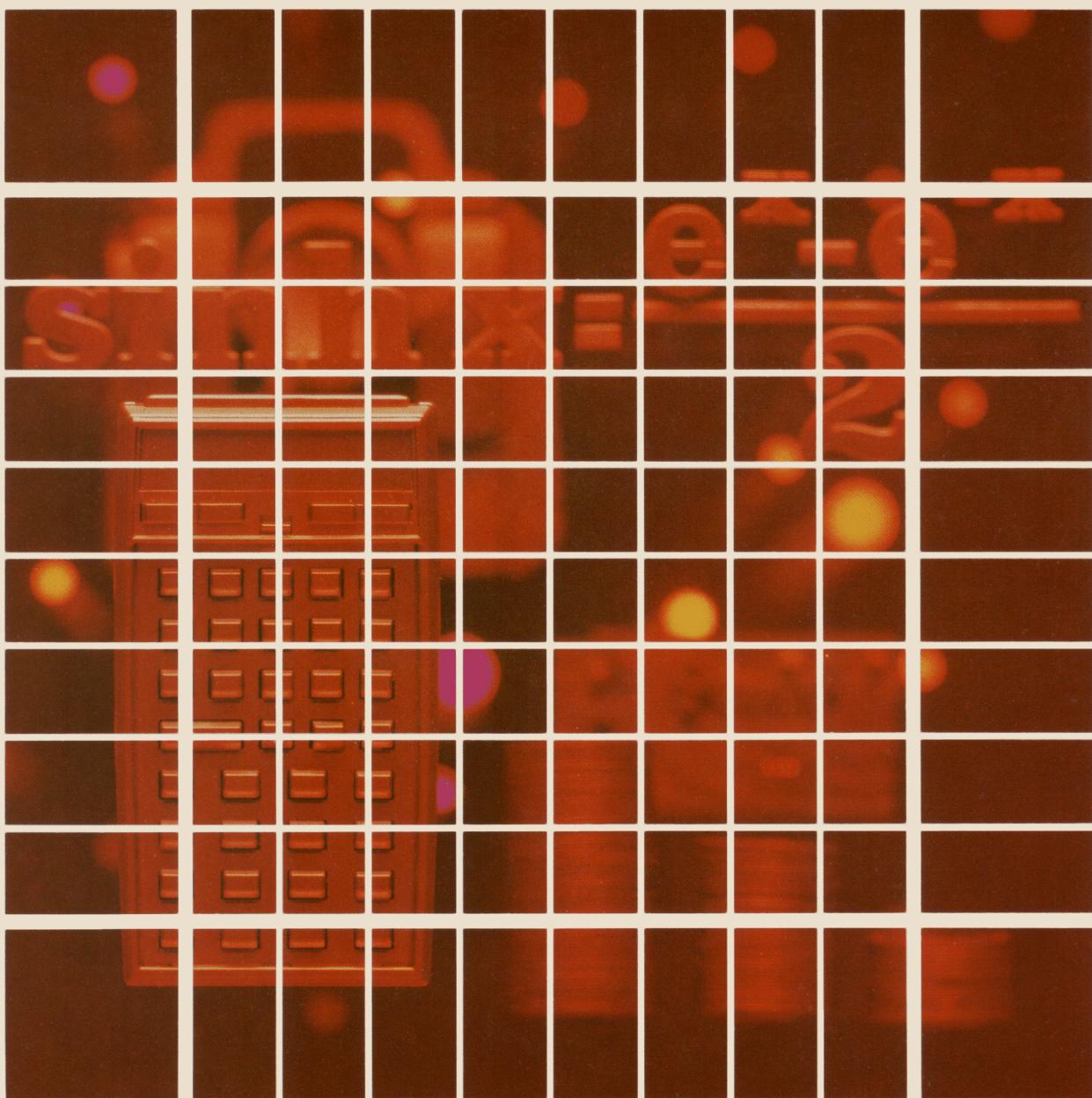


HEWLETT-PACKARD

HP-41C

USERS'
LIBRARY SOLUTIONS
Mechanical Engineering



NOTICE

The program material contained herein is supplied without representation or warranty of any kind. Hewlett-Packard Company therefore assumes no responsibility and shall have no liability, consequential or otherwise, of any kind arising from the use of this program material or any part thereof.

INTRODUCTION

This HP-41C Solutions book was written to help you get the most from your calculator. The programs were chosen to provide useful calculations for many of the common problems encountered.

They will provide you with immediate capabilities in your everyday calculations and you will find them useful as guides to programming techniques for writing your own customized software. The comments on each program listing describe the approach used to reach the solution and help you follow the programmer's logic as you become an expert on your HP calculator.

KEYING A PROGRAM INTO THE HP-41C

There are several things that you should keep in mind while you are keying in programs from the program listings provided in this book. The output from the HP 82143A printer provides a convenient way of listing and an easily understood method of keying in programs without showing every keystroke. This type of output is what appears in this handbook. Once you understand the procedure for keying programs from the printed listings, you will find this method simple and fast. Here is the procedure:

1. At the end of each program listing is a listing of status information required to properly execute that program. Included is the SIZE allocation required. Before you begin keying in the program, press **XEQ ALPHA SIZE ALPHA** and specify the allocation (three digits; e.g., 10 should be specified as 010).

Also included in the status information is the display format and status of flags important to the program. To ensure proper execution, check to see that the display status of the HP-41C is set as specified and check to see that all applicable flags are set or clear as specified.

2. Set the HP-41C to PRGM mode (press the **PRGM** key) and press **■ GTO □ □** to prepare the calculator for the new program.
3. Begin keying in the program. Following is a list of hints that will help you when you key in your programs from the program listings in this handbook.

- a. When you see " (quote marks) around a character or group of characters in the program listing, those characters are ALPHA. To key them in, simply press **ALPHA**, key in the characters, then press **ALPHA** again. So "SAMPLE" would be keyed in as **ALPHA "SAMPLE" ALPHA**.
- b. The diamond in front of each LBL instruction is only a visual aid to help you locate labels in the program listings. When you key in a program, ignore the diamond.
- c. The printer indication of divide sign is /. When you see / in the program listing, press **+**.
- d. The printer indication of the multiply sign is ×. When you see × in the program listing, press **×**.
- e. The † character in the program listing is an indication of the **APPEND** function. When you see †, press **■ APPEND** in ALPHA mode (press **■** and the K key).
- f. All operations requiring register addresses accept those addresses in these forms:

nn (a two-digit number)

IND nn (INDIRECT: **■**, followed by a two-digit number)

X, Y, Z, T, or L (a STACK address: **□** followed by X, Y, Z, T, or L)

IND X, Y, Z, T or L (INDIRECT stack: **■ □** followed by X, Y, Z, T, or L)

Indirect addresses are specified by pressing **■** and then the indirect address. Stack addresses are specified by pressing **□** followed by X, Y, Z, T, or L. Indirect stack addresses are specified by pressing **■ □** and X, Y, Z, T, or L.

Printer Listing

```
01 ♦LBL "SAM  
PLE"  
02 "THIS IS  
A"  
03 ♦I-SAMPLE  
"  
04 AVIEW  
05 6  
06 ENTER↑  
07 -2  
08 /  
09 ABS  
10 STO IND  
L  
11 "R3="  
12 ARCL 03  
13 AVIEW  
14 RTN
```

Keystrokes

```
■ LBL ALPHA SAMPLE ALPHA  
ALPHA THIS IS A ALPHA  
ALPHA ■ APPEND SAMPLE  
■ AVIEW ALPHA  
6  
ENTER↑  
2 CHS  
+  
XEQ ALPHA ABS ALPHA  
STO ■ □ L  
ALPHA R3= ■ ARCL 03  
■ AVIEW  
ALPHA  
■ RTN
```

Display

```
01 LBLT SAMPLE  
02T THIS IS A  
03T I-SAMPLE  
04 AVIEW  
05 6  
06 ENTER ↑  
07 -2  
08 /  
09 ABS  
10 STO IND L  
11T R3=  
12 ARCL 03  
13 AVIEW  
14 RTN
```

TABLE OF CONTENTS

1. GEAR FORCES	1
Computes the forces on helical, bevel, or worm gears.	
2. STRESS ON AN ELEMENT	9
Performs Mohr circle stress analysis, and reduces rosette strain gage data.	
3. EQUATIONS OF STATE	19
Solves both the ideal gas and Redlich-Kwong equations of state.	
4. SODERBERG'S EQUATION FOR FATIGUE	27
Given six variables, this program will calculate the seventh unknown.	
5. SPRING CONSTANT	34
Solves any general linear equation of the form $(y - y_0) = m(x - x_0)$.	
6. PROGRESSION OF A SLIDER CRANK	39
Calculates displacement, velocity, acceleration, and other quantities.	
7. FREE VIBRATIONS	48
Solves the differential equation $m\ddot{x} + c\dot{x} + kx = 0$.	
8. INTERFERENCE FITS	56
Calculates contact pressure or interference for concentric cylinders.	
9. LINEAR OR ANGULAR DEFORMATION	64
Solves for linear deflection under tensile load or angular deflection under torque.	
10. CONSTANT ACCELERATION	69
Calculates displacement, acceleration, initial velocity and time for an object undergoing constant acceleration.	

GEAR FORCES

This program computes three mutually perpendicular forces resulting from input torque on helical, bevel, or worm gears.

Helical gear equations:

$$F_t = \frac{T}{r}$$

$$F_{gs} = F_t \tan \phi$$

$$F_{gax} = F_t \tan \alpha$$

$$\tan \phi = \frac{\tan \phi_n}{\cos \alpha}$$

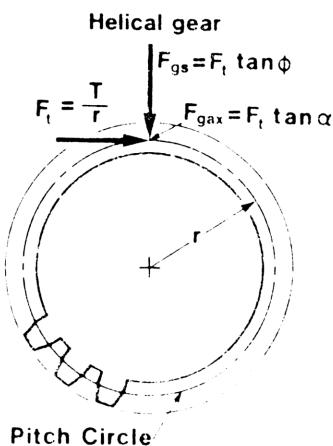


Figure 1-Helical Gear

where:

T is the input torque;

r is the pitch radius of the input gear;

F_t is the tangential force;

α is the helix angle measured from the axis of the gear (for spur gears $\alpha = 0$);

ϕ_n is the pressure angle measured perpendicular to the gear tooth;

ϕ is the pressure angle measured perpendicular to the gear axis;

F_{gs} is the radial force trying to separate the gears;

F_{gax} is the force parallel to the gear axis.

Bevel gear equations:

$$F_t = \frac{T}{r}$$

$$F_{bpax} = F_t \left(\frac{\tan \phi_n \sin (\text{cone}\angle)}{\cos \alpha} + \tan \alpha \cos (\text{cone}\angle) \right)$$

$$F_{bgax} = F_t \left(\frac{\tan \phi_n \cos (\text{cone}\angle)}{\cos \alpha} - \tan \alpha \sin (\text{cone}\angle) \right)$$

$$\tan \phi = \frac{\tan \phi_n}{\cos \alpha}$$

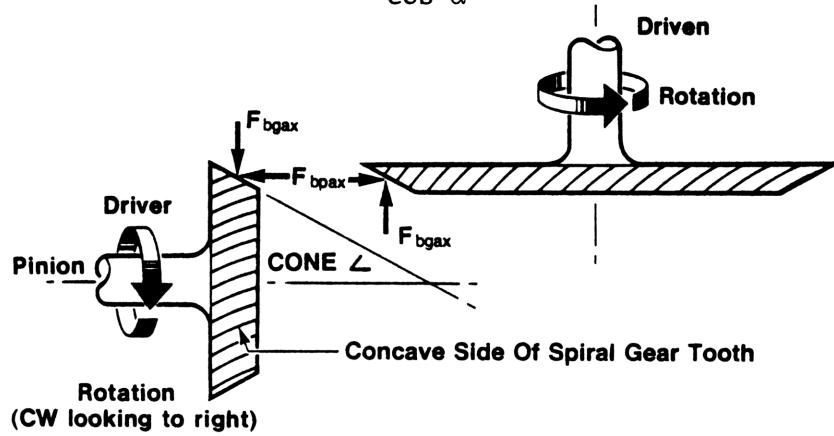


Figure 2—Spiral Bevel Gears

where:

- T is the input (pinion) torque;
- r is the pitch radius of the pinion gear;
- F_t is the tangential force;
- α is the pinion spiral angle (zero for straight tooth bevel gears);
- ϕ_n is the pressure angle measured perpendicular to the gear tooth;
- ϕ is the pressure angle measured perpendicular to the gear axis;
- Cone \angle is the pitch cone angle of the pinion;
- F_{bpax} is the force along the axis of the bevel pinion;
- F_{bgax} is the force along the axis of the bevel gear.

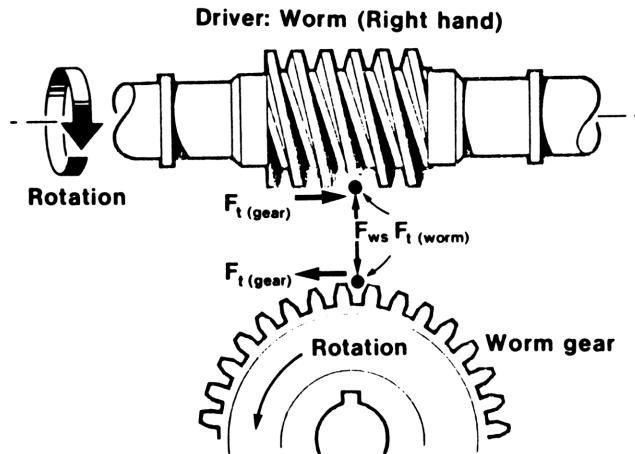
Worm gear equations:

$$F_t = \frac{T}{r}$$

$$F_{ws} = F_t \left(\frac{\sin \phi_n}{\cos \phi_n \sin \alpha + f \cos \alpha} \right)$$

$$F_{gax} = F_t \frac{1 - \frac{f \tan \alpha}{\cos \phi_n}}{\tan \alpha + \frac{f}{\cos \phi_n}}$$

$$\tan \phi = \frac{\tan \phi_n}{\cos \alpha}$$



**Figure 3
WORM GEAR**

where:

- T is the input (worm) torque;
- n is the pitch radius of the worm;
- F_t is the tangential force on the worm;
- α is the lead angle of the worm ($\alpha = \tan^{-1}(L/2\pi r)$, where L is the lead of the worm);
- ϕ_n is the pressure angle measured perpendicular to the worm teeth;
- ϕ is the pressure angle measured parallel to the worm axis;
- f is the coefficient of friction;
- F_{ws} is the separating force between the worm and gear;
- F_{gax} is the force parallel to the gear axis.

Example 1:

A helical gear with pitch radius 12 cm has a torque applied to it of 450,000 dyne-cm. The helix angle is 30° and the normal pressure angle, measured perpendicular to a tooth is 17.5° . Find the tangential, separating, and thrust forces.

Keystrokes:

```
[XEQ] [ALPHA] SIZE [ALPHA] 013
[XEQ] [ALPHA] GEAR [ALPHA]
450000 [R/S]
12 [R/S]
[R/S]
30 [R/S]
17.5 [R/S]
[XEQ] [ALPHA] HEL [ALPHA]
[R/S]
```

Display:

```
T?
R?
F=37,500.00
ALPHA?
AN?
FGS=13,652.84
FBGAX=21,650.64
```

Example 2:

A spiral pinion with mean radius 1.73 inches is subjected to a torque of 745 in-lb. The pinion is cut with a normal pressure angle of 20° , a spiral angle of 35° , and a pitch cone of 18° . Find the forces acting on the pinion. Rotation is in the direction of the concave side of the pinion teeth, so α is positive 35° .

Keystrokes:

```
[XEQ] [ALPHA] GEAR [ALPHA]
745 [R/S]
1.73 [R/S]
[R/S]
35 [R/S]
20 [R/S]
[XEQ] [ALPHA] BEV [ALPHA]
18 [R/S]
[R/S]
```

Display:

```
T?
R?
F=430.64
ALPHA?
AN?
CONE $\angle$ ?
FBPAX=345.90
FBGAX=88.80
```

User Instructions

STEP	INSTRUCTIONS	INPUT	FUNCTION	SIZE: 013
				DISPLAY
1	Load program			
2	Begin execution		[XEQ] GEAR	T?
3	Input values:			
	Torque	T	[R/S]	R?
	Pitch radius	r	[R/S]	F=(F _t)
			[R/S]	ALPHA?
4	Input helix angle for helical gears, or spiral angle for spiral bevel gears, or lead angle for worm gears	α	[R/S]	AN?
5	Input normal pressure angle	ϕ_n	[R/S]	
	OR			
	Input pressure angle	ϕ	[XEQ] N	
6	Execute appropriate program			
	For helical gears		[XEQ] HEL	FGS=
			[R/S]	FGAX=
	For bevel gears		[XEQ] BEV	CONE α ?
	input cone angle	CONE α	[R/S]	FBPAX=
			[R/S]	FBGAX=
	For worm gears		[XEQ] WORM	F?
	input coefficient of friction	f	[R/S]	FWS=
			[R/S]	FGAX=

Program Listings

01♦LBL "GEA R" 02 FIX 2 03 "T?" 04 PROMPT 05 "R?" 06 PROMPT 07 / 08 STO 06 09 "F=" 10 XEQ 05 11 "ALPHA?" 12 PROMPT 13 STO 11 14 "AN?" 15 PROMPT 16♦LBL 10 17 STO 05 18 STOP 19♦LBL "HEL " 20 RCL 06 21 RCL 11 22 TAN 23 * 24 RCL 05 25 TAN 26 RCL 11 27 COS 28 / 29 RCL 06 30 * 31 "FGS=" 32 XEQ 05 33 X<>Y 34 "FGAX=" 35 GTO 05 36♦LBL "BEV " 37 "CONE?" 38 PROMPT 39 STO 12 40 RCL 12 41 RCL 05 42 TAN 43 RCL 11 44 COS 45 / 46 RCL 06 47 * 48 P-R 49 RCL 06	Initialization	50 RCL 11 51 TAN 52 * 53 RCL 12 54 X<>Y 55 P-R 56 RT 57 + 58 RDH 59 - 60 RT 61 "FBPAX=" 62 XEQ 05 63 X<>Y 64 "FBGAX=" 65 GTO 05 66♦LBL "WOR M"	Worm gears
20 RCL 06 21 RCL 11 22 TAN 23 * 24 RCL 05 25 TAN 26 RCL 11 27 COS 28 / 29 RCL 06 30 * 31 "FGS=" 32 XEQ 05 33 X<>Y 34 "FGAX=" 35 GTO 05 36♦LBL "BEV	Helical gears	67 "F?" 68 PROMPT 69 STO 04 70 RCL 05 71 SIN 72 LASTX 73 COS 74 RCL 11 75 SIN 76 * 77 RCL 11 78 COS 79 RCL 04 80 * 81 + 82 / 83 RCL 06 84 *	
37 "CONE?" 38 PROMPT 39 STO 12 40 RCL 12 41 RCL 05 42 TAN 43 RCL 11 44 COS 45 / 46 RCL 06 47 * 48 P-R 49 RCL 06	Bevel gears	85 "FWS=" 86 XEQ 05 87 1 88 RCL 11 89 TAN 90 RCL 04 91 * 92 RCL 05 93 COS 94 / 95 - 96 RCL 11 97 TAN 98 RCL 04 99 RCL 05	

Program Listings

100 COS		51	
101 /			
102 +			
103 *			
104 RCL 06			
105 *			
106 "FGAX="			
107+LBL 05	-----		
108 ARCL X	Display		
109 AVIEW		60	
110 STOP			
111 RTN			
112+LBL "N"	-----		
113 TAN	Calculate θ from θ		
114 RCL 11			
115 COS			
116 *			
117 ATAN		70	
118 GTO 10			
119 .END.			
30		80	
40		90	
50		00	

REGISTERS, STATUS, FLAGS, ASSIGNMENTS

DATA REGISTERS			STATUS				
00		50	SIZE	013	TOT. REG.	43	USER MODE
05	f		ENG		FIX	2	SCI
	fn	55	DEG	X	RAD		ON OFF X
	Ft		FLAGS				
10		60	#	INIT S/C	SET INDICATES	CLEAR INDICATES	
	α						
	Cone	.					
15		65					
20		70					
25		75					
30		80					
35		85					
ASSIGNMENTS							
40		90		FUNCTION	KEY	FUNCTION	KEY
45		95					

STRESS ON AN ELEMENT

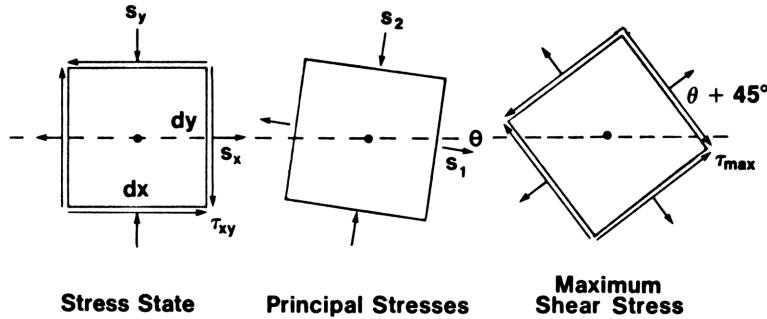
This program reduces data from rosette strain gage measurements and/or performs Mohr circle stress analysis calculations.

Correlations for rectangular and equiangular rosette configurations are included.

Strain Gage Equations:

CONFIGURATION CODE	1	2
TYPE OF ROSETTE	RECTANGULAR	DELTA (EQUIANGULAR)
PRINCIPAL STRAINS: ϵ_1, ϵ_2	$\frac{1}{2} [\epsilon_a + \epsilon_c \pm \sqrt{2(\epsilon_a - \epsilon_b)^2 + 2(\epsilon_b - \epsilon_c)^2}]$	$\frac{1}{3} [\epsilon_a + \epsilon_b + \epsilon_c \pm \sqrt{2(\epsilon_a - \epsilon_b)^2 + 2(\epsilon_b - \epsilon_c)^2 + 2(\epsilon_c - \epsilon_a)^2}]$
CENTER OF MOHR CIRCLE: $\frac{s_1 + s_2}{2}$	$\frac{E(\epsilon_a + \epsilon_c)}{2(1 - \nu)}$	$\frac{E(\epsilon_a + \epsilon_b + \epsilon_c)}{3(1 - \nu)}$
MAXIMUM SHEAR STRESS: τ_{\max}	$\frac{E}{2(1 + \nu)} \sqrt{2(\epsilon_a - \epsilon_b)^2 + 2(\epsilon_b - \epsilon_c)^2 + 2(\epsilon_c - \epsilon_a)^2}$	$\frac{E}{3(1 + \nu)} \sqrt{2(\epsilon_a - \epsilon_b)^2 + 2(\epsilon_b - \epsilon_c)^2 + 2(\epsilon_c - \epsilon_a)^2}$
ORIENTATION OF PRINCIPAL STRESSES	$\tan^{-1} \left[\frac{2\epsilon_b - \epsilon_a - \epsilon_c}{\epsilon_a - \epsilon_c} \right]$	$\tan^{-1} \left[\frac{\sqrt{3} (\epsilon_a - \epsilon_b)}{(2\epsilon_a - \epsilon_b - \epsilon_c)} \right]$

The Mohr circle portion of the program converts an arbitrary stress configuration to principal stresses, maximum shear stress and rotation angle. It is then possible to calculate the state of stress for an arbitrary orientation θ' .



Mohr Circle Equations:

$$\tau_{max} = \sqrt{\left(\frac{s_x - s_y}{2}\right)^2 + \tau_{xy}^2}$$

$$s_1 = \frac{s_x + s_y}{2} + \tau_{max}$$

$$s_2 = \frac{s_x + s_y}{2} - \tau_{max}$$

$$\theta = \frac{1}{2} \tan^{-1} \left(\frac{2\tau_{xy}}{s_x - s_y} \right)$$

$$s = \frac{s_1 + s_2}{2} + \tau_{max} \cos 2\theta'$$

$$\tau = \tau_{max} \sin 2\theta'$$

where:

s is the normal stress, and τ is the shear stress.

ϵ_a , ϵ_b , and ϵ_c are the strains measured using rosette gages;

S_x is the stress in the x direction for Mohr circle input;

S_y is the stress in the y direction for Mohr circle input;

τ_{xy} is the shear stress on the element for Mohr circle input;

ϵ_1 and ϵ_2 are the principal strains;

S_1 and S_2 are the principal normal stresses;

τ_{max} is the maximum shear stress;

θ is the counterclockwise angle of rotation from the specified axis to the principal axis. Note that this is opposite to the normal Mohr circle convention.

θ' is an arbitrary rotation angle from the original (x,y) axis;

E is modulus of elasticity.

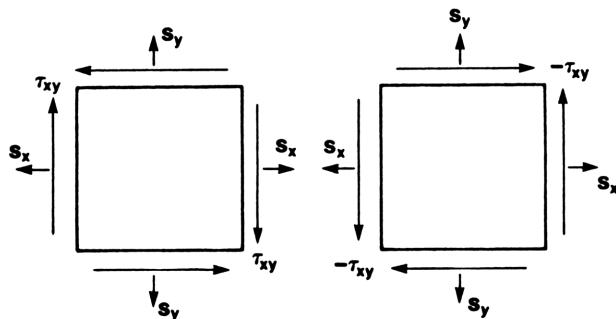
Reference:

Spotts, M.F., Design of Machine Elements; Prentice-Hall, 1971.

Beckwith, T. G., Buck, N. L., Mechanical Measurements; Addison-Wesley, 1969.

Remarks:

Negative stresses and strains indicate compression. Positive and negative shear are represented below:



Example:

A rectangular rosette measures the strains below. What are the principal strains and principal stresses?

$$\epsilon_a = 90 \times 10^{-6} \quad \epsilon_b = 137 \times 10^{-6} \quad \epsilon_c = 305 \times 10^{-6}$$

$$\nu = 0.3 \quad E = 30 \times 10^6 \text{ psi}$$

Keystrokes:

[USER]

[XEQ [ALPHA] SIZE [ALPHA] 016

[XEQ] [ALPHA] ROSETTA [ALPHA]

Y [R/S]

30 [EEX] 6 [R/S]

.3 [R/S]

90 [EEX] 6 [CHS] [R/S]

137 [EEX] 6 [CHS] [R/S]

305 [EEX] 6 [CHS] [R/S]

[R/S]

[R/S]

[R/S]

[R/S]

[R/S]

[R/S]

Display:

RECT? Y/N

E?

RATIO?

EA?

EB?

EC?

E=320.9E-6

E=74.14E-6

$\angle=14.69E0$

S=11.31E3

S=5.618E3

TMAX=2.847E3

$\angle=14.69E0$

User Instructions

				SIZE: 016
STEP	INSTRUCTIONS	INPUT	FUNCTION	DISPLAY
1	Load program and set USER mode.		[USER]	
2	For Mohr's circle go to step 9.			
3	Initialize for rosetta strain gage data.		[XEQ] ROSETTA	RECT? Y/N
4	Choose configuration:			
	rectangular OR	Y	[R/S]	E?
	equiangular	N	[R/S]	E?
5	Input modulus of elasticity.	E	[R/S]	RATIO?
6	Input poisson's ratio.	v	[R/S]	EA?
7	Input strains and calculate principal strains and rotation angle.	ϵ_a	[R/S]	EB?
		ϵ_b	[R/S]	EC?
		ϵ_c	[R/S]	$E = (\epsilon_1)$
			[R/S]	$E = (\epsilon_2)$
			[R/S]	$\angle = (\theta)$
8	Calculate Mohr's circle data from strain gage data.		[R/S]	$S = (S_1)$
			[R/S]	$S = (S_2)$
			[R/S]	TMAX=
			[R/S]	$\angle = (\theta)$
9	Initialize for Mohr's circle.		[XEQ] MOHR	SX?
10	Input stresses and calculate principal stresses and rotation angle.	s_x	[R/S]	SY?
		s_y	[R/S]	TXY?
		τ_{xy}	[R/S]	$S = (S_1)$
			[R/S]	$S = (S_2)$
			[R/S]	TMAX=
			[R/S]	$\angle = (\theta)$
11	Optional: calculate stress at specific angle.		[E]	$\angle ?$

User Instructions

Program Listings

```

01♦LBL "ROS
ETTA"
02 ENG 3
03 1
04 STO 10
05 "RECT? Y
/N"
06 RON
07 PROMPT
08 ROFF
09 ASTO X
10 "N"
11 ASTO Y
12 X=Y?
13 ISG 10
14♦LBL 00
15 "E?"
16 PROMPT
17 STO 15
18 "RATIO?"
19 PROMPT
20 STO 09
21 "EA?"
22 PROMPT
23 STO 11
24 "EB?"
25 PROMPT
26 STO 12
27 "EC?"
28 PROMPT
29 STO 13
30 RCL 11
31 GTO IND
10
32♦LBL 02
33 RCL 12
34 +
35♦LBL 01
36 RCL 13
37 +
38 STO 06
39 0
40 GTO IND
10
41♦LBL 02
42 RCL 13
43 RCL 11
44 -
45♦LBL 01
46 RCL 12
47 RCL 13
48 -

```

Initialization

```

49 R-P
50 RCL 11
51 RCL 12
52 -
53 R-P
54 2
55 SQRT
56 *
57 STO 05
58 2
59 GTO IND
10
60♦LBL 02
61 1
62 +
63♦LBL 01
64 ST/ 05
65 ST/ 06
66 "E="
67 ASTO 00
68 XEQ 05
69 RCL 15
70 RCL 09
71 1
72 +
73 /
74 ST* 05
75 RCL 15
76 1
77 RCL 09
78 -
79 /
80 ST* 06
81 RCL 13
82 RCL 12
83 -
84 3
85 SQRT
86 GTO IND
10
87♦LBL 01
88 2
89 RCL 12
90 *
91 RCL 11
92 -
93 RCL 13
94 -
95 RCL 11
96 RCL 13
97 GTO 04
98♦LBL 02

```

Calculate τ_{\max}
and $\frac{s_1 + s_2}{2}$
from strains

Program Listings

```

99 *
100 2
101 RCL 11
102 *
103 RCL 12
104 -
105 RCL 13
106*LBL 04
107 -
108 XEQ 06
109 RDN
110 "Z="
111 ARCL X
112 AVIEW
113 STOP
114 GTO D
115*LBL "MOH
R"
116 "SX?"
117 PROMPT
118 "SY?"
119 PROMPT
120 "TXY?"
121 PROMPT
122 RT
123 RT
124 STO 03
125 STO 06
126 RT
127 ST+ 06
128 -
129 STO 04
130 2
131 ST/ 06
132 /
133 RT
134 STO 02
135 ST+ 02
136 R-P
137 STO 05
138 RCL 02
139 CHS
140 RCL 04
141 XEQ 06
142 GTO D
143*LBL 06
144 X#0?
145 /
146 ATAN
147 STO 02
148 2
149 /

```

```

150 0
151 RTN
152*LBL D
153 "S="
154 ASTO 00
155 XEQ 05
156 RCL 05
157 "TMAX="
158 ARCL X
159 AVIEW
160 STOP
161 RCL 02
162 2
163 /
164 "Z="
165 ARCL X
166 AVIEW
167 STOP
168*LBL E
169 "Z?"
170 PROMPT
171 ENTER↑
172 +
173 RCL 02
174 -
175 RCL 05
176 P-R
177 RCL 06
178 +
179 "S="
180 ARCL X
181 AVIEW
182 STOP
183 X<>Y
184 "T="
185 ARCL X
186 AVIEW
187 STOP
188*LBL 05
189 RCL 06
190 RCL 05
191 +
192 ARCL X
193 AVIEW
194 STOP
195 RCL 06
196 RCL 05
197 -
198 CLA
199 ARCL 00
200 ARCL X
201 AVIEW

```

Output S_1 , S_2 ,
 τ_{\max} and θ'

Calculate ϵ_1 ,
 ϵ_2 , or S_1 , S_2

Program Listings

202 STOP		51	
203 RTH			
204 .END.			
10		60	
20		70	
30		80	
40		90	
50		00	

REGISTERS, STATUS, FLAGS, ASSIGNMENTS

DATA REGISTERS			STATUS			
00	Temp: ALPHA Storage	50	SIZE 016 ENG 3 DEG X	TOT. REG. 61	USER MODE ON <input checked="" type="checkbox"/> OFF _____	
	2θ			FIX _____	SCI _____	ON <input checked="" type="checkbox"/> OFF _____
	Sx			RAD _____	GRAD _____	
	Sx-Sy			FLAGS		
05	τ_{max}	55	# INIT S/C	SET INDICATES	CLEAR INDICATES	
	$(S_1 + S_2) / 2$					
	v					
	Control	60				
10	ϵ_a					
	ϵ_b					
	ϵ_c					
	E	65				
15						
20		70				
25		75				
30		80				
35		85				
ASSIGNMENTS						
40		90	FUNCTION	KEY	FUNCTION	KEY
45		95				

EQUATIONS OF STATE

This program provides both ideal gas and Redlich-Kwong equations of state. Given four of the five state variables, the fifth is calculated. For the Redlich-Kwong solution, the critical pressure and temperature of the gas must be known. They are not needed for ideal gas solutions.

Values of the Universal Gas Constants

Value of R	Units of R	Units of P	Units of V	Units of T
8.314	N-m/g mole-K	N/m ²	m ³ /g mole	K
83.14	cm ³ -bar/g mole-K	bar	cm ³ /g mole	K
82.05	cm ³ -atm/g mole-K	atm	cm ³ /g mole	K
0.7302	atm-ft ³ /lb mole-°R	atm	ft ³ /lb mole	°R
10.73	psi-ft ³ /lb mole-°R	psi	ft ³ /lb mole	°R
1545	psf-ft ³ /lb mole-°R	psf	ft ³ /lb mole	°R

Critical Temperatures and Pressures

Substance	T _c , K	T _c , °R	P _c , ATM
Ammonia	405.6	730.1	112.5
Argon	151	272	48.0
Carbon dioxide	304.2	547.6	72.9
Carbon monoxide	133	239	34.5
Chlorine	417	751	76.1
Helium	5.3	9.5	2.26
Hydrogen	33.3	59.9	12.8
Nitrogen	126.2	227.2	33.5
Oxygen	154.8	278.6	50.1
Water	647.3	1165.1	218.2
Dichlorodifluoromethane	384.7	692.5	39.6
Dichlorofluoromethane	451.7	813.1	51.0
Ethane	305.5	549.9	48.2
Ethanol	516.3	929.3	63
Methanol	513.2	923.8	78.5
n-Butane	425.2	765.4	37.5
n-Hexane	507.9	914.2	29.9
n-Pentane	469.5	845.1	33.3
n-Octane	568.6	1023.5	24.6
Trichlorofluoromethane	471.2	848.1	43.2

Equations:**Ideal gas:**

$$PV = nRT$$

Redlich-Kwong:

$$P = \frac{nRT}{(V - b)} - \frac{a}{T^{1/2} V (V + b)}$$

$$a = 4.934 b nRT_c^{1.5}$$

$$b = 0.0867 \frac{nRT_c}{P_c}$$

where:

P is the absolute pressure;

V is the volume;

n is the number of moles present;

R is the universal gas constant;

T is the absolute temperature;

T_c is the critical temperature;P_c is the critical pressure.**Remarks:**

P, V, n and T must have units compatible with R.

At low temperatures or high pressures, the ideal gas law does not represent the behavior of real gases.

No equation of state is valid for all substances over an infinite range of conditions. The Redlich-Kwong equation gives moderate to good accuracy for a variety of substances over a wide range of conditions. Results should be used with caution and tempered by experience.

Solutions for V, n, R and T, using the Redlich-Kwong equation, require an iterative technique. Newton's method is employed using the ideal gas law to generate the initial guess. Iteration time is generally a function of the amount of deviation from ideal gas behavior. For extreme cases, the routine may fail to converge entirely, resulting in a "DATA ERROR".

Example 1:

0.63 g moles of air are enclosed in a 25,000 cm³ space at 1200 K. What is the pressure in bars? Assume an ideal gas.

Keystrokes:

[XEQ] [ALPHA] SIZE [ALPHA] 015

[XEQ] [ALPHA] ID [ALPHA]

0 [R/S]

25,000 [R/S]

0.63 [R/S]

83.14 [R/S]

1200 [R/S]

Display:

P?

V?

N?

R?

T?

P=2.52

Example 2:

The specific volume of a gas in a container is 800 cm³/g mole. The temperature will reach 400K. What will the pressure be, according to the Redlich-Kwong relation?

$$P_c = 48.2 \text{ atm}$$

$$T_c = 305.5 \text{ K}$$

$$R = 82.05 \text{ cm}^3 - \text{atm/g mole-K}$$

Keystrokes:

[XEQ] [ALPHA] RK [ALPHA]

305.5 [R/S]

48.2 [R/S]

0 [R/S]

800 [R/S]

1 [R/S]

82.05 [R/S]

400 [R/S]

Display:

TC?

PC?

P?

V?

N?

R?

T?

P=36.27

User Instructions

SIZE: 015

Program Listings

```

01♦LBL "ID"
02 0
03 SF 00
04 GTO 00
05♦LBL "RK"
06 1
07 CF 00
08 "TC?"
09 PROMPT
10 STO 13
11 "PC?"
12 PROMPT
13 STO 14
14♦LBL 00
15 SF 02
16 CF 01
17 FIX 2
18 "P?"
19 PROMPT
20 5
21 XEQ 00
22 "V?"
23 PROMPT
24 6
25 XEQ 00
26 "N?"
27 PROMPT
28 7
29 XEQ 00
30 "R?"
31 PROMPT
32 8
33 XEQ 00
34 "T?"
35 PROMPT
36 CF 02
37 9
38♦LBL 00
39 CF 01
40 STO 01
41 RDH
42 STO IND
01
43 X#0?
44 GTO 01
45 RT
46 STO 10
47 1
48 STO IND
01
49♦LBL 01

```

- - - - -
Initialization

```

50 FS? 02
51 RTN
52 GTO IND
10
53♦LBL 05
54 "P="
55 GTO 00
56♦LBL 06
57 "V="
58♦LBL 00
59 RCL 07
60 RCL 08
61 *
62 RCL 09
63 *
64 RCL 05
65 RCL 06
66 *
67 /
68 STO IND
10
69 GTO 00
70♦LBL 07
71 SF 01
72 "N="
73 GTO 01
74♦LBL 08
75 SF 01
76 "R="
77 GTO 01
78♦LBL 09
79 "T="
80 SF 01
81♦LBL 01
82 RCL 05
83 RCL 06
84 *
85 RCL 07
86 /
87 RCL 08
88 /
89 RCL 09
90 /
91 STO IND
10
92♦LBL 00
93 FS? 00
94 GTO 10
95 XEQ 01
96 GTO 00
97♦LBL 02
98 FS? 01

```

- - - - -
Calculate
unknown

- - - - -
If ideal,
display

Program Listings

99 XEQ 01		146 X†2	
100♦LBL 00		147 /	
101 RCL 00		148 RCL 00	
102 RCL 09		149 RCL 09	
103 *	-----	150 *	
104 RCL 06	Calculate using	151 RCL 04	
105 RCL 12	Redlich-Kwong	152 X†2	
106 -	equations	153 /	
107 STO 04		154 -	
108 /		155 RTN	
109 RCL 11		156♦LBL 09	$\frac{\partial P}{\partial T}$
110 RCL 09		157 RCL 00	
111 SQRT		158 RCL 04	
112 /		159 /	
113 STO 02		160 RCL 02	
114 RCL 06		161 Z	
115 /		162 /	
116 LASTX		163 RCL 09	
117 RCL 12		164 /	
118 +		165 RCL 06	
119 STO 03		166 /	
120 /		167 RCL 03	
121 -		168 /	
122 RCL 05		169 +	
123 -		170 RTN	
124 XEQ IND		171♦LBL 07	$\frac{\partial P}{\partial n}$ or $\frac{\partial P}{\partial R}$
10		172♦LBL 08	
125 /		173 RCL 09	
126 ST- IND		174 RCL 06	
10		175 *	
127 RCL IND		176 RCL 04	
10		177 X†2	
128 /		178 /	
129 ABS		179 RCL 06	
130 1 E-4		180 ENTER†	
131 X<=Y?		181 +	
132 GTO 02		182 RCL 12	
133 RCL IND		183 +	
10		184 RCL 00	
134 GTO 10	-----	185 /	
135♦LBL 06	$\frac{\partial P}{\partial V}$	186 RCL 06	
136 RCL 06		187 /	
137 ENTER†		188 RCL 03	
138 +		189 X†2	
139 RCL 12		190 /	
140 +		191 RCL 02	
141 RCL 02		192 *	
142 *		193 -	
143 RCL 03		194 RCL 00	
144 RCL 06		195 *	
145 *		196 RCL IND	
		10	

Program Listings

197 /		51	
198 RTN			
199+LBL 05			
200 LASTX			
201 +			
202 STO 05	-----		
203 GTO 10	Calculate a, b		
204+LBL 01			
205 RCL 07			
206 RCL 08		60	
207 *			
208 STO 00			
209 .0867			
210 RCL 14			
211 /			
212 X<>Y			
213 RCL 13			
214 *			
215 *			
216 STO 12		70	
217 LASTX			
218 *			
219 RCL 13			
220 SQRT			
221 *			
222 4.934			
223 *			
224 STO 11	-----		
225 RTN	Display		
226+LBL 10			
227 ARCL X		80	
228 AVIEW			
229 STOP			
230 .END.			
40		90	
50		00	

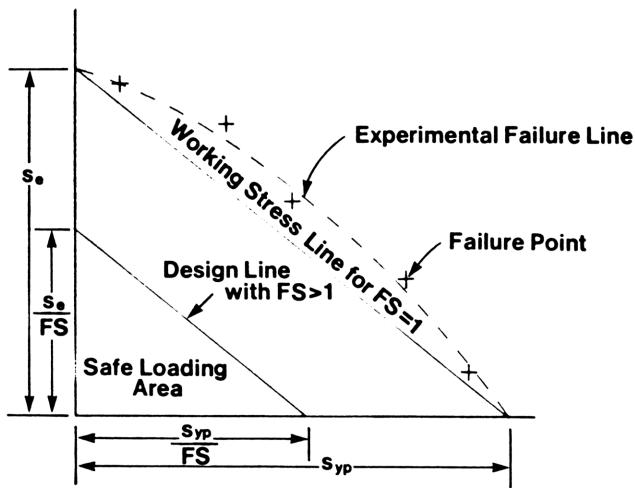
REGISTERS, STATUS, FLAGS, ASSIGNMENTS

DATA REGISTERS				STATUS				
00	NR	50		SIZE	015	TOT. REG.	61	USER MODE
	Temp storage index a/T ^{1/2} (V+b) (V-b)			ENG	FIX 2	SCI	ON OFF X	DEG RAD GRAD
05	P	55		FLAGS				
	V			INIT #	S/C	SET INDICATES	CLEAR INDICATES	
	n			00		Ideal	Redlich-Kwong	
	R			01		Calculate a,b	Don't calculate a,b	
	T			02		Input data	Calculate	
10	control	60						
	a							
	b							
	Tc							
	Pc							
15		65						
20		70						
25		75						
30		80						
35		85						
ASSIGNMENTS								
40		90		FUNCTION	KEY	FUNCTION	KEY	
45		95						

SODERBERG'S EQUATION FOR FATIGUE

This program will calculate the seventh variable from the other six values in Soderberg's equation. It is useful in sizing parts for cyclic loading, calculating factors of safety, choosing materials based on size constraints and estimating the fatigue resistance of available parts. Soderberg's equation is graphically represented in figure 1.

Equations:



Working Stress Diagram
Figure 1

$$\frac{s_{yp}}{FS} = \frac{s_{max} + s_{min}}{2} + K \left(\frac{s_{yp}}{s_e} \right) \left(\frac{(s_{max} - s_{min})}{2} \right)$$

$$\frac{s_{max} + s_{min}}{2} = \frac{P_{max} + P_{min}}{2A}$$

$$\frac{s_{max} - s_{min}}{2} = \frac{P_{max} - P_{min}}{2A}$$

where:

S_{yp} is the yield point stress of the material;

S_e is the material endurance stress from reversed bending tests;

K is the stress concentration factor for the part;

FS is the factor of safety ($FS \geq 1.00$)

S_{max} is the maximum stress;

S_{min} is the minimum stress;

P_{max} is the maximum load;

P_{min} is the minimum load;

A is the cross sectional area of the part over which the force is evenly distributed.

Reference:

Spots, M. F., Design of Machine Elements; Prentice-Hall, Inc., 1971.

Baumeister, T., Marks Standard Handbook for Mechanical Engineers; McGraw-Hill Book Company, 1967.

Remarks:

If S_{max} and S_{min} are to be input or calculated instead of P_{max} or P_{min} , simply use 1.00 for the value of area.

This implementation of Soderberg's equation is for ductile materials only.

Values of stress concentration factors and material endurance limits may be found in the referenced sources.

In the presence of corrosive media, or for rough surfaces, fatigue effects may be much more significant than predicted by this program.

Example:

What is the maximum permissible cyclic load for a part if the minimum load is 2000 pounds and the area is 0.5 square inches?

$$S_{yp} = 70000 \text{ psi}$$

$$S_e = 25000 \text{ psi}$$

$$K = 1.25$$

$$FS = 2.0$$

Keystrokes:

[XEQ] [ALPHA] SIZE [ALPHA] 010

[XEQ] [ALPHA] SEF [ALPHA]

70000 [R/S]

25000 [R/S]

0.5 [R/S]

1.25 [R/S]

[R/S]

2000 [R/S]

2.0 [R/S]

[XEQ] [ALPHA] PMAX [ALPHA]

Display:

SYP?

SE?

A?

K?

PMAX?

PMIN?

FS?

PMAX=8.889E3

If P_{max} is changed to 10000 pounds, what will S_e have to be?

10000 [STO] 03

[XEQ] [ALPHA] SE [ALPHA]

SE=30.43E3

User Instructions

SIZE: 010

STEP	INSTRUCTIONS	INPUT	FUNCTION	DISPLAY
1	Load program			
2	Begin execution		[XEQ] SEF	SYP?
3	Input values (skip unknown by pressing [R/S])	Syp Se Area* K $P_{max}(S_{max})^*$ $P_{min}(S_{min})^*$ FS	[R/S] [R/S] [R/S] [R/S] [R/S] [R/S]	SE? A? K? PMAX? (S_{max})? PMIN? (S_{min})? FS?
4	Calculate unknown			
	Syp		[XEQ] SYP	SYP=
	Se		[XEQ] SE	SE=
	Area		[XEQ] A	A=
	K		[XEQ] K	K=
	P_{max}		[XEQ] PMAX	PMAX=
	P_{min}		[XEQ] PMIN	PMIN=
	S_{max}		[XEQ] SMAX	SMAX=
	S_{min}		[XEQ] SMIN	SMIN=
	FS		[XEQ] FS	FS=
5	To change a value (then go to step 4)			
	Syp		[STO] 08	
	Se		[STO] 09	
	A		[STO] 01	
	K		[STO] 02	
	$P_{max} (S_{max})$		[STO] 03	
	$P_{min} (S_{min})$		[STO] 04	
	FS		[STO] 05	

*If S_{max} and S_{min} are to be input or calculated, use 1.00 for the value of area.

Program Listings

01♦LBL "SEF"		51 RCL 08
"		52 *
02 ENG 3		53 RCL 02
03 "SYP?"		54 *
04 PROMPT	Initialization	55 XEQ 01
05 STO 08		56 CHS
06 "SE?"		57 RCL 08
07 PROMPT		58 RCL 05
08 STO 09		59 /
09 "A?"		60 +
10 PROMPT		61 /
11 STO 01		62 STO 09
12 1		63 "SE="
13 -		64 GTO 05
14 CF 00		65♦LBL "SMA"
15 X=0?	X"	S _{max}
16 SF 00		66♦LBL "PMA"
17 "K?"	X"	P _{max}
18 PROMPT		67 RCL 01
19 STO 02		68 ENTER↑
20 "PMAX?"		69 +
21 FS? 00		70 RCL 08
22 "SMAX?"		71 *
23 PROMPT		72 RCL 05
24 STO 03		73 /
25 "PMIN?"		74 RCL 02
26 FS? 00		75 RCL 08
27 "SMIN?"		76 *
28 PROMPT		77 RCL 09
29 STO 04		78 /
30 "FS?"		79 1
31 PROMPT		80 -
32 STO 05		81 RCL 04
33 STOP		82 *
34♦LBL "SYP"	S _{yp}	83 +
"		84 RCL 02
35 XEQ 01		85 RCL 08
36 XEQ 02		86 *
37 RCL 02		87 RCL 09
38 *		88 /
39 RCL 09		89 1
40 /		90 +
41 CHS		91 /
42 RCL 05		92 STO 03
43 1/X		93 "PMAX="
44 +		94 FS? 00
45 /		95 "SMAX="
46 STO 08		96 GTO 05
47 "SYP="		97♦LBL A
48 GTO 05		98 1
49♦LBL "SE"		99 STO 01
50 XEQ 02	S _e	100 XEQ 01

Area

Program Listings

101 XEQ 02	150 1	
102 RCL 02	151 RCL 02	
103 *	152 RCL 08	
104 RCL 08	153 *	
105 *	154 RCL 09	
106 RCL 09	155 /	
107 /	156 -	
108 +	157 /	
109 RCL 08	158 STO 04	
110 /	159 "PMIN="	
111 RCL 05	160 FS? 00	
112 *	161 "SMIN="	
113 STO 01	162 GTO 05	-----
114 "A="	163 *LBL "FS"	FS
115 GTO 05	164 XEQ 01	
116 *LBL "K"	165 XEQ 02	
117 RCL 08	166 RCL 02	
118 RCL 05	167 *	
119 /	168 RCL 08	
120 XEQ 01	169 *	
121 -	170 RCL 09	
122 XEQ 02	171 /	
123 RCL 08	172 +	
124 *	173 RCL 08	
125 RCL 09	174 /	
126 /	175 1/X	
127 /	176 STO 05	
128 STO 02	177 "FS="	
129 "K="	178 GTO 05	-----
130 GTO 05	179 *LBL 02	Q=P _{max} -P _{min}
131 *LBL "SMI	180 RCL 03	
N"	181 RCL 04	
132 *LBL "PMI	182 CHS	
N"	183 GTO 00	
133 RCL 01	184 *LBL 01	-----
134 ENTER↑	185 RCL 03	Q=P _{max} -P _{min}
135 +	186 RCL 04	
136 RCL 08	187 *LBL 00	-----
137 *	188 +	Q
138 RCL 05	189 RCL 01	2A
139 /	190 /	
140 RCL 02	191 2	
141 RCL 08	192 /	
142 *	193 RTN	-----
143 RCL 09	194 *LBL 05	Display
144 /	195 ARCL X	
145 1	196 AVIEW	
146 +	197 STOP	
147 RCL 03	198 .END.	
148 *		
149 -		

REGISTERS, STATUS, FLAGS, ASSIGNMENTS

DATA REGISTERS			STATUS				
00	A	50	SIZE	010	TOT. REG.	61	USER MODE
	K		ENG	3	FIX	SCI	ON OFF X
	P _{max} (S _{max})		DEG		RAD	GRAD	
	P _{min} (S _{min})		FLAGS				
	FS	55	#	INIT S/C	SET INDICATES		CLEAR INDICATES
05			00		S _{max} , S _{min}		P _{max} , P _{min}
	S _{yp}						
	S _e						
10		60					
15		65					
20		70					
25		75					
30		80					
35		85					
ASSIGNMENTS							
40		90	FUNCTION	KEY	FUNCTION	KEY	
45		95					

SPRING CONSTANT

This program calculates the value of any variable (X_1 , F_1 , X_2 , F_2 , k) given the other four in the spring equation. It may be used to solve any general linear equation of the form $y - y_0 = m(x - x_0)$. It is also useful for linear interpolation in tables. Computed values are automatically stored to provide an interchangeable solution.

X_1 = Spring length

F_1 = Force required to retain spring at length X_1

X_2 = Spring length

F_2 = Force required to retain spring at length X_2

k = Spring constant

Equations:

$$k = \frac{F_1 - F_2}{X_2 - X_1}$$

$$F_1 = F_2 + k(X_2 - X_1)$$

$$F_2 = F_1 + k(X_1 - X_2)$$

$$X_1 = \frac{F_2 - F_1}{k} + X_2$$

$$X_2 = \frac{F_1 - F_2}{k} + X_1$$

Example 1:

A compression spring is 4.0 inches long under no compressive forces. A force of 270 lbf compresses the spring to a length of 2.8 inches. The solid height of the spring is 2.5 inches. Find the spring constant and the force required to fully compress the spring.

Keystrokes:	Display:
[XEQ] [ALPHA] SIZE [ALPHA] 005	
[XEQ] [ALPHA] SP [ALPHA]	X1?
4 [R/S]	F1?
0 [R/S]	X2?
2.8 [R/S]	F2?
270 [R/S]	K?
[R/S]	
[XEQ] [ALPHA] K [ALPHA]	K=225.00
2.5 [STO] 02	
[XEQ] [ALPHA] F2 [ALPHA]	F2=377.50

Example 2:

10.00%	10.25%	?
215.93	222.60	219.97

From the table shown, find the linear approximation to a value of 219.9749.

Keystrokes:	Display:
[XEQ] [ALPHA] SP [ALPHA]	X1?
10 [R/S]	F1?
215.93 [R/S]	X2?
10.25 [R/S]	F2?
222.60 [R/S]	K?
[R/S]	
[XEQ] [ALPHA] K [ALPHA]	K=-26.68
219.97 [STO] 03	
[XEQ] [ALPHA] X2 [ALPHA]	X2=10.15

User Instructions

Program Listings

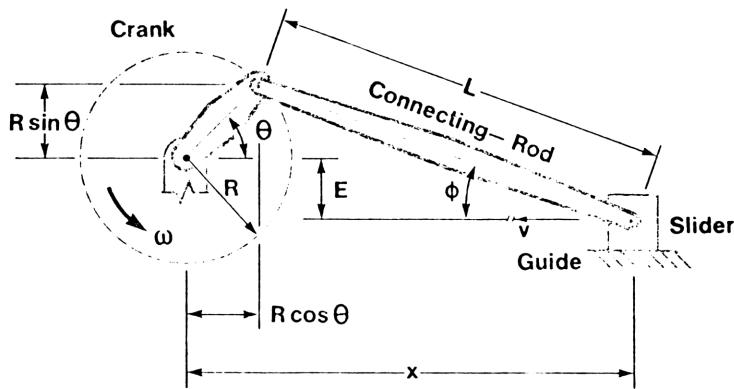
01♦LBL "SP"		52 "F2="	
02 FIX 2		53 GTO 05	
03♦LBL A	-----	54♦LBL "K"	k
04 "X1?"	Initialization	55 RCL 01	
05 PROMPT		56 RCL 03	
06 STO 00		57 -	
07♦LBL B		58 RCL 02	
08 "F1?"		59 RCL 00	
09 PROMPT		60 -	
10 STO 01		61 /	
11♦LBL C		62 STO 04	-----
12 "X2?"		63 "K="	Display
13 PROMPT		64♦LBL 05	
14 STO 02		65 ARCL X	
15♦LBL D		66 AVIEW	
16 "F2?"		67 STOP	-----
17 PROMPT		68♦LBL 01	$F_2 - F_1$
18 STO 03		69 RCL 03	k
19♦LBL E		70 RCL 01	
20 "K?"		71 -	
21 PROMPT		72 RCL 04	
22 STO 04		73 /	
23 STOP	-----	74 RTN	
24♦LBL "X1"	X ₁	75♦LBL 00	k(X ₂ - X ₁)
25 XEQ 01		76 RCL 02	
26 RCL 02		77 RCL 00	
27 +		78 -	
28 STO 00		79 RCL 04	
29 "X1="		80 *	
30 GTO 05	-----	81 RTN	
31♦LBL "F1"	F ₁	82 .END.	
32 XEQ 00			
33 RCL 03			
34 +			
35 STO 01			
36 "F1="			
37 GTO 05	-----		
38♦LBL "X2"	X ₂		
39 XEQ 01			
40 CHS			
41 RCL 00		90	
42 +			
43 STO 02			
44 "X2="			
45 GTO 05	-----		
46♦LBL "F2"	F ₂		
47 XEQ 00			
48 CHS			
49 RCL 01			
50 +			
51 STO 03		00	

REGISTERS, STATUS, FLAGS, ASSIGNMENTS

DATA REGISTERS			STATUS				
00	X ₁	50	SIZE	005	TOT. REG.	27	USER MODE
	F ₁		ENG		FIX	2	SCI
	X ₂		DEG		RAD		GRAD
	F ₂						
	k						
05		55	FLAGS				
#	INIT S/C	SET INDICATES			CLEAR INDICATES		
10							
15		65					
20		70					
25		75					
30		80					
35		85					
ASSIGNMENTS							
		FUNCTION	KEY	FUNCTION	KEY		
40	90						
45	95						

PROGRESSION OF A SLIDER CRANK

In a slider crank mechanism (e.g., the piston, wrist pin and connecting rod in an internal combustion engine), for given crank radius, connecting rod length, slider offset, crankshaft speed (RPM) and crank position, this program calculates the following: the displacement, velocity, and acceleration of the slider; the connecting rod angle, velocity and acceleration; the maximum and minimum displacements, and the maximum and minimum angular values for ϕ .



Equations:

$$\omega = \frac{\pi N}{30}$$

$$x = R \cos \theta + L \cos \phi$$

$$x_{\max} = (R + L) \cos \left[\sin^{-1} \left(\frac{E}{R + L} \right) \right]$$

$$x_{\min} = (L - R) \cos \left[\sin^{-1} \left(\frac{E}{L - R} \right) \right]$$

$$\Delta x = x_{\max} - x_{\min}$$

$$\phi = \sin^{-1} \left(\frac{E + R \sin \theta}{L} \right)$$

$$v = \frac{dx}{dt} = R\omega \left(\frac{-\sin(\theta + \phi)}{\cos \phi} \right)$$

$$a = \frac{d^2x}{dt^2} = R\omega^2 \left(\frac{-\cos(\theta + \phi)}{\cos \phi} - \frac{R \cos^2 \theta}{L \cos^3 \phi} \right)$$

$$\phi_{\max} = \sin^{-1} \left(\frac{E + R}{L} \right)$$

$$\phi_{\min} = \sin^{-1} \left(\frac{E - R}{L} \right)$$

$$\Delta\phi = \phi_{\max} - \phi_{\min}$$

$$\dot{\phi} = \frac{d\phi}{dt} = \omega \frac{R \cos \theta}{L \cos \phi}$$

$$\ddot{\phi} = \frac{d^2\phi}{dt^2} = \omega^2 \left[\left(\frac{d\phi}{d\theta} \right)^2 \tan \phi - \frac{R \sin \theta}{L \cos \phi} \right]$$

where:

N is crankshaft speed in RPM;

E is slider offset;

L is connecting rod length;

R is crank radius;

ω is crank angular velocity in radians/sec;

θ is crank angle;

x is slider displacement;

x_{\max} is maximum slider displacement

x_{\min} is minimum slider displacement;

Δx is stroke;

v is slider velocity;

a is slider acceleration;

ϕ is connecting rod angular displacement;

ϕ_{\max} is maximum connecting rod angular displacement;

ϕ_{\min} is minimum connecting rod angular displacement;

$\Delta\phi$ is total angular throw of connecting rod;

$\dot{\phi}$ is angular velocity of connecting rod;

$\ddot{\phi}$ is angular acceleration of connecting rod.

References:

H. A. Rothbart, Mechanical Design and Systems Handbook; McGraw-Hill, 1964.

V. M. Faires, Kinematics, McGraw-Hill; 1959.

Example 1:

For an in-line slider crank mechanism ($E = 0$), turning at 4800 RPM, having a crank radius of 2.0 inches and connecting rod length of 7.0 inches, find:

- (1) x_{\max} , x_{\min} and ϕ_{\max} , ϕ_{\min}
- (2) x , v , and a of the wrist pin in the slider
- (3) $\dot{\phi}$, $\ddot{\phi}$, and $\ddot{\phi}$ of the connecting rod for $\theta = 0^\circ, 15^\circ, 45^\circ, 90^\circ, 135^\circ, 180^\circ, 225^\circ$

The table below was produced by the following keystrokes:

θ°	x (in)	ϕ°	v (in/sec)	$\dot{\phi}$ (rad/sec)	a (in/sec 2)	$\ddot{\phi}$ (rad/sec 2)
0	9.00	0.00	0.00	143.62	-649701.96	0.00
15	8.91	4.24	-332.20	139.10	-614226.44	-17300.41
45	8.27	11.66	-857.50	103.69	-360454.40	-49902.29
90	6.71	16.60	-1005.31	0.00	150658.43	-75329.22
135	5.44	11.66	-564.22	-103.69	354181.29	-49902.29
180	5.00	0.00	0.00	-143.62	360945.53	0.00
225	5.44	-11.66	564.22	-103.69	354181.29	49902.29

Keystrokes:

[USER]	Display:		
[XEQ] [ALPHA] SIZE [ALPHA] 016	(set USER mode)		
[XEQ] [ALPHA] PSC [ALPHA]	N?		
4800 [R/S]	E?		
0 [R/S]	L?		
7 [R/S]	R?		
2 [R/S]	W=502.65		
[A]	X=9.00	(X_{\max})	
[R/S]	X=5.00	(X_{\min})	
[B]	$\underline{L}=16.60$	(max)	
[R/S]	$\underline{L}=-16.60$	(min)	
[C]	$\underline{L}?$		
0 [R/S]	X=9.00		

Keystrokes:	Display:
[R/S]	$\underline{A}=0.00$
[D]	$V=0.00$
[R/S]	$d\underline{A}=143.62$
[E]	$A=-649,701.96$
[R/S]	$d\uparrow 2 \underline{A}=0.00$
[C]	$\underline{A}?$
15 [R/S]	$X=8.91$
[R/S]	$\underline{A}=4.24$
[D]	$V=-332.20$
[R/S]	$d\underline{A}=139.10$
:	:
[C]	$\underline{A}?$
225 [R/S]	$X=5.44$
[R/S]	$\underline{A}=-11.66$
[D]	$V=564.22$
[R/S]	$d\underline{A}=-103.69$
[E]	$A=354,181.29$
[R/S]	$d\uparrow 2 \underline{A}=49,902.29$

User Instructions

STEP	INSTRUCTIONS	INPUT	FUNCTION	DISPLAY
1	Load program (USER mode)		[USER]	
2	Begin execution and input:		[XEQ] PSC	N?
	Key in crankshaft speed in RPM	N	[R/S]	E?
	Key in slider offset	E	[R/S]	L?
	Key in connecting rod length	L	[R/S]	R?
	Key in crank radius	R	[R/S]	$W = (\omega)$
3	For X_{\max} , X_{\min}		[A]	$X = (X_{\max})$
			[R/S]	$X = (X_{\min})$
4	For ϕ_{\max} , ϕ_{\min}		[B]	$\underline{\Delta} = (\theta_{\max})$
			[R/S]	$\underline{\Delta} = (\theta_{\min})$
5	For X , ϕ , V , $\dot{\phi}$, a , $\ddot{\phi}$:		[C]	$\underline{\Delta}?$
	Key in θ	θ	[R/S]	$X =$
			[R/S]	$\underline{\Delta} = (\phi)$
	then		[D]	$V =$
			[R/S]	$d\underline{\Delta} = (\phi)$
	then		[E]	$A =$
			[R/S]	$d^2 \underline{\Delta} = (\phi)$
6	For automatic display:		[F]	$\underline{\Delta}2?$
	Key in ending θ	θ_2	[R/S]	$\underline{\Delta}1?$
	Key in beginning θ	θ_1	[R/S]	N?
	Key in the number of sectors you wish	n	[R/S]	$\underline{\Delta} = (\theta)$
	to divide the interval into		[R/S]	$X =$
			[R/S]	$\underline{\Delta} = (\phi)$
			[R/S]	$V =$
			[R/S]	$d\underline{\Delta} = (\phi)$
			[R/S]	$A =$
	(continue pressing [R/S] until done.)		[R/S]	$d^2 \underline{\Delta} = (\phi)$

Program Listings

<pre> 01♦LBL "PSC" "-----" 02 FIX 2 03 "N?" 04 PROMPT 05 STO 11 06 "E?" 07 PROMPT 08 STO 12 09 "L?" 10 PROMPT 11 STO 13 12 "R?" 13 PROMPT 14 STO 14 15 RCL 11 16 PI 17 * 18 30 19 / 20 STO 15 21 "W=" 22 GTO 05 23♦LBL A 24 CF 01 25 XEQ 00 26 SF 01 27 STO 04 28 XEQ 00 29 STOP 30♦LBL B 31 1 32 ASIN 33 XEQ 02 34 STO 04 35 "Z=" 36 XEQ 05 37 -1 38 ASIN 39 XEQ 02 40 "Z=" 41 GTO 05 42♦LBL C 43 "Z?" 44 PROMPT 45 STO 00 46♦LBL 07 47 XEQ 02 48 COS 49 STO 04 50 RCL 13 </pre>	<pre> ----- Initialization -----</pre>	<pre> 51 * 52 RCL 00 53 COS 54 STO 03 55 RCL 14 56 * 57 + 58 STO 01 59 "X=" 60 XEQ 05 61 RCL 00 62 XEQ 02 63 "Z=" 64 GTO 05 65♦LBL D 66 RCL 00 67 RCL 02 68 + 69 SIN 70 CHS 71 RCL 04 72 / 73 RCL 15 74 * 75 RCL 14 76 * 77 "V=" 78 XEQ 05 79 RCL 04 80 RCL 13 81 * 82 RCL 14 83 / 84 1/X 85 STO 05 86 RCL 03 87 * 88 RCL 15 89 * 90 "dZ=" 91 GTO 05 92♦LBL E 93 RCL 03 94 X↑2 95 RCL 14 96 * 97 RCL 13 98 / 99 RCL 04 100 3 101 Y↑X 102 / </pre>	<pre> ----- Calculate V + Ø -----</pre>
---	--	---	---

Program Listings

103 RCL 00	154 STO 06
104 RCL 02	155♦LBL 04
105 +	156 RCL 07
106 COS	157 "A="
107 RCL 04	158 ARCL X
108 /	159 AVIEW
109 +	160 STOP
110 CHS	161 STO 00
111 RCL 14	162 CF 01
112 *	163 XEQ 07
113 RCL 15	164 XEQ D
114 X†2	165 XEQ E
115 *	166 SF 01
116 "A="	167 DSE 10
117 XEQ 05	168 GTO 03
118 RCL 02	169 RTN
119 TAN	170♦LBL 03
120 RCL 05	171 RCL 06
121 RCL 03	172 ST+ 07
122 *	173 GTO 04
123 X†2	174 RTN
124 *	175♦LBL 05
125 RCL 00	176 ARCL X
126 SIN	177 AVIEW
127 RCL 05	178 STOP
128 *	179 RTN
129 -	180♦LBL 00
130 RCL 15	181 RCL 12
131 X†2	182 RCL 13
132 *	183 RCL 14
133 "d†24="	184 FS? 01
134 GTO 05	185 CHS
135♦LBL F	186 +
136 "A2?"	187 /
137 PROMPT	188 ASIN
138 STO 08	189 COS
139 "A1?"	190 RCL 13
140 PROMPT	191 RCL 14
141 STO 07	192 FS? 01
142 "N?"	193 CHS
143 PROMPT	194 +
144 STO 09	195 *
145 STO 10	196 "X="
146 ISG 10	197 XEQ 05
147 RDN	198 RTN
148 RCL 07	199♦LBL 02
149 RCL 08	200 SIN
150 -	201 RCL 14
151 CHS	202 *
152 RCL 09	203 RCL 12
153 /	204 +

display

Calculate Xmax +
Xmin

Program Listings

205 RCL 13		51	
206 /			
207 ASIN			
208 STO 02			
209 RTN			
210 .END.			
20		60	
30		70	
40		80	
50		90	
		00	

REGISTERS, STATUS, FLAGS, ASSIGNMENTS

DATA REGISTERS			STATUS				
			SIZE	016	TOT. REG.	61	USER MODE
			ENG		FIX	2	ON X OFF
			DEG		RAD	GRAD	
				FLAGS			
				#	INIT S/C	SET INDICATES	CLEAR INDICATES
				01		Maximum	Minimum
00	θ	50					
	x						
	ϕ						
	COS θ						
	COS ϕ						
05	$R/(L \cos \phi)$	55					
	$(\theta_1 - \theta_2)/h$						
	θ_1						
	θ_2						
	n						
10	Index	60					
	N						
	E						
	L						
	R						
15	ω	65					
20		70					
25		75					
30		80					
35		85					
ASSIGNMENTS							
			FUNCTION	KEY	FUNCTION	KEY	
40		90					
45		95					

FREE VIBRATIONS

This program provides an exact solution to the differential equation for a damped oscillator vibrating freely: $mx'' + cx' + kx = 0$.

The user inputs the mass m , spring constant k , and damping constant c . The output will be:

1. ω for an underdamped system, i.e. $c < c_{\text{crit}}$ • c_{crit} is calculated by pressing [B].
2. 0 for a critically damped system, i.e. $c = c_{\text{crit}}$.
3. -1 for an overdamped system, i.e. $c > c_{\text{crit}}$.

The initial conditions are the displacement and velocity at time zero (x_0 and \dot{x}_0).

Equations:

$$c_{\text{crit}} = 2 \sqrt{km}$$

$$\omega = \sqrt{\frac{k}{m} - \left(\frac{c}{2m} \right)^2}$$

$$\ddot{x} = -(c\dot{x} + kx)/m$$

Underdamping $(c^2 - 4km < 0)$

$$x(t) = Re^{-\frac{c}{2m}t} \cos(\omega t - \delta)$$

$$\dot{x}(t) = -R\omega e^{-\frac{c}{2m}t} \sin(\omega t - \delta) - \frac{c}{2m} Re^{-\frac{c}{2m}t} \cos(\omega t - \delta)$$

where:

$$R \cos \delta = x_0$$

$$R \sin \delta = \frac{1}{\omega} \left[\dot{x}_o + \frac{c}{2m} x_o \right]$$

Critical damping

$$(c = c_{crit}, \text{ or } c^2 = 4km)$$

$$x(t) = (A_{cr} + B_{cr}t)e^{-\frac{c}{2m}t}$$

$$\dot{x}(t) = \left[B_{cr} - \frac{c}{2m} (A_{cr} + B_{cr}t) \right] e^{-\frac{c}{2m}t}$$

where:

$$A_{cr} = x_o$$

$$B_{cr} = \dot{x}_o + \frac{c}{2m} x_o$$

Overdamping

$$(c^2 - 4km > 0)$$

$$\dot{x}(t) = A_{ov}e^{r_1 t} + B_{ov}e^{r_2 t}$$

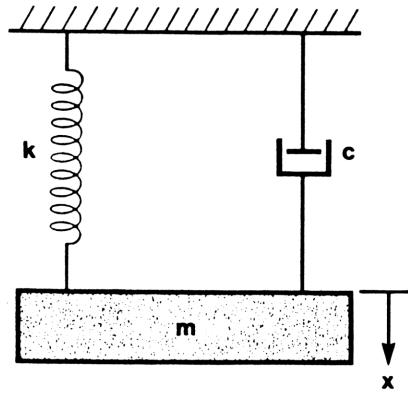
$$x(t) = A_{ov}r_1 e^{r_1 t} + B_{ov}r_2 e^{r_2 t}$$

where:

$$r_1, r_2 = -\frac{c}{2m} \pm \sqrt{\left(\frac{c}{2m}\right)^2 - \frac{k}{m}}$$

$$A_{ov} = x_o - B_{ov}$$

$$B_{ov} = \frac{\dot{x}_o - r_1 x_o}{r_2 - r_1}$$



Reference:

Boyce, W. E. and DiPrima, R. C., Elementary Differential Equations;
John Wiley and Sons, 1969.

Remarks:

For overdamping, ω has no meaning and is, in fact, an imaginary number.

For $c = c_{\text{crit}}$, $\omega = 0$.

This program sets the angular mode of the calculator to radians. Erroneous answers will occur if the degrees mode is inadvertently set.

Example:

A mass of 20 g stretches a spiral spring 10 cm. The mass is pulled down an additional 4 cm, held, and then released. Find the mass displacement and velocity at 0.1 second intervals up to 1 second for $c = 50$ dyne-sec/cm.

$$k = \frac{F}{x} = \frac{mg}{x} = \frac{20g (980 \text{ cm/s}^2)}{10 \text{ cm}} = \frac{20 \times 980}{10} \text{ dyne/cm}$$

Keystrokes:

[USER]
 [XEQ] [ALPHA] SIZE [ALPHA] 016
 [XEQ] [ALPHA] VIB [ALPHA]
 20 [R/S]
 50 [R/S]
 20 [ENTER[↑]] 980 [x] 10 [÷] [R/S]
 [B]
 [C]
 4 [R/S]
 0 [R/S]
 0 [R/S]
 [R/S]
 [R/S]
 [D]
 0.1 [R/S]
 [R/S]
 [R/S]
 [D]
 :::
 (Repeat for t = .2, .3 ... 1.0)
 :::
 1 [R/S]
 [R/S]
 [R/S]

Display:

M?
 C?
 K?
 W=9.820
 C=395.980
 X?
 dX?
 T?
 X=4.000
 dX=-1.000E-9
 d2X=-392.000
 T?
 X=2.334
 dX=-29.296
 d2X=-155.493
 T?
 :::
 X=-1.114
 dX=4.406
 d2X=98.132

User Instructions

Program Listings

<pre> ~ 01*LBL "VIB" " 02 RAD 03 FIX 3 04 "M?" 05 PROMPT 06 STO 02 07 "C?" 08 PROMPT 09 STO 03 10 "K?" 11 PROMPT 12 STO 04 13 RCL 02 14 / 15 STO 15 16 RCL 03 17 RCL 02 18 2 19 * 20 / 21 STO 14 22 X†2 23 - 24 RND 25 X<0? 26 GTO 01 27 X=0? 28 STOP 29 SQRT 30 STO 05 31 "W=" 32 GTO 05 33*LBL 01 34 STO 05 35 -1 36 STOP 37*LBL B 38 RCL 04 39 RCL 02 40 * 41 SQRT 42 2 43 * 44 "C=" 45 GTO 05 46*LBL C 47 "X?" 48 PROMPT 49 STO 00 50 "dX?"</pre>	<p>Initialization</p> <p>Calculate $\left[\frac{k}{m} - \left(\frac{c}{2m} \right)^2 \right]^{\frac{1}{2}}$</p> <p>Calculate C_{cr}</p> <p>Initial conditions</p>	<pre> 51 PROMPT 52 STO 01 53*LBL D 54 "T?" 55 PROMPT 56 STO 07 57 RCL 14 58 * 59 CHS 60 E†X 61 STO 13 62 RCL 05 63 X<0? 64 GTO c 65 X=0? 66 GTO b 67*LBL a 68 RCL 14 69 RCL 00 70 * 71 RCL 01 72 + 73 RCL 05 74 / 75 RCL 00 76 R-P 77 STO 11 78 RDN 79 STO 12 80 RCL 05 81 RCL 07 82 * 83 - 84 CHS 85 STO 09 86 COS 87 RCL 13 88 * 89 RCL 11 90 * 91 STO 02 92 "X=" 93 XEQ 05 94 RCL 14 95 * 96 RCL 09 97 SIN 98 RCL 13 99 * 100 RCL 05 101 * 102 RCL 11</pre> <p>----- Calculate $x(t)$, $\dot{x}(t)$ and $\ddot{x}(t)$ ----- $c > c_{cr}$</p>
--	---	---

Program Listings

103 *		154 RCL 03	
104 +		155 -	
105 CHS		156 /	
106 "dX="		157 STO 12	
107 XEQ 05		158 RCL 00	
108 GTO e	- - - - -	159 -	
109♦LBL b	C = C _{cr}	160 CHS	
110 RCL 14		161 STO 11	
111 RCL 00		162 RCL 03	
112 *		163 RCL 07	
113 RCL 01		164 *	
114 +		165 E↑X	
115 STO 12		166 RCL 11	
116 RCL 07		167 *	
117 *		168 STO 02	
118 RCL 00		169 RCL 03	
119 +		170 *	
120 RCL 13		171 STO 08	
121 *		172 RCL 04	
122 STO 02		173 RCL 07	
123 "X="		174 *	
124 XEQ 05		175 E↑X	
125 RCL 14		176 RCL 12	
126 *		177 *	
127 CHS		178 ST+ 02	
128 RCL 12		179 RCL 04	
129 RCL 13		180 *	
130 *		181 ST+ 08	
131 +		182 RCL 02	
132 "dX="		183 "X="	
133 XEQ 05		184 XEQ 05	
134 GTO e	- - - - -	185 RCL 08	
135♦LBL c	C < C _{cr}	186 "dX="	
136 CHS		187 XEQ 05	
137 SQRT		188♦LBL e	- - - - -
138 ENTER↑		189 RCL 14	Calculate acceleration
139 ENTER↑		190 2	
140 RCL 14		191 *	
141 -		192 *	
142 STO 03		193 RCL 02	
143 X<>Y		194 RCL 15	
144 2		195 *	
145 *		196 +	
146 -		197 CHS	
147 STO 04		198 "d2X="	- - - - -
148 RCL 01		199♦LBL 05	Display
149 RCL 03		200 ARCL X	
150 RCL 00		201 AVIEW	
151 *		202 STOP	
152 -		203 RTN	
153 RCL 04		204 .END.	

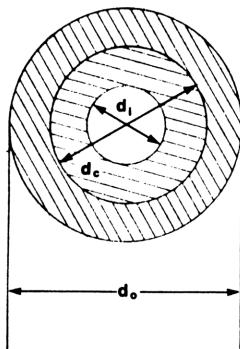
REGISTERS, STATUS, FLAGS, ASSIGNMENTS

DATA REGISTERS			STATUS			
REG	NAME	VAL	SIZE	TOT. REG.	USER MODE	
00	x ₀	50	.016	56		
	dot x ₀		ENG	3	ON	X OFF
	m, x(t)		DEG	RAD	SCI	
	c, r ₁			GRAD		
	k, r ₂					
05	$\omega, \sqrt{(-\omega)^2}$	55				
	(t ₁ -t ₂)/n					
	t, t ₁					
	dot x(t)					
	wt- δ					
10	n	60				
	R, A _{cr} , A _{ov}					
	δ, B_{cr}, B_{ov}					
	$e^{-c_2 m t}$					
	c/2m					
15	k/m	65				
20		70				
25		75				
30		80				
35		85				
ASSIGNMENTS						
			FUNCTION	KEY	FUNCTION	KEY
40		90				
45		95				

INTERFERENCE FITS

This program can be used to determine contact pressure or interference for concentric cylinders. Once the contact pressure has been determined, the actual tangential stresses at the surfaces of the cylinders can be determined. These stresses may be used in maximum shear theory of failure analysis. Modified tangential stresses for use in maximum strain theory of failure analysis can also be computed.

Concentric Cylinders



Equations:

for contact pressure:

$$\delta = d_c P_c \left[\frac{d_c^2 + d_i^2}{E_i(d_c^2 - d_i^2)} + \frac{d_o^2 + d_c^2}{E_o(d_o^2 - d_c^2)} - \frac{\mu_i}{E_i} + \frac{\mu_o}{E_o} \right]$$

for actual tangential stresses:

$$s_{to} = \frac{2P_c d_c^2}{d_o^2 - d_c^2}$$

$$s_{tco} = P_c \left(\frac{d_o^2 + d_c^2}{d_o^2 - d_c^2} \right)$$

$$s_{tci} = -P_c \left(\frac{d_c^2 + d_i^2}{d_c^2 - d_i^2} \right)$$

$$S_{ti} = \frac{-2P_c d_c^2}{d_c^2 - d_i^2}$$

for modified tangential stresses:

$$S'_{to} = \frac{2P_c d_c^2}{d_o^2 - d_c^2}$$

$$S'_{tco} = P_c \left(\frac{d_o^2 + d_c^2}{d_o^2 - d_c^2} + \mu_o \right)$$

$$S'_{tci} = -P_c \left(\frac{d_c^2 + d_i^2}{d_c^2 - d_i^2} - \mu_i \right)$$

$$S'_{ti} = \frac{-2P_c d_c^2}{d_c^2 - d_i^2}$$

where:

δ is the total interference;

P_c is the contact pressure;

d_i is the inside diameter;

d_c is the contact surface diameter;

d_o is the outside diameter;

μ_o is Poisson's ratio for the outside material;

μ_i is Poisson's ratio for the inside material;

E_o is the modulus of elasticity for the outside material;

E_i is the modulus of elasticity for the inside material;

S_{to} is the tangential stress of the outer surface;

s_{tco} is the tangential stress in the outer material at the contact surface;

s_{tci} is the tangential stress in the inner material at the contact surface;

s_{ti} is the stress at the inner surface of the inner cylinder;

s'_{to} , s'_{tco} , s'_{tci} , and s'_{ti} are the modified tangential stresses corresponding to the actual stresses just described.

Reference:

Hall, Holowenko, Laughlin,
Theory and Problems of Machine Design; Schaum's Outline Series,
McGraw Hill Co., 1961.

Example 1:

The choke at the end of a shotgun barrel is to be attached using an interference fit. If 5000 pounds per square inch must be developed to hold the choke in place, what is the minimum allowable interference? What are the values or actual stress?

$$d_i = 0.75 \text{ in}$$

$$d_c = 0.9375 \text{ in}$$

$$d_o = 1.125 \text{ in}$$

$$\mu_o = \mu_i = 0.3$$

$$E_o = E_i = 30 \times 10^6 \text{ psi}$$

Keystrokes:

[XEQ] [ALPHA] SIZE [ALPHA] 009

[XEQ] [ALPHA] INTER [ALPHA]

0.75 [R/S]

0.9375 [R/S]

1.125 [R/S]

0.3 [R/S]

Display:

DI?

DC?

DO?

RATIO O?

EO?

Keystrokes:	Display:
30 [EEX] 6 [R/S]	RATIO I?
0.3 [R/S]	EI?
30 [EEX] 6 [R/S]	Pc?
5000 [R/S]	DELTA=1.578E-3
[R/S]	STO=22.73E3
[R/S]	STCO=27.73E3
[R/S]	STCI=-22.78E3
[R/S]	STI=-27.78E3

User Instructions

STEP	INSTRUCTIONS	INPUT	FUNCTION	DISPLAY
1	Load program			
2	Begin execution			
	to find interference		[XEQ] INTER	DI?
	to find contact pressure		[XEQ] PRES	DI?
3	Input data			
		d_i	[R/S]	DC?
		d_c	[R/S]	DO?
		d_o	[R/S]	RATIO O?
		μ_o	[R/S]	EO?
		E_o	[R/S]	RATIO I?
		μ_i	[R/S]	EI?
		E_i	[R/S]	DELTA? or P_c ?
		δ or P_c	[R/S]	
4	Output is automatic			$P_c=$
				(or)
				DELTA=
5	Continue to find:			
	S_{to}		[R/S]	STO=
	S_{tco}		[R/S]	STCO=
	S_{tci}		[R/S]	STCI=
	S_{ti}		[R/S]	STI=
	S'_{to}		[R/S]	S:TO=
	S'_{tco}		[R/S]	S:TCO=
	S'_{tci}		[R/S]	S:TCI=
	S'_{ti}		[R/S]	S:TI=

Program Listings

```

01♦LBL "INT
ER"
02 SF 00
03 GTO 00
04♦LBL "PRE
S"
05 CF 00
06♦LBL 00
07 ENG 3
08 "DI?"
09 PROMPT
10 STO 01
11 "DC?"
12 PROMPT
13 STO 02
14 "DO?"
15 PROMPT
16 STO 03
17 "RATIO 0
?
18 PROMPT
19 STO 04
20 "EO?"
21 PROMPT
22 STO 05
23 "RATIO I
?
24 PROMPT
25 STO 06
26 "EI?"
27 PROMPT
28 STO 07
29 FS? 00
30 GTO A
31 "DELTA?"
32 PROMPT
33 STO 08
34 XEQ E
35 RCL 08
36 X<>Y
37 /
38 STO 08
39 "Pc="
40 XEQ d
41 GTO B
42♦LBL A
43 "Pc?"
44 PROMPT
45 STO 08
46 XEQ E
47 RCL 08
48 *

```

Initialization

```

49 "DELTA="
50 XEQ d
51 GTO B
52♦LBL E
53 RCL 01
54 X†2
55 RCL 02
56 X†2
57 +
58 LASTX
59 RCL 01
60 X†2
61 -
62 /
63 RCL 07
64 /
65 RCL 02
66 X†2
67 RCL 03
68 X†2
69 +
70 LASTX
71 RCL 02
72 X†2
73 -
74 /
75 RCL 05
76 /
77 +
78 RCL 06
79 RCL 07
80 /
81 -
82 RCL 04
83 RCL 05
84 /
85 +
86 RCL 02
87 *
88 RTN
89♦LBL B
90 0
91 RCL 03
92 XEQ E
93 RCL 08
94 *
95 2
96 *
97 "STO="
98 XEQ d
99 RCL 03
100 RCL 03

```

Subroutine "E"

Outputs

Program Listings

```

101 XEQ E
102 RCL 08
103 *
104 "STC0=""
105 XEQ d
106 RCL 01
107 RCL 01
108 XEQ E
109 RCL 08
110 *
111 "STCI=""
112 XEQ d
113 0
114 RCL 01
115 XEQ E
116 RCL 08
117 *
118 2
119 *
120 "STI=""
121 XEQ d
122 0
123 RCL 03
124 XEQ E
125 RCL 08
126 *
127 2
128 *
129 "S:TO=""
130 XEQ d
131 RCL 03
132 RCL 03
133 XEQ E
134 RCL 04
135 +
136 RCL 08
137 *
138 "S:TC0=""
139 XEQ d
140 RCL 01
141 RCL 01
142 XEQ E
143 RCL 06
144 +
145 RCL 08
146 *
147 "S:TCI=""
148 XEQ d
149 0
150 RCL 01
151 XEQ E
152 RCL 08

```

```

153 *
154 2
155 *
156 "S:TI="
157 GTO d
158♦LBL E
159 X↑2
160 X<>Y
161 X↑2
162 RCL 02
163 X↑2
164 +
165 X<>Y
166 LASTX
167 -
168 /
169 RTN
170♦LBL d
171 ARCL X
172 PROMPT
173 RTN
174 END

```

Subroutine "E"

Display

80

90

00

REGISTERS, STATUS, FLAGS, ASSIGNMENTS

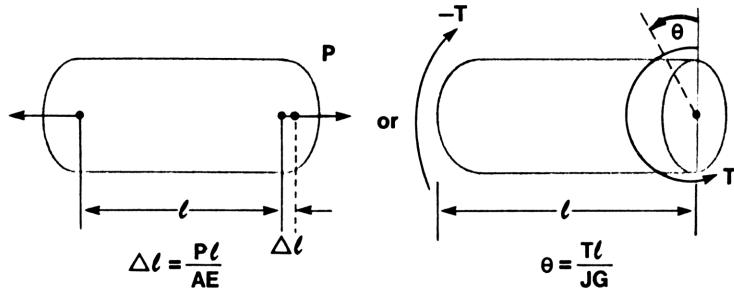
63

DATA REGISTERS				STATUS					
00		50		SIZE 009	TOT. REG. 5 7	USER MODE			
d _i				ENG 3	FIX ____ SCI _____	ON _____	OFF X		
d _c				DEG _____	RAD _____ GRAD _____				
μ_0				FLAGS					
05 E ₀		55		# INIT S/C	SET INDICATES	CLEAR INDICATES			
d ₀				00	Calculate δ	Calculate P _c			
P _c , δ									
10		60							
15		65							
20		70							
25		75							
30		80							
35		85							
ASSIGNMENTS									
40		90		FUNCTION	KEY	FUNCTION	KEY		
45		95							

LINEAR OR ANGULAR DEFORMATION

This program solves for linear deflection under tensile load, or the analogous angular deflection under torque, using an interchangeable solution. Given four of the five variables, the unknown is calculated.

Equations:



where:

Δl is the change in length;

P is the applied load;

ℓ is the length;

A is the cross sectional area;

E is the modulus of elasticity;

θ is the deflection angle in radians;

T is the applied torque;

J is the polar moment of the section;

G is the modulus of elasticity in shear.

Remarks:

This program is not applicable for non-elastic media or elastic media where stress exceeds the elastic limit. Materials must be isotropic. The equation for angular deflection is not valid in the neighborhood of the applied torque.

Example:

Steel bars affixed to the roof are to be used to support the end of a cantilever balcony. The load on each bar will be 50,000 newtons. If the maximum allowable deflection is 0.001 meters, what should the area of the bars be? $\ell = 10$ meters $E = 2.068 \times 10^7$ N/m²

Keystrokes:

[XEQ] [ALPHA] SIZE [ALPHA] 006

[XEQ] [ALPHA] LINEAR [ALPHA]

[R/S]

.001 [R/S]

10 [R/S]

50,000 [R/S]

2.068 [EEX] 11 [R/S]

Display:

A

DELTA

L

P

E

A=2.418E-3

User Instructions

Program Listings

```

01♦LBL "LIN
EAR"
02 CF 00
03 CLX
04 "A"
05 PROMPT
06 X=0?
07 XEQ b
08 STO 01
09 CLX
10 "DELTA"
11 PROMPT
12 X=0?
13 XEQ b
14 STO 02
15 CLX
16 "L"
17 PROMPT
18 X=0?
19 XEQ a
20 STO 03
21 CLX
22 "P"
23 PROMPT
24 X=0?
25 XEQ a
26 STO 04
27 CLX
28 "E"
29 PROMPT
30 X=0?
31 XEQ b
32 STO 05
33 GTO 00
34♦LBL "ANG
ULAR"
35 CF 00
36 CLX
37 "J"
38 PROMPT
39 X=0?
40 XEQ b
41 STO 01
42 CLX
43 "Z"
44 PROMPT
45 X=0?
46 XEQ b
47 STO 02
48 CLX
49 "L"
50 PROMPT

```

```

----- Initialization "LINEAR"
51 X=0?
52 XEQ a
53 STO 03
54 CLX
55 "T"
56 PROMPT
57 X=0?
58 XEQ a
59 STO 04
60 CLX
61 "G"
62 PROMPT
63 X=0?
64 XEQ b
65 STO 05
66 GTO 00
67♦LBL a
68 SF 00
69♦LBL b
70 ASTO 00
71 1
72 RTN
73♦LBL 00
74 ENG 3
75 RCL 04
76 RCL 03
77 *
78 RCL 02
79 RCL 01
80 *
81 RCL 05
82 *
----- Initialization "ANGULAR"
83 /
84 FS? 00
85 1/X
86 CLA
87 ARCL 00
88 "F="
89 ARCL X
90 AVIEW
91 STOP
92 .END.

```

REGISTERS, STATUS, FLAGS, ASSIGNMENTS

DATA REGISTERS				STATUS				
00	OUTPUT LABEL	50		SIZE	006	TOT. REG.	28	USER MODE
	A(J)			ENG	3	FIX		ON
	DELTA (Δ)			DEG		SCI		OFF X
	L					RAD		GRAD
05	P(T)			FLAGS				
	E(G)	55		#	INIT S/C	SET INDICATES	CLEAR INDICATES	
				00		Invert	Don't Invert	
10		60						
15		65						
20		70						
25		75						
30		80						
35		85						
ASSIGNMENTS								
40		90		FUNCTION	KEY	FUNCTION	KEY	
45		95						

CONSTANT ACCELERATION

This program calculates an interchangeable solution among the variables displacement, acceleration, initial velocity, and time, for an object that undergoes constant acceleration. The motion may be either circular or linear. Final velocity as a function of initial velocity, acceleration, and displacement may also be computed.

Equations:

	Linear	Angular
Displacement	$x = v_o t + \frac{1}{2} at^2$	$\theta = \omega_o t + \frac{1}{2} \alpha t^2$
Initial velocity	$v_o = \frac{x}{t} - \frac{1}{2} at$	$\omega_o = \frac{\theta}{t} - \frac{1}{2} \alpha t$
Acceleration	$a = \frac{x - v_o t}{\frac{1}{2} t^2}$	$\alpha = \frac{\theta - \omega_o t}{\frac{1}{2} t^2}$
Time	$t = \frac{\sqrt{v_o^2 + 2ax} - v_o}{a}$	$t = \frac{\sqrt{\omega_o^2 + 2\alpha\theta} - \omega_o}{\alpha}$
Final velocity	$v = \sqrt{v_o^2 + 2ax}$	$\omega = \sqrt{\omega_o^2 + 2\alpha\theta}$

Remarks:

Any consistent set of units may be used.

Displacement, acceleration, and velocity should be considered as signed (vector) quantities. For example, if initial velocity and acceleration are in opposite directions, one should be positive and the other negative.

All equations assume that the initial displacement, x_o or θ_o , equals zero.

Example 1:

An automobile accelerates for 4 seconds from a speed of 35 mph and covers a distance of 264 feet. Assuming constant acceleration, what is the acceleration in ft/sec^2 ? If the acceleration continues to be constant, what distance is covered in the next second?

Keystrokes:	Display:
[XEQ] [ALPHA] SIZE [ALPHA] 004	
[XEQ] [ALPHA] CA [ALPHA]	X, \angle ?
264 [R/S]	V0,W0?
35 [ENTER] 5280 [*] 3600 [÷] [R/S]	T?
4 [R/S]	A?
[R/S]	
[XEQ] A	A=7.733
5 [STO] 02	
[XEQ] X	X, \angle =348.33 (total distance)
264 [-]	84.33

User Instructions

Program Listings

```

01♦LBL "CA"
02 FIX 2
03 "X,Δ?"
04 PROMPT
05 STO 00
06 "V0,W0?"
07 PROMPT
08 STO 01
09 "T?"
10 PROMPT
11 STO 02
12 "A?"
13 PROMPT
14 STO 03
15 STOP
16♦LBL "X"
17 RCL 02
18 X↑2
19 RCL 03
20 *
21 2
22 /
23 RCL 02
24 RCL 01
25 *
26 +
27 STO 00
28 "X,Δ="
29 GTO 05
30♦LBL "V0"
31 RCL 00
32 RCL 02
33 /
34 RCL 02
35 RCL 03
36 *
37 2
38 /
39 -
40 STO 01
41 "V0,W0="
42 GTO 05
43♦LBL A
44 RCL 00
45 RCL 01
46 RCL 02
47 *
48 -
49 RCL 02
50 X↑2
51 2
52 /

```

Initialization

Displacement

Initial velocity

Acceleration

```

53 /
54 STO 03
55 "A="
56 GTO 05
57♦LBL "T"
58 CF 01
59 XEQ 01
60 RCL 01
61 -
62 RCL 03
63 /
64 STO 02
65 "T="
66 GTO 05
67♦LBL "V"
68 SF 01
69♦LBL 01
70 RCL 01
71 X↑2
72 RCL 03
73 RCL 00
74 *
75 2
76 *
77 +
78 SQRT
79 FC? 01
80 RTN
81 "V,W="
82♦LBL 05
83 ARCL X
84 AVIEW
85 STOP
86 .END.

```

	90
	00

Time

Final velocity

Display

REGISTERS, STATUS, FLAGS, ASSIGNMENTS

DATA REGISTERS			STATUS				
00	X, θ V or ω t A, α	50	SIZE	004	TOT. REG.	25	USER MODE
			ENG	2	SCI		ON OFF X
			DEG		RAD	GRAD	
05		55	FLAGS				
			#	INIT S/C	SET INDICATES	CLEAR INDICATES	
10		60					
15		65					
20		70					
25		75					
30		80					
35		85					
ASSIGNMENTS							
40		90	FUNCTION	KEY	FUNCTION	KEY	
45		95					

HEWLETT-PACKARD

HP-41C

USERS' LIBRARY SOLUTIONS

Bar Codes

Mechanical Engineering

MECHANICAL ENGINEERING

GEAR FORCES.....	1
STRESS ON AN ELEMENT.....	2
EQUATIONS OF STATE.....	4
SODERBERG'S EQUATION FOR FATIGUE.....	6
SPRING CONSTANT.....	8
PROGRESSION OF A SLIDER CRANK.....	9
FREE VIBRATIONS.....	11
INTERFERENCE FITS.....	13
LINEAR OR ANGULAR DEFORMATION.....	15
CONSTANT ACCELERATION.....	16

NOTICE

The program material contained herein is supplied without representation or warranty of any kind. Hewlett-Packard Company therefore assumes no responsibility and shall have no liability, consequential or otherwise, of any kind arising from the use of this program material or any part thereof.

GEAR FORCES

PROGRAM REGISTERS NEEDED: 31

ROW 1 (1 - 3)



ROW 2 (4 - 10)



ROW 3 (11 - 14)



ROW 4 (15 - 21)



ROW 5 (22 - 31)



ROW 6 (31 - 35)



ROW 7 (36 - 37)



ROW 8 (37 - 49)



ROW 9 (50 - 61)



ROW 10 (61 - 64)



ROW 11 (64 - 66)



ROW 12 (67 - 77)



ROW 13 (78 - 86)



ROW 14 (86 - 97)



ROW 15 (98 - 106)



ROW 16 (106 - 113)



ROW 17 (114 - 119)



STRESS ON AN ELEMENT

PROGRAM REGISTERS NEEDED: 46

ROW 1 (1 - 2)



ROW 2 (3 - 6)



ROW 3 (7 - 15)



ROW 4 (15 - 20)



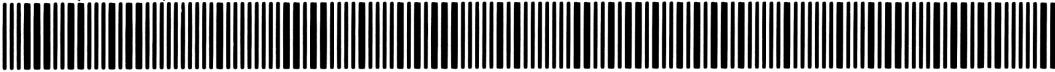
ROW 5 (21 - 27)



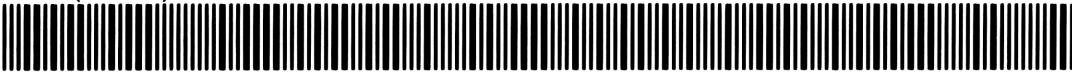
ROW 6 (27 - 36)



ROW 7 (37 - 48)



ROW 8 (49 - 60)



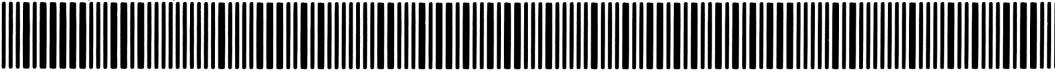
ROW 9 (61 - 68)



ROW 10 (68 - 78)



ROW 11 (79 - 89)



ROW 12 (90 - 101)



ROW 13 (102 - 110)



ROW 14 (111 - 115)



ROW 15 (115 - 120)



ROW 16 (120 - 128)



ROW 17 (129 - 139)



ROW 18 (140 - 148)



STRESS ON AN ELEMENT

ROW 19 (149 – 155)



ROW 20 (156 – 162)



ROW 21 (163 – 169)



ROW 22 (170 – 180)



ROW 23 (180 – 189)



ROW 24 (190 – 200)



ROW 25 (200 – 204)



EQUATIONS OF STATE

PROGRAM REGISTERS NEEDED: 47

ROW 1 (1 - 5)



ROW 2 (5 - 10)



ROW 3 (11 - 17)



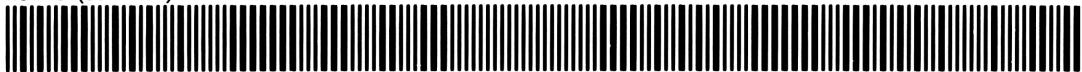
ROW 4 (18 - 24)



ROW 5 (25 - 30)



ROW 6 (30 - 37)



ROW 7 (38 - 47)



ROW 8 (48 - 55)



ROW 9 (55 - 65)



ROW 10 (66 - 73)



ROW 11 (73 - 79)



ROW 12 (80 - 91)



ROW 13 (91 - 98)



ROW 14 (98 - 108)



ROW 15 (109 - 121)



ROW 16 (122 - 130)



ROW 17 (130 - 138)



ROW 18 (139 - 151)



EQUATIONS OF STATE

ROW 19 (152 – 164)



ROW 20 (165 – 177)



ROW 21 (178 – 190)



ROW 22 (191 – 202)



ROW 23 (203 – 210)



ROW 24 (211 – 222)



ROW 25 (222 – 230)



ROW 26 (230 – 230)



SODERBERG'S EQUATION FOR
FATIGUE
PROGRAM REGISTERS NEEDED: 52

ROW 1 (1 - 3)



ROW 2 (3 - 10)



ROW 3 (11 - 19)



ROW 4 (20 - 22)



ROW 5 (22 - 27)



ROW 6 (27 - 33)



ROW 7 (34 - 36)



ROW 8 (37 - 47)



ROW 9 (47 - 50)



ROW 10 (51 - 61)



ROW 11 (62 - 65)



ROW 12 (65 - 69)



ROW 13 (70 - 82)



RCW 14 (83 - 93)



ROW 15 (93 - 96)



ROW 16 (97 - 104)



ROW 17 (105 - 115)

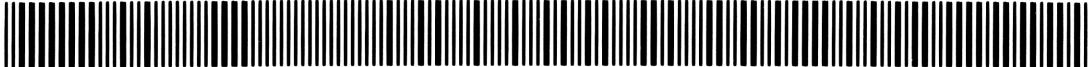


ROW 18 (115 - 121)



SODERBERG'S EQUATION FOR
FATIGUE

ROW 19 (122 - 130)



ROW 20 (130 - 132)



ROW 21 (132 - 141)



ROW 22 (142 - 154)



ROW 23 (155 - 161)



ROW 24 (161 - 163)



ROW 25 (164 - 172)



ROW 26 (173 - 181)



ROW 27 (182 - 193)

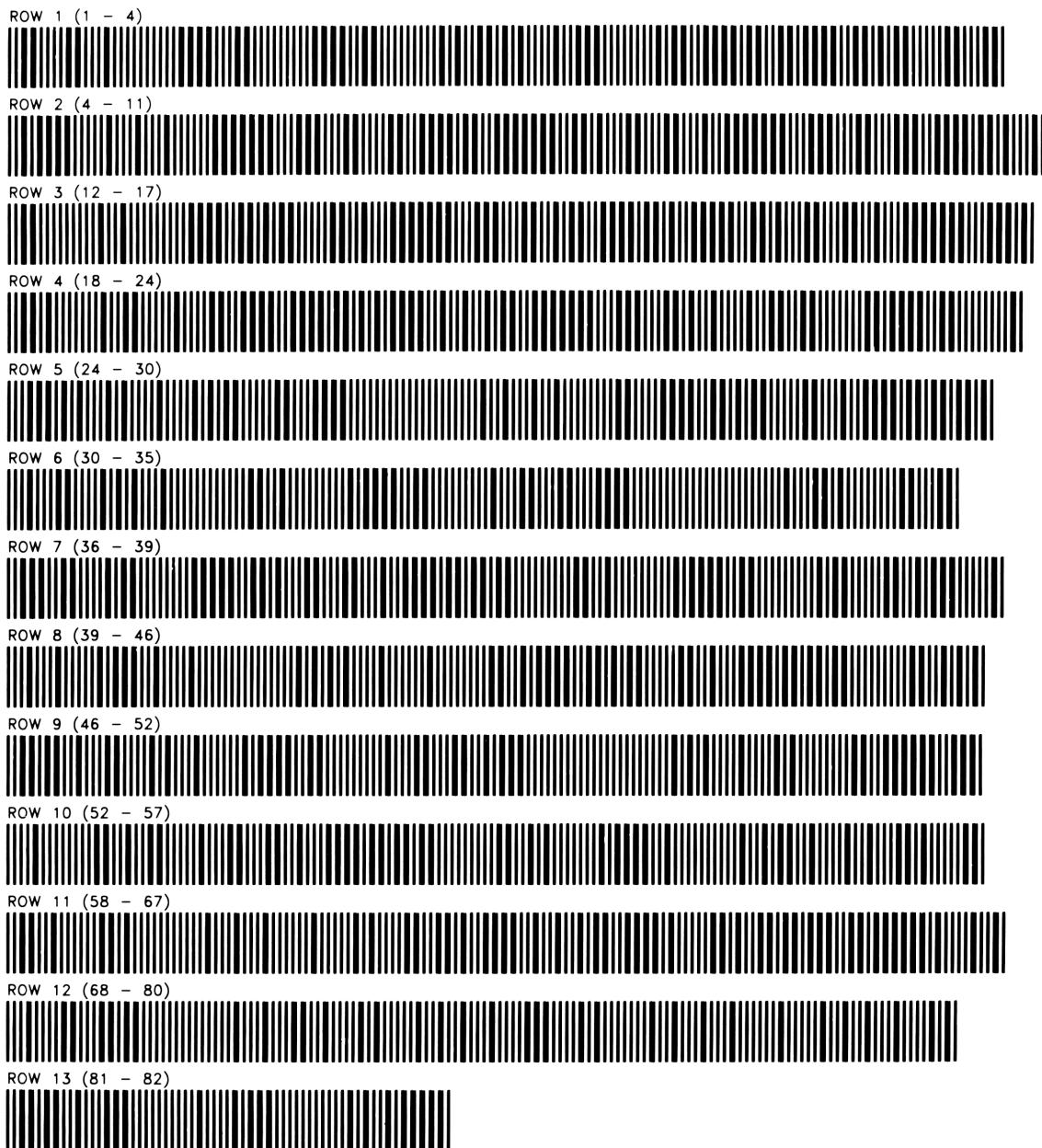


ROW 28 (194 - 198)



SPRING CONSTANT

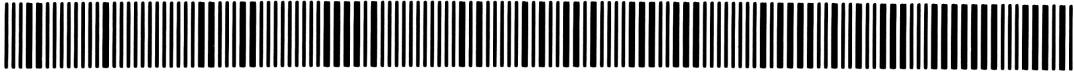
PROGRAM REGISTERS NEEDED: 23



PROGRESSION OF A SLIDER CRANK

PROGRAM REGISTERS NEEDED: 46

ROW 1 (1 - 4)



ROW 2 (5 - 12)



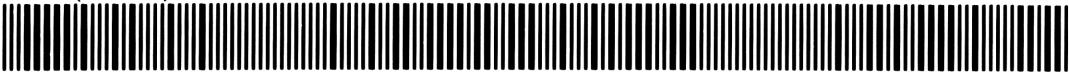
ROW 3 (12 - 21)



ROW 4 (22 - 28)



ROW 5 (28 - 35)



ROW 6 (35 - 40)



ROW 7 (41 - 47)



ROW 8 (48 - 59)



ROW 9 (59 - 64)



ROW 10 (65 - 76)



ROW 11 (77 - 85)



ROW 12 (86 - 93)



ROW 13 (94 - 106)



ROW 14 (107 - 117)



ROW 15 (117 - 128)



ROW 16 (129 - 135)



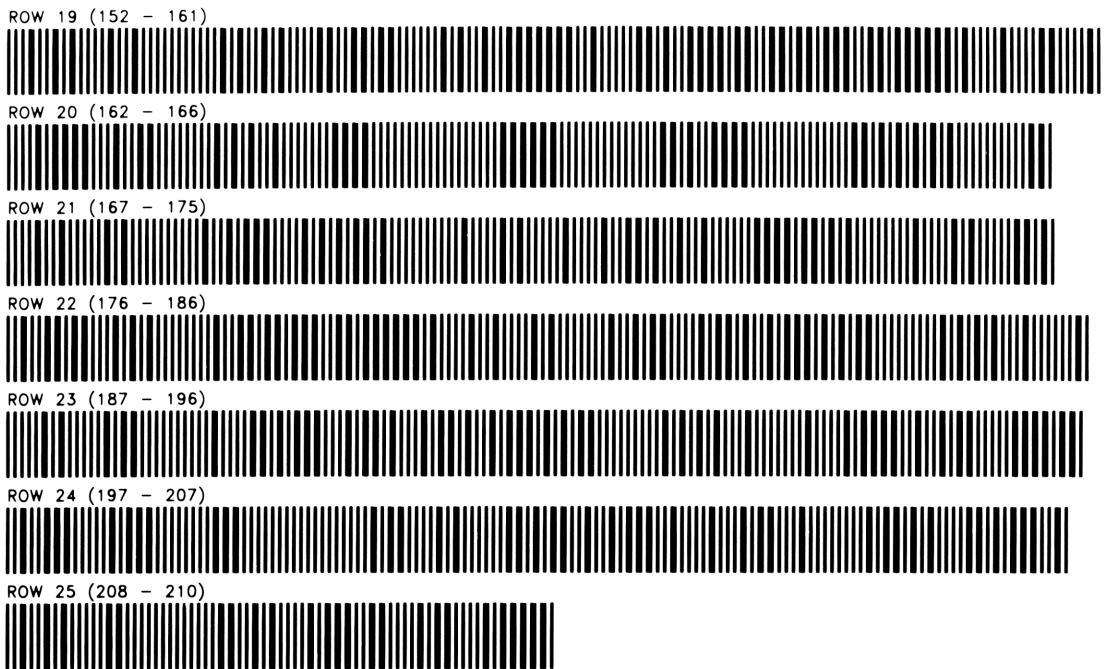
ROW 17 (135 - 141)



ROW 18 (142 - 151)



PROGRESSION OF A SLIDER CRANK



FREE VIBRATIONS

PROGRAM REGISTERS NEEDED: 41

ROW 1 (1 - 4)



ROW 2 (5 - 13)



ROW 3 (14 - 26)



ROW 4 (26 - 35)



ROW 5 (35 - 44)



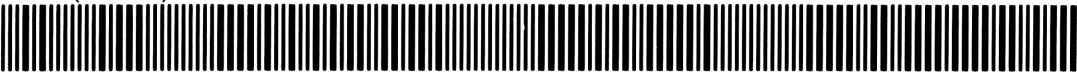
ROW 6 (45 - 50)



ROW 7 (51 - 60)



ROW 8 (61 - 68)



ROW 9 (69 - 81)



ROW 10 (82 - 92)



ROW 11 (93 - 103)



ROW 12 (104 - 109)



ROW 13 (109 - 121)



RCW 14 (122 - 130)



ROW 15 (131 - 135)



ROW 16 (136 - 148)



ROW 17 (149 - 161)



ROW 18 (162 - 174)



FREE VIBRATIONS

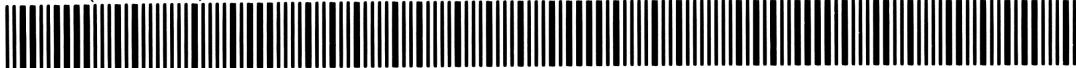
ROW 19 (175 - 183)



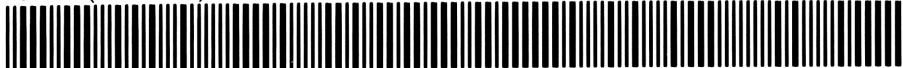
ROW 20 (184 - 188)



ROW 21 (189 - 198)



ROW 22 (198 - 204)



INTERFERENCE FITS

PROGRAM REGISTERS NEEDED: 49

ROW 1 (1 - 3)



ROW 2 (4 - 7)



ROW 3 (8 - 14)



ROW 4 (14 - 17)



ROW 5 (17 - 23)



ROW 6 (23 - 28)



ROW 7 (29 - 32)



ROW 8 (33 - 40)



ROW 9 (40 - 45)



ROW 10 (46 - 50)



ROW 11 (50 - 58)



ROW 12 (59 - 71)



ROW 13 (72 - 84)



ROW 14 (85 - 94)



ROW 15 (95 - 101)



ROW 16 (101 - 105)



ROW 17 (106 - 111)



ROW 18 (112 - 120)



INTERFERENCE FITS

ROW 19 (120 – 125)



ROW 20 (126 – 131)



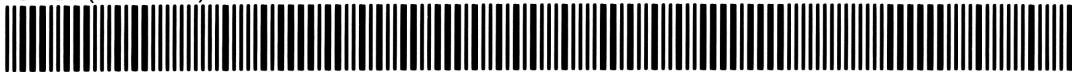
ROW 21 (132 – 138)



ROW 22 (138 – 145)



ROW 23 (146 – 150)



ROW 24 (151 – 156)



ROW 25 (157 – 166)



ROW 26 (167 – 174)



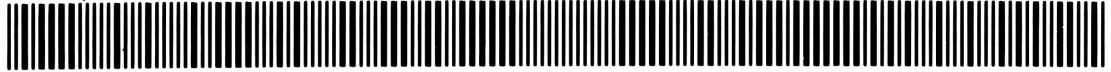
LINEAR OR ANGULAR DEFORMATION

PROGRAM REGISTERS NEEDED: 23

ROW 1 (1 - 3)



ROW 2 (4 - 10)



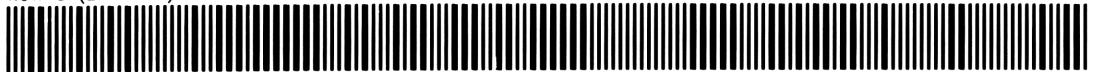
ROW 3 (10 - 18)



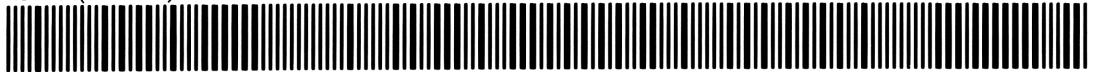
ROW 4 (19 - 26)



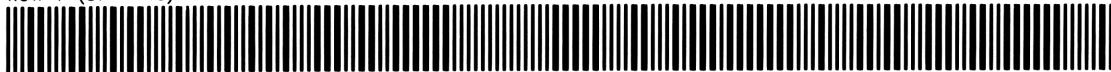
ROW 5 (27 - 34)



ROW 6 (34 - 37)



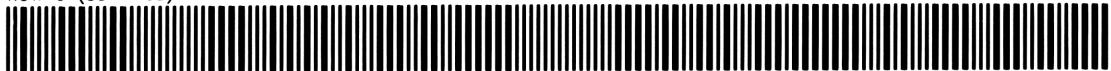
ROW 7 (37 - 46)



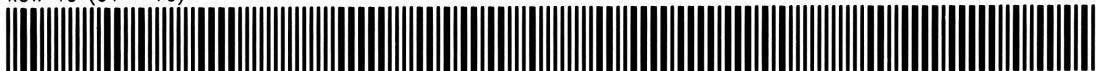
ROW 8 (46 - 54)



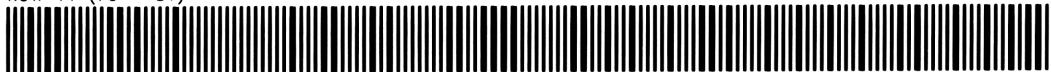
ROW 9 (55 - 63)



ROW 10 (64 - 70)



ROW 11 (70 - 81)



ROW 12 (82 - 89)



ROW 13 (90 - 92)



CONSTANT ACCELERATION

PROGRAM REGISTERS NEEDED: 22

ROW 1 (1 - 3)



ROW 2 (4 - 9)



ROW 3 (9 - 16)



ROW 4 (16 - 28)



ROW 5 (28 - 31)



ROW 6 (32 - 41)



ROW 7 (41 - 49)



ROW 8 (50 - 57)



ROW 9 (57 - 65)



ROW 10 (65 - 70)



ROW 11 (71 - 81)



ROW 12 (81 - 86)



NOTES



HEWLETT
PACKARD

Hewlett-Packard Software

In terms of power and flexibility, the problem-solving potential of the HP-41C programmable calculator is nearly limitless. And in order to see the practical side of this potential, HP has different types of software to help save you time and programming effort. Every one of our software solutions has been carefully selected to effectively increase your problem-solving potential. Chances are, we already have the solutions you're looking for.

Application Pacs

To increase the versatility of your HP-41C, HP has an extensive library of "Application Pacs". These programs transform your HP-41C into a specialized calculator in seconds. Included in these pacs are detailed manuals with examples, minature plug-in Application Modules, and keyboard overlays. Every Application Pac has been designed to extend the capabilities of the HP-41C.

You can choose from:

Aviation	Structural Analysis	Home Management
Clinical Lab	Surveying	Machine Design
Circuit Analysis	Securities	Navigation
Financial Decisions	Statistics	Real Estate
Mathematics	Stress Analysis	Thermal and Transport Science
	Games	

Users' Library

The Users' Library provides the best programs from contributors and makes them available to you. By subscribing to the HP-41C Users' Library you'll have at your fingertips literally hundreds of different programs from many different application areas.

*** Users' Library Solutions Books**

Hewlett-Packard offers a wide selection of Solutions Books complete with user instructions, examples, and listings. These solution books will complement our other software offerings and provide you with a valuable tool for program solutions.

You can choose from:

Business Stat/Marketing/Sales	Civil Engineering
Home Construction Estimating	Heating, Ventilating & Air Conditioning
Lending, Saving and Leasing	Mechanical Engineering
Real Estate	Solar Engineering
Small Business	Calendars
Geometry	Cardiac/Pulmonary
High-Level Math	Chemistry
Test Statistics	Games
Antennas	Optometry I (General)
Chemical Engineering	Optometry II (Contact Lens)
Control Systems	Physics
Electrical Engineering	Surveying
Fluid Dynamics and Hydraulics	

* Some books require additional memory modules to accomodate all programs.

MECHANICAL ENGINEERING

GEAR FORCES
STRESS ON AN ELEMENT
EQUATIONS OF STATE
SODERBERG'S EQUATION FOR FATIGUE
SPRING CONSTANT
PROGRESSION OF A SLIDER CRANK
FREE VIBRATIONS
INTERFERENCE FITS
LINEAR OF ANGULAR DEFORMATION
CONSTANT ACCELERATION

