUN
D" 10.0000 RUN
S=100.0000
C-S=5.1640
Fb'=1,729.0000 PSI

INTRM
Fc'b=1,060.4621 PSI

INTRM
Fc'd=997.0495 PSI

J=0.5708
INTRXN-d=2.0404-NG

Le-b" RUN
Le-d" RUN
B" 6.0000 RUN
D" 14.0000 RUN
S=196.0000
C-S=6.1101
Fb'=1,699.6381 PSI

INTRM
Fc'b=1,060.4621 PSI

INTRM
Fc'd=1,087.6455 PSI

J=0.1889
INTRXN-d=1.0049-NG

Le-b"

Say OK

	X	Y	Z	T	L		X	Y	Z	T	L
01*LBL "WBM"						50*LBL 00					
02 SF 09						51 E					
03 11						52 "LDF"					
04 GTO 00						53 PROMPT					
						54 STO 05					
05*LBL "WCOL"						55 .6					
06 CF 09						56 SQRT	$\sqrt{.6}$				
07 18						57 FS? 04					
						58 .811					
08*LBL 00						59 RCL 07	F_b	E			
09 SF 08						60 RCL 01					
10 "NDS"						61 /					
11 XROM "AA"						62 SQRT	C_k				$C_k = .411 \sqrt{E/F_c}$
						63 *					
12*LBL F						64 STO 10					
13 "Fc"						65 "C-K"					
14 FS? 09						66 FC? 10					
15 "F-V"						67 XEQ 05					
16 "t PSI"						68 FS? 09					
17 PROMPT	$F_c \text{ or } F_v$					69 GTO 1	E				
18 STO 09						70 RCL 07	F_c				
19 "Fb PSI"						71 RCL 09					
20 PROMPT						72 /	LDF				
21 STO 01	F_b					73 RCL 05					
22 "E PSI"						74 /					
23 PROMPT						75 SQRT					
24 STO 07	E					76 .671					
25 FC? 08						77 *	K				$K = .671 \sqrt{\frac{E}{LDF \cdot F_c}}$
26 GTO 10						78 STO 18					
						79 "K"					
27*LBL E						80 XEQ 05					
28 CF 10						81 SF 14					
29 XROM "L"											
30 FS? 09						82*LBL D					
31 GTO 00						83 XROM "LL"					
32 "P, K"						84 SF 06					
33 PROMPT						85 CF 07					
34 STO 12						86 "Le-b"	Le				Default - Previous Le
35 CLX						87 RCL 02					
36 "ECCd"						88 PROMPT					
37 PROMPT						89 STO 02					
38 STO 14						90 RCL 11					
39 "LATd"						91 "Le-d"	Le_d				
40 XROM "M"						92 FC? 09					
41 PROMPT						93 PROMPT					
42 *						94 STO 11					
43 STO 13	$M_d^{''4}$					95 XROM "BD"	D	B			
44 RCL 12	P					96 X=0?					
45 RCL 14	ECC_d	P	M			97 GTO 02	D^2				
46 *						98 X+2					
47 +	ΣM					99 STO 08					
48 X=0?											
49 SF 10											

	X	Y	Z	T	L		X	Y	Z	T	L
100 LASTX	D					153+LBL 10					
101 RCL 03	B					154 FS? 10					
102 X<Y?						155 GTO 12					
103 CF 06						156 RCL 02	$L_e \cdot b$				
104 *	Area					157 RCL 04	d				
105 STO 06						158 *					
106 RCL 04	D					159 RCL 03	b^2				
107 *						160 X12	b				
108 6						161 /					
109 /	S_x					162 SQR	C_s				
110 E	1					163 "C-S"					
111 GTO 03						164 XEQ 05					
						165 10	C_s	10			
112+LBL 02						166 X<Y					
113 SF 07						167 X<=Y?					
114 "DIA"						168 FC? 06					
115 PROMPT						169 .					
116 STO 03											
117 .866						170 RCL 10	C_x				
118 *	eff. d					171 X<=Y?					
119 STO 04	diem.					172 GTO 00					
120 X< 03						173 XEQ 04	Factor				
121 ENTER						174 RCL 08	$C_F \cdot C_F$				
122 X12	d^2	d				175 X>Y?					
123 PI						176 X<Y					
124 *						177 RCL 01	F_b				
125 4						178 *	F'_b				
126 /	area					179 RCL 05	LDF				
127 STO 06						180 *					
128 STO 08						181 GTO 01					
129 8	ϕ	area	d								
130 /						182+LBL 00					
131 *	S					183 CLX	50	C_s			
132 1.18	1.18					184 50					
						185 X<Y?	0?				
133+LBL 03						186 CLX					
134 X<Y	S	C_F				187 X=0?					
135 "S"						188 GTO 01	E				
136 STO 00						189 RCL 07					
137 FC? 10						190 FC? 04	.4	E	50	C_s	
138 XEQ 05						191 .4					
139 12	12					192 FS? 04					
140 RCL 08	d^2	12	S	C_F		193 .438					
141 SQR	eff. d					194 *	.4 E				
142 /	$12/d$					195 R1	C_s				
143 9						196 X12					
144 1/X	1/9	$12/d$	S	C_F		197 /	$\frac{.4 E}{(C_s)^2}$				
145 Y1X											
146 E	1	$(\frac{12}{d})^4$	S	C_F		198+LBL 01					
147 X>Y?						199 "Fb"					
148 X<Y	C_F					200 XEQ 06					
149 R1	C_F					201 FC? 09					
150 *	$C_F \cdot C_F$					202 GTO 12					
151 STO 08						203 RCL 00	S	F'_b			
152 CF 08						204 *					
						205 XROM "M="					

$$C_s = \sqrt{\frac{L_e \cdot b \cdot d}{b^2}}$$

Short

Intermed.

$$S = \frac{\pi d^3}{32}$$

Long

0226

206 2
207 AVIEW
208 3
209 FS? 07
210 4
211 /
212 E3
213 /
214 RCL 09
215 *
216 RCL 05
217 *
218 "ALL V="

222+LBL 12
223 STO 17
224 RCL 02
225 RCL 03
226 "Fc*b"

227 XEQ "MFC"
228 STO 16
229 RCL 11
230 RCL 04
231 "Fc*d"

232 XEQ "MFC"
233 FC? 10
234 GTO 00
235 RCL 16
236 X>Y?
237 X<Y
238 "t P="

239 XEQ 07

240 RCL 12
241 GTO 11

242+LBL 07
243 RCL 06
244 *
245 ARCL X
246 "t K"

247 RTN

248+LBL 00
249 RCL 11
250 RCL 04
251 /
252 11
253 -
254 X<0?
255 CLX
256 "J"

$$d \square b \quad f_v = \frac{3}{2} \frac{V}{bd}$$

$$\text{circle with } f_v \text{ and } V \quad f_v = \frac{4}{3} \frac{V}{\text{area}}$$

Note: if from
line 155, R.17
does not contain
 F_b . In this case
(no moment), F_b
is not used.

257 RCL 18
258 LASTX
259 -
260 /
261 ABS
262 SIGN
263 LASTX
264 X=Y?
265 X<Y
266 RDN
267 XEQ 05
268 STO \\\n269 1.5
270 *
271 6
272 +
273 RCL 14
274 RCL 04
275 /
276 *
277 RCL 12
278 RCL 06
279 /
280 STO 15
281 *
282 RCL 13
283 RCL 00
284 /
285 +
286 RCL \\\n287 RCL 15
288 *
289 RCL 17
290 -
291 CHS
292 /
293 RCL 15
294 RT
295 /
296 +
297 RCL 15
298 RCL 16
299 /
300 "INTRXN-"
301 X=Y?
302 "t d="

303 X>Y?

304 "t b="

305 X=Y?

306 X<Y

307 SIGN

308 LASTX

309 ARCL X

K
11
11-K
J
J
1
J
greater
J
J
J
K
eccen.
d
e/d
K
P
area
 f_c
Mink
S
 f_b
num.
J
 f_c
J
 f_c
denom
Z
 f_c
Z
 f_c
Interaction-d
 f_c
Interaction-b
greater
1
Interaction

$$\text{Interaction} = \frac{f_c}{F_c} + \frac{f_b + f_c(6+5J)}{F_c - Jf_c}$$

	X	Y	Z	T	L		X	Y	Z	T	L	
310+LBL 11												
311 CF 07												
312 X>Y?												
313 SF 07												
314 XROM "OK"												
315 AVIEW												
316 GTO D												
317+LBL 04	K	$\frac{L}{b}$ or 0	11	F_b								
318 /	$\frac{L}{b}$											
319 X†2												
320 X†2	() ⁴											
321 3												
322 /	1											
323 SIGN												
324 LASTX												
325 -	1-() ⁴	11	F_b	F_b								
326 RTH												
327+LBL 05												
328 "t="												
329 ARCL X												
330 FC? 00												
331 AVIEW												
332 RTH												
333+LBL "Wfc"	b	L_b										
334 ASTO 15												
335 "LONG"												
336 /	$\frac{L}{b}$											
337 50	50	$\frac{L}{b}$										
338 X<=Y?												
339 CLX												
340 X<=Y?												
341 GTO 08												
342 CLX												
343 11	11	$\frac{L}{b}$										
344 X<>Y	$\frac{L}{b}$	11										
345 X<=Y?												
346 .	0											
347 RCL 18	K	0 or $\frac{L}{b}$ or 11										
348 X<=Y?												
349 GTO 09												
350 XEQ 04	1-() ⁴											
351 "INTRM"												
352 E	1											
353 X<>Y	1-() ⁴	1										
354 X=Y?												
355 "SHORT"												
356 RCL 09	F_c											
357 *												
358 RCL 05	LDF											
359 *	Allow. F_c'											
360 GTO 08												
361+LBL 09												
362 RCL 07	K	$\frac{L}{b}$										"LONG" column
363 .3	E	E	K	$\frac{L}{b}$								
364 *	.3											
365 R†	.3E											
366 X†2	$\frac{L}{b}$											
367 /	F_c'											Allow. $F_c' = \frac{.3E}{(\frac{L}{b})^2}$
368+LBL 08												
369 FC? 00												
370 AVIEW												
371 CLA												
372 ARCL 15	$F_c'-b$											" $F_c'b$ "
373+LBL 06												
374 "t="												
375 ARCL X												
376 E3	1000											
377 /	$F_c' \text{ ksi}$											
378 "t PSI"												
379 FC? 00												
380 AVIEW												
381 ADV												
382 "ALL"												
383 END	$F_c' \text{ ksi}$											
<p>"$F_c'b$" stored</p> <p>$\frac{L}{b} > 50 \Rightarrow \text{Allow. stress} = 0$</p> <p>If $\frac{L}{b} < 11$, $K > 0 \Rightarrow \text{don't GTO 09}$</p> <p>If $\frac{L}{b} > 11$, then if $K < \frac{L}{b}$, GTO 09.</p>												

SIMPLE SPAN TRUCK MOMENTS

xeq T2 for Type 2 (H15 or H20) truck* (SIZE 013)
T3 for Type 3 straight truck
T4 for Type 4 straight truck (triple axle)
2S1 for HS20 configuration*
3S2 for Type 3S2 semi-trailer truck
3S3 for Type 3S3 semi-trailer truck (triple axle)

*Generic trucks only--lane loading not calculated by these routines. RAM program "TRUCK" contains a routine which calculates and compares lane loadings.

SOFTKEYS (USER mode ON)

No softkeys are used by the truck routines

USER FLAGS

No user flags are used by the truck routines

INPUT/OUTPUT SUMMARY

These routines have no prompts or output explanation. All the required data must be stored in the following registers by a calling program, and the results displayed by that program. The truck routines are simple numerical calculation routines; output consists of the governing moment for that truck in the X-register and register 12. An example calling program which generates the moments for Minnesota trucks is included in the main documentation for use or reference.

<u>Register</u>	<u>Input</u>
00	Span in feet
01	P1 (front wheel load), kips
02	P2 (second wheel load), kips
03	P3 (third wheel load), kips
04	L2 (distance between the second and third wheel of a type 2S1 truck) in feet
05	L1 (distance between the front two wheels), feet
06	L2 (distance between the middle and rear sets of wheels), feet

PROGRAM FLAGS

No program flags are used by the truck routines

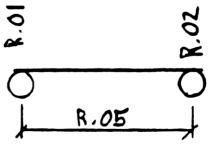
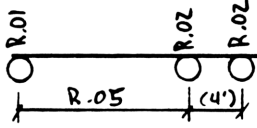
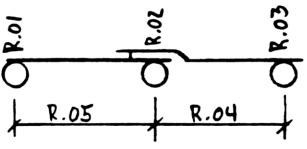
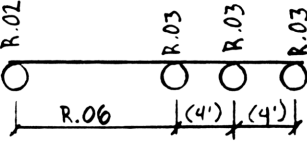
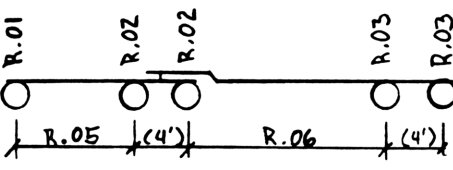
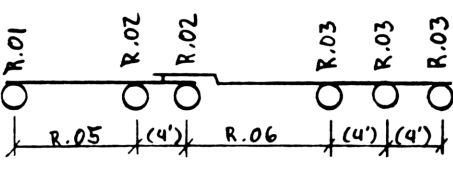
STORAGE REGISTER USE

<u>Register</u>	<u>Value</u>
00	Span in feet
01	P1 (front wheel load), kips
02	P2 (second wheel load), kips
03	P3 (third wheel load), kips
04	L2 (distance between the second and third wheel of a type 2S1 truck) in feet
05	L1 (distance between the front two wheels), feet
06	L2 (distance between the middle and rear sets of wheels), feet
07	$P \times L$
08	Minimum P
09	$-(\sum P \times L) / \sum P$
10	Maximum P
11	$\sum P$
12	Governing Moment

Storage Register Use - Input

Each truck type, and corresponding program label, is named for the number of axles. Semi-trailer truck designations have the number of axles in both the tractor and trailer.

Storage register use is as shown below. The user-written calling program must fill these registers with the appropriate values. Where a dimension is assumed by the programs, it is enclosed in parenthesis.

<u>Truck</u>		<u>Assumptions made</u>
T2		None.
T3		Rear wheels identical, 4' apart. (But, see 2S1)
2S1		None.
T4		Rear wheels identical, 4' apart. R.06 length assumed long enough that last 3 axles govern over first 3.
3S2		Wheels which are 4' apart are identical.
3S3		Wheels which are 4' apart are identical.

The span in feet must be stored in R.00 for all routines.

Program TRUCK For Minnesota Truck Moments 198 bytes Page 1 of 1

01*LBL "TRUCK"
02 CF 07
03 12
04 XROM "S"

05*LBL 00
06 "SPAN"
07 PROMPT
08 STO 00
09 "I="

10 50
11 X<>Y
12 125

13 +
14 /
15 .3

16 X>Y?
17 X<>Y
18 ARCL X
19 FC? 00
20 AVIEW
21 ADV

22 7
23 STO 01
24 8.5
25 STO 02
26 12
27 STO 05
28 XEQ "T3"

29 "T3"
30 XEQ 01
31 4.65
32 STO 01
33 8

34 STO 02
35 STO 03
36 10
37 STO 05
38 25

39 STO 06
40 XEQ "3S2"

41 "3S2U"
42 XEQ 01
43 4.5
44 STO 01
45 6.5
46 STO 03

47 XEQ "3S3"

48 "3S3"
49 XEQ 01
50 ADV
51 4

52 STO 01
53 16
54 STO 02
55 STO 03

56 14
57 STO 04
58 STO 05
59 XEQ "T2"
60 "H 20"
61 XEQ 03
62 XEQ "2S1"
63 "HS20"
64 XEQ 03
65 XROM "LL"
66 GTO 00

67*LBL 03
68 CF 07
69 RCL 00
70 4
71 *

72 225
73 +
74 E2
75 /

76 RCL 00
77 *
78 RCL 12
79 X>Y?
80 GTO 01
81 X<>Y
82 STO 12
83 SF 07

84*LBL 01
85 "I: M="

86 ARCL X
87 XROM "FK"
88 FS? 07
89 "I(L)"

90 FC? 00
91 AVIEW
92 END

L₁
L₂
M₂

M_{2S1}

Span

4S

Span
Lane M
Truck M

Lane M

XEQ "TRUCK"
9:03:25 AM 02/08/89

TITLE:
MINNESOTA TRUCKS RUN

SPAN' 60.0000 RUN
I=0.2703

T3: M=301.4340 'K
3S2U: M=297.9510 'K
3S3: M=292.8094 'K
H 20: M=279.0000 'K(L)
HS20: M=403.2667 'K

SPAN'

Lane Loading:
 $M = \frac{1}{2} \cdot (.64 \cdot S^2/8 + 18k \cdot S/4)$

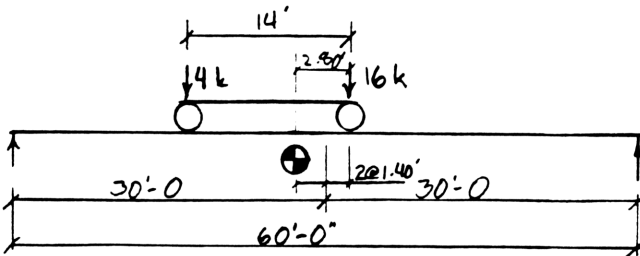
(L) → Lane
loading
governs

P₂ = 8 - still stored

0232

Truck T2

Analyze the moment
due to an H2O truck
configuration on a
60'-0" span.



One wheel

$$\frac{16k \times 60'}{4} = \underline{240.0''k}$$

Two wheels

$$C.G. = \frac{4k \times 14'}{4k + 16k} = 2.80' \text{ from } 16k \text{ wheel}$$

$$R_{\text{Right}} = \left(\frac{60'}{2} - \frac{2.80'}{2} \right) \times \frac{20k}{60'} = 9.53k$$

$$M = \left(\frac{60'}{2} - \frac{2.80'}{2} \right) \times 9.53k = \underline{272.65''k} \leftarrow \text{Governs}$$

Check lane loading:

$$W = .64 k/\text{foot per lane}$$

$$P = 16.0 k/\text{lane}$$

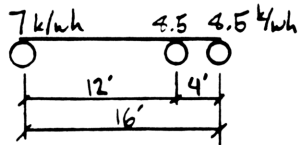
$$\begin{aligned} \text{Per wheel: } & .32 \times 60^2/8 = 144.0 \\ & 9.0 \times 60/4 = 135.0 \\ & 279.00''k/\text{wheel} - \text{Lane load governs} \end{aligned}$$

Note: Routine T2 does not figure lane loads. However,
user program TRUCK does.

Analyze the moment due to a Minnesota Type 3 truck as shown.

Truck T3

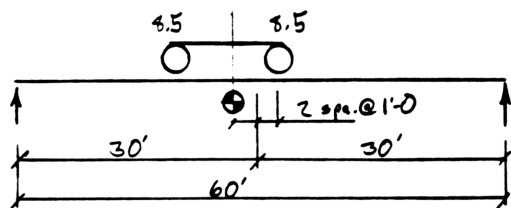
Span = 60'-0", simple.



One wheel

$$M = \frac{Pl}{4} = \frac{4.5k \cdot 60'}{4} = 127.50''k$$

Two wheels



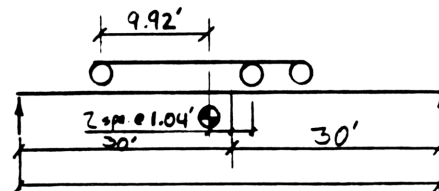
$$C.G. = \frac{4.5k \cdot 4'}{2 \cdot 4.5k} = 2.0'$$

$$R_R = (30' - \frac{2.0'}{2}) \times 17.0k / 60' = 8.22k$$

Max. M under right tire:

$$M = 8.22k \cdot (30' - \frac{2.0'}{2}) = 238.28''k$$

Three wheels



$$C.G. = \frac{4.5k \cdot 12' + 4.5k \cdot 16'}{2 \cdot 4.5k + 7} = 9.92'$$

$$R_R = \frac{(30' - \frac{2.08'}{2})}{60'} \times 24k = 11.583k$$

Max. M under middle tire:

$$M = 11.583k \cdot (30' - \frac{2.08'}{2})$$

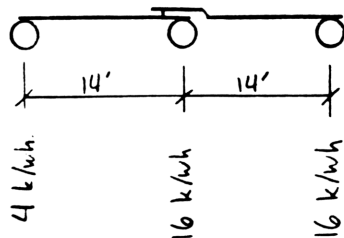
$$- 4.5k \cdot 4'$$

$$= 301.43''k$$

Max. = 301.43''k/wheel line

Analyze the moment due to an HS20 truck configuration as shown.

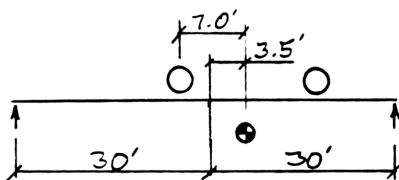
Span = 60'-0, simple.



One wheel

$$M = \frac{Pl}{4} = \frac{16k \times 60'}{4} = 240.0''k$$

Two wheels



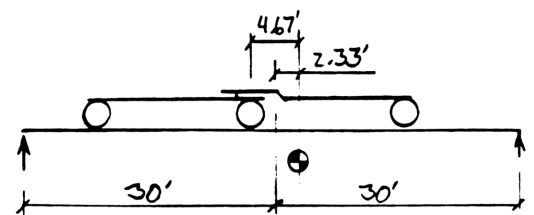
By inspection, C.G. = 7.0'

$$R_L = (30' - \frac{7.0'}{2}) \times \frac{32k}{60'} = 14.13 k$$

$$M = (30' - 3.5') \times 14.13 = \underline{374.53''k}$$

Truck 2S1

Three wheels



$$C.G. = \frac{14' \times 16k - 14' \times 4k}{36k} = 4.67'$$

$$R_L = (30' - \frac{4.67'}{2}) \times \frac{36k}{60'} = 16.60 k$$

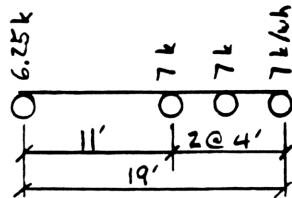
$$M = (30' - \frac{4.67'}{2}) \times 16.60k - 4k \times 14' = \underline{403.27''k}$$

Max. M = 403.27''k

Analyze the moment due to an Iowa Type 4 truck as shown.

Truck T4

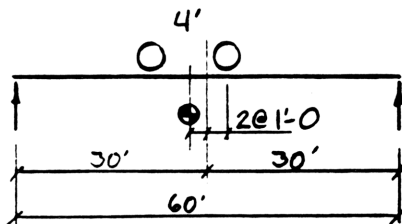
Span = 60'-0, simple.



One wheel

$$\frac{P}{4} = 7k \times 60/4 = \underline{105.00''k}$$

Two wheels



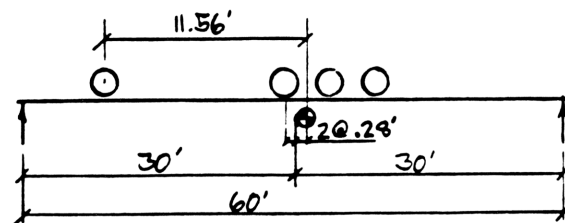
By inspection, C.G. @ 2.0'

$$R_R = \frac{(30' - \frac{2.0'}{2})}{60'} \times (2 \times 7k) = 6.76 k$$

Max. M under right tire:

$$M = (30' - \frac{2.0'}{2}) \times 6.76 k = \underline{196.23''k}$$

Four wheels



$$C.G. = \frac{21k \times 15'}{27.25k} = 11.56' \text{ from front}$$

$$R_L = (30' - \frac{5.6'}{2}) \times 27.25k / 60' = 13.50 k$$

Moment under 2nd tire:

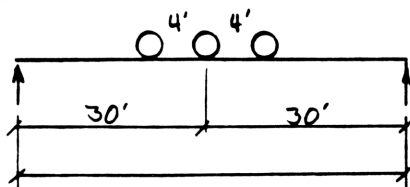
$$M = (30' - \frac{5.6'}{2}) \times 13.50 = 401.16 \\ - 6.25k \times 11' = \underline{-62.75} \\ 332.41''k$$

Check for max. M under 3rd tire.

$$R_R = (30' - \frac{3.44'}{2}) \times \frac{27.25k}{60'} = 12.84 k$$

$$M = (30' - \frac{3.44'}{2}) \times 12.84 - 7k \times 4' = \underline{335.22''k}$$

Three wheels



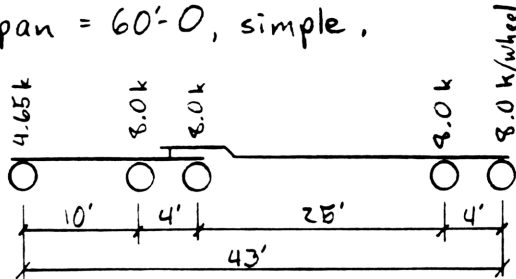
By inspection, $R_R = R_L = \frac{3 \times 7k}{2} = 10.5 k$

$$M = 30' \times 10.5 k - 4' \times 7k = \underline{287.00''k}$$

Max. M = 335.22''k/wheel line

Analyze the moment due to a Minnesota Type 3S2 truck as shown.

Span = 60'-0, simple.



In the interest of brevity, calculations similar to those of previous pages will be omitted or shortened.

One wheel $M = 120.0''k$

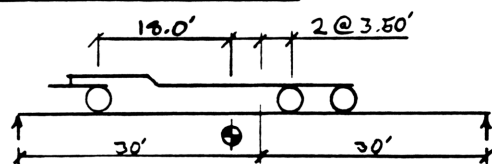
Two wheels $M = 224.27''k$

Three wheels (tractor)

$$C.G. = \frac{8.0 \times (10 \times 14)}{20.65k} = 9.30' \text{ from front wh.}$$

$$M = (30' - \frac{7.0}{2}) \times \frac{20.65k}{60'} - 8k \times 4' = \underline{270.54''k}$$

Three wheels (trailer)



$$C.G. = \frac{8.0 \times (25' - 29')}{24.0k} = 18.0'$$

$$R_R = (30' - \frac{7.0}{2}) \times \frac{24.0k}{60'} = 10.60k$$

$$M = (30' - 3.5') \times 10.60 - 8k \times 4' = \underline{248.90''k}$$

Truck 3S2

Four wheels (rear)

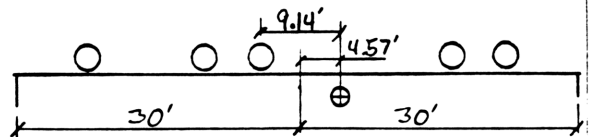
By inspection, C.G. = 16.5' from front

$$R_{min} = (30' - \frac{12.5}{2}) \times \frac{32.0k}{60'} = 12.67k$$

$$M = (30' - \frac{12.5}{2}) \times 12.67k - 8 \times 4' = \underline{268.93''k}$$

Five wheels

Max. M under 3rd wheel.



$$C.G. = \frac{8.0 \times (25' - 29') - 8 \times 4 - 4.65 \times 14}{36.65} = 9.14'$$

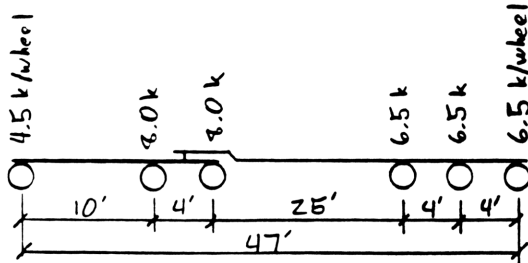
$$R_L = (30' - \frac{9.14}{2}) \times \frac{36.65k}{60'} = 15.53k$$

$$M = (30' - \frac{9.14}{2}) \times 15.53k - (8k \times 4' + 4.65k \times 14') = \underline{297.95''k}$$

$$\underline{\underline{Max. M = 297.95''k}}$$

Analyze the moment due to a Minnesota Type 353 truck as shown.

Span = 60'-0, simple.



In the interest of brevity, calculations similar to those of previous pages will be omitted or shortened.

One wheel $M = 120.0''^k$

Two wheels $M = 224.27''^k$

Three wheels (tractor)

$$C.G. = \frac{8.0 \cdot (10' + 14')}{20.50k} = 9.37'$$

$$M = (30' - \frac{9.37'}{2})^2 \times \frac{20.50k}{60'} - 8k \cdot 4' = \underline{269.03''^k}$$

Rear three wheels

$$M = 30' \cdot \frac{3 \cdot 6.5k}{2} - 6.5k \cdot 4' = \underline{266.50''^k}$$

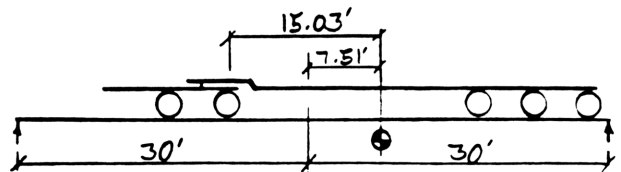
Rear four wheels

$$(as for Truck T4) : M = \underline{278.66''^k}$$

(D38)

Truck 353

Rear five wheels



Under 2nd wheel:

$$C.G. = \frac{3 \cdot 6.5k \cdot 29' - 8k \cdot 4'}{35.5k} = 15.03'$$

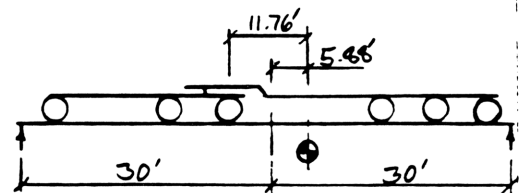
$$R_L = (30' - \frac{15.03'}{2}) \times 35.5k/60' = 13.30k$$

$$M = (30' - 7.51') \cdot 13.30k - 8k \cdot 4' = \underline{267.16''^k}$$

Under 3rd wheel : $C.G. = 25' - 15.03' = 9.97'$

$$M = (30' - \frac{9.97'}{2})^2 \times \frac{35.5k}{60'} - 6.5k \cdot (4' + 8') = \underline{292.21''^k}$$

Six wheels - Max. under 3rd wheel



$$C.G. = \frac{3 \cdot 6.5 \cdot 29' - 8 \cdot 4' - 4.5 \cdot 14'}{40.0k} = 11.76'$$

$$R_L = (30' - 5.88') \cdot \frac{40k}{60'} = 16.08k$$

$$M = (30' - 5.88') \cdot 16.08 - (8 \cdot 4' + 4.5 \cdot 14') = \underline{292.81''^k}$$

$$\underline{\underline{Max. M = 292.81''^k}}$$

	X	Y	Z	T	L		X	Y	Z	T	L
01*LBL "3S2"						57 RCL 07	ΣP_{front}				
02 XEQ 13	M_1					58 -	M_5				
03 XEQ 03	M_3				M_3 for front	59 GTO 22					
04 RCL 03	P_3				3 wheels						
05 RCL 02	P_2				(tractor)	60*LBL "3S3"					
06 X>Y?						61 XEQ 13	M_1				
07 X<Y	P_{min}					62 XEQ 03	M_3				
08 STO 08						63 XEQ 04	M_4				M_4 for rear
09 X<Y						64 RCL 03	P_3				3 wheels
10 STO 10	P_{max}	P_{min}				65 6	(w-z')				(trailer)
11 XEQ 12	M_2					66 *	$\Sigma P_3 l_3$				
12 RCL 10	P_{max}					67 RCL 06	l_2				
13 RCL 10	✓					68 2					
14 RCL 08	P_{min}					69 +					
15 RCL 06	l_2	P_{min}	P_{max}	P_{max}		70 RCL 02	P_2				
16 XEQ 14	M_3				M_3 for rear	71 *					
17 RCL 10					3 wheels	72 -	$\Sigma P \cdot d_5$				
18 ST+ X	ΣP_{max}					73 ST+ X					
19 RCL 06	l_2					74 RCL 11	ΣP_{rear}				
20 2						75 RCL 02	P_2				
21 +	$l_2 - 2$					76 +	ΣP_{rear}	$\Sigma P d_5$			
22 RCL 08	P_{min}					77 XEQ 10	M_5				
23 *						78 RCL 03	P_3				
24 -	$\Sigma P d$					79 12	(s-u')				
25 RCL 10						80 *					
26 RCL 08						81 -	M_{5A}				M_{5A} under
27 +	ΣP	$-\Sigma P d$				82 XEQ 22	l_2				1st axle of
28 XEQ 10	$M_{1/2}$					83 RCL 06					triple
29 ST+ X	M'					84 RCL 09	$-d_{5A}$				
30 RCL 10	P_{max}					85 +					
31 4	$4'$					86 CHS	d'_{5B}				
32 *	$4' \cdot P$					87 XEQ 05	M'_{5B}				
33 -	M_4				M_4 for rear	88 RCL 02	P_2				
34 XEQ 22					4 wheels	89 4					
35 RCL 02	P_2					90 *	$4' \cdot P_2$				
36 4						91 STO 07					
37 *	$4P_2$					92 -	M_{5B}				M_{5B} under
38 RCL 05	l_1					93 XEQ 22					2nd axle
39 LASTX	4					94 RCL 05	l_1				of tractor
40 +	$l_1 \cdot 4$					95 4					dual
41 RCL 01	P_1					96 +					
42 *						97 RCL 01	P_1				
43 +	$P_1(l_1 - 4)$					98 *	$P_1(l_1 - 4)$				
44 STO 07	$\Sigma P d_{\text{front}}$					99 ST+ 07					$\Sigma P d_{\text{front}}$ in R.O.
45 RCL 06	l_2					100 RCL 09	$-d_{5A}$				
46 ST+ X						101 RCL 06	l_2				
47 4						102 +	d_{5B}				
48 +						103 RCL 11	ΣP_5				
49 RCL 03	P_3					104 *					
50 *	$P_3(l_1 - 4/2)$					105 -	$-\Sigma P d_6$				
51 -	$\Sigma P \cdot d$					106 RCL 11	ΣP_5				
52 RCL 11	ΣP_4					107 RCL 01	P_1				
53 ST+ X	ΣP_4					108 +	ΣP_6				
54 RCL 01	P_1					109 XEQ 10	M'_6				
55 +	ΣP_5					110 RCL 07					
56 XEQ 10	M'_5					111 -					
						112 GTO 22	M_6				

	X	Y	Z	T	L		X	Y	Z	T	L
113*LBL "T4"											
114 RCL 03	P ₂					168*LBL 14	l ₁	P ₁	P ₂	P ₂₊₃	
115 XEQ 11	M ₁					169 X<>Y	P ₁	l ₁			
						170 ST+ Z			ΣP		
116*LBL 04	P ₂					171 *	P ₁ l ₁	ΣP	P ₂₊₃	P ₂₊₃	
117 RCL 03	M ₂					172 STO 07					
118 XEQ 12	P ₂					173 RCL 04	l ₂				
119 RCL 03	Span					174 R↑	P ₂₊₃	l ₂	P ₁ l ₁	ΣP ₂	
120 RCL 00						175 ST+ T				ΣP	
121 3						176 *	P ₁ l ₂	P ₁ l ₁			
122 *						177 -	-ΣP ₁	ΣP			
123 4						178 X<>Y	ΣP	-ΣP ₁			
124 /	35/4	P ₂				179 XEQ 10	M ₃				
125 LASTX	4					180 RCL 07	P ₁ l ₁				
126 -	35-4					181 -	M ₃				
127 *						182 GTO 22					
128 XEQ 22	M _{triples}										
129 RCL 02	P ₁					183*LBL "2S1"					
130 RCL 06	l					184 XEQ 13	M ₁				
131 4						185 RCL 03					
132 +	l-4					186 RCL 02					
133 *	P ₁ (l-4)					187 RCL 01					
134 CHS	-P ₁ d					188 RCL 05	l ₁	P ₁	P ₂	P ₃	
135 RCL 03	P ₂					189 XEQ 14	M ₃				
136 3						190 RCL 03					
137 *	ΣP _{triples}					191 RCL 02					
138 RCL 02	P ₁					192 RCL 04	l ₂	P ₂	P ₃		
139 +	ΣP	-ΣP ₁ d				193 XEQ 02	M ₂ max				
140 XEQ 10	M ₁					194 RCL 01					
141 RCL 03	P ₂					195 RCL 02					
142 4						196 RCL 05	l ₁	P ₂	P ₁		
143 *	4P					197 GTO 02					
144 -	M _{4A}										
145 XEQ 22						198*LBL 12	P				
146 RCL 09	-l _A					199 ENTER↑					
147 4						200 ENTER↑	P	P	P		
148 +	-l _B					201 4	4	P	P		
149 CHS						202 GTO 02					
150 XEQ 05	M ₁										
151 RCL 02	P ₁					203*LBL "T2"					
152 RCL 06	l ₁					204 RCL 01	P ₁				
153 *	P ₁ l ₁					205 XEQ 11					
154 -	M _{4B}					206 RCL 01	P ₁				
155 GTO 22						207 RCL 02	P ₂				
						208 RCL 05	l				
156*LBL "T3"											
157 RCL 01						209*LBL 02					
158 XEQ 11	M ₁					210 RDN	P ₂	P ₁		l	
						211 X>Y?					
159*LBL 03						212 X<>Y	P _{min}	P _{max}		l	
160 4						213 ST+ Y	P _{min}	ΣP			
161 STO 04	l ₂					214 R↑	l				
162 RCL 02	P ₂					215 *	P _{min} l	ΣP			
163 XEQ 12	M ₂					216 CHS					
164 RCL 02						217 X<>Y	ΣP	-P ₁ l			
165 RCL 02						218 XEQ 10	M ₂				
166 RCL 01											
167 RCL 05	l ₁	P ₁	P ₂	P ₂							

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	X	Y	Z	T	L		X	Y	Z	T	L
219+LBL 22	M					268+LBL "ACO"					
220 RCL 12	M _{max}	M				269 STO ↑					
221 X<=Y?						270 ACOS					
222 X<>Y						271 GTO 08					
223 STO 12	M _{max}										
224 RTN						272+LBL "ASI"					
						273 STO ↑					
225+LBL 10	ΣP	-Pl				274 ASIN					
226 STO 11	-l					275 GTO 08					
227 /											
228 STO 09						276+LBL "ATA"					
						277 STO ↑					
229+LBL 05						278 ATAN					
230 RCL 00											
231 +	Sp _u -l					279+LBL 08					
232 2	Σ-l					280 FS? 03					
233 /	Σ-l					281 HMS					
234 X>0?	() ²										
235 X+2	ΣP					282+LBL 07	result				angle
236 RCL 11						283 X<> L	ang _u .dec				result
237 *						284 CLX					
238 RCL 00	span					285 X<> ↑	orig.ang _u				note: clears Reg. P
239 /						286 X<> L	result				orig.ang _u
240 RTN	M'					287 RTN					
241+LBL 13	P ₁					288+LBL 09					
242 RCL 01	P ₃					289 STO ↑	angle				
243 RCL 03						290 FS? 03					
244 X<Y?						291 HR	Dec.ang				
245 X<>Y	P _{max}					292 END					
246+LBL 11	P										
247 RCL 02	P ₂										
248 X<Y?											
249 X<>Y	P _{max}										
250 RCL 00	span										
251 *											
252 4	P _l /4										
253 /	M ₁										
254 STO 12											
255 RTN											
256+LBL "CO"											
257 XEQ 09	Dec.ang										
258 COS											
259 GTO 07											
260+LBL "SI"											
261 XEQ 09	Dec.ang										
262 SIN											
263 GTO 07											
264+LBL "TA"											
265 XEQ 09	Dec.ang										
266 TAN											
267 GTO 07											

RETAINING WALL ANALYSIS

xeg RW (SIZE 016)

SOFTKEYS (USER mode ON)

d Jump directly to wall DEPTH prompt
F New footing dimensions
H New height and dimensions

Note: softkeys A, B, and D are used by program SSD and are active while RW is being run

USER FLAGS

Flag 00	clear	Full output
	set	Partial vert. loads and moments skipped
Flag 03	clear	Slope and Phi angles in decimal degrees
	set	Angles in degrees-minutes-seconds

INPUT SUMMARY

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
SLOPE	Slope angle from horizontal	Zero
PHI	Angle of internal friction	No default
VERT E PCF	Vertical earth weight in pounds per cubic foot	No default
SUR PSF	Vertical surcharge in pounds per square foot	No default
LAT E PCF	Lateral earth load expressed as an equivalent fluid, in pcf	$C_a \times \text{Vert.}$
SUR PSF	Lateral surcharge in psf	No default
FRIC μ	Coeff. of sliding friction, μ	No default
HT'	Height from <u>bottom</u> of footing to top of backfill, in feet	No default
F TH'	Footing thickness in feet	No default
COV'	Soil depth over the toe (front) of the footing, measured from the top of the footing	No default
T/W TH'	Thickness of the cantilever wall at top and bottom of the wall	No default
B/W TH'		

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
P, K	Additional vertical load applied to the top of the stem, in kips (e.g. rail DL or part of the wall above top of backfill)	No default
P CG'	Location of the load P, measured back from the front face of the <u>stem</u> , in feet	No default
FTG W'	Footing width in feet	No default
TOE'	Distance from the front of the footing to the front face of the stem (wall)	No default
D'	Location along the wall (depth) at which to figure moment and shear, measured down from top of backfill	No default

OUTPUT SUMMARY (Items in parenthesis are displayed only if flag 00 is clear)

Note: all moments caused by vertical loads are taken about the toe of the footing.

C_a	Coefficient of active pressure
(CONC: V,M)	Vertical load and moment due to the footing and stem, in kips and foot-kips
(P: V,M)	Vertical load and moment due to the load applied to the top of the wall
(VERT E: V,M)	Vertical load and moment due to vertical earth load
(SUR: V,M)	Vertical load and moment due to vertical surcharge. Only if prompt was answered.
(ΣV , M, RES)	Sum of the vertical loads and the moments due to these loads about the toe of the footing, and the position of the resultant measured in feet from the toe.
SLIDE FS	Sliding factor of safety, neglecting passive resistance and any key present
TOE P HEEL P	Pressure under the heel and toe for a spread footing
LAT E: H,M SUR: H,M $\Sigma H,M$	Horizontal shear and moment due to lateral earth pressure and surcharge, and their summations.

PROGRAM FLAGS

RW uses no program flags.

STORAGE REGISTER USE

<u>Register</u>	<u>Value</u>
00	Summation of vertical forces
01	Summation of moments
02	Lateral earth equiv. fluid pressure (k/cf stored)
03	HT' (height from bottom of footing)
04	F TH'
05	Footing width
06	TOE' (distance from front of footing to the stem)
07	ST TH' (stem thickness)
08	Lateral surcharge (ksf stored)
09	Coefficient of friction, mu
10	P, K (rail DL)
11	C.G. P (point of application of rail DL from the front face of the wall)
12	COV' (Soil cover over the toe)
13	Vertical earth weight (k/cf stored)
14	Vertical surcharge (ksf stored)
15	Working register

RETAINING WALL ANALYSIS

Program RW calculates the loads and reactions due to selfweight, soil, and surcharge (if any) on cantilever retaining walls with cohesionless backfill. Input consists of the wall and footing dimensions, soil parameters, and surcharge loadings. RW calculates the resulting moments and horizontal and vertical forces at bottom of footing, bottom of wall, and any desired points along the wall, as well as the heel and toe reactions and sliding factor of safety (assuming a spread footing).

SOIL PROPERTIES

RW prompts first for the backfill angle from the horizontal (see sketch), the internal friction angle ϕ (phi) of the soil, and the coefficient of friction (μ) between soil and footing concrete. Typical values for μ , from the 1958 AREA specifications, are as follows:

<u>Soil</u>	<u>Mu</u>
Sound rock, rough surface	0.60
Coarse-grained, no silt	0.55
Coarse-grained, with silt	0.45
Silt	0.35

The program solves the Rankine/Coulomb formula for the active pressure coefficient C_a for lateral loading. This value is not used directly by the program; however, it is multiplied by the soil density (usually about 120 pcf) to calculate the default value for the equivalent fluid density used to calculate lateral pressure (usually about 35 pcf). If the designer agrees with the assumptions involved in this formula (cohesionless soil; essentially vertical pressure surface with a prescribed amount of friction between soil and wall), the default lateral earth pressure may be used. The formula used is:

$$C_a = \cos(B) \times \frac{\cos(B) - \sqrt{\cos(B) - \cos(\phi)}}{\cos(B) + \sqrt{\cos(B) - \cos(\phi)}}$$

B = backfill angle (0° = horizontal)
 ϕ = angle of internal friction
cohesion assumed zero

The user inputs the actual soil weight for vertical loading, and the equivalent fluid weight for horizontal loading, using the default if desired. If the user wants to check a range of soil values (e.g. a lighter soil weight for potential heel uplift), separate runs must be made.

The user may also input surcharge loading, both vertical and horizontal, if desired. This is input in psf.

OUTPUT

The program calculates a force (vertical or horizontal) and a moment due to each component of load. These are output separately (in order to allow for either working stress or load factor methods of design) and then summed. If flag 00 is set, only the sums are displayed. An average wall thickness is used in the calculations; this leads to a minor error in the calculation of vertical surcharge loading.

After calculating vertical and horizontal forces and moments about the bottom of footing, the program calculates the sliding factor of safety and the toe and heel bearing stresses. These are valid, of course, only for spread footings. The sliding factor of safety is calculated using the input friction factor " μ " (μ) or the value of $\tan \phi$, whichever is less. The former represents sliding between the footing and the soil; the latter represents shear failure of the soil under the footing. (Thus, the PHI prompt should not be bypassed, since it affects sliding as well as the default lateral load value.) The safety factor does not include passive resistance in front of the toe, or the effect of any key. RW ignores all vertical friction forces.

After outputting resulting forces, moments, and reactions at top and bottom of footing, the program calculates the horizontal force and moment at any point on the wall, e.g. to aid in reinforcing design and cutoff. RW will continue to do this until the DEPTH' prompt is bypassed, at which point a new footing size will be prompted for. (Softkey F will do the same thing.) Thus, many footing trials may be run easily.

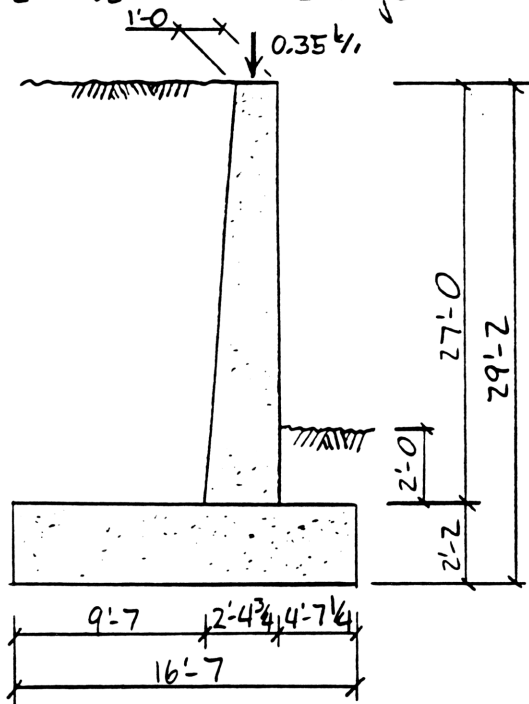
SOFTKEYS

For design of sheetpile walls or to check reinforcing in a concrete wall, allow the program to prompt for soil parameters, then use softkey d (lowercase) to jump to the DEPTH' prompt for different points along the wall.

As mentioned, softkey F will bring up prompts for a new footing. Softkey H (for Height) will bring up prompts for a whole new wall configuration. This is the same as re-executing the program except it saves re-inputting soil parameters.

Retaining Wall Analysis

Analyze this wall for
2' live load surcharge.



Vert. E = 120 psf

Vert. Surch. = 0

Horiz. E = 33 psf

Horiz. Surch. = 66 psf (LL)

$\phi = 35^\circ$

Backslope is flat

Friction $\mu = .45$

Parapet DL = $2.33' \times 1' \times .15 = .35\%$

Vert. Load ΣM about toe

$$\begin{array}{l} \text{Ftg. } 2.17' \times 16.58' \times .15 = 5.39 \text{ k} \times \frac{16.58}{2} = 44.69 \text{ k}' \\ \text{Wall } \frac{1}{2} (1' + 2.40') \times 27' \times .15 = 6.89 \times 5.45' = 37.52 \\ \text{(Partial Summation) (12.26)} \end{array} \quad \begin{array}{l} M \\ (42.21) \end{array}$$

$$\text{Heel E aug. } 10.28' \times 27' \times .12 = 33.32 \times 11.44' = 381.21$$

$$\text{Toe E } 4.60' \times 2' \times .12 = 1.10 \times 2.30' = 2.54$$

$$\text{Parapet } \frac{.35}{47.05 \text{ k}} \times 5.10' = \frac{1.78}{467.74 \text{ k}'}$$

$$\text{eccentricity} = \frac{467.74}{47.05} = 9.94'$$

Horiz. Load ΣM about bottom/ftg

$$\text{Earth } \frac{1}{2} \times .033 \times 29.17^2 = 14.04 \times \frac{29.17}{3} = 136.47$$

$$\text{Surch } .066 \times 29.17 = \frac{1.93 \times 29.17}{2} = \frac{28.07}{164.54}$$

Overturning Moments about toe

$$F.S. = \frac{\text{Resist. } M}{\text{O.T. } M} = \frac{467.74}{164.54} = 2.84 > 2.0$$

OK

Note: "RW" does not calculate
overturning F.S. The designer
must do this.

Sliding Neglect passive toe resist.

silty coarse-grained soil $\rightarrow \mu = .45$

$$\tan(\phi = 35^\circ) = .70 > .45 \quad \text{OK}$$

$$F.S. = \frac{\mu \times V}{H} = \frac{.45 \times 47.05}{15.97} = 1.33$$

$1.33 < 1.50$ desired but say OK

with passive resist. and shear key

Bearing

$$\text{Ftg } S = \frac{16.58^2}{6} = 45.83 \text{ ft}^2/\text{ft}$$

$$\text{Vert. } 47.05 / 16.58 = 2.84$$

$$\text{Vert. } M = 47.05 \times (9.94 - \frac{16.58}{2}) / 45.83 = -1.69$$

$$\text{Horiz. } M = 169.54 / 45.83 = 3.59$$

$$\text{Toe} \rightarrow 2.84 - 1.69 + 3.59 = 4.74 \text{ ksf}$$

$$\text{Heel} \rightarrow 2.84 + 1.69 - 3.59 = 0.94 > 0$$

Need ~ 5.0 ksf bearing

Retaining Wall Analysis

Stem Analysis

@ Top of Ftg. :

$$E \frac{1}{2} \times .033 \times 27^2 = \frac{H}{12.03 \text{ k}} \times \frac{27}{3} = \frac{M}{108.26 \text{ 'k}}$$

$$\text{Surch. } .066 \times 27 = \frac{1.78}{13.81 \text{ k}} \times \frac{27}{2} = \frac{24.06}{132.32 \text{ 'k}}$$

Design stem reinf. for 132.32' k

Check for 13.81 k shear

@ 10' Depth for bar cutoff

$$E \frac{1}{2} \times .033 \times 10^2 = \frac{H}{1.65} \times \frac{10}{3} = \frac{M}{5.50}$$

$$\text{Surch. } .066 \times 10 = \frac{.66}{2.31 \text{ k}} \times \frac{10}{2} = \frac{3.30}{4.80 \text{ 'k}}$$

Run 'RW' with Flag 00 set (Short Form)

	SF 00	
	XEQ "RW"	No title block
SLOPE (dd)		
	RUN	
PHI (dd)		
35.0000	RUN	
Ca=0.2710		
VERT E PCF		
120.0000	RUN	
SUR PSF		
0.0000	RUN	Input soil properties
LAT E PCF		
33.0000	RUN	
SUR PSF		
66.0000	RUN	
FRIC "		
.4500	RUN	

HT'	29.1667	RUN
F TH'	2.1667	RUN
COV'	2.0000	RUN
T/W TH'	1.0000	RUN
B/W TH'	2.4000	RUN
P, K	.3500	RUN
P CG'	.5000	RUN

FTG W'	16.5833	RUN
TOE'	4.6000	RUN

Input wall parameters

ΣV=47.0465 K
M=467.7482 'K
RES09.9422'

B/FTG:

ΣH=15.9615 K
M=164.5390 'K

SLIDE FS=1.3264

TOE P=4.7326 KSF
HEEL P=0.9414 KSF

T/FTG:

ΣH=13.8105 K
M=132.3135 'K

D' 10.0000 RUN

ΣH=2.3100 K
M=8.8000 'K

D'

Formatting is a bit loose with Flag 00 set

No partial sums

Retaining Wall Analysis

Full output - Flag 00 clear

XEQ "RW"

4:45:09 PM 02/02/90

TITLE:

LARP. AVE. WALL RUN

SLOPE (dd)

0.0000 RUN

PHI (dd)

35.0000 RUN

Ca=0.2710

VERT E PCF

120.0000 RUN

SUR PSF

0.0000 RUN

LAT E PCF

33.0000 RUN

SUR PSF

66.0000 RUN

FRIC μ

.4500 RUN

HT'

29.1667 RUN

F TH'

2.1667 RUN

COV'

2.0000 RUN

T/W TH'

1.0000 RUN

B/W TH'

2.4000 RUN

P, K

.3500 RUN

P CG'

.5000 RUN

FTG W'

16.5833 RUN

TOE'

4.6000 RUN

CONC: V=12.2747 K

M=82.2124 'K

P: V=0.3500 K

M=1.7850 'K

TOE E: V=1.1040 K

M=2.5392 'K

HEEL E: V=33.3179 K

M=381.2117 'K

ΣV=47.0465 K

M=467.7482 'K

RES09.9422'

B/FTG:

LAT E: H=14.0365 K

M=136.4660 'K

SUR: H=1.9250 K

M=28.0730 'K

ΣH=15.9615 K

M=164.5390 'K

SLIDE FS=1.3264

TOE P=4.7326 KSF

HEEL P=0.9414 KSF

T/FTG:

LAT E: H=12.0285 K

M=108.2565 'K

SUR: H=1.7820 K

M=24.0570 'K

ΣH=13.8105 K

M=132.3135 'K

D'

10.0000 RUN

LAT E: H=1.6500 K

M=5.5000 'K

SUR: H=0.6600 K

M=3.3000 'K

ΣH=2.3100 K

M=8.8000 'K

42 381 50 SHEETS 3 SQUARE
42 382 100 SHEETS 3 SQUARE
42 383 200 SHEETS 3 SQUARE
42 384 300 SHEETS 3 SQUARE



	X	Y	Z	T	L		X	Y	Z	T	L
01*LBL "SSD"							162*LBL H				
114 GTO 06							163 XROM "L"				
115*LBL "RW"							164 "HT"				
116 15							165 PROMPT				
117 XROM "S"							166 STO 03				
118 CLX							167 "F TH"				
119 "SLOPE"							168 PROMPT				
120 XROM "Pd"							169 STO 04				
121 COS							170 "COV"				
122 ENTER↑							171 PROMPT				
123 ENTER↑							172 STO 12				
124 X↑2							173 "T/W TH"				
125 "PHI"							174 PROMPT				
126 XROM "Pd"							175 STO 07				
127 STO 09							176 "B/W TH"				
128 COS							177 PROMPT				
129 X↑2							178 RCL 07				
130 -							179 +				
131 SQR							180 2				
132 ST- Z							181 /				
133 +							182 STO 07				
134 /							183 "P, K"				
135 *							184 PROMPT				
136 STO 05							185 STO 10				
137 "Ca="							186 "P CG"				
138 ARCL X							187 PROMPT				
139 AVIEW							188 STO 11				
140 XROM "L"							189*LBL F				
141 "VER"							190 XROM "L"				
142 XEQ 04							191 "FTG W"				
143 STO 14							192 PROMPT				
144 RDN							193 STO 05				
145 STO 13							194 "TOE"				
146 LASTX							195 PROMPT				
147 *							196 STO 06				
148 RCL 05							197 XROM "L"				
149 *							198 RCL 04				
150 "LA"							199 RCL 05				
151 XEQ 04							200 *				
152 STO 08							201 STO 00				
153 RDN							202 LASTX				
154 STO 02							203 *				
155 "FRIC μ"							204 RCL 03				
156 PROMPT							205 RCL 04				
157 RCL 09							206 -				
158 TAN							207 RCL 07				
159 X>Y?							208 *				
160 X<>Y							209 ST+ 00				
161 STO 09							210 RCL 06				
							211 ST+ X				
							212 LASTX				
							213 +				
							214 *				
							215 +				
							216 .15				

	X	Y	Z	T	L		X	Y	Z	T	L
217 ST* 00											
218 *						272 RCL 05					
219 2						273 2					
220 /						274 /					
221 STO 01						275 -					
222 RCL 00						276 RCL 00					
223 X<>Y						277 *					
224 "CONC"						278 STO 01					
225 XEQ 14						279 XROM "LL"					
226 ADV						280 "B"					
227 RCL 10						281 RCL 03					
228 ST+ 00						282 XEQ 05					
229 RCL 06						283 RCL 00					
230 RCL 11						284 RCL 09					
231 +						285 *					
232 RCL 10						286 X<>Y					
233 *						287 /					
234 ST+ 01						288 ADV					
235 "P"						289 "SLIDE FS="					
236 XEQ 14						290 ARCL X					
237 ADV						291 AVIEW					
238 RCL 13						292 ADV					
239 RCL 12						293 RCL 00					
240 *						294 RCL 05					
241 RCL 06						295 /					
242 *						296 RCL 01					
243 ENTER↑						297 6					
244 ST+ 00						298 RCL 05					
245 LASTX						299 X↑2					
246 *						300 /					
247 2						301 *					
248 /						302 ST+ Z					
249 ST+ 01						303 -					
250 "TOE E"						304 "TOE"					
251 XEQ 14						305 XEQ 10					
252 RCL 03						306 X<>Y					
253 RCL 04						307 "HEEL"					
254 -						308 XEQ 10					
255 RCL 13						309 XROM "L"					
256 *						310 RCL 03					
257 "HEEL E"						311 RCL 04					
258 XEQ 11						312 -					
259 RCL 14						313 "T"					
260 "SUR"						314 XEQ 05					
261 XEQ 11											
262 XROM "L"						315+LBL d					
263 "ΣV="						316 XROM "LL"					
264 RCL 01						317 "D"					
265 RCL 00						318 CLX					
266 XEQ 13						319 PROMPT					
267 /						320 X=0?					
268 "RES0"						321 GTO F					
269 ARCL X						322 XEQ 02					
270 XEQ "FT"						323 GTO d					
271 AVIEW											

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	X	Y	Z	T	L		X	Y	Z	T	L
324*LBL 05						375*LBL 14					
325 *F/FTG:-						376 X<>Y					
326 AVIEW						377 FC? 00					
						378 X=0?					
327*LBL 02						379 RTN					
328 ENTER↑						380 *F: V=-					
329 ENTER↑						381 GTO 13					
330 X↑2											
331 2						382*LBL 12					
332 /						383 ST- 01					
333 STO 15						384 X<>Y					
334 RCL 02						385 FC? 00					
335 *						386 X=0?					
336 ST* Y						387 RTN					
337 X<>Y						388 *F: H=-					
338 3											
339 /						389*LBL 13					
340 *LAT E-						390 ARCL X					
341 XEQ 12						391 *F K-					
342 ADV						392 AVIEW					
343 R↑						393 *M=-					
344 RCL 08						394 ARCL Y					
345 *						395 XROM *FK-					
346 RCL 15						396 AVIEW					
347 ST* L						397 RTN					
348 X<> L											
349 *SUR-						398*LBL 04					
350 XEQ 12						399 *F T E PCF-					
351 ADV						400 PROMPT					
352 ST+ Z						401 *SUR PSF-					
353 X<> T						402 PROMPT					
354 +						403 E3					
355 X<>Y						404 ST/ Z					
356 *ΣH=-						405 /					
357 GTO 13						406 RTN					
358*LBL 11						407*LBL 10					
359 RCL 05						408 *F P=-					
360 RCL 06						409 ARCL X					
361 -						410 *F KSF-					
362 RCL 07						411 X<0?					
363 -						412 *F:UPL-					
364 STO T						413 AVIEW					
365 *						414 END					
366 ST+ 00											
367 RCL 05											
368 R↑											
369 2											
370 /											
371 -											
372 *											
373 ST+ 01											
374 ADV											

	X	Y	Z	T	L		X	Y	Z	T	L
01*LBL "SSD"						166 RCL 04	Ftg				
114 GTO 06						167 -	ht'				
115*LBL "RW"						168 RCL 07	Stem Th				
116 15						169 *	Stem A				
117 XROM "S"						170 ST+ 00	Toe				
118 "LA"						171 RCL 06	Z-Toe				
119 XEQ 04						172 ST+ X	Stem Th				
120 STO 08	Lat Sur					173 LASTX	Dist.				
121 RDM						174 +	Stem M				
122 STO 02	Lat E					175 *	M				
123 FS? 00						176 +	conc wt				
124 GTO 03						177 .15					
125 "VER"						178 ST* 00					
126 XEQ 04						179 *					
127 STO 14	Vert Sur					180 2	M				
128 RDM						181 /					
129 STO 13	Vert E					182 STO 01	V	M			
130*LBL H						183 RCL 00					
131 XROM "L"						184 "CONC"					
132 "HT"						185 XEQ 14	P				
133 PROMPT						186 RCL 10	(ZY)				
134 STO 03	Ht.					187 ST+ 00	Toe				
135 "F TH"						188 RCL 06	C.G.				
136 PROMPT						189 RCL 11	C.G.				
137 STO 04	F th					190 +	M				
138 "COV"						191 *					
139 PROMPT						192 ST+ 01	P	M			
140 STO 12	Cover					193 RCL 10					
141 "ST TH"						194 "P"					
142 PROMPT						195 XEQ 14	.120				
143 STO 07	Stem th					196 RCL 13	Cover				
144 "P, K"						197 RCL 12	Toe				
145 PROMPT						198 *	Toe V				
146 STO 10	P					199 RCL 06	Toe	V			
147 "P CG"						200 *	Toe M				
148 PROMPT						201 STO 15					
149 STO 11	C.G.					202 LASTX					
150*LBL 01						203 *					
151 XROM "L"						204 STO 09	Ht.				
152 "FTG W"	Ftg W					205 RCL 03	Ftg Th				
153 PROMPT						206 RCL 04	ht.				
154 STO 05						207 -	.120				
155 "TOE"						208 RCL 13	Earth V				
156 PROMPT						209 *					
157 STO 06	Toe					210 "VERT E"					
158 XROM "LL"						211 XEQ 11					
159 RCL 04	Ftg th					212 CLX					
160 RCL 05	Ftg W					213 STO 09					
161 *						214 STO 15					
162 STO 00	Ft. A					215 RCL 14	Sur V				
163 LASTX	Ftg W					216 "SUR"					
164 *	M'					217 XEQ 11					
165 RCL 03	Ht					218 XROM "L"					
						219 "SV"	M				
						220 RCL 01	V				
						221 RCL 00					
						222 XEQ 13	Res.				
						223 /					

	X	Y	Z	T	L		X	Y	Z	T	L
224 *RESE*						273*LBL 11	W				
225 ARCL X						274 RCL 05	FtgW				
226 AVIEW						275 RCL 06	Toe				
227 XROM *LL*						276 -					
228 *B*						277 RCL 07	StemTh				
229 RCL 03	Ht.					278 -	Rael				
230 XEQ 05						279 STO T	Heel	W		heel	
231 RCL 03	Ht.					280 *	V				
232 RCL 04	Ftg+H					281 ST+ 15	($\leq V$)				
233 -	Stem.Ht					282 RCL 05	FtgW				
234 *T*						283 Rt	heel				
235 XEQ 05						284 2					
236*LBL 03						285 ST/ 09					
237 *D**						286 /	heel/z	FtgW	V		
238 CLX						287 -	Arm	V			
239 PROMPT	Depth					288 *	M				
240 X=0?						289 RCL 09	MToe				
241 SF 14						290 +	$\leq M$				
242 X=0?						291 ST+ 01					
243 GTO 01						292 RCL 15	$\leq V$				
244 XEQ 02						293 ST+ 00					
245 GTO 03						294*LBL 14					
246*LBL 05						295 X=0?					
247 *T/FTG**						296 RTN					
248 AVIEW						297 ADV					
249*LBL 02	h					298 AVIEW					
250 ENTER↑	h					299 *V=*					
251 ENTER↑	h	h	h			300 GTO 13					
252 X↑2	h ²					301*LBL 12	M	H			
253 2	Z	h ²	h	h		302 X<>Y	H	M			
254 /	h ^{1/2}					303 X=0?					
255 STO 15						304 RTN					
256 RCL 02	Lat.E	h ^{1/2}	h			305 AVIEW					
257 *	wh ^{1/2}					306 *H=*					
258 ST* Y		wh ^{3/2}	h	h		307*LBL 13	V or H	M			
259 X<>Y						308 ARCL X					
260 3						309 *T K*					
261 /	wh ^{3/6}	wh ^{1/2}	h	h		310 AVIEW					
262 *LAT E*						311 *H=*					
263 XEQ 12						312 ARCL Y					
264 Rt	h					313 XROM *FK*					
265 RCL 15	h ^{1/2}					314 AVIEW					
266 RCL 08	Lat.Sur	h ^{1/2}	h			315 RTN					
267 ST* Z			wh			316*LBL 04					
268 *	wh ^{1/2}	wh				317 *T E PCF*					
269 *SUR*						318 PROMPT					
270 XEQ 12						319 *SUR PSF*	Surch	Earth			
271 XROM *L*						320 PROMPT					
272 RTN						321 E3	1000				
						322 ST/ Z					
						323 /	Surch	Earth			
						324 END					

ELASTOMERIC BEARING PAD ANALYSIS

xeq NEO (SIZE 017)

SOFTKEYS (USER mode ON)

D Input new shim thickness and pad dimensions

USER FLAGS

Flag 00	clear set	Full output Skips output of shape factor, dead load stress, and reduced pier forces
Flag 02	clear set	Warning BEEP disabled Warning BEEP enabled
Flag 03	clear set	Pier SKEW in decimal degrees SKEW in degrees-minutes-seconds
Flag 04	clear set	AASHTO AASHTO (ignores flag 04)

INPUT SUMMARY

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
HARD	Neoprene durometer hardness	No default
LOW G PSI HI G PSI	Low- and high-end values of shear modulus. Default per AASHTO table 14.2.2A.	AASHTO
DEL-T"	Movement due to temperature (one way), in inches (zero for a fixed pad)	No default
DL K/PAD	Dead load, kips per pad (reduced by live load uplift, if any)	No default
D+L K/PAD	Maximum dead and live load	No default
# PADS	Number of bearing pads on one pier. If bypassed (or one is entered), pier properties are not prompted and pier flexibility is neglected.	One
PIER HT'	Pier height from point of fixity to beam seat, in feet. If zero is entered, pier properties are ignored.	No default

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
SKEW (dd or DMS)	Skew of the pier	
# COLS	Number of pier columns. Zero (or bypass) for a T-pier.	0 (T-pier)
DIA' (If #COLS is not zero)	Column diameter in feet	No default
SH TH' (T-piers)	Shaft thickness in feet	No default
W' (T-piers only)	Shaft width, feet	No default
SHIM TH 16'S	Metal shim thickness in sixteenths of an inch. May bypass for unreinforced pads. Answer 2 for 1/8"shims, 3 for 3/16".	Two (1/8")
PAD L"	Pad length in inches	No default
W"	Pad width in inches. Interchangeable with length.	No default
#SHIMS	Number of shims. Bypass for unreinforced pads.	No shims
LAM TH"	Lamination thickness in inches (for the lams between the shims). Program assumes 1/4" thick cover layers outside the shims.	No default
TH"	Thickness of the (unreinforced) pad in inches, if #SHIMS prompt was answered zero or bypassed	No default

OUTPUT SUMMARY (Items in parenthesis are displayed only if flag 00 is clear)

TH	Thickness of the (reinforced) pad in inches if SHIM TH was not answered zero (assumes 1/4" cover layers)
TH>MAX ***	Warns that bearing is too thick per AASHTO 14.2.7 (TH > W/3 or W/5)
(SH F)	Shape factor (Area/Perimeter*th.)
ALL STRS	Allowable compressive stress in kips per square inch. Fixed bearings are allowed a 10% increase (DEL-T" input as zero).
(DL STRS)	Dead load stress acting.
D+L STRS	Combined dead and live load stress acting.

DL>MAX ***	Warns that dead load stress is more than 500 psi (guideline carried over from <u>1978 AASHTO</u>)
DL<.2 ***	Warns that dead load is less than 200 psi (guideline carried over from 1978 AASHTO concerning anchorage of pads)
D+L>MAX ***	Warns that compressive stress is greater than calculated allowable
STRN	Compressive strain in percent. DuPont recommends strain be kept under 7%.
STR>7% ***	Warns against high strain. Reduce stress or increase shape factor.
TH<2T ***	Warns pad is too thin for shear deformation needed, per AASHTO 14.2.6
BOND ***	Warns pad needs to be secured against horizontal movement per AASHTO 14.2.9. No account is taken of pier flexibility for this warning.
F	Shear force in kips developed by the thermal deflection (DEL-T") input, considering the pier rigid
(Fe-11 and Fe-1)	Effective shear force developed parallel and perpendicular to the long axis of the pier, respectively, <u>taking into account the flexibility of the pier</u>
OK or NG	Trial pad design is OK or no good with respect to the above-mentioned checks

PROGRAM FLAGS

Flag 05	clear set	Thermal movement Fixed pad (DEL-T" = 0)
Flag 06	clear set	No pier properties Consider pier flexibility
Flag 07	clear set	OK NG
Flag 09	clear set	Multiple shims One shim (used rarely)

STORAGE REGISTER USE

<u>Register</u>	<u>Value</u>
00	DL K/PAD (Minimum load combination)
01	D+L K/PAD (Maximum load combination)
02	Lesser of pad length or width; pad area
03	Low-end shear modulus G (ksi stored)
04	Number of shims; 3 or 5 x Total thickness
05	Lamination thickness
06	Durometer hardness
07	Combined DL + LL stress
08	DEL-T (Thermal movement, one way)
09	Total elastomer thickness
10	High-end shear modulus x 1.9 (cold) x 5
11	Greater of pad length and width; shape factor
12	Skew (decimal degrees stored)
13	Pier height; $L^3/3EI$ of pier shaft or columns (C_p)
14	Number of pads per pier
15	Allowable compressive stress
16	Shim thickness (inches stored)

ELASTOMERIC BEARING PAD ANALYSIS

Program NEO calculates properties and allowable stresses for elastomeric (neoprene) bearing pads by AASHTO Specifications, 1985 Revision. It uses AASHTO's compressive stress:strain curves. Default shear moduli are the high- and low-end values from AASHTO Table 14.2.2A; a cold-weather constant of 1.9 (recommended by DuPont for cold regions with temperatures down to -20°F) is included in the shear-force calculations (high-end values) but not the allowable stress calculations (low-end values). The program allows elastomer durometer hardness values between 50 and 70; any other values may be assigned incorrect shear moduli.

PIER INTERACTION

NEO will analyze a bearing pad alone, or it will analyze a system of bearing pads and pier. If a thermal movement $\text{DEL-T}''$ ($\text{delta}_{\text{temp}}$, inches) of zero is input, the bearing is assumed fixed and its allowable compressive stress is increased by 10%. If $\text{DEL-T}''$ greater than zero is input, the program prompts for pier dimensions in order to calculate the effective shear force considering pier flexibility. The following assumptions are made:

- 1) Superstructure movement is purely longitudinal; deck does not rotate or deflect laterally on skewed piers.
- 2) Deck and beams are rigid in all directions.
- 3) Pier deflects only perpendicular to its long axis; it is assumed rigid along its long axis (parallel direction).
- 4) Pier is fixed at bottom. Definition of "bottom" depends on $\text{PIER HT}'$ (pier height in feet) input by user, but is usually taken from top of footing.
- 5) The elastomeric material is at a temperature of -20°F and its stiffness has increased 90% over tabulated values. This may be somewhat over-conservative for warmer climates.
- 6) The pier concrete has a proportionality constant of $N=8$, and the gross moment of inertia of the pier shaft or columns is used.
- 7) The program assumes that a single pier shaft is rectangular, and that multiple columns are round. It considers the shaft or columns to be of uniform

stiffness all the way to the bearings, ignoring the effect of the pier cap perpendicular to the long axis of the pier. (See the INPUT SUMMARY section.)

All but the last of these is conservative. If the bridge design involves a short T-pier, the designer may wish to input a PIER HT less than actual in recognition of the greater stiffness of the cap. For pier concrete of somewhat greater stiffness than indicated by N=8, assumption 6) is probably still conservative since the columns usually suffer at least minor cracking, decreasing their stiffness. The assumption of rectangular shape is pretty good for round-nosed T-pier shafts; just input a shaft length less than the actual by about one-half the shaft width. For rectangular column bents the user will have to calculate the DIA' (diameter in feet) of a round column of equal stiffness. For a single round column, answer the #COL prompt with 1; for a rectangular or roundnosed shaft respond with a zero.

DEFAULT SHEAR MODULI

The default values of G in psi are from AASHTO table 14.2.2A (except for 70 duro) and are as follows:

<u>Duro</u>	<u>Low G</u>	<u>High G</u>
50	85	110
55	102.5	132.5
60	120	155
70	155	200
(AASHTO 70)	(160)	(260)

SHEAR FORCE

For a non-fixed pier (one whose bearings absorb movement), NEO first calculates the shear force assuming the pads take all the thermal movement, i.e. the substructure is assumed rigid. If the #PADS prompt is bypassed (R/S is hit without keying in a number) or the PIER HT' prompt is answered zero, the program leaves its shear-force calculations at that. This, for instance, is appropriate for abutment bearings and very conservative pier designers. If these prompts are answered, however, NEO goes on to calculate the effective force developed considering the piers flexible using the above-noted assumptions.

Pad anchorage and shear checks are made for non-fixed piers using the full rigid-pier shear force (i.e. pier flexibility is not taken into account) whether or not pier information has been input. These checks are not made for fixed piers since the anchor bolts or

other "fixing" mechanism are assumed to absorb the entire shear force.

The program assumes that elastomeric material harder than 60 durometer is used only in fixed, unreinforced pads, in accordance with AASHTO 14.2.2. If an expansion bearing (assumed reinforced) of 70 durometer material is analyzed using the default high-end shear modulus, the shear-force calculations will be about 25% low due to an error in the shear modulus calculation algorithm. If the user chooses to ignore 14.2.2, the default high-end modulus should not be used. The preceding section lists the defaults.

PAD GEOMETRY

The program works only with rectangular bearing pads, not round ones. It is not able to accommodate holes in pads. Pad length and width are interchangeable.

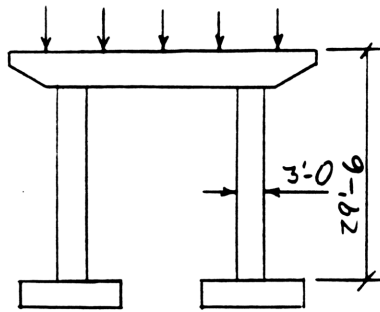
EXCEPTIONS

All AASHTO specifications are checked except 14.2.5, dealing with rotation, and 14.2.2, which forbids laminated pads of greater than 60 durometer hardness. If a pad fails to meet a given criterion, a message is displayed. At the end of the run, if a design has met all the checks, the program displays OK. If it has failed one or more of the checks, the program displays NG (no good) and, if flag 02 is set, a BEEP sounds.

OBSOLETE CHECKS

NEO checks against dead load stress of over 500 psi or under 200 psi; and total compressive strain of over 7%. The former are carried over as a guide from the 1978 AASHTO Specifications, while the latter comes from DuPont's 1959 guide to elastomeric bearing pad design. The program will label a design "NG" if it violates either of these; however, the design may still be valid by present (1989) AASHTO Specifications.

Design pads for 90' span (steel girders)



$$\text{One-way } \Delta_T = (90' \times 12) \times (6.5 \times 10^{-6}) \times 75^\circ \text{F} = \underline{.5265''}$$

$$DL = 169.5 \text{ k/beam}$$

$$\text{Max LL} = \frac{107.99 \text{ k/beam}}{277.49}$$

$$\text{No uplift} \Rightarrow \text{min. LL} = 0.0$$

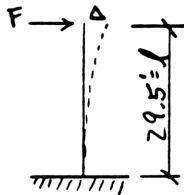
5 beams/pier ; pier @ 10° skew

Assume:

- 1) columns fixed @ Top/Ftg.
- 2) deck rigid (beam spacing and orientation constant; no lateral movement by beams)
- 3) pier rigid parallel to frame
- 4) constant column EI all the way to the top.

*1-3 are conservative; *4 is slightly non-cons.

Derive pier flexural properties



$$\Delta = F \times \frac{l^3}{3EI}$$

$$l = 29.5' = 354''$$

$$\text{Use } E = \frac{29,000}{n=8} = 3625 \text{ ksi}$$

$$I = 2 \text{ col.} \times \frac{\pi}{64} (36'')^4 = 164,896''^4$$

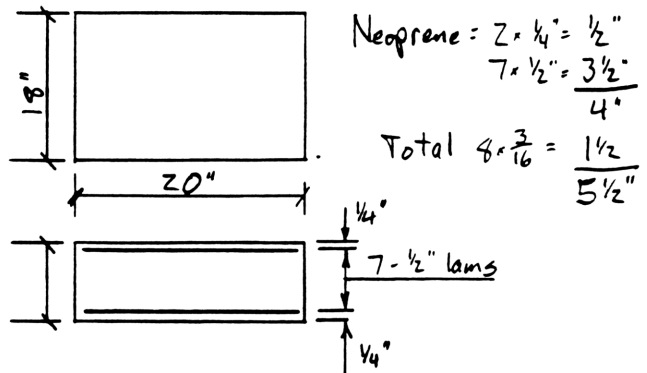
$$\therefore \Delta_{\text{pier}} = F \times \frac{(354'')^3}{3 \times 3625 \times I}$$

$$= F_{\text{kips}} \times .0247\%$$

(D262)

Use 55-durometer per Mn/DOT specs.

Try 18" x 20" pad w/ $\frac{7}{16}$ " shims, 7- $\frac{1}{2}$ " lams



$$\text{Neoprene} = 2 \times \frac{1}{4} = \frac{1}{2}''$$

$$7 \times \frac{1}{2} = \frac{3 \frac{1}{2}}{4}''$$

$$\text{Total } 8 \times \frac{3}{16} = \frac{1 \frac{1}{2}}{5 \frac{1}{2}}''$$

$$\text{Stability (AASHTO 14.2.8)} \quad \frac{W}{3} = 6'' > 5 \frac{1}{2}'' \quad \underline{\text{OK}}$$

$$\text{Anchorage (AASHTO 14.2.9)} \quad F_{\text{horiz}} = G \times A \times \Delta_T / T$$

$$55 \text{ duro} \Rightarrow G = \frac{1}{2} (110 + 155) = 132.5 \text{ psi}$$

$$A = 18'' \times 20'' = 360''^2 ; \Delta_T = .5265''$$

$$T = \text{neoprene thickness} = 4''$$

$$\therefore F_H = \frac{132.5 \times 360 \times .5265 \times 1.9^*}{4''} = 11.93 \text{ k}$$

$$\text{Minimum Vert. } F = 169.5 \text{ k} > 5 \times 11.93 \quad \underline{\text{OK}}$$

$$\text{Shear (AASHTO 14.2.6)} \quad T = 4'' > 2 \times .5265'' \quad \underline{\text{OK}}$$

$$\text{Comp. Stress (AASHTO 14.2.3)} \quad \text{Actual} = 277.49 \text{ k} / 360''^2 = .7708 \text{ ksi}$$

$$\text{Shape Factor} = \frac{18'' \times 20''}{2 \times (18'' + 20'') \times \frac{1}{2}} = 9.4737$$

$$\text{Allowable stress} = \frac{\frac{1}{2} (65 + 120) \times 9.4737}{(\beta = 1.0) \times 1000} = .9711 \text{ ksi}$$

$$.9711 > .7708 \quad \underline{\text{OK}}$$

* Cold-weather factor for -20°F .

Comp. Strain From tables 4.2.4,
(AASHTO 4.2.4) for stress of
770 psi:

$$\begin{aligned} 50 \text{ duro} &\rightarrow 3.70\% \\ 60 \text{ duro} &\rightarrow 3.25\% \\ \text{avg.} &\sim 3.48\% \end{aligned}$$

As a guide, old AASHTO
allowed max. 7%

$$3.48\% < 7\% \quad \underline{\text{OK}}$$

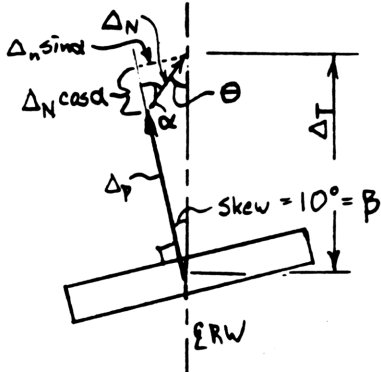
Also as a guide, check
DL stress against 500 psi:

$$DL = \frac{169.5 \text{ k}}{360 \text{ sq.}} = .47 \text{ ksi} < .5 \quad \underline{\text{OK}}$$

all OK.

\therefore USE 18" x 20" x 5 1/2" PAD

Calculate actual pier force



$$\begin{aligned} \Delta_{Neo} &= F \times \left(\frac{1000 \times 4''}{132.5 \times 360 \times 1.9 \times 5 \text{ psi}} \right) \\ &= F \times .0066\% \end{aligned}$$

$$\Delta_{Pier} = F \times .0247$$

$$\begin{aligned} \Delta_N \sin \alpha &= \Delta_T \sin(\text{skew}) = .5265'' \times \sin 10^\circ \\ &= .0914'' \end{aligned}$$

$$\begin{aligned} F_{e-11} &= .0914'' \times .0066\% \\ &= \underline{\underline{10.357 \text{ k}}} \end{aligned}$$

$$\Delta_N \cos \alpha = \Delta_T \cos \beta - \Delta_P$$

$$\text{and } \frac{\Delta_N \cos \alpha}{C_N} = \frac{\Delta_P}{C_P}$$

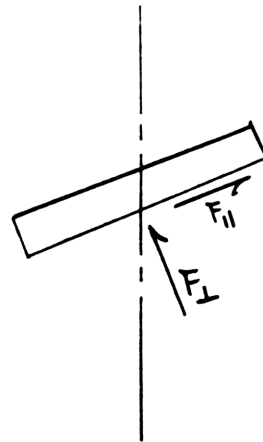
$$\text{or } \Delta_P = \frac{C_P \cdot \Delta_N \cos \alpha}{C_N}$$

$$\therefore \Delta_T \cos \beta = \Delta_N \cos \alpha \left(1 + \frac{C_P}{C_N} \right)$$

$$C_N \Delta_T \cos \beta = \Delta_N \cos \alpha (C_N + C_P)$$

$$\begin{aligned} F_{e-t} &= \frac{\Delta_N \cos \alpha}{C_N} = \frac{\Delta_T \cos \beta}{C_N + C_P} \\ &= \frac{.5265'' \times \cos 10^\circ}{.0247 + .0066\%} \\ &= \underline{\underline{15.447 \text{ k}}} \end{aligned}$$

Design pier for these
effective thermal forces



XEQ "NEO"
5:36:25 PM 02.03.89

TITLE:
I 394 PAD E-5 RUN

HARD

55.0000 RUN

DEL-T*
.5265 RUN

DL K/PAD
169.5000 RUN

MAX LL K/PAD
107.9900 RUN

MIN LL K/PAD
RUN

PADS
5.0000 RUN

PIER HT*
29.5000 RUN

SKEW (dd)
10.0000 RUN

COLS
2.0000 RUN

DIA*
3.0000 RUN

SHIM TH 16'S
3.0000 RUN

PAD L*
18.0000 RUN

W*
20.0000 RUN

#3/16" SHIMS
8.0000 RUN

LAM TH*
.5000 RUN

TH=5.5000"

SH F=9.4737
ALL STRS=0.9711 KSI

DL STRS=0.4708 KSI
D+L STRS=0.7708 KSI
STRAIN=3.4549%

F=11.9315 K/PAD
=59.6577 K/PIER
Fe-11=10.3595 K/PIER
Fe-t=15.4179 K/PIER

OK

total thickness
shape factor
allowable

est. strain

} equivalent forces
taking pier flex.
into account

design is OK

PAD L*
18.0000 RUN

W*
20.0000 RUN

#3/16" SHIMS
2.0000 RUN

LAM TH*
.5000 RUN

TH=1.3750"

SH F=9.4737
ALL STRS=0.9711 KSI

DL STRS=0.4708 KSI
D+L STRS=0.7708 KSI
STRAIN=3.4549%

TH<2T ***

BOND ***

F=47.7262 K/PAD
=238.6309 K/PIER
Fe-11=41.4378 K/PIER
Fe-t=19.1960 K/PIER

NG

PAD L*
18.0000 RUN

W*
20.0000 RUN

#3/16" SHIMS
4.0000 RUN

LAM TH*
.7500 RUN

TH=3.5000"

SH F=6.3158
ALL STRS=0.6474 KSI

DL STRS=0.4708 KSI
D+L STRS=0.7708 KSI
D+L>MAX ***

STRAIN=3.9629%

F=17.3550 K/PAD
=86.7749 K/PIER
Fe-11=15.0683 K/PIER
Fe-t=16.7952 K/PIER

NG

PAD L*
18.0000 RUN

W*
20.0000 RUN

#3/16" SHIMS
4.0000 RUN

LAM TH*
.7500 RUN

TH=3.5000"

SH F=6.3158
ALL STRS=0.6474 KSI

DL STRS=0.4708 KSI
D+L STRS=0.7708 KSI
D+L>MAX ***

STRAIN=3.9629%

F=17.3550 K/PAD
=86.7749 K/PIER
Fe-11=15.0683 K/PIER
Fe-t=16.7952 K/PIER

NG

Try another
design

too thin

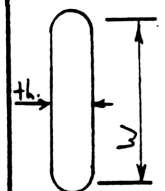
is design is no good

Try a third
design



Load too high

	X	Y	Z	T	L		X	Y	Z	T	L
01*LBL "NEO"						53 SF 06					
02 CF 05						54 22					
03 16						55 /	#/ZZ				
04 XROM "S"						56 3	(L ³ /const)				
05 "HARD"	duro					57 Y1X					
06 PROMPT						58 STO 13					
07 STO 06						59 "SKEW"					
08 .286						60 XROM "Pd"					
09 /						61 STO 12	Skew				
10 89.8						62 CLX					
11 -						63 "# COLS"					
12 "LOW"	Low G					64 PROMPT					
13 XEQ 07						65 X=0?					
14 STO 03	Low G					66 GTO 00	#Col.				
15 RCL 06	duro					67 "DIA"					
16 4.5						68 PROMPT					
17 *						69 X12	d ⁴	#col.			
18 115						70 X12	Pier I				
19 -						71 *	.6	Pier I			
20 "HI"						72 .6					
21 XEQ 07	High G					73 GTO 13					
22 9.5	1.9x5					74*LBL 07					
23 *						75 "G PSI"					
24 STO 10						76 PROMPT	G psi				
25 "DEL-T"						77 E3	1000				
26 PROMPT						78 /					
27 STO 08	Δ ₁					79 RTH	G ksi				
28 X=0?											
29 SF 05						80*LBL 00					
30 CF 06						81 "SH TH"					
31 "DL"						82 PROMPT					
32 XEQ 03						83 3					
33 PROMPT						84 Y1X					
34 STO 00	Min.load					85 "H"					
35 "D+L"						86 PROMPT	W	th ³			
36 XEQ 03											
37 PROMPT						87*LBL 13					
38 STO 01	Max.load					88 *	Pier I				
39 FS? 05						89 ST/ 13					
40 GTO D											
41 E						90*LBL D					
42 ENTER↑	1	1				91 2	2				
43 "# PADS"						92 "SHIM TH 16"S"					
44 PROMPT	*Pads	1				93 PROMPT	Sl.th.				
45 STO 14						94 16					
46 X=Y?						95 /					
47 GTO D						96 STO 16	th, index				
48 "PIER HT"											
49 PROMPT						97*LBL 22					
50 STO 13	Height					98 XROM "LL"					
51 X=0?						99 CF 07					
52 GTO D						100 CF 09					
						101 "PAD L"					
						102 PROMPT					
						103 STO 02	L				

$$\Delta = \frac{Pl^3}{3EI}$$



$$\frac{l^3}{3EI} \text{ in R.13}$$

	X	Y	Z	T	L		X	Y	Z	T	L
104 "H"						157 5					
105 PROMPT	W					158 *					
106 RCL 02	L	W				159 STO 04	5+total				14.2.7
107 X>Y?						160 .8					
108 X<Y	lessor	greater				161 STO 15	.8				500 psi max
109 STO 02						162 E	1				
110 X<Y						163 +	β				$\beta = 1.8$
111 STO 11	greater						β				
112 CLX						164 LBL 11	β				
113 "#SHIMS"						165 RCL 02	L				
114 PROMPT	# of shims					166 RCL 11	W	L	β		
115 X=0?						167 *	area				
116 GTO 10						168 LASTX	L				
117 STO 04	# shims					169 RCL 02	W	L	area	β	
118 E	1					170 +	L+W				
119 X=Y?					one shim \rightarrow SF09	171 ST+ X	Perim.	area	β		
120 SF 09						172 /	area/p				
121 X<Y	# shims					173 RCL 05	lan.th				
122 RCL 16	ch.th.					174 /	Shape F	β			Shape Factor in R.11
123 *	metal th.					175 X<Y 11	L				
124 ENTER	metal	metal				176 RCL 02	W	L	β		
125 "LAM TH"						177 RCL 04	max.th.				
126 FC? 09						178 "TH"					
127 PROMPT						179 X>Y?					14.2.7
128 STO 05	lan.th.	metal				180 XEQ 05	W	L	β		
129 RCL 04	# shims		.	.		181 RDN	area	β			
130 E	1					182 *	area				
131 -	# lams	lan.th.	.	.		183 STO 02	β				
132 *	2 lams	.				184 RDN	β				
133 .5	Z = 1/4"	.				185 FS? 09					$\beta = 1.0$ for lam.
134 +	elect.th.	.				186 1.4	β				= 1.4 for cover
135 STO 09	elect.th.	metal th.				187 RCL 11	Sh.F				= 1.8 for plain
136 +	total th.					188 "SH F="					
137 "TH="						189 ARCL X					
138 ARCL X						190 FC? 00					
139 3						191 AVIEW	β	\leq			
140 *	3+th.				14.2.7	192 X<Y					
141 STO 04						193 /					
142 "T"						194 RCL 03	Low G				
143 AVIEW						195 *	GS/ β				Allow. compr. stress
144 XROM "L"						196 RCL 15	max allow				= GS/ β
145 4						197 X>Y?					14.2.3
146 1/X	1/4					198 X<Y	allow.				10% increase for fire
147 FS? 09						199 1.1					
148 STO 05	cover layer					200 FC? 05					
149 E	1				$\beta = 1$	201 SIGN					
150 STO 15	max				1000 psi max.	202 *	allow.				
151 GTO 11						203 STO 15					
						204 "ALL"					
152 LBL 10					Unrein. pad	205 XROM "K"					
153 "TH"	Th.					206 XEQ "PV"					
154 PROMPT						207 ADV					
155 STO 05	lan.th.					208 .5	500 psi				
156 STO 09	elect.th.					209 RCL 00	DL				
						210 RCL 02	area				
						211 /	DL stress				

DL66

	X	Y	Z	T	L		X	Y	Z	T	L
212 "DL"											
213 XROM "K"						268*LBL 12					
214 FC? 00						269 "STRN="					
215 XEQ "PV"						270 ARCL X					
216 "DL"						271 "t2"					
217 X>Y?						272 AVIEW					
218 XEQ 05						273 7	7% strain				Dupont rule limits strain to 7%
219 "t<.2"						274 "STR>7%"					
220 .2	200psi	DL				275 X<=Y?					
221 X>Y?						276 XEQ 06					
222 XEQ 06						277 FS? 05					
223 RCL 01	Max DblL					278 GTO 08					
224 RCL 02	area					279 ADV					
225 /	Max. strg					280 RCL 08	ΔT				
226 STO 07						281 ST+ X	$Z \Delta T$				
227 "B+L"						282 RCL 09	elast. th.				
228 XROM "K"						283 "TH<2T"					14.2.6
229 XEQ "PV"						284 X<=Y?					
230 "B+L"						285 XEQ 06					
231 RCL 15	allow.	actual				286 RCL 00	DL				
232 X<=Y?						287 RCL 10	5.6 Gai				
233 XEQ 05						288 RCL 02	area				
234 6	6					289 *	ΔT				$F = \frac{\Delta L G}{\text{elast. th.}}$
235 RCL 11	Sh.F.					290 RCL 08					
236 X<Y?						291 *					
237 GTO 09						292 RCL 09	elast. th.				
238 LH						293 /	Temp. F	DL			
239 RCL 06	duro	ln(Sh.F)				294 "BOND"					14.2.9
240 27.3						295 X>Y?					
241 /						296 XEQ 06					
242 3.64						297 5					
243 -						298 /	Temp. F				Force for one pad
244 *						299 "F="					
245 RCL 06	duro					300 ARCL X					
246 7.16						301 XEQ 03					
247 /						302 XEQ "PV"					FC? 06 or FS? 00
248 -						303 FS? 06					
249 14.56						304 FS? 00					
250 +						305 GTO 08					
251 RCL 07	atbl. strngs					306 RCL 14	# Pads				
252 *						307 *	Z F				Total force on pier if rigid
253 .97						308 " = "					
254 +						309 XEQ 14	Z Force				
255 GTO 12	Strain%					310 RCL 08	ΔT				
						311 ENTER↑	ΔT	ΔT	F		see example:
256*LBL 09	Sh. factor					312 "Fe-11="					
257 18						313 RCL 12	skew				
258 *						314 SIN	sin	ΔT	ΔT	F	
259 214						315 R↑	F				
260 X<>Y						316 *	Fe-11	ΔT	ΔT	ΔT	
261 -						317 XEQ 14					
262 RCL 06	duro					318 X<> L	F	ΔT			
263 -						319 /	C neo				
264 .1						320 RCL 13	C pier				
265 *						321 +	Z C	ΔT			
266 RCL 07	stress										
267 *	Strain%										

	X	Y	Z	T	L		X	Y	Z	T	L
322 /	$\frac{\Delta T}{\Delta C}$										
323 RCL 12	skew										
324 COS	cos										
325 *	Fc.t										
326 "Fe-t="											
327 XEQ 14											
328*LBL 08											
329 ADV											
330 CLA											
331 XROM "OK"											
332 AVIEW											
333 GTO 22											
334*LBL 03											
335 "t K/PAD"											
336 RTN											
337*LBL 14											
338 ARCL X	Force										
339 "t K/PIER"											
340 XEQ "PV"											
341 RTN											
342*LBL 05											
343 "t>MAX"											
344*LBL 06											
345 "t ***"											
346 AVIEW											
347 XROM "L"											
348 SF 07											
349 END											

SF 07 for any warning

AASHTO T-PIER ANALYSIS

xeq PIER (SIZE 043)

SOFTKEYS (USER mode ON)

PIER uses no softkeys

USER FLAGS

PIER recognizes no user flags

INPUT SUMMARY

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
T/C EL	Top of cap elevation	No default
CAP HT'	Cap height in feet	No default
FTG HT'	Footing height (thickness), feet	No default
B/F EL	Bottom of footing elevation	No default
DLW	Design low water elevation. Bypass (or enter zero) to signify dry-land construction.	Zero
DHW (if DLW prompt was answered)	Design high water elevation. Used to calculate ice elevation.	No default
GR EL (if DLW was bypassed)	Ground elevation, for landlocked piers	No default
SH TH'	Shaft thickness in feet	No default
SH L'	Shaft length measured along centerline of pier, in feet	No default
CAP TH'	Cap thickness in feet	No default
CAP L'	Cap length in feet	No default
WEDGE	Length of the triangular chamfer cut from the underside of the cap cantilever, in feet	No default
WEDGE HT'	Height of the cantilever chamfer which when subtracted from the CAP HT' leaves the height of the cap at the end of the cantilever	No default

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
SPAN'	Span length in feet. For different spans about a pier, use the average of their lengths.	No default
DW O-O'	Deck width, out to out, in feet	No default
RW W'	Roadway width between curbs, for figuring live load lane eccentricities	No default
SKEW	Skew of the pier. A "square" bridge is defined to have 0° skew.	No default
DECK EL	Elevation of the bridge deck	No default
BM+R HT'	Height from low beam to top of rail, for wind calculation, in feet	No default
BRG HT'	Bearing height; the difference in elevation between low beam and top of cap.	No default
ICE Fc PSI (if DLW prompt answered)	Ice crushing strength in psi	No default

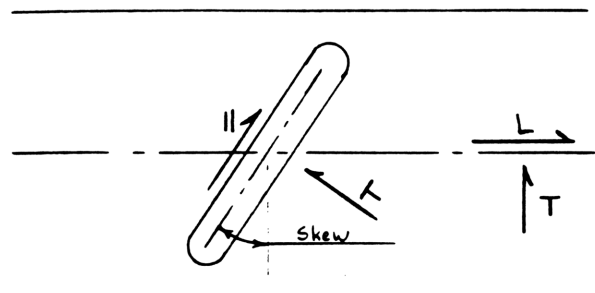
OUTPUT SUMMARY (Flag 00 has no effect on PIER)

MIN A-S	Minimum area of vertical reinforcing for the pier shaft (1/2 %) in square inches
SH DL	Shaft dead load in kips (assuming round ends)
CAP DL	Cap dead load in kips (assuming square ends)
SU DL EST	An estimate of the superstructure dead load
1L LL EST	An estimate of one lane of live load, without impact. Assumes continuous construction.
1L ARM 2L ARM	Eccentricity of one and two lanes of live load with respect to centerline of roadway (or CL pier) <u>measured along the pier cap</u> , in feet. Will not display 2-lane value if RW W' is less than 22'-0.
DLW ICE	Values which follow are for Design Low Water (or Ice) elevation. Press R/S to continue. If DLW prompt was bypassed (i.e. landlocked pier), no such identifier is displayed.
WOS	Wind on substructure. See next page for explanation of postscripts.

SH ARM	Position of center of action of load with respect to top of footing, for shaft design
FTG ARM	Position of center of action of load with respect to bottom of footing, for footing design; in feet.
ICE	Ice force along centerline of pier
WOF	Wind overturning force (uplift), in kips
WOF-11	Wind overturning moment, in foot-kips
W	Wind on superstructure
WL	Wind on live load

Postscripts (see below)

⊥	Perpendicular to the long axis of the pier
11	Parallel to the long axis of the pier
T	Transverse to centerline of roadway
L	Longitudinal to (along) centerline of roadway



PROGRAM FLAGS

Flag 07	clear set	Transverse force Longitudinal force
Flag 08	clear set	Landlocked pier (DLW prompt bypassed) Pier in water
Flag 09	clear set	WOS for DLW stage WOS for ICE stage, or if landlocked
Flag 10	clear set	WOS-⊥ (perpendicular) WOS-11 (parallel)

STORAGE REGISTER USE

<u>Register</u>	<u>Value</u>
00	Height of shaft
01	Top of cap elevation
02	DLW elevation or zero
03	ICE elevation or Ground elevation
04	Top of footing elevation
05	Footing thickness
06	Skew (degrees)
07	Shaft length
08	Cap height
09	Wedge height
10	Wedge length
11	Cap width (thickness)
12	Shaft thickness
13	Span
14	Deck width out to out
15	Cap length
16	Beam + rail (+ slab) height
17	WOS-11
18	WOS-11 Arm
19	WOS--
20	WOS-- Arm
} — DLW Elev.	
21	WOS-11
22	WOS-11 Arm
23	WOS-1
24	WOS-1 Arm
} — Ice or Ground Elev.	
25	Bearing height
26	Pier dead load (shaft + cap)
27	WOF, K
28	WOF-11, 'K
29	ICE force (11)
30	ICE arm

<u>Register</u>	<u>Value</u>
31	Estimated Live load (one lane, no impact)
32	Estimated superstructure dead load
33	Height from top of cap to 6'-0 above top of deck (WL-T Arm)
34	W-T (11)
35	W-T (┐)
36	W-T Arm
37	W-L (11)
38	W-L (┐)
39	WL-T (11)
40	WL-T (┐)
41	WL-L (11)
42	WL-L (┐)

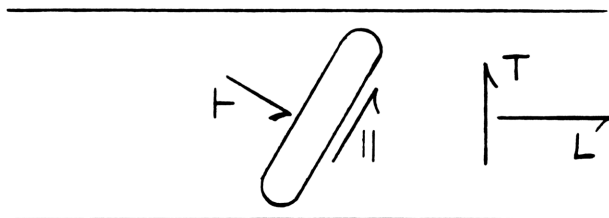
T-PIER ANALYSIS

Program PIER is a fast, simple way to calculate or check the AASHTO loads on a bridge pier. The program is written to analyze T-piers, but most of the output can also be used for column-bent piers by considering the columns as one shaft of equivalent width for purposes of wind load. In order to shorten the program, a few simplifying conditions were set:

- 1) The superstructure, pier, and footing are assumed to be symmetric with respect to the centerline of roadway.
- 2) The program does not recognize the presence of a sidewalk when calculating live load lane eccentricity, nor does it calculate sidewalk live load.
- 3) PIER does not calculate earth loads, loads due to earthquake or streamflow, or longitudinal forces due to live load.
- 4) Elastomeric bearing pads are assumed used. That is, the force due to temperature change is not the limit of the longitudinal force able to reach the pier; it is merely another additive force.

CONVENTION of DIRECTION of FORCES

This program, along with programs FTG and SHFTG (footing analysis programs) use a simple convention to differentiate directions of applied forces. Directions with respect to the superstructure or centerline of roadway are designated with a "T" for transverse forces and an "L" for longitudinal forces (forces acting along the centerline of roadway). Directions with respect to the centerline of the pier are designated with a "⊥" for perpendicular forces and a "||" for parallel forces (forces acting along the long axis of the pier). For bridges with no skew ("square" bridges), of course, $F_{\perp} = F_L$ and $F_{||} = F_T$. For skewed bridges, the following diagram will provide the simplest description.



See the example for further clarification of the translation process between the two axes.

POINT of APPLICATION of FORCES

Program PIER calculates the magnitude and point of action of the following forces, according to AASHTO Chapter 3:

W	Wind on superstructure
WOF	Wind overturning force
WOS	Wind on substructure
WL	Wind on live load
ICE	Ice force

The point of application of each component of these forces is given as a distance, or moment arm, both from bottom of footing (FTG ARM) for footing design; and from top of footing (SH ARM) for shaft design.

Wind on superstructure (W) is calculated by the simplified method of AASHTO 3.15.2.1.3; PIER does not check the requirement that the span be less than 125 feet. W and WL (wind on LL) are calculated as transverse and longitudinal forces with respect to centerline of roadway. Since a transverse force effectively is felt at its point of application while a longitudinal force can act only through friction at the beam seat (due to the restraint of rotation by the beams longitudinally), different moment arms are given for each component. Then each of these forces is normalized (split) into components parallel (||) and perpendicular (⊥) to the long axis of the pier. See the example for clarification.

WOF (wind overturning force) is calculated and displayed as an uplifting force in kips and an overturning moment parallel to the long axis of the pier (WOF-||). There is no component perpendicular to the pier.

WOS (wind on substructure) is calculated as a force of 40 psf on each face of the pier. Those who would argue that the wind can't blow four directions at once will have to reduce these values with their sines and cosines. This program uses the simple reasoning that it's already blowing from two different directions on the superstructure; two more directions isn't that many more. WOS is actually calculated twice: once on the portion of the pier above DLW (design low water elevation), and once on the portion of the pier above the Ice elevation. PIER assumes that the bottom of the pier cap is above both these elevations; if this is not the case, the WOS values will be in error (they will be calculated with the full cap and a negative length of shaft exposed to the wind).

The ICE elevation is not input directly; it is assumed to be midway between DLW and DHW (design high water). Since the DHW elevation is not used for anything else, the ICE elevation can be set to any desired

value by changing the DHW elevation. Although this may seem like a backward way of doing things, most hydrology information is presented in terms of DLW and DHW; only rarely is an ICE elevation given directly. The ice force is calculated as twelve inches of ice, of the crushing strength input, times the thickness of the pier shaft.

LANDLOCKED PIERS

For landlocked piers, bypass the DLW prompt and you will be prompted for GR EL, the ground elevation. WOS is calculated only once for this case, on the portion of the pier above ground. Again, the ground elevation must be lower than the top of the cap.

DEAD LOADS and ESTIMATES

Besides the AASHTO wind and ice loads, PIER also calculates the following quantities:

- 1) The pier dead load (shaft and cap separately), assuming a round-ended shaft, square-ended cap, and concrete at 150 pounds per cubic foot.
- 2) The minimum reinforcing required in the shaft (1/2% by AASHTO 4.4.9.4).
- 3) An estimate of the superstructure dead load, assuming a 7" thick deck, jersey rails, steel beams spaced at about 7'-0", and continuous construction. Naturally, all these assumptions place limits on the accuracy of such an estimate; still, experience with a given type of construction will lead to an expected deviation from this estimate, so it still can be useful as a check.
- 4) An estimate of reaction due to one lane of live load, assuming AASHTO lane loading, continuous construction, and no impact.
- 5) The effective moment arm (eccentricity) of one and two lanes of live load, measured along the pier cap from centerline of roadway. See the example for assumptions.

USE as SUBROUTINE

Although written as a stand-alone program, features are incorporated that allow PIER to be called as part of a user-written program; all of its output is stored for use by such a program. For example, a program to calculate and display AASHTO load groups could be written:

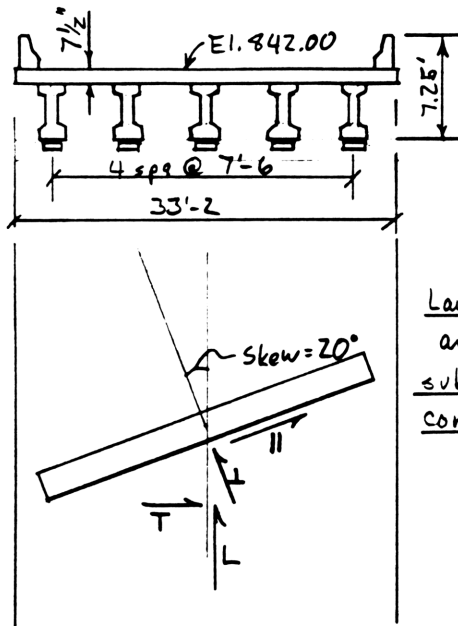
```
01   LBL GROUP
02   XROM "PIER" (called with XEQ)
03   "1 LANE LL"
04   PROMPT
    etc.
```

Such a program was not included in this module for two reasons: differing interpretations of several AASHTO

provisions (for example, the application of WOS to a skewed pier was alluded to); and lack of memory space in the module (a working version took up about 750 bytes). Also important is the fact that output from PIER is essentially an intermediate step in the pier design procedure; an engineer has to digest it to get useful results, i.e. AASHTO load groups. If these results were calculated and output by a program written by a third party (this author), the results would be susceptible to use without checking or even understanding how they came about. As mentioned, many details of pier design by AASHTO are open to debate, and most design offices settle on certain methods; these would doubtless be different than the reasoning this author would employ.

Calculate loads for a T-Pier.

Span = 60'-0" Upstation
and 60'-0" Downstation



Layout
and
subscript
conventions

$$\text{Cap length} = 4 \times 7.5' / \cos 20^\circ = 31.93'$$

USE 36'-0" CAP

Superst. DL:

$$\text{Beams } 5 \times .45 \text{ k/ft} \times 70' = 157.50 \text{ k}$$

$$\text{Diaph } 29' \times 3.75' \times 25' \times .15 = 40.40$$

$$\text{Int. Diaph } 29' \times .43' \times 3' \times .15 = 10.40$$

$$\text{Deck } .625 \times 33.17' \times 70' \times .15 = 217.70$$

$$\text{FWS } .02 \times 30' \times 70' = 42.0$$

$$\text{Rails } 2 \times 37.4' \times 70' = 51.4$$

$$\underline{520.6 \text{ k}}$$

Est. Superst. LL

$$(.64 \text{ k/ft} \times 70' \times 1.25) + 26 \text{ k} = 82.0 \text{ k/line}$$

continuity

Estimate cap depth by WSD:

$$\frac{2}{5} (520.6) + \frac{1}{2} (82.0) + 4' \times 3.33' \times 12' \times .15$$

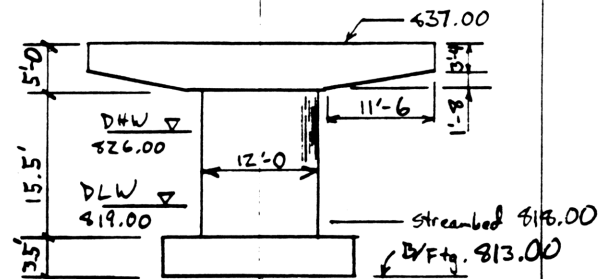
say = 273.24 k

Allow. $V_c = 150 \text{ psi}$ w/ stirrups

Use CAP TH = 3'-4"

$$\therefore \text{Need } d = \frac{273.24 \text{ k}}{.15 \text{ ksi} \times 3.33 \times 12} = 46"$$

USE CAP HT. = 5'-0"



Elevations

Beam	3.75'
Deck	.67'
Haunch	.17'
	<u>4.58'</u>

$$\text{PG Elev. } 842.00$$

$$4.58'$$

$$\text{Brg } .42'$$

$$837.00 \quad \text{T/Cap El.}$$

$$\text{Streambed elev.} \approx 813.00$$

$$\Rightarrow \text{Set B/Ftg Elev.} = \underline{813.00}$$

$$\text{T/Cap } 837.00$$

$$\text{Cap } 5'-0"$$

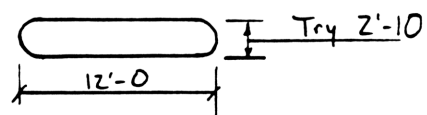
$$\text{Ftg. : est. } 3.5'$$

$$\text{B/Ftg } \underline{813.00}$$

$$15.5' = \text{Shaft}$$

Shaft

$$\text{Cap} = 36'-0" \Rightarrow \text{USE } 12'-0" \text{ SHAFT}$$



(D279)

Cap DL

$$(36' \times 5' - 11.5' \times 1.67') \times 3.33' \times .15 = \underline{40.42 \text{ k}}$$

Shaft DL

$$(12' - 2.43') \times 2.43' = 25.97 \text{ sq'}$$

$$(2.43')^2 \times \pi/4 = \underline{6.30}$$

$$32.27$$

$$32.27 \text{ sq'} \times 15.5' \times .15 = \underline{75.04 \text{ k}}$$

Wind on Substructure

WOS _⊥ :	Area	d	Ad
cap	140.0	2.5'	450
Δ	-19.17	4.44'	-85.19
	160.83 sq'		364.81

@ Design Low Water (DLW):

	160.83		364.81
Shaft	156.00	11.5'	1794.00
	316.83 sq'		2158.81

$$C.G. = 6.41' \text{ From T/Cap} = \underline{13.69' \text{ From T/Ftg. (DLW)}}$$

$$WOS_{\perp} @ DLW = 316.83 \text{ sq'} \times .040 = \underline{12.67 \text{ k}}$$

@ Ice Elev. (= $\frac{DLW + DHW}{2} = 422.50$)

Cap	160.83		364.81
Shaft	114.00	9.75'	1111.50
	274.83		1476.31

$$C.G. @ 5.37' \text{ From T/Cap} = \underline{15.13' \text{ From T/Ftg (Ice)}}$$

$$WOS_{\perp} @ \text{Ice} = 274.83 \text{ sq'} \times .040 \text{ ksf} = \underline{10.99 \text{ k}}$$

WOS_{||} @ DLW

	Area	d	Ad
Cap	16.67 sq'	2.5'	41.67
Shaft	36.83	11.5'	423.53
	53.50 sq'		465.20

$$C.G. = 4.70' \text{ from T/Cap} = \underline{11.40' \text{ from T/Ftg.}}$$

$$WOS_{||} @ DLW = 53.50 \text{ sq'} \times .040 = \underline{2.14 \text{ k}}$$

WOS_{||} @ Ice Elev (= 422.50)

	Area	d	Ad
Cap	16.67		41.67
Shaft	26.92	9.75'	262.44
	43.59 sq'		304.11

$$C.G. @ 6.94' \text{ from T/Cap} = \underline{13.52' \text{ from T/Ftg}}$$

$$WOS_{||} @ \text{Ice} = 43.59 \text{ sq'} \times .04 = \underline{1.74 \text{ k}}$$

Wind on Superstructure

Beam	3.75'
Haunch	.17'
Deck	.67'
Rail	2.67'
	7.25'

$$\text{Area} = 7.25' \times \frac{60' + 40'}{2} = 507.5'$$

$$W_T = 507.5 \text{ sq'} \times .050 \text{ ksf} = \underline{25.38 \text{ k}}$$

$$W_{T,||} = 25.38 \times \cos 20^\circ = \underline{23.84 \text{ k}}$$

$$W_{T,\perp} = \quad \times \sin 20^\circ = \underline{8.64 \text{ k}}$$

Arm : Shaft = 15.5'

Cap = 5.0'

Brg = .42'

Super. $\frac{7.25}{2} = 3.63'$

24.55' To T/Ftg

Wind on Super. (cont'd)

$$W_L = 507.5 \text{ ft} \times .012 \text{ ksf} = \underline{6.09 \text{ k}}$$

$$W_{L_{\parallel}} = 6.09 \times \sin 20^\circ = \underline{2.04 \text{ k}}$$

$$W_{L_{\perp}} = 6.09 \times \cos 20^\circ = \underline{5.72 \text{ k}}$$

$$\text{Arm} = 5.0' + 15.5' = \underline{20.5' \text{ to T/Ftg.}}$$

Wind on Live Load

$$WLL_L = 70' \times .04 \text{ k/ft} = \underline{2.80 \text{ k}}$$

$$WLL_{L_{\perp}} = 2.80 \times \cos 20^\circ = \underline{2.63 \text{ k}}$$

$$WLL_{L_{\parallel}} = 2.80 \times \sin 20^\circ = \underline{.96 \text{ k}}$$

$$\text{Arm} = \underline{20.5' \text{ to T/Ftg.}}$$

$$WLL_T = 70' \times .10 \text{ k/ft} = \underline{7.00 \text{ k}}$$

$$WLL_{T_{\perp}} = 7.0 \times \sin 20^\circ = \underline{2.39 \text{ k}}$$

$$WLL_{T_{\parallel}} = 7.0 \times \cos 20^\circ = \underline{6.54 \text{ k}}$$

$$\begin{aligned} \text{Arm} &= 842.00 - 816.50 + 6.0' \\ &= \underline{31.50' \text{ to T/Ftg.}} \end{aligned}$$

Wind Overturning Force

$$WOF = 70' \times 33.17' \times .020 = \underline{46.43 \text{ k} \uparrow}$$

$$\begin{aligned} WOF_{\parallel} &= 46.43 \text{ k} \times \frac{33.17'}{4} / \cos 20^\circ \\ &= \underline{409.72 \text{ k}} \end{aligned}$$

$$WOF_{\perp} = 0.0$$

Ice @ 300 psi (varies depending on state)

$$\text{Ice}_{\parallel} = 2.43' \times 1' \times 144 \times .30 = \underline{122.26 \text{ k}}$$

$$\text{Ice}_{\perp} = .15 \times 122.26 = \underline{18.34 \text{ k}}$$

$$\text{Arm} = 822.50 - 816.50 = \underline{6.00' \text{ from T/Ftg.}}$$

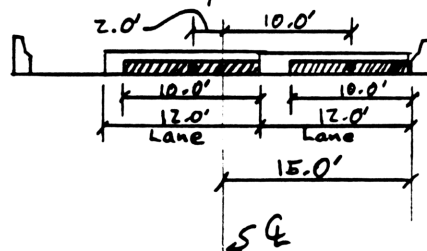
Thermal Forces

Obtain forces from Elastomeric Bearing Pad design.

$$\text{Arm} = 837.00 - 816.50 = \underline{20.5'}$$

Live Load

From sh. 1, 1 lane = 82.0 k (est.)



$$1\text{-Lane Arm} = 10.0' / \cos 20^\circ = \underline{10.64'}$$

$$2\text{-Lane Arm} = \frac{10.0 - 2.0}{2} / \cos 20^\circ = \underline{4.26'}$$

Note: this is the method used by the program. Some states keep the second lane on its side of ϕ . Thus the arm would be calculated:

$$2\text{-Lane Arm} = \frac{10.0 - 5.0}{2} / \cos 20^\circ = \underline{2.66'}$$

The program's method is the more conservative.

XEQ "PIER"

7:43:17 AM 09/27/88

TITLE:
PIERS 1 AND 3 RUN

T/O EL

837.0000 RUN

CAP HT'

5.0000 RUN

FTG HT'

3.5000 RUN

B/F EL

813.0000 RUN

DLW

819.0000 RUN

DHW

826.0000 RUN

SH TH'

2.8333 RUN

SH L'

12.0000 RUN

CAP TH'

3.3333 RUN

CAP L'

36.0000 RUN

WEDGE

11.5000 RUN

WEDGE HT'

1.6667 RUN

SPAN'

70.0000 RUN

DW'0-0

33.1667 RUN

RW W'

30.0000 RUN

SKEW

20.0000 RUN

DECK EL

842.0000 RUN

BM+R HT'

7.2500 RUN

BRG HT'

.4167 RUN

ICE Fc PSI

300.0000 RUN

MIN A-S=23.2394 SQ'

SH DL=75.0439 K

CAP DL=80.4157 K

SU DL EST=513.0727 K

1L LL EST=82.0000 K

1L ARM=10.6418'

2L ARM=4.2567'

Top of Cap

Footing Thickness

Bottom of Footing

Design Low Water El.

Design High Water El.

Shaft Thickness

Shaft Length

Cap Thickness

Cap Length

Triangular Wedge
Length

Avg. span on pier

deck width out-ast

Roadway width

Height of beam, rail,
and deck

bearing height

Min. shaft reinf.
shaft DL
cap DL

} estimates

1 lane }
2 lane } LL eccentricity

DLW

WOS-t=12.6733 K

SH ARM=13.6863'

FTG ARM=17.1863'

WOS-11=2.1400 K

SH ARM=11.8037'

FTG ARM=15.3037'

ICE

WOS-t=10.9933 K

SH ARM=15.1283'

FTG ARM=18.6283'

WOS-11=1.7433 K

SH ARM=13.5225'

FTG ARM=17.0225'

ICE=122.3986 K

SH ARM=6.0000'

FTG ARM=9.5000'

WOF=46.4334 K

11 M=409.7196 'K

M-T=25.3750 K

F-11=23.8447 K

F-t=8.6788 K

SH ARM=24.5417'

FTG ARM=28.0417'

M-L=6.0900 K

F-11=2.0829 K

F-t=5.7227 K

SH ARM=20.5000'

FTG ARM=24.0000'

ML-T=7.0000 K

F-11=6.5778 K

F-t=2.3941 K

SH ARM=31.5000'

FTG ARM=35.0000'

ML-L=2.8000 K

F-11=0.9577 K

F-t=2.6311 K

SH ARM=20.5000'

FTG ARM=24.0000'

Wind on Sub-
structure
@ Design Low Water

WOS @ Ice

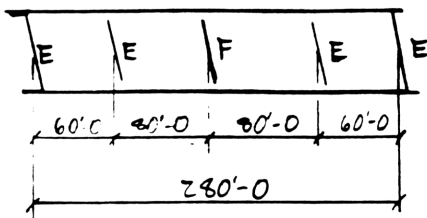
Ice Force

Wind Overturning

Wind on
Superstructure

Wind on LL

Design Bearings for Piers 1,3



Skew 20° RHF
5 Beams @ 7'-6"

$$DL = \frac{520.6}{10 \text{ pads}} = 52.06 \text{ k/pad (avg.)}$$

Live Load

$$\text{Dist.} = \frac{S}{5.5} = \frac{7.5}{5.5} = 1.3636 \text{ ft/bm}$$

$$82.0 \text{ k/line} \cdot \frac{1 \text{ lane}}{2 \text{ wh}} \cdot 1.3636 \frac{\text{wh}}{\text{bm}} \cdot \frac{1 \text{ beam line}}{2 \text{ pads}} = 27.95 \text{ k/pad}$$

Uplift (from computer output or tables)

$$4.70 \text{ k/line} \times \frac{1}{2} \times 1.3636 \times \frac{1}{2} = 1.60 \text{ k/pad}$$

Temperature - One-way ΔT

$$(40' \times 12) \times (6.5 \times 10^{-6}) \times 45^\circ \text{F (concrete beams)} \\ = .281''$$

$$\text{Pier height} = (437.0 - 413.0) - 3.5' = 20.5'$$

Prestr. bms \Rightarrow 10 brg. pads

Desire Length = 1'-4 1/2" for prestr. beams

Use 55 durometer neoprene

Try 10" x 16 1/2" Pads w/ 4-1/2" lams

XROM "NEO"
5:45:52 PM 09/18/88

TITLE:
PIERS 1 AND 3 RUN

HARD

55.0000 RUN

DEL-T°

.2810 RUN

DL K/PAD

52.0600 RUN

MAX LL K/PAD

27.9500 RUN

MIN LL K/PAD

-1.6000 RUN

PADS

10.0000 RUN

PIER HT'

20.5000 RUN

SKEW (DMS)

20.0000 RUN

COLS

RUN

SH TH'

2.8333 RUN

W'

11.0000 RUN

SHIM TH 16'S

2.0000 RUN

PAD L'

16.5000 RUN

W'

10.0000 RUN

#1/8" SHIMS

5.0000 RUN

LAM TH'

.5000 RUN

TH=3.1250"

SH F=6.2264

ALL STRS=0.6382 KSI

DL STRS=0.3155 KSI

B+L STRS=0.4849 KSI

STRAIN=2.8641%

F=4.6736 K/PAD

=46.7359 K/PIER

Fe-11=15.9846 K/PIER

Fe-t=28.5575 K/PIER

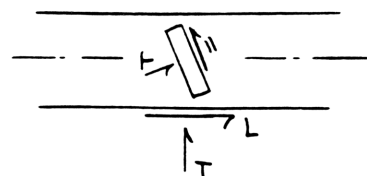
OK

PAD L'

0243

Calculations For FOOTING DESIGN

Skew = 20° RHF



GROUP	P (k)	M _{II} 'k	M _T 'k	GROUP	P (k)	M _{II} 'k	M _T 'k
<u>IA</u>				<u>IV A</u>			
Superst. DL	520.60	—	—	Group IA	835.03	698.60	—
Pier DL 75.04 × 80.42	155.46	—	—	Thermal (from NEO) @ 24'	—	383.63	685.38
Pier Sh. Bury 3227 × 2.5' × 0.624	-5.03	—	—	Group IV A @ 125%	835.03	1082.73	685.38
2 Lanes LL 2 × 82.042	164.00	698.6	—	<u>IV B</u>			
Group IA @ 100%	435.03	698.6	—	Gr. IB	753.03	472.50	—
				Thermal @ 24'	—	383.63	685.38
<u>IB</u>				Group IV B @ 125%	753.03	1256.13	685.38
DL	671.03	—	—	<u>V</u>			
One Lane LL @ 24'	82.0	872.5	—	Group II	624.60	1061.11	598.51
Group IB @ 100%	753.03	872.5	—	T	—	383.63	685.38
<u>II</u>				Group V @ 140%	624.60	1444.74	1283.89
DL	671.03	—	—	<u>VI A</u>			
WOS @ DLW	—	32.75	217.80	Gr. 3A	821.10	1224.17	326.49
WOF	-46.43	409.72	—	T	—	383.63	685.38
W Longit. @ 24.0'	—	-50.00	137.34	Group VI A @ 140%	821.10	1607.80	1011.87
W Transv. @ 28.04'	—	668.64	243.37	<u>VIB</u>			
Group II @ 125%	624.60	1061.11	598.51	Gr. III B	739.10	1398.07	326.49
<u>III A</u>				T	—	383.63	685.38
Gr. IA	835.03	698.60	—	Group VI B @ 140%	739.10	1781.70	1011.87
.3 (W + WOF + WOS)	-13.93	318.33	179.55	<u>VIII A</u>			
WLL Longit. @ 24.0'	—	-22.98	63.15	Gr. IA	835.03	698.60	—
WLL Transv. @ 35.0'	—	230.22	83.79	Ice 122.40 × 9.5'	—	1162.80	174.42
Group III A @ 125%	821.10	1224.17	326.49	Group VIII A @ 140%	835.03	1461.40	174.42
<u>III B</u>				<u>VIII B</u>			
Gr. IB	753.03	472.50	—	Gr. IB	753.03	472.50	—
-13.93	525.57	326.49	—	—	—	1162.80	174.42
Group III B @ 125%	739.10	1398.07	326.49	Group VIII B @ 140%	753.03	2035.30	174.42
				<u>IX</u>			
				Gr. II	624.60	1061.11	598.51
				Ice	—	1162.80	174.42
				Group IX @ 150%	624.60	2223.91	772.93

01+LBL "PIER"

02 42

03 XROM "S"

04 CF 09

05 CF 10

06 "T/C EL"

07 PROMPT

08 STO 01

09 "CAP"

10 XEQ 08

11 STO 08

12 "FTG"

13 XEQ 08

14 STO 05

15 "B/F EL"

16 PROMPT

17 RCL 05

18 +

19 STO 04

20 CLX

21 "DLW"

22 PROMPT

23 STO 02

24 X=0?

25 GTO 00

26 CF 08

27 "DHW"

28 PROMPT

29 RCL 02

30 +

31 2

32 /

33 GTO 03

34+LBL 00

35 "GR EL"

36 PROMPT

37 SF 08

38 SF 09

39+LBL 03

40 STO 03

41 XROM "L"

42 "SH TH"

43 PROMPT

44 STO 12

45 "SH L"

46 PROMPT

47 STO 07

48 "CAP TH"

49 PROMPT

50 STO 11

51 "CAP L"

52 PROMPT

53 STO 15

54 "WEDGE"

55 PROMPT

56 STO 10

57 XEQ 08

58 STO 09

59 XROM "L"

60 "SPAN"

61 PROMPT

62 STO 13

63 "DW'0-0"

64 PROMPT

65 STO 14

66 "RW W"

67 PROMPT

68 STO 17

69 "SKEW"

70 PROMPT

71 STO 06

72 "DECK EL"

73 PROMPT

74 RCL 04

75 -

76 6

77 +

78 STO 33

79 "BM+R"

80 XEQ 08

81 STO 16

82 "BRG"

83 XEQ 08

84 STO 25

85 "ICE Fc PSI"

86 FC? 08

87 PROMPT

88 STO 29

89 XROM "LL"

90 RCL 07

91 RCL 12

92 4.66

93 /

94 -

95 RCL 12

96 *

97 .72

98 *

99 "MIN"

100 XROM "AS"

101 AVIEW

102 4.8

103 /

104 RCL 01

105 RCL 04

106 -

107 STO 00

108 RCL 08

109 -

110 STO 19

Wedge L

Wedge W

Span

Deck W

Footway

Skew

Deck El

T/Fg El

6'

HT to LL

Bm-Rail W

Brg W

Ice Fc

Sh. L

Sh. Th

Sh. Area

Area As

Sh. W. H.

T/C El

T/Fg El

Shed Area

Cap W.

Cap L



$$(p=.005) \cdot 444 = .72$$

$$\frac{.15 \text{ k/cf}}{.72} = \frac{1}{4.8}$$

D245

	X	Y	Z	T	L		X	Y	Z	T	L
111 *	Sh. DL					168 RCL 06	Shaw				
112 "SH DL"						169 COS					
113 XEQ 10						170 /					
114 STO 26						171 "1L"					
115 RCL 15	Cap L					172 XEQ 13	Arm to 1 Lane LL				
116 RCL 08	Cap Ht					173 6	Lane W/2				
117 *						174 LASTX	cos(Shaw)				
118 RCL 09	Wedge Ht					175 /					
119 RCL 10	Wedge L					176 -					
120 *						177 "2L"	Arm to 2 Lanes LL				
121 -	Cap Area					178 X>0?					
122 RCL 11	Cap Th					179 XEQ 13					
123 *						180 XROM "LL"					
124 .15						181 "DLW"					
125 *	Cap DL					182 FC? 08					
126 ST+ 26						183 AVIEW					
127 "CAP DL"											
128 XEQ 10						184+LBL 18					
129 ADV						185 RCL 02	DLWEL				
130 RCL 13	Span					186 FS? 09					
131 75						187 RCL 03	Ice Fl.				
132 /						188 RCL 01	T/Cap Fl.				
133 .42						189 -					
134 -						190 CHS	Expanded Sh.				
135 RCL 14						191 RCL 08	Cap Ht.				
136 *						192 -	Expanded Sh. Ht.				
137 7						193 FS? 10					
138 /						194 RCL 12					
139 .87						195 FC? 10					
140 +						196 RCL 07	Sh. L or Th.				
141 RCL 14						197 FC? 10					
142 10						198 RCL 15					
143 /						199 FS? 10					
144 +						200 RCL 11	Cap L or Th.				
145 RCL 13						201 RCL 08	Cap Wt.	Cap L or Th.	Sh. L or Th.	Sh. Ht.	
146 *						202 CLA					
147 RCL 14						203 XROM "AY"	Area	C.G. Gravity			
148 1.5						204 STO [
149 *						205 *	A \bar{Y}				
150 +						206 RCL 08	Cap Ht.				
151 STO 32						207 RCL 09	Wedge Ht.				
152 "SU D"	Spec. DL Estimate					208 3	C.B. pos				
153 XEQ 04						209 /					
154 RCL 13						210 -					
155 .8						211 RCL 09	Wedge Ht.				
156 *						212 RCL 10	Wedge L				
157 26	Lane Length					213 FS? 10					
158 +						214 CLX					
159 STO 31						215 *	Wedge A				
160 "1L L"						216 ST- [
161 XEQ 04	LL cat.					217 *	Wedge A \bar{Y}	A \bar{Y}			
162 ADV						218 -	Σ A \bar{Y}				
163 RCL 17	Roundup					219 RCL [Σ A				
164 2						220 /	\bar{Y}				
165 /						221 LASTX	Σ A				
166 5						222 25					
167 -						223 /	.04 * A	\bar{Y} \leftarrow neg			

Pier DL in R.26

.644 * 1.25 = .8

	X	Y	Z	T	L		X	Y	Z	T	L
224 RCL 00	SL. HT.					276 -					
225 RCL Z	Y	SL. HT.	WOS	Y		277 STO 30	Ice Area				
226 -	WOS Area					278 XEQ 11					
227 "WOS"						279 XROM "L"					
228 FS? 10											
229 GTO 02						280+LBL 06					
230 "H"						281 RCL 14	Deck W				
231 STO 24	Area					282 3					
232 X<>Y						283 *					
233 STO 23	WOS					284 RCL 13	Span				
234 GTO 07						285 RCL 14	Deck V				
						286 *	Area				
235+LBL 02						287 50					WOF = 20 psf
236 "H"						288 /					
237 STO 22	Area					289 STO 27	WOF A	Width 3			
238 X<>Y						290 "WOF"					
239 STO 21	WOS					291 XEQ 10					
						292 *					
240+LBL 07						293 RCL 06	Skew				
241 XEQ 10	WOS					294 COS					
242 X<>Y						295 /					
243 XEQ 11	Area					296 "H"					
244 FS?C 10						297 XROM "M"					"M" divides by 10
245 GTO 14						298 STO 28					
246 SF 10						299 AVIEW					
247 ADV						300 XROM "LL"					
248 GTO 18						301 CF 07					
						302 RCL 13	Span				
249+LBL 14						303 RCL 16	Br. HT.				
250 XROM "L"						304 *	Area				Wind = 50 psf
251 FS? 08						305 20					
252 GTO 06						306 /					
253 "ICE"						307 "W-T"					
254 FS?C 09						308 XEQ 09					
255 GTO 05						309 STO 35	W-T				
256 STO 18	Area					310 X<>Y					
257 X<>Y						311 STO 34	W-T				
258 STO 17	WOS					312 RCL 00	Skew H.				
259 RCL 23						313 RCL 16	Br. HT.				
260 STO 19	WOS					314 2					
261 RCL 24						315 /	Br. C. Br.				
262 STO 20	Area					316 +					
263 SF 09						317 RCL 25	Br. HT.				
264 AVIEW						318 +	W-T Area				
265 GTO 18						319 STO 36					
						320 XEQ 11					
266+LBL 05						321 SF 07					
267 RCL 29	Ice E.					322 RCL 13					
268 .144	WOS					323 RCL 16					
269 *						324 *	Area				
270 RCL 12	SL. Th.					325 .012					Wind-L = 12 psf
271 *						326 *					
272 STO 29	Ice Area					327 ADV					
273 XEQ 10						328 "W-L"					
274 RCL 03	Ice E.					329 XEQ 09					
275 RCL 04	T/Ety					330 STO 38	W-L				

	X	Y	Z	T	L		X	Y	Z	T	L
331 X<>Y						380+LBL 09	force				
332 STO 37	W-L _H					381 XEQ 10					
333 RCL 00	SL-4+					382 RCL 06	skew				
334 XEQ 11						383 X<>Y	force	skew			
335 CF 07						384 P-R	f _y	f _L			
336 RCL 13	Span					385 FS? 07					
337 .1						386 X<>Y					
338 *					WL = 100 1/4 ft.	387 " F-11"					
339 XROM "L"						388 XEQ 10	f _L	f _H			
340 "WL-T"						389 X<>Y					
341 XEQ 09						390 " F-T"					
342 STO 40	WL-T _H					391+LBL 10					
343 X<>Y						392 "t="					
344 STO 39	WL-T _H					393 ARCL X					
345 RCL 33	Arm					394 "t K"					
346 XEQ 11						395 AVIEW					
347 SF 07						396 END					
348 RCL 13	Span										
349 25											
350 /					WL-L = 40 ft.						
351 ADV											
352 "WL-L"											
353 XEQ 09											
354 STO 42	WL-L _H										
355 X<>Y											
356 STO 41	WL-L _L										
357 RCL 00	SL. Ht.										
358 XEQ 11											
359 XROM "L"											
360 RTN											
361+LBL 08											
362 "t HT"											
363 PROMPT											
364 RTN											
365+LBL 04											
366 "tL EST"											
367 GTO 10											
368+LBL 11											
369 " SH"	SL. Arm										
370 XEQ 13											
371 RCL 05	Ftg. TL										
372 +											
373 " FTG"	Ftg Arm										
374+LBL 13											
375 "t ARM"											
376 ARCL X											
377 XEQ "FT"											
378 AVIEW											
379 RTN											

FOOTING ANALYSIS PROGRAMS

xeq FTG for iterative footing analysis (SIZE 4N+17 for N groups)

SOFTKEYS (USER mode ON)

- C Correct a load group input incorrectly
- F New footing configuration

USER FLAGS

Flag 00	clear	Full output, including MAX ₁₁ and MAX ₊
	set	Skips the MAX for each side

INPUT SUMMARY

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
#GRPS	Number of load groups to be input and stored	One
P-DL	Vertical force due to dead load. Bypass to input Group 1 directly and skip the load breakdown.	No default
M-11	Moment parallel to the long axis of the pier	Zero
M- \perp	Moment perpendicular to the long axis of the pier	Zero
P-LL	Vertical force due to live load	No default
P-2, etc.	Vertical force (parallel moment, perpendicular moment) in Group 2	No default
DIV 1.25,1.4	The load group divisor (from column 14 of AASHTO Table 3.22.1A)	1.0
LIMIT	Allowable pile load or bearing pressure above which the resulting pressure calculated should be displayed. If bypassed (or zero is input), all results are displayed. If a value greater than zero is input, only displays MAX's greater than LIMIT and MIN values less than zero.	Zero
# PILES	Number of piles. Bypass for a spread footing.	Zero

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
S-11 (piles)	Section modulus of a pile footing parallel to the long axis of the pier, in feet	No default
S-⊥ (piles)	Section modulus of a pile footing perpendicular to the long axis of the pier, in feet	No default
L' (11) (spread)	Length of the footing in the direction parallel to the long axis of the pier, in feet	No default
W' (⊥) (spread)	Width of the footing in the direction perpendicular to the long axis of the pier	No default
FTG DL, K	Footing deadload in kips. Should include the weight of earth on the footing and the effect of buoyancy on the footing, if any.	No default

OUTPUT SUMMARY (Items in brackets { } are displayed only if flag 00 is clear)

1 MAX	The maximum pressure or pile load due to Group 1, in kips per square foot or kips respectively
{ (11) MAX }	The average maximum pressure on the long side of the footing
{ (⊥) MAX }	The average maximum pressure on the short side of the footing
1 MIN	The minimum pressure or pile load due to Group 1; If less than zero, appends UPL to denote uplift.
GR _ GOV:	Displays the group producing the largest MAX
DL LL OT Σ	The dead load, live load, and overturning components and total load of the governing group, in <u>tons</u> per square foot or tons. This is the full unDIVided load.

PROGRAM FLAGS

Flag 05	clear set	LIMIT of zero (all groups displayed) Only display MAX's greater than LIMIT and MIN's less than zero
Flag 06	clear set	Inputting DL and LL DL and LL have been input

Flag 08	clear set	FTG SHFTG
Flag 09	clear set	Spread footing Pile footing
Flag 10	clear set	Have DL and LL--do breakdown No DL or LL--no breakdown

STORAGE REGISTER USE

<u>Register</u>	<u>Value</u>
00	LIMIT
01	# PILES or area
02	S_{\perp}
03	S_{11}
04	FTG DL, K
05	Counter
06	Governing MAX
07	Governing Group
08	(11) MAX
09	(\perp) MAX
10	Governing MAX, unDIVided
11	DL P
12	DL M-11
13	DL M- \perp
14	LL P
15	LL M-11
16	LL M- \perp
17	Group 1 P
18	Group 1 M-11
19	Group 1 M- \perp
20	Group 1 L.F. (1.0)
21	Group 2 P
22	Group 2 M-11
23	Group 2 M- \perp
24	Group 2 L.F.
25	Group 3 P etc.

FOOTING ANALYSIS PROGRAM

Program FTG is a general-purpose pier and column footing analysis program, an extended version of SHFTG. It analyzes both spread footings on rock and pile footings. It accepts any number of load groups, then prompts for and analyzes different footing configurations until a satisfactory solution is found. This is the biggest difference between FTG and SHFTG, which accepts one footing configuration and then analyzes different load groups on that footing.

FTG first asks for the number of load groups to be input, in order to check the SIZE (storage register) requirement. It then begins prompting for load groups. A load group consists of a vertical load, moments in two planes, and a "load factor" by which these are divided. See the example. The vertical load should not include the weight of the footing itself (FTG DL), since this is input with each trial footing's properties. This way, if the footing size changes from the original estimate the results are still correct.

LOAD GROUPS

FTG assumes that Load Group 1 consists of only dead load and live load. The vertical load and moments due to each of these are input separately and combined by the program to comprise its Group 1; it then prompts for Group 2 and so on. Group 1 is thus not input directly. The DL and LL are also stored separately so that a breakdown may be made of the governing load group into DL, LL, and the remainder, OT (overturning, i.e. wind and other lateral forces). After the program has analyzed a given footing configuration for all the load groups, it displays the governing group and calculates this breakdown for that group; these loads have not been divided by the DIV factor input. This is made clear in the example. If this breakdown is not wanted, bypass the dead load prompt (P-DL). The program will then prompt for Group 1 in its entirety, then Group 2 and so on as before. All this is also made more clear in the examples. Note that the program assumes the governing group includes the Live Load (axial and moments) input at the beginning of the run. If this is not true, the load breakdown at the end will be incorrect.

If there is only one load group to be analyzed, the P-DL prompt may be bypassed and the group entered directly, or (if it consists of just DL and LL) the component loads can be input, and the program will give the breakdown at the end.

CORRECTING INPUT ERRORS

If a mistake is made when inputting load groups, press softkey C to correct it. This must be done before finishing inputting the next group. If C is pressed during the moment or DIV input, the program returns to the beginning prompt (P, K) for that load group; if before or during input of the vertical load, it returns to the previous group's prompt. Softkey C may be pressed more than once to get back to a previous load group. However, any load information input after that group will have to be re-input. Softkey C will not return to the P-DL or P-LL prompt from the P-2 prompt sequence; it will prompt for P-1, etc., not correcting the DL or LL values input.

If a mistake is made inputting footing properties, press softkey F. Softkey F can be used at any time; it will interrupt an analysis and jump immediately to the footing input section.

"LIMIT:" LONG and SHORT OUTPUT

FTG has two basic modes of operation, controlled by the user's input to the LIMIT prompt. If a "limit" of zero is input (the default value), the program calculates and displays the results of all the load groups upon the given footing, as explained in the next several paragraphs. If, on the other hand, the designer wants to check out several different footing configurations quickly, this full output becomes rather time-consuming. If, for instance, the maximum allowable pile load is 80 kips, input a LIMIT of, say, 75. Then the program will only display the load groups which cause a maximum pile load over 75 kips or a minimum pile load of less than zero (uplift). Thus, a number of footing trials can be run quickly. This also works, of course, for spread footings; the LIMIT is then in kips per square foot.

Assuming the LIMIT prompt is bypassed (or zero input), FTG calculates and displays the maximum and minimum corner or pile load for each load group; if the minimum is less than zero (i.e. an uplift exists) this is noted by UPL displayed after the pressure value. The program assumes that the footing is effective in uplift. That is, its calculations are based on the full footing properties whether or not uplift exists. In reality, of course, if a spread footing undergoes uplift, a neutral axis shift occurs and the maximum corner pressure increases. FTG does not calculate this increase.

If flag 00 is clear, the program also displays the average maximum load along each side as defined by

AASHTO 4.4.6.1 and ACI 15.4.1 for reinforcing design. See the examples. As mentioned, it then displays the governing load group, and, if the dead load and live load forces and moments were input separately, the breakdown of forces for that group.

FLAG 00

If flag 00 is set, display of the side maximums (11)MAX and (1-)MAX is skipped. This is the only function of flag 00; most of the output control is handled by the LIMIT feature. FTG is thus one of the few programs in this package whose normal mode of operation will probably be with flag 00 set.

SPREAD and PILE FOOTINGS

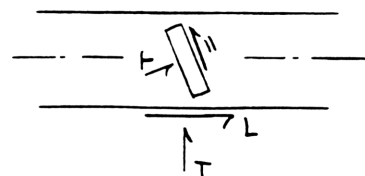
Spread footings and pile footings are handled differently in the input sequence. For a pile footing, the designer must calculate the section (moment) properties in each direction, and then input the number of piles (# PILES) and these properties. For a spread footing, the # PILES prompt is bypassed (press R/S without keying anything in). The program then prompts for the footing length and width, and figures the section properties itself. For either of these, the designer needs to calculate and input the FTG DL, the selfweight of the footing being analyzed (unless it was included in all the load groups--this takes a lot of confidence). An extra calculator is handy for this, as well as pile property calculations. However, these calculations can all be done on the HP-41 after the prompt if a few extra numbers on the printout aren't objectionable. For instance, for a 9'x12' footing 3'-0 thick, after the FTG DL prompt, key in:

```
12
Enter/
9
*
3
*
.15
*
R/S
```

If not using a printer, of course, this is no problem at all.

The calculations for pile group properties (S-11 and S-1-) can also be done during a run, as no information is being carried in the stack. However, it is often more convenient to use another calculator or work out a few trials beforehand. Although this sounds awkward, in practice it works quite well.

Skew = 20° RHF



GROUP	P (k)	M _{II} 'k	M _T 'k	GROUP	P (k)	M _{II} 'k	M _T 'k
<u>IA</u>				<u>IV A</u>			
Superst. DL	520.60	—	—	Group IA	435.03	698.60	—
Pier DL 75.04' 80.42	155.46	—	—	Thermal (from NEO) @ 24'	—	383.63	645.38
Pier Sh. Buoy 3227' ± 2.5' 0.624	-5.03	—	—	Group IV A @ 125%	435.03	1082.23	645.38
2 Lanes LL 2 × 82.0426'	164.00	698.6	—	<u>IV B</u>			
Group IA @ 100%	435.03	698.6	—	Gr. IB	753.03	472.50	—
				Thermal @ 24'	—	383.63	645.38
<u>IB</u>				Group IV B @ 125%	753.03	1256.13	645.38
DL	671.03	—	—	<u>V</u>			
One Lane LL @ 10.44'	42.0	872.5	—	Group II	624.60	1061.11	598.51
Group IB @ 100%	753.03	872.5	—	T	—	383.63	645.38
<u>II</u>				Group V @ 140%	624.60	1444.74	1283.89
DL	671.03	—	—	<u>VI A</u>			
WOS @ DLW	—	32.75	217.80	Gr. 3A	821.10	1224.17	326.49
WOF	-46.43	409.72	—	T	—	383.63	645.38
W Longit. @ 24.0'	—	-50.00	137.34	Group VI A @ 140%	821.10	1607.80	1011.87
W Transv. @ 28.04'	—	668.64	243.37	<u>VI B</u>			
Group II @ 125%	624.60	1061.11	598.51	Gr. III B	739.10	1398.07	326.49
<u>III A</u>				T	—	383.63	645.38
Gr. IA	435.03	698.60	—	Group VI B @ 140%	739.10	1781.70	1011.87
.3 (W + WOF + WOS)	-13.93	318.33	179.55	<u>VIII A</u>			
WLL Longit. @ 24.0'	—	-22.98	63.15	Gr. IA	835.03	698.60	—
WLL Transv. @ 35.0'	—	230.22	83.79	Ice 122.40' ± 9.5'	—	1162.80	174.42
Group III A @ 125%	821.10	1224.17	326.49	Group VIII A @ 140%	835.03	1461.40	174.42
<u>III B</u>				<u>VIII B</u>			
Gr. IB	753.03	472.50	—	Gr. IB	753.03	472.50	—
-13.93	525.57	326.49	—	—	—	1162.80	174.42
Group III B @ 125%	739.10	1398.07	326.49	Group VIII B @ 140%	753.03	2035.30	174.42
				<u>IX</u>			
				Gr. II	624.60	1061.11	598.51
				Ice	—	1162.80	174.42
				Group IX @ 150%	624.60	2223.91	772.93

Design a footing for the T-Pier loads given. (See documentation for program PIER.)

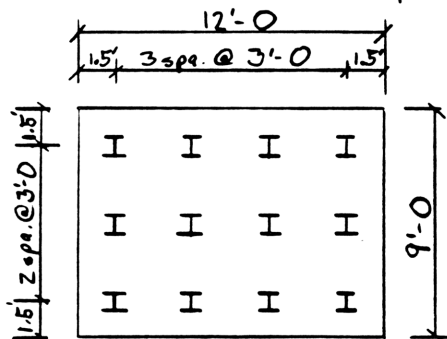
Check Groups IA (1)
 II (2)
 IIIA (3)
 VIA (4)
 and VI B (5)

Use HP 12x53 piling @ 7.0 ksi:

$$15.5" \times 7 = 108.5, \text{ say } \underline{110 \text{ kips}}$$

$$\text{Group I} \rightarrow \frac{835 \text{ k}}{110} = 7.6 \rightarrow 8 \text{ piles}$$

Estimate 12 piles: try minimum footing configuration. Estimate 3.0' thick. Neglect earth on ftg. Consider under full buoyancy.



$$\text{Footing DL} = 12 \times 9 \times 3 \times (1.5 - 0.0624) \\ = \underline{28.34 \text{ k}}$$

$$S_{II} = (6 \times 1.5^2 + 6 \times 4.5^2) / 4.5' = 30.0'$$

$$S_{\perp} = 8 \times 3' = 24.0'$$

0296

XEQ "FTG"		
# GRPS	5.0000	RUN
	8:22:40 AM	03/21/89
TITLE:		
T-PIER 20d SKEW		RUN

P-DL K	671.0300	RUN
M-11 'K	0.0000	RUN
M-T 'K	0.0000	RUN

P-LL K	164.0000	RUN
M-11 'K	698.6000	RUN
M-T 'K		RUN

P-2 K	624.6000	RUN
M-11 'K	1,061.1100	RUN
M-T 'K	598.5100	RUN
DIV 1.25, 1.4	1.2500	RUN

P-3 K	821.1000	RUN
M-11 'K	1,224.1700	RUN
M-T 'K	326.4900	RUN
DIV 1.25, 1.4	1.2500	RUN

P-4 K	821.1000	RUN
M-11 'K	1,607.8000	RUN
M-T 'K	1,011.8700	RUN
DIV 1.25, 1.4	1.4000	RUN

P-5 K	739.1000	RUN
M-11 'K	1,781.7000	RUN
M-T 'K	1,011.8700	RUN
DIV 1.25, 1.4	1.4000	RUN

LIMIT		

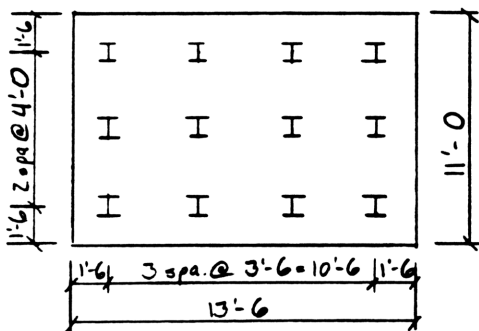
Run preliminary trial with
LIMIT of 100 k (slightly
below allowable of 110 kips) to
check for overload, uplift

LIMIT
100.0000 RUN
PILES
12.0000 RUN
S-11
30.0000 RUN
S-T
24.0000 RUN
FTG DL, K
28.3800 RUN

2 MIN=-5.8933 K: UPL
3 MAX=100.1595 K
4 MAX=118.9604 K
4 MIN=-17.8318 K: UPL
5 MAX=118.2199 K
5 MIN=-26.8533 K: UPL

GR 4 GOV:

Both maximum downward
load and uplift are too high -
increase footing size and
re-run. Since these were
close, expect next trial to
work \Rightarrow set LIMIT = 0.



Greater moment arm \Rightarrow make 3'-6 thick

$$\text{Ftg DL} = 13.5' \times 11' \times 3.5' \times (15 - 0.629) = 45.53 \text{ k}$$

$$S_{11} = 6 \times (1.75^2 + 5.25^2) / 5.25' = 35.0'$$

$$S_{\perp} = 4 \times (4')^2 / 4' = 32.0'$$

LIMIT
0.0000 RUN
PILES
12.0000 RUN
S-11
35.0000 RUN
S-T
32.0000 RUN
FTG DL, K
45.5300 RUN

1 MAX=93.3400 K
(11) MAX=93.3400 K
(T) MAX=73.3800 K
1 MIN=53.4200 K

2 MAX=83.8920 K
(11) MAX=68.9293 K
(T) MAX=59.6381 K
2 MIN=5.4586 K

3 MAX=93.9186 K
(11) MAX=85.7564 K
(T) MAX=65.9376 K
3 MIN=21.6321 K

4 MAX=106.9837 K
(11) MAX=84.3974 K
(T) MAX=74.1715 K
4 MIN=-3.8135 K: UPL

5 MAX=105.6518 K
(11) MAX=83.0654 K
(T) MAX=69.2906 K
5 MIN=-12.2434 K: UPL

GR 4 GOV:
DL=29.8567 T
LL=16.8133 T
OT=28.2186 T

 $\Sigma = 74.8886 \text{ T}$

Check Group 4:

$$\text{DL: } \frac{671.03 + 45.53}{12} = 59.71 \text{ k}$$

$$\text{LL: } \frac{164.00}{12} + \frac{698.6'^k}{35.0'} = 33.63 \text{ k}$$

$$\Sigma: \frac{821.1 + 45.53}{12} + \frac{1607.90'^k}{35.0'} + \frac{1011.87'^k}{32.0'} = 149.78 \text{ k}, \div 1.40 = 106.98 \text{ k}$$

$$\text{OT} = \text{difference} = 149.78 - 59.71 - 33.63 = 56.44 \text{ k}$$

Check the same footing
as a spread footing.

$$L = 13.5'; W = 11.0'$$

$$\Sigma = \frac{821.10 + 45.53}{148.50 \text{ sf}} = 5.436 \text{ ksf}$$

$$\frac{1607.80 \text{ ft}^3}{334.125} = 4.812$$

$$\frac{1011.87 \text{ ft}^3}{272.25} = 3.717$$

$$\begin{aligned} \text{Max } \Sigma &= 14.365 \text{ ksf} \\ &\div 1.40 \\ &= 10.26 \text{ ksf} \end{aligned}$$

$$\text{Min } \Sigma = 5.436 - (4.812 + 3.717) / 1.40$$

$$= -2.69 \text{ ksf} / 1.40$$

$$= -1.92 \text{ ksf (UPLIFT)}$$

(Note that for a spread footing, a corner in uplift renders the above analysis invalid. Most designers would accept only a small amount of uplift. This footing should thus be upsized)

Had flag 00 been clear, the average side loads would be:

$$(11) \text{ Max} = (5.436 + 4.812) / 1.40$$

$$= 7.61 \text{ ksf}$$

$$(1) \text{ Max} = (5.436 + 3.717) / 1.40$$

$$= 6.82 \text{ ksf}$$

Again check group 4:

$$\text{Area} = 13.5' \times 11' = 148.50 \text{ sf}$$

$$S_{11} = 13.5^2 \times 11 / 6 = 334.125 \text{ ft}^3$$

$$S_1 = 11^2 \times 13.5 / 6 = 272.25 \text{ ft}^3$$

These are used for the methods described in ACI 15.4.1 and AASHTO 4.4.6.1 if these are used by the designer.

50 SHEETS 3 SQUARE
42 381 100 SHEETS 3 SQUARE
42 389 200 SHEETS 3 SQUARE



FOOTING ANALYSIS PROGRAMS

xeq SHFTG for simple footing analysis (SIZE 004)

SOFTKEYS

SHFTG uses no softkeys. However, the softkeys C and F are active in FTG when SHFTG is in use, and should be avoided.

USER FLAGS

Flag 00	clear	Full output
	set	Title block not printed; no other effect

INPUT SUMMARY

<u>Prompt</u>	<u>Input</u>	<u>Default</u>
# PILES	Number of piles. Bypass for a spread footing.	Zero
S-11 (piles)	Section modulus of a pile footing parallel to the long axis of the pier, in feet	No default
S-⊥ (piles)	Section modulus of a pile footing perpendicular to the long axis of the pier, in feet	No default
L' (11) (spread)	Length of the footing in the direction parallel to the long axis of the pier, in feet	No default
W' (⊥) (spread)	Width of the footing in the direction perpendicular to the long axis of the pier	No default
P	Vertical force in kips	No default
M-11	Moment parallel to the long axis of the pier	Zero
M-⊥	Moment perpendicular to the long axis of the pier	M-11

OUTPUT SUMMARY

MAX The maximum pressure or pile load, in kips per square foot or kips respectively

MIN The minimum pressure or pile load; if less than zero, appends UPL to denote uplift.

PROGRAM FLAGS

Flag 08	clear	FTG
	set	SHFTG
Flag 09	clear	Spread footing
	set	Pile footing

STORAGE REGISTER USE

<u>Register</u>	<u>Value</u>
00	not used
01	# PILES or area
02	S_{\perp}
03	S_{11}

COMPARISON BETWEEN PROGRAMS

<u>Feature</u>	<u>SHFTG</u>	<u>FTG</u>
FTG DL	Must be included in loads	Input separately
Divisor	Must be included in loads	Input separately
Loads	Not stored	Stored
Best use	Checking/analysis	Design
SIZE	004	$4N + 17$

SHORT FOOTING ANALYSIS PROGRAM

Program SHFTG is a simple pier and column footing analysis program, an much-simplified version of FTG. It analyzes both spread footings on rock and pile footings. It prompts for a footing configuration, then analyzes any number of load groups; these load groups, however, are not stored and must be re-input for each footing trial. This is the main difference between SHFTG and FTG, which accepts all the load groups up front, and then analyzes different footing configurations on all the groups input. SHFTG is thus best used for checking a given footing configuration, while program FTG is better suited for iterating to the best configuration. See the preceding page for a comparison.

INPUT

SHFTG begins by asking for the footing configuration; the input sequence is the same as for program FTG. It then begins prompting for load groups. A load group consists of a vertical load and moments in two planes. See the example. The vertical load must include the weight of the footing; SHFTG does not allow input of the footing deadload separately as FTG does. It must also have been divided by the "load factor" (1.0, 1.25, 1.40, or 1.50) as SHFTG does not do this itself. The moments should both be positive. If one is negative, pressures or pile loads will be at non-extreme corners of the footing; if both are negative, MAX and MIN will be switched.

MAXIMUMS and MINIMUMS

SHFTG calculates and displays the maximum and minimum corner or pile load for each load group; if the minimum is less than zero (i.e. an uplift exists) this is noted by UPL displayed after the pressure value. The program assumes that the footing is effective in uplift. That is, its calculations are based on the full footing properties whether or not uplift exists. In reality, of course, if a spread footing undergoes uplift, a neutral axis shift occurs and the maximum corner pressure increases. SHFTG does not calculate this increase.

FOOTING TYPES

Spread footings and pile footings are input differently. For a pile footing, the designer must calculate the section (moment) properties in each direction, and then input the number of piles and these properties; an extra calculator is handy for this. The

calculations can be done on the HP-41 during the run; no data is being carried in the stack. See the program listing and stack analysis for clarification.

For a spread footing, on the other hand, the # PILES prompt is bypassed (press R/S without keying anything in). The program then prompts for the footing length and width, and figures the section properties itself.

FOOTING SHAPE

A spread footing must be rectangular in order that its section properties will be calculated correctly by the program. A pile footing, on the other hand, has all its properties input by the user. Thus, if a non-rectangular spread footing is to be analyzed, its properties must be calculated by hand and input as for a pile footing. All output will be in Kips and Tons, instead of KSF and TSF as for a normal spread footing.

	X	Y	Z	T	L		X	Y	Z	T	L
01*LBL "FTG"						56 4					
02 CF 08						57 /					
03 E	1					58 INT					
04 "# GRPS"						59 4					
05 PROMPT	* Grps					60 *	New count	17.5	Frc		
06 4						61 +					
07 +						62 INT	New Count	Frc			
08 4											
09 *	20,24,40,					63*LBL 07	Counter				
10 ENTER↑	Reg.	Reg.				64 +					
11 XROM "S"	Reg.					65 STO 05					
12 E3	1000					66 GTO 06					
13 /											
14 11						67*LBL 01					
15 +	11.00R					68 SF 10					
16 STO 05	counter					69 6					
17 SF 06						70 ST+ 05	Counter				
18 "DL"											
19 ASTO X						71*LBL 06					
20 CF 22						72 CF 06					
21 XEQ 12						73 RCL 05	counter				
22 XROM "L"						74 4					
23 "LL"						75 /					
24 ASTO X						76 3					
25 XEQ 12						77 -	Group*				
26 CF 10						78 XROM "L"					
27 RCL 14	LL P										
28 RCL 11	DL P					79*LBL 12	Group*				
29 +						80 "P--"					
30 STO 17	G-I P					81 XROM "AX"					
31 RCL 15	LL M _u					82 "I K"	P				
32 RCL 12	DL M _u					83 PROMPT					
33 +						84 FC? 22					
34 STO 18	G-I M _u					85 GTO 01					
35 RCL 16	LL M _u					86 STO IND 05					
36 RCL 13	DL M _u					87 ISG 05					
37 +						88 CLX					
38 STO 19	G-I M _u					89 XEQ 05	M _u				
39 SIGN	1					90 STO IND 05					
40 STO 20	G-I M _u					91 ISG 05					
41 3						92 CLX					
42 ST+ 05						93 PROMPT	M _u				
43 ISG 05						94 STO IND 05					
44 GTO 06						95 ISG 05					
45 GTO F						96 FS? 06					
						97 RTN					
						98 SIGN	1				
46*LBL C	Counter					99 "DIV 1.25,1.4"	Divisor				
47 RCL 05	.00 Reg					100 PROMPT					
48 FRC	Counter	Frc.				101 STO IND 05					
49 LASTX						102 ISG 05					
50 17.5						103 GTO 06					
51 -											
52 LASTX	17.5					104*LBL F					
53 X<>Y	Count=17.5	17.5	Frc			105 XROM "L"					
54 X<0?						106 CF 05					
55 GTO 07						107 CLX					
						108 STO 06	Mex				

	X	Y	Z	T	L		X	Y	Z	T	L	
109 *LIMIT*						164*LBL 23						
110 PROMPT						165 FC? 05						
111 STO 00	Limit					166 ADV						
112 X>0?												
113 SF 05						167*LBL 02	P					
						168 RCL IND 05	Fig DL					
114*LBL 08						169 RCL 04	ΣP					
115 CF 09						170 +	#Piles or Area					
116 CLX						171 RCL 01	P/A					
117 *# PILES*						172 /	P/A					
118 PROMPT						173 STO 08	M ₁₁					
119 X=0?						174 STO 09	S ₁₁					
120 GTO 09	#Piles					175 ISG 05	M/S ₁₁					
121 STO 01						176 RCL IND 05	(Max ₁₁)					
122 SF 09						177 RCL 03	M ₁					
123 *S-11*						178 /	S ₁					
124 PROMPT						179 ST+ 08	M/S ₁					
125 STO 03	S ₁₁					180 ISG 05	Max					
126 *S-t*						181 RCL IND 05	Min					
127 PROMPT						182 RCL 02	Countor					
128 STO 02	S _t					183 /	4					
129 GTO 02						184 ST+ 09	Countor					
						185 +	Max					
130*LBL 09						186 ST- 2	Min					
131 *L'(11)*						187 +	Max					
132 PROMPT	Length					188 RCL 05	Countor					
133 ENTER↑	L	L	L			189 4	4					
134 ENTER↑						190 /						
135 *W'(t)*						191 4						
136 PROMPT	W	L	L			192 -						
137 *	Area					193 CLA	Group*					
138 STO 01						194 XROM "AX"	"Group"					
139 LASTX	W					195 RDN	Max					
140 *	W ² L					196 ISG 05	Min					
141 X<>Y	L					197 RCL IND 05	Divisor					
142 RCL 01	Area					198 ST/ 08	Max					
143 *	WL ²					199 ST/ 09	Min					
144 6						200 ST/ 2						
145 /	WL ² / ₆					201 /	Max					
146 STO 03	S ₁₁					202 RCL 00	Limit					
147 X<>Y	W ² L					203 X>Y?						
148 LASTX	6					204 FC? 05						
149 /						205 X<0?						
150 STO 02	S _t					206 GTO 10						
						207 ASTO X	"Group"					
151*LBL 02						208 X<>Y	Max					
152 FS? 08						209 XEQ 14	"#"					
153 RTN						210 RCL 06	MAX					
154 *FTG DL, K*						211 X>Y?	Max					
155 PROMPT						212 GTO 02	Max					
156 STO 04	Fig DL					213 RDN	"Gr"					
157 XROM *L*						214 STO 06	Min					
158 RCL 05						215 ST* L						
159 FRC	.00 Reg					216 RDN	"Gr"					
160 17						217 STO 07	Min					
161 +						218 LASTX	Max (unDIVided)					
162 STO 05	Countor					219 STO 10						
163 GTO 02						220 RT						

x>y?
and } gto 10
FS:05 }

MAX is largest
of Max's

Max is new MAX

Div. = (Max/Div)
in Last X

0304

	X	Y	Z	T	L		X	Y	Z	T	L
221+LBL 02						274 RCL 01	Area				
222 FC? 05						275 /	LL P/A				
223 FS? 00						276 RCL 15	LL M ₁₁				
224 GTO 02						277 RCL 03	S ₁₁	M ₁₁	P/A	Max	
225 - (11)-						278 /	M/S	P/A	Max		
226 X(>) 08	Max ₁₁					279 +	Σ LL				
227 XEQ 14						280 RCL 16	LL M ₁₁				
228 - (1)-						281 RCL 02	S ₁₁	M ₁₁	Σ LL	Max	
229 X(>) 09	Max ₁₁					282 /	Σ LL	Max			
230 XEQ 14						283 +					
						284 ST- 10					
231+LBL 02			"Gr#"	Min		285 "LL"					
232 CLA						286 XEQ 13	Σ LL				
233 ARCL 2			"Gr#"			287 RCL 10	Reminder				Reminder
						288 RND	OT	Σ LL	Max	Max	= Overturning
						289 "OT"					
234+LBL 10						290 XEQ 13					
235 R+	Min					291 XROM "L"					
236 X>0?					X < 0?	292 "Σ"					
237 FC? 05					or	293 R+	Max (undivided)				
238 XEQ 03					FC? 05	294 XEQ 13					
						295 GTO F					
239+LBL 11						296+LBL 05					
240 ISG 05						297 - M-11 "K"					
241 GTO 23						298 PROMPT	M ₁₁				
242 RCL 05	counter					299 - M-1 "K"					
243 FRC						300 RTH					
244 17											
245 +						301+LBL "SHFTG"					
246 STO 05	counter					302 SF 08					
247 XROM "LL"						303 3					
248 RCL 10	undivided Max					304 XROM "S"	Reg.				
249 RCL 04	Fig DL					305 XEQ 08					
250 RCL 11	DL P					306 XROM "L"					
251 +	Σ DL P	Max									
252 RCL 01	Area										
253 /	DL P/A					307+LBL 04					
254 RCL 12	DL M ₁₁					308 "P, K"					
255 RCL 03	S ₁₁	M ₁₁	DL P/A	Max		309 PROMPT	P				
256 /	DL M ₁₁	P/A	Max			310 RCL 01	Area				
257 +	Σ DL	Max				311 /	P/A				
258 RCL 13	DL M ₁₁					312 XEQ 05	M ₁₁				
259 RCL 02	S ₁₁	M ₁₁	Σ DL	Max		313 RCL 03	S ₁₁				
260 /	Σ DL					314 /	M/S ₁₁	P/A			
261 +						315 PROMPT	M ₁₁				
262 ST- 10						316 RCL 02	S ₁₁	M ₁₁	M/S ₁₁	P/A	
263 "GR"						317 /	M/S ₁₁				
264 ARCL 07						318 +	Σ M/S	P/A			
265 "T GOV."						319 +	Max				
266 AVIEW						320 CLA					
267 FC? 10						321 XEQ 14					
268 FS? 05						322 LASTX	Σ M/S	Max			
269 GTO F						323 ST+ X					
270 "DL"						324 -	Min				
271 XEQ 13	Σ DL					325 CLA					
272 RDN	Max					326 XEQ 03					
273 RCL 14	LL P					327 XROM "LL"					
						328 GTO 04					

	X	Y	Z	T	L		X	Y	Z	T	L
329+LBL 13	Load, k										
330 2						367+LBL -RP-	X	Y			
331 /	Load, Tms					368 R-P	d	0'			
332 -t=-						369 STO a	d				X
333 ARCL X						370 X<> L	X	0'			d
334 -t T-						371 X<>Y	0'	X			
335 GTO 07						372 FS? 03					
						373 HMS	0	X			
336+LBL 03						374 X<>Y	X	0			
337 -t MIN=-						375 X<> L	d-0'	0			X
338 GTO 02						376 CLX	0				
						377 X<> a	d	0			X
339+LBL 14						378 RTN					
340 -t MAX=-											
						379+LBL -PR-	d	0'			
341+LBL 02						380 X<>Y	0'	d			
342 ARCL X						381 FS? 03					
						382 HR	0	d			
343+LBL 00						383 X<>Y	d	0			
344 -t K-						384 P-R	X	Y			d
						385 END					
345+LBL 07											
346 FC? 09											
347 -tSF-											
348 X<0?	Load										
349 -t: UPL-											
350 XEQ -PV-											
351 RTN											
352+LBL -YZ-	X	Y	Z	T							
353 RDN	Y	Z	T	X							
354 X<>Y	Z	Y	T	X							
355 R↑											
356 RTN	X	Z	Y	T							
357+LBL -YT-	X	Y	Z	T							
358 RDN	Y	Z	T	X							
359 X<> Z	T	Z	Y	X							
360 R↑											
361 RTN	X	T	Z	Y							
362+LBL -ZT-	X	Y	Z	T							
363 R↑	T	X	Y	Z							
364 X<> T	Z	X	Y	T							
365 RDN											
366 RTN	X	Y	T	Z							

