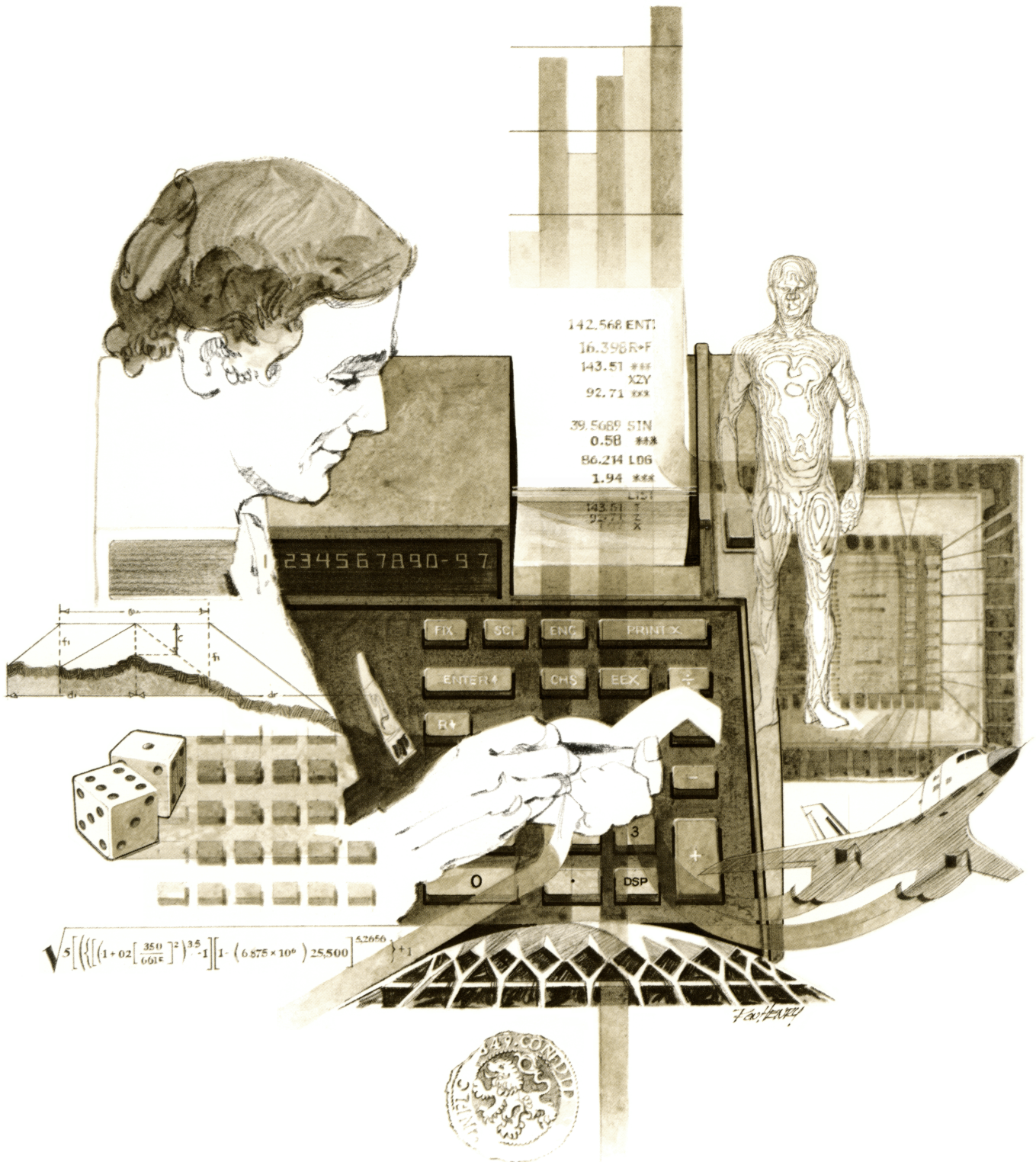


HEWLETT-PACKARD

HP-67/HP-97

Users' Library Solutions
Energy Conservation



INTRODUCTION

In an effort to provide continued value to its customers, Hewlett-Packard is introducing a unique service for the HP fully programmable calculator user. This service is designed to save you time and programming effort. As users are aware, Programmable Calculators are capable of delivering tremendous problem solving potential in terms of power and flexibility, but the real genie in the bottle is program solutions. HP's introduction of the first handheld programmable calculator in 1974 immediately led to a request for program **solutions** — hence the beginning of the HP-65 Users' Library. In order to save HP calculator customers time, users wrote their own programs and sent them to the Library for the benefit of other program users. In a short period of time over 5,000 programs were accepted and made available. This overwhelming response indicated the value of the program library and a Users' Library was then established for the HP-67/97 users.

To extend the value of the Users' Library, Hewlett-Packard is introducing a unique service—a service designed to save you time and money. The Users' Library has collected the best programs in the most popular categories from the HP-67/97 and HP-65 Libraries. These programs have been packaged into a series of low-cost books, resulting in substantial savings for our valued HP-67/97 users.

We feel this new software service will extend the capabilities of our programmable calculators and provide a great benefit to our HP-67/97 users.

A WORD ABOUT PROGRAM USAGE

Each program contained herein is reproduced on the standard forms used by the Users' Library. Magnetic cards are not included. The Program Description I page gives a basic description of the program. The Program Description II page provides a sample problem and the keystrokes used to solve it. The User Instructions page contains a description of the keystrokes used to solve problems in general and the options which are available to the user. The Program Listing I and Program Listing II pages list the program steps necessary to operate the calculator. The comments, listed next to the steps, describe the reason for a step or group of steps. Other pertinent information about data register contents, uses of labels and flags and the initial calculator status mode is also found on these pages. Following the directions in your HP-67 or HP-97 **Owners' Handbook and Programming Guide**, "Loading a Program" (page 134, HP-67; page 119, HP-97), key in the program from the Program Listing I and Program Listing II pages. A number at the top of the Program Listing indicates on which calculator the program was written (HP-67 or HP-97). If the calculator indicated differs from the calculator you will be using, consult Appendix E of your **Owner's Handbook** for the corresponding keycodes and keystrokes converting HP-67 to HP-97 keycodes and vice versa. No program conversion is necessary. The HP-67 and HP-97 are totally compatible, but some differences do occur in the keycodes used to represent some of the functions.

A program loaded into the HP-67 or HP-97 is not permanent—once the calculator is turned off, the program will not be retained. You can, however, permanently save any program by recording it on a blank magnetic card, several of which were provided in the Standard Pac that was shipped with your calculator. Consult your **Owner's Handbook** for full instructions. A few points to remember:

The Set Status section indicates the status of flags, angular mode, and display setting. After keying in your program, review the status section and set the conditions as indicated before using or permanently recording the program.

REMEMBER! To save the program permanently, **clip** the corners of the magnetic card once you have recorded the program. This simple step will protect the magnetic card and keep the program from being inadvertently erased.

As a part of HP's continuing effort to provide value to our customers, we hope you will enjoy our newest concept.

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Program calculates one of any four quantities in the design of an air cooling system for electronic equipment. Given the ambient temperature, the power dissipation in the enclosure, and the worst case maximum temperature, the program calculates the required blower rating in cubic feet per minute.	
BLACK BODY THERMAL RADIATION	6
Calculates wave length of maximum emissive power, total emissive power, monochromatic emissive power, emissive power from zero to a specified wave length, for black radiating surfaces.	
ECONOMIC INSULATION THICKNESS	14
Can be used to determine the economic thickness of insulation given the thermal properties of the insulation, the cost of energy, hours of operation, cost of insulation, and the temperature difference.	
HEAT TRANSFER THROUGH COMPOSITE CYLINDERS AND WALLS	18
Can be used to calculate the overall heat transfer coefficient for composite tubes and walls from individual section conductances and surface coefficients.	
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Computes heat duty, heat load, transfer area, logarithmic mean temperature difference, heat capacity, transfer coefficient and mass flow rate.	
SUN ALTITUDE, AZIMUTH, SOLAR POND ABSORPTION	27
Computes sun's altitude, azimuth, and the fraction of the sun's radiation which will penetrate the surface of a solar pond given index of refraction of pond fluid, latitude, number of days after spring equinox, and number of hours before or after solar noon.	
TOTAL DAILY AMOUNT OF SOLAR RADIATION	31
Computes length of day and total amount of solar radiation received by a horizontal surface of unit area as a function of latitude and declination of the sun.	
TEMPERATURE OR CONCENTRATION PROFILE FOR A SEMI-INFINITE SOLID	35
May be used to find the temperature (or concentration profile) at a specified time for a semi-infinite solid with constant surface temperature (or concentration) the profiles is assumed to be uniform when time equals zero.	

TRANSIENT TEMPERATURE DISTRIBUTION IN A SEMI-INFINITE SOLID WITH CONVECTION BOUNDARY CONDITION	40
Computes factor enabling calculation of temperature in a semi- infinite solid for data including distance from surface, time, thermal conductivity, thermal diffusivity, and heat transfer coefficient.	
CONSERVATION OF ENERGY.	44
This is a two card set which may be used to solve a variety of energy conservative flow problems (Bernoulli's equation). Card one accepts English units while card two is for metric units.	

Program Description I

1

Program Title Air Cooling System Design

Contributor's Name Hewlett-Packard

Address 1000 N.E. Circle Blvd.

City Corvallis **State** Oregon **Zip Code** 97330

Program Description, Equations, Variables

Define: H_i = Molal enthalpy of air at T_i and P_i (BTU/LB Mole)

P_i = Power in kilowatts (3413 kW-Hr / BTU)

T_i = Temperature in degrees Rankine ($^{\circ}\text{R}$)

\dot{V}_i = Volumetric flow rate

\dot{N}_i = Molal flow rate

$i = 1$ for outside enclosure

p_i = pressure (in atmospheres)

$i = 2$ for inside enclosure

C_p = specific heat at constant pressure =
6.953 (BTU/lb-mole $^{\circ}\text{R}$)

- Molar volume for air at a temperature T and pressure p is $V = (0.35905$
 $(0.35905 \times 10^3 \text{ Ft}^3/\text{lb-mole}) (\text{Atm}/p)(T/491.7^{\circ}\text{R}) = .7302$

- Energy balance at steady state

$$H_1 \dot{N}_1 + (P_1 - P_2) = H_2 \dot{N}_2$$

- Molal volume for an ideal gas has a flow rate $\dot{V}_1 = \left(\frac{p_2}{p_1}\right) \left(\frac{T_2}{T_1}\right) \dot{V}_2$

- Specific heat equation $H_2 - H_1 = C_p (T_2 - T_1)$

- Neglect pressure difference $p_i = p_2$

Continued on next page →

Operating Limits and Warnings

1. Calculation assumes steady state, treats air as an ideal gas, neglects humidity, etc.
2. The "E" key should not be used as it contains several values needed to cram the program into the limited memory.

This program has been verified only with respect to the numerical example given in *Program Description II*. User accepts and uses this program material AT HIS OWN RISK, in reliance solely upon his own inspection of the program material and without reliance upon any representation or description concerning the program material.

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Program Description I

Program Title Air Cooling System Design
Contributor's Name Hewlett-Packard
Address 1000 N.E. Circle Blvd.
City Corvallis **State** Oregon **Zip Code** 97330

Program Description, Equations, Variables

Then $\dot{V}_2 = \frac{K(P_2 - P_1)T_2}{T_2 - T_1} = \text{Ft}^3/\text{min} = \text{Req'd flow rate}$ $k = 5.974$

or $P_2 - P_1 = \frac{\dot{V}_2}{k} \left(\frac{T_2 - T_1}{T_2} \right) = \text{kW} = \text{Power Input}$

or $T_2 = \frac{\dot{V}_2 T_1}{V_2 - k(P_2 - P_1)} = \text{Max. Enclosure Temperature}$

or $T_1 = \frac{T_2}{V_2} (V_2 - k(P_2 - P_1)) = \text{Ambient Temperature}$

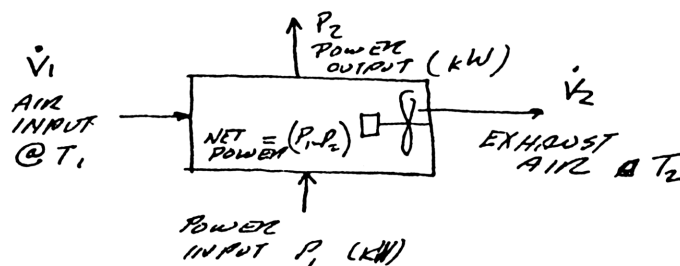
Operating Limits and Warnings

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Program Description II

Sketch(es)



Sample Problem(s)

$$t_1 = 72^\circ\text{F}$$

$$t_2 = 85^\circ\text{F}$$

$$\dot{V} = 250 \text{ ft}^3/\text{min}$$

$$\text{Calculate: } P_{1N} = 0.9982 \text{ kW}$$

Now assume you wish to change the ambient to 68°F with the same power and see what effect it has upon the blower rating. After having first pressed "D"

just key in 68 in "A" and press "C". To use the calculated P_{1N} $\dot{V} = 191.1765$

Now change \dot{V} to 200 and P_{1N} to 1 kW and resolve for $t_1 = 68.7209^\circ\text{F}$. Now change t_1 to 65°F and calculate t_2 for the $\dot{V} = 200$ and $P_{1N} = 1$ by entering 65° in "A" and pressing "B" to obtain $t_2 = 81.1646$

If you forget one of the other three variables in using the program for optimizing just look in R_1, R_2, R_3 or R_4 For T_1, T_2, V or P_{1N} as may be of interest.

Solution(s)

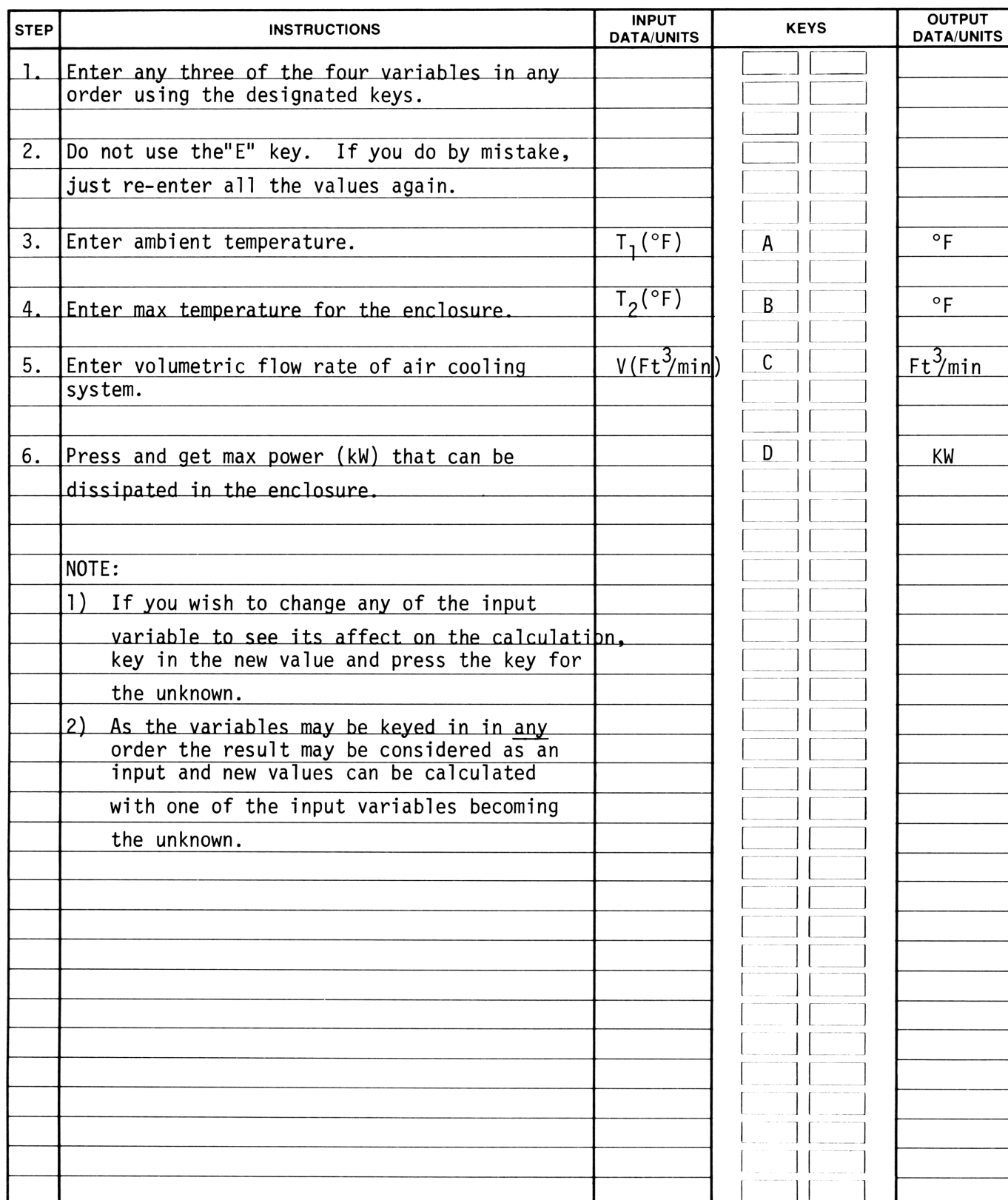
Keystrokes:

Outputs

72[A] 85[B] 250[C] [D] -----	0.9882
[D] 68[A] [C] -----	191.1765
200[C] 1[D] [A] -----	68.7209
65[A] [B] -----	81.1646

Reference(s) V.M. Faires, Thermodynamics, 5th Edition, MacMillan Co., New York, 1970; page 453.

This program is a translation of the HP-65 User's Library program
#02001A submitted by Todd A.C. Heard.



97 Program Listing I

5

STEP	KEY ENTRY	KEY CODE	COMMENTS	STEP	KEY ENTRY	KEY CODE	COMMENTS
001	*LBLA	21 11	Calls constants,etc	057	*LBLD	21 14	Computed P_{1N} (kW)
002	STO1	35 01		058	STO4	35 04	
003	0	00		059	0	00	
004	X#Y?	16-32		060	X#Y?	16-32	
005	RTN	24		061	RTN	24	
006	GSBE	23 15		062	GSBE	23 15	
007	RCL4	36 04		063	RCL5	36 05	
008	RCL6	36 06		064	RCL3	36 03	
009	CHS	-22		065	X	-35	
010	RCL3	36 03		066	RCL6	36 06	
011	+	-55	067	÷	-24	Constant 5.974	
012	RCL3	36 03	068	ENT↑	-21		
013	÷	-24	069	RCL8	36 08		
014	ENT↑	-21	070	÷	-24		
015	RCL8	36 08	071	RTN	24		
016	X	-35	072	*LBLB	21 15		
017	RCL7	36 07	073	5	05		
018	-	-45	074	.	-62		
019	RTN	24	075	9	09		
020	*LBLB	21 12	076	7	07		
021	STO2	35 02	077	4	04	Constant 460	
022	0	00	078	STO6	35 06		
023	X#Y?	16-32	079	4	04		
024	RTN	24	080	6	06		
025	GSBE	23 15	081	0	00		
026	RCL1	36 01	082	STO7	35 07		
027	RCL7	36 07	083	RCL2	36 02		
028	+	-55	084	RCL1	36 01		
029	RCL3	36 03	085	-	-45		
030	X	-35	086	STO5	35 05		(t_2-t_1)
031	ENT↑	-21	087	RCL2	36 02		
032	RCL4	36 04	088	RCL7	36 07		
033	RCL6	36 06	089	+	-55		
034	X	-35	090	STO8	35 08		
035	CHS	-22	091	RTN	24		
036	RCL3	36 03					
037	+	-55					
038	÷	-24					
039	RCL7	36 07					
040	-	-45					
041	RTN	24					
042	*LBLC	21 13					
043	STO3	35 03					
044	0	00	100				
045	X#Y?	16-32					
046	RTN	24					
047	GSBE	23 15					
048	RCL8	36 08					
049	RCL4	36 04					
050	X	-35					
051	RCL6	36 06					
052	X	-35					
053	ENT↑	-21					
054	RCL5	36 05					
055	÷	-24					
056	RTN	24					
			Computed $t_2(^{\circ}\text{F})$				

Program Description I

Program Title Black Body Thermal Radiation

Contributor's Name Hewlett-Packard

Address 1000 N. E. Circle Blvd.

City Corvallis

State Oregon

Zip Code 97330

Program Description, Equations, Variables

Bodies with finite temperatures emit thermal radiation. The higher the absolute temperature, the more thermal radiation emitted. Bodies which emit the maximum possible amount of energy at every wavelength for a specified temperature are said to be black bodies. While black bodies do not actually exist in nature, many surfaces may be assumed to be black for engineering considerations.

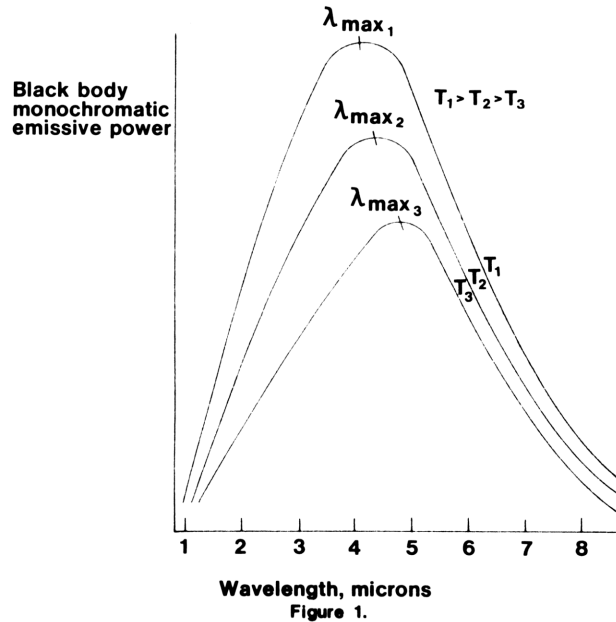


Figure 1.

(continued
next page)

Operating Limits and Warnings

A minute or more may be required to obtain $E_{b(0-\lambda)}$ or $E_{b(\lambda_1-\lambda_2)}$ since the integration is numerical.

Sources differ on values for constants. This could yield small discrepancies between published tables and program outputs.

This program has been verified only with respect to the numerical example given in *Program Description II*. User accepts and uses this program material AT HIS OWN RISK, in reliance solely upon his own inspection of the program material and without reliance upon any representation or description concerning the program material.

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Figure 1 is a representation of black body thermal emission as a function of wavelength. Note that as temperature increases, the area under the curves (total emissive power $E_{b(0-\infty)}$) increases. Also note that the wavelength of maximum emissive power λ_{\max} shifts to the left as temperature increases.

This program calculates the wavelength of maximum emissive power for a given temperature, the temperature for which a given wavelength would be the wavelength of maximum emissive power, the total emissive power over all wavelengths, the emissive power at a particular wavelength, the emissive power from zero to a specified wavelength, and the emissive power between specified wavelengths.

Equations:

$$\lambda_{\max} T_{\lambda_{\max}} = c_3$$

$$E_{b(0-\infty)} = \sigma T^4$$

$$E_{b\lambda} = \frac{2\pi c_1}{\lambda^5 (e^{c_2/\lambda T} - 1)}$$

$$E_{b(0-\lambda)} = \int_0^{\lambda} E_{b\lambda} d\lambda$$

$$= 2\pi c_1 \sum_{k=1}^{\infty} \frac{-T/kc_2}{e^{-kc_2/\lambda T}} \left[\left(\frac{1}{\lambda} \right)^3 + \frac{3T}{\lambda^2 kc_2} + \frac{6}{\lambda} \left(\frac{T}{kc_2} \right)^2 + 6 \left(\frac{T}{kc_2} \right)^3 \right]$$

$$E_{b(\lambda_1 - \lambda_2)} = E_{b(0-\lambda_2)} - E_{b(0-\lambda_1)}$$

where

λ_{\max} is the wavelength of maximum emissivity in microns;

T is the absolute temperature in $^{\circ}\text{R}$ or K;

$E_{b(0-\infty)}$ is the total emissive power in Btu/hr-ft^2 or Watts/cm^2 ;

$E_{b\lambda}$ is the emissive power at λ in $\text{Btu/hr-ft}^2\text{-}\mu\text{m}$ or $\text{Watts/cm}^2\text{-}\mu\text{m}$;

$E_{b(0-\lambda)}$ is the emissive power for wavelengths less than λ in Btu/hr-ft^2 or Watts/cm^2 ;

$E_{b(\lambda_1-\lambda_2)}$ is the emissive power for wavelengths between λ_1 and λ_2 in Btu/hr-ft^2 or Watts/cm^2 .

$$\begin{aligned} c_1 &= 1.8887982 \times 10^7 \text{ Btu-}\mu\text{m}^4/\text{hr-ft}^2 \\ &= 5.9544 \times 10^3 \text{ W}\mu\text{m}^4/\text{cm}^2 \end{aligned}$$

$$c_2 = 2.58984 \times 10^4 \mu\text{m-}^{\circ}\text{R} = 1.4388 \times 10^4 \mu\text{m-K}$$

$$c_3 = 5.216 \times 10^3 \mu\text{m-}^{\circ}\text{R} = 2.8978 \times 10^3 \mu\text{m-K}$$

$$\sigma = 1.713 \times 10^{-9} \text{ Btu/hr-ft}^2\text{-}^{\circ}\text{R}^4 = 5.6693 \times 10^{-12} \text{ W/cm}^2\text{-K}^4$$

$$\sigma_{\text{exp}} = 1.731 \times 10^{-9} \text{ Btu/hr-ft}^2\text{-}^{\circ}\text{R}^4 = 5.729 \times 10^{-12} \text{ W/cm}^2\text{-K}^4$$

Program Description II

9

Sketch(es)

Sample Problem(s) Example 1:

What percentage of the radiant output of a lamp is in the visible range (0.4 to 0.7 microns) if the filament of the lamp is assumed to be a black body at 2400 K? What is the percentage at 2500 K?

Keystrokes:

[f] [B]-----
2400 [A] .4 [B] .7 [f] [E] [C] [÷] 100 [x]-----
2500 [A] .7 [f] [E] [C] [÷] 100 [x]-----

Outputs:

$5.669 \times 10^{-12} \text{ W/cm}^2\text{-K}^4$
2.641%
3.337%

Example 2:

If the human eye was designed to work most efficiently in sunlight and the visible spectrum runs from about 0.4 to 0.7 microns, what is the sun's temperature in degrees Rankine? Assume that the sun is a black body. Using the temperature calculated, find the fraction of the sun's total emissive power which falls in the visible range. Find the percentage of the sun's radiation which has a wavelength less than 0.4 microns.

Keystrokes:

[f] [A]-----
Compute mean of visible range.
.4 [+] .7 [+] 2 [÷]-----
Compute temperature of sun.
[B]-----

Outputs:

$1.713 \times 10^{-9} \text{ Btu/hr-ft}^2\text{-}^\circ\text{R}^4$
 $550.0 \times 10^{-3} \mu\text{m}$
 $9.484 \times 10^3 \text{ }^\circ\text{R}$

(continued)

Reference(s)

Robert Siegel and John R. Howell, *Thermal Radiation Heat Transfer*,
Volume 1, National Aeronautics and Space Administration, 1968.

Compute percentage of power in visible range.

[A] .4 [B] .7 [E] [C] [÷] 100 [x]-----→ $33.70 \times 10^0 \%$

Compute percentage of power under 0.4 microns.

[E] [C] [÷] 100 [x]-----→ **8.433%**

11

[illegible]

STEP	KEY ENTRY	KEY CODE	COMMENTS	STEP	KEY ENTRY	KEY CODE	COMMENTS												
001	*LBLc	21 16 11	Store English constants.	057		05	Convert to experimental σ .												
002	.	01		058	.	-62													
003	0	08		059	6	06													
004	0	08		060	5	06													
005	0	08		061	9	09													
006	7	07		062	3	03													
007	3	09		063	EEX	-23													
008	0	08		064	CHS	-22													
009	2	02		065	1	01													
010	ST01	35 01		066	2	02													
011	2	02		067	ST04	35 04													
012	5	05		068	RTN	24													
013	0	08		069	*LBLc	21 16 13	Store T and calculate λ_{\max} .												
014	0	09		070	1	01													
015	0	08		071	.	-62													
016	.	-62		072	0	00													
017	4	04		073	1	01													
018	ST02	35 02		074	0	00													
019	5	05		075	5	05													
020	2	02		076	ST*4	35-35 04													
021	1	01		077	RCL4	36 04													
022	6	06		078	RTN	24													
023	ST03	35 03		079	*LBLA	21 11	Store λ and calculate T for which λ would be λ_{\max} .												
024	.	-62		080	ST05	35 05													
025	1	01		081	RCL3	36 03													
026	7	07		082	X*Y	-41													
027	1	01		083	÷	-24													
028	3	03		084	RTN	24													
029	1	01		085	*LBLB	21 12													
030	2	02		086	ST06	35 06													
031	EEX	-23	087	RCL3	36 03														
032	CHS	-22	088	X*Y	-41														
033	0	08	089	÷	-24														
034	ST04	35 04	090	RTN	24														
035	RTN	24	091	*LBLC	21 13	Calculate $E_{b(0-\infty)}$.													
036	*LBLb	21 16 12	092	RCL5	36 05														
037	5	05	093	Y ²	53														
038	9	09	094	X ²	53														
039	5	05	095	RCL4	36 04														
040	4	04	096	X	-35														
041	.	-62	097	RTN	24														
042	4	04	098	*LBLD	21 14														
043	ST01	35 01	099	RCL1	36 01														
044	1	01	100	ENT†	-21														
045	4	04	101	+	-55	Calculate $E_{b\lambda}$.													
046	3	03	102	Pi	16-24														
047	0	08	103	X	-35														
048	0	08	104	RCL6	36 06														
049	ST02	35 02	105	5	05														
050	2	02	106	Y*	31														
051	0	08	107	÷	-24														
052	9	09	108	RCL2	36 02														
053	7	07	109	RCL6	36 06														
054	.	-62	110	÷	-24														
055	0	08	111	RCL5	36 05														
056	ST03	35 03	112	÷	-24														
REGISTERS																			
0	λ	1	c_1	2	c_2	3	c_3	4	σ	5	T	6	λ, λ'	7	sum	8	kc_2/T	9	
S0		S1		S2		S3		S4		S5		S6		S7		S8		S9	
A		B		C		D		E		F		G		H		I		J	

97 Program Listing II

13

STEP	KEY ENTRY	KEY CODE	COMMENTS	STEP	KEY ENTRY	KEY CODE	COMMENTS
113	e ^x	33	Calculate E _{b(0-λ)} .	169	XZYZ	16-35	Calculate E _{b(λ-λ')} .
114	1	01		170	GT01	22 01	
115	-	-45		171	R↓	-31	
116	÷	-24		172	CLX	-51	
117	RTN	24		173	RCL7	36 07	
118	*LBL6	21 15		174	ENT↑	-21	
119	0	00		175	+	-55	
120	ST08	35 08		176	Pi	16-24	
121	ST07	35 07		177	×	-35	
122	*LBL1	21 01		178	RCL1	36 01	
123	R↓	-31		179	×	-35	
124	CLX	-51		180	RTN	24	
125	RCL8	36 08		181	*LBL6	21 16 15	
126	RCL2	36 02		182	ENT↑	-21	
127	RCL5	36 05		183	ENT↑	-21	
128	÷	-24		184	GSPE	23 15	
129	-	-45		185	XZY	-41	
130	ST08	35 08		186	RCL6	36 06	
131	3	03		187	ST00	35 00	
132	XZY	-41		188	R↓	-31	
133	÷	-24	189	ST06	35 06		
134	RCL6	36 06	190	GSBE	23 15		
135	X ²	53	191	-	-45		
136	÷	-24	192	ABS	16 31		
137	LSTX	16-63	193	RCL0	36 00		
138	1/X	52	194	ST06	35 06		
139	RCL6	36 06	195	R↓	-31		
140	÷	-24	196	RTN	24		
141	-	-45					
142	6	06					
143	RCL6	36 06					
144	÷	-24	200				
145	RCL8	36 08					
146	X ²	53					
147	÷	-24					
148	-	-45					
149	6	06					
150	RCL8	36 08					
151	X ²	53					
152	÷	-24					
153	RCL8	36 08					
154	÷	-24	210				
155	+	-55					
156	RCL8	36 08					
157	RCL6	36 06					
158	÷	-24					
159	e ^x	33					
160	×	-35					
161	RCL8	36 08					
162	÷	-24					
163	ST+7	35-55 07					
164	RCL7	36 07	220				
165	÷	-24					
166	EE%	-23					
167	DHS	-22					
168	5	05					

LABELS					FLAGS	SET STATUS										
A	T→λ _{max}	B	λ→T(λ _{max})	C	→E _{b(0-∞)}	D	→E _{b λ}	E	→E _{b(0-λ)}	0		FLAGS		TRIG	DISP	
a	Eng	b	SI	c	Exp σ	d		e	λ'→E _{b(λ-λ')}	1		ON OFF		DEG	FIX	
0		1	E _{b(0-λ)}	2		3		4		2		0	<input checked="" type="checkbox"/>	<input checked="" type="checkbox"/>	<input type="checkbox"/>	
5		6		7		8		9		3		1	<input type="checkbox"/>	<input checked="" type="checkbox"/>	GRAD	SCI
												2	<input type="checkbox"/>	<input checked="" type="checkbox"/>	RAD	ENG
												3	<input type="checkbox"/>	<input checked="" type="checkbox"/>		n

Program Description I

Program Title Economic Insulation Thickness
Contributor's Name Hewlett-Packard
Address 1000 N.E. Circle Blvd.
City Corvallis **State** Oregon **Zip Code** 97330

Program Description, Equations, Variables

$$I = 3.46 \times 10^{-3} \sqrt{Y(\Delta T) M k/b} - 6k$$

Where:

I = thickness of insulation in inches

Y = hours per year

k = conductivity of insulation BTU/ft²°F/ft.

ΔT = temperature difference, °F

M = cost of energy \$ per 10⁶ BTU

b = cost of insulation \$ per ft² per in. thickness

Operating Limits and Warnings

Insulation is assumed to be protected from moisture saturation possibilities.

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Program Description II

Sketch(es)

Sample Problem(s)

A. What thickness of insulation is economic for a prefab wall used in a structure with the following conditions?

$$k = 0.15, \Delta T = 72^\circ - 32^\circ F$$

$$Y = 24 \text{ hrs per day per year}(8760),$$

$$M = \$1.00 \text{ per } 10^6 \text{ BTU}$$

$$b = \$0.20 \text{ per sq. ft. per inch thickness}$$

B. What if the energy price is \$2.50/million BTU?

Solution(s)

$$0.15 [k] 8760 [Y] 40 [C] 1 [D] 0.2 [E] [A] [R/S] \text{ -----} > 0.87 \text{ inches}$$

A Ans. 0.87 inches

$$2.5 [D] [A] [R/S] \text{ -----} > 1.90$$

B Ans. 1.90 inches

Reference(s) Mechanical Engineers Handbook, L. Marks, McGraw-Hill 1941, pg 404.

This program is a translation of the HP-65 Users' Program #01621A submitted by John R. Feemster.

[illegible]

97 Program Listing I

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STEP	KEY ENTRY	KEY CODE	COMMENTS	STEP	KEY ENTRY	KEY CODE	COMMENTS
001	*LBLA	21 11		057	=	-24	
002	ST01	35 01		058	ST04	35 04	
003	3	03		059	RTN	24	
004	.	-62		060	*LBLB	21 15	
005	4	04	Const. 3.46×10^{-3}	061	ST07	35 07	
006	6	06		062	R/S	51	
007	EEY	-23		063	RCL2	36 02	
008	3	03		064	RCL5	36 05	
009	CHS	-22		065	X	-35	
010	ST06	35 06		066	RCL4	36 04	
011	RCL1	36 01		067	X	-35	
012	R/S	51	Calculate I	068	RCL3	36 03	
013	RCL2	36 02		069	X	-35	
014	RCL3	36 03		070	RCL1	36 01	
015	X	-35		071	RCL5	36 05	
016	RCL4	36 04		072	6	06	
017	X	-35		073	X	-35	
018	RCL5	36 05		074	+	-55	
019	X	-35		075	RCL6	36 06	
020	RCL7	36 07		076	=	-24	
021	=	-24		077	X ²	53	
022	YX	54		078	=	-24	
023	RCL6	36 06		079	ST07	35 07	
024	X	-35		080	RTN	24	
025	6	06					
026	RCL5	36 05					
027	X	-35					
028	-	-45					
029	ST01	35 01					
030	RTN	24					
031	*LBLB	21 12					
032	ST02	35 02					
033	R↓	-31					
034	ST05	35 05					
035	RTN	24		090			
036	*LBLC	21 13					
037	ST03	35 03					
038	RTN	24					
039	*LBLD	21 14					
040	ST04	35 04					
041	R/S	51					
042	RCL1	36 01					
043	RCL5	36 05					
044	6	06					
045	X	-35		100			
046	+	-55					
047	RCL6	36 06					
048	=	-24					
049	X ²	53					
050	RCL2	36 02					
051	RCL5	36 05					
052	X	-35					
053	RCL3	36 03					
054	X	-35		110			
055	RCL7	36 07					
056	=	-24					

SET STATUS

FLAGS		TRIG	DISP
ON	OFF		
0	<input checked="" type="checkbox"/>	DEG <input checked="" type="checkbox"/>	FIX <input checked="" type="checkbox"/>
1	<input type="checkbox"/>	GRAD <input type="checkbox"/>	SCI <input type="checkbox"/>
2	<input checked="" type="checkbox"/>	RAD <input type="checkbox"/>	ENG <input type="checkbox"/>
3	<input checked="" type="checkbox"/>		n <u>2</u>

REGISTERS

0	1 Ins.	2 hrs/yr	3 ΔT	4 M	5 k	6 Const	7 b	8	9
S0	S1	S2	S3	S4	S5	S6	S7	S8	S9
A	B	C	D	E	I				

Program Description I

Program Title Heat Transfer Through Composite Cylinders and Walls

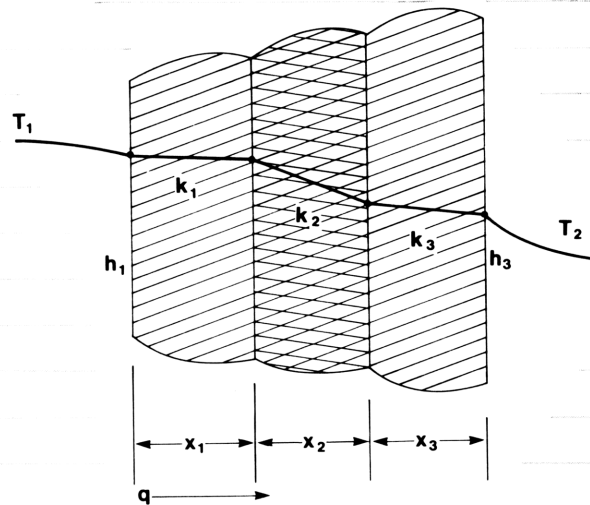
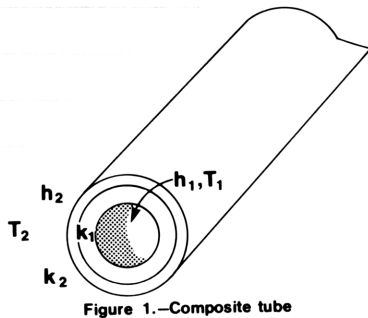
Contributor's Name Hewlett-Packard

Address 1000 N.E. Circle Blvd.

City Corvallis **State** Oregon **Zip Code** 97330

Program Description, Equations, Variables

This program can be used to calculate the overall heat transfer coefficient for composite tubes and walls from individual section conductances and surface coefficients.



Operating Limits and Warnings

These equations are for steady state heat transfer through materials with constant properties in all directions.

Inputs must start with the inside convective coefficient and work out in the case of composite cylinders.

Zero is an invalid input for D , k , and h .

Dimensional consistency must be maintained.

This program has been verified only with respect to the numerical example given in *Program Description II*. User accepts and uses this program material AT HIS OWN RISK, in reliance solely upon his own inspection of the program material and without reliance upon any representation or description concerning the program material.

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Program Description I

Program Title Heat Transfer Through Composite Cylinders and Walls

Contributor's Name Hewlett-Packard

Address 1000 N.E. Circle Blvd.

City Corvallis

State Oregon

Zip Code 97330

Program Description, Equations, Variables

The overall heat transfer coefficient U is defined by:

$$q/L = U \Delta T$$

or

$$q/A = U \Delta T$$

where ΔT is the total temperature difference ($T_2 - T_1$), q/L is the heat transfer per unit length of pipe, and q/A is the heat transfer per unit area of wall.

For cylinders

$$U = \frac{2\pi}{\frac{2}{h_1 D_1} + \frac{\ln D_2/D_1}{k_1} + \frac{\ln D_3/D_2}{k_2} + \dots + \frac{2}{h_n D_n}}$$

For walls

$$U = \frac{1}{\frac{1}{h_1} + \frac{x_1}{k_1} + \frac{x_2}{k_2} + \dots + \frac{1}{h_n}}$$

where

h is the convective surface coefficient;

D_n is the outside diameter of the annulus;

k is the conductive coefficient;

x is the thickness of a wall section.

This program has been verified only with respect to the numerical example given in *Program Description II*. User accepts and uses this program material AT HIS OWN RISK, in reliance solely upon his own inspection of the program material and without reliance upon any representation or description concerning the program material.

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Program Description II

Sketch(es)

Sample Problem(s)

Example 1:

A steel pipe with an inside diameter of 4 inches and a thickness of 0.5 inches has a conductivity of 25 Btu/ft-hr-°F. Two inches of asbestos (k = 0.1 Btu/hr-ft-°F) enclose the pipe bringing the total diameter to 9 inches. If the inside convective coefficient is 1000 Btu/hr-ft²-°F and the outside coefficient is 5 Btu/hr-ft²-°F, what is the overall heat transfer coefficient? What is the heat loss for 100 feet of pipe if ΔT is 115°F?

Keystrokes

See Displayed

4 [↑] 12 [÷] 1000 [A] 5 [↑] 12 [÷] 25 [B] 9 [↑] 12 [÷]
0.1 [B] 9 [↑] 12 [÷] 5 [A] [C] → 0.98
Btu/hr-ft-°F
115 [X] → 112.44
Btu/hr-ft
100 [X] → 11244.20
Btu/hr

Solution(s)

Example 2:

A wall is composed of 1 foot of brick (k = 0.4 Btu/hr-ft-°F), and 1 inch of wood (k = 0.12 Btu/hr-ft-°F). The convective coefficient on one side is 23 Btu/hr-ft²-°F. The convective coefficient of the other side is 5 Btu/hr-ft²-°F. What is the overall coefficient? What is the heat flux if the temperature difference is 70°F?

Keystrokes

See Displayed





[RTN] 1 [↑] 0.4 [E] 1 [↑] 12 [÷] .12 [E] 23 [D] 5 [D] [C] → 0.29
Btu/ft²-hr-°F
70 [X] → 20.36
Btu/ft²-hr

Reference (s)

User Instructions

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COMPOSITE CYLINDERS AND WALLS

 RTN=START
  D \leftrightarrow k
  \rightarrow U
 h
  x \leftrightarrow k

STEP	INSTRUCTIONS	INPUT DATA/UNITS	KEYS	OUTPUT DATA/UNITS
1	Enter program		<input type="text"/> <input type="text"/>	
2	For a composite wall go to		<input type="text"/> <input type="text"/>	
	step 9.		<input type="text"/> <input type="text"/>	
3	Input the inner diameter	D_{in}	<input type="text"/> \uparrow <input type="text"/>	D_{in}
4	Input the inner convective		<input type="text"/> <input type="text"/>	
	coefficient	h_{in}	<input type="text"/> A <input type="text"/>	$2/hD$
5	Input next diameter value	D	<input type="text"/> \uparrow <input type="text"/>	D
	and corresponding coefficient	k or h	<input type="text"/> B <input type="text"/>	
6	Go to step 5 for next surface		<input type="text"/> <input type="text"/>	
	or go to step 3 for outside		<input type="text"/> <input type="text"/>	
	surface*		<input type="text"/> <input type="text"/>	
7	Calculate overall heat transfer		<input type="text"/> <input type="text"/>	
	coefficient		<input type="text"/> C <input type="text"/>	U
8	To calculate another overall		<input type="text"/> <input type="text"/>	
	coefficient, go to step 2		<input type="text"/> <input type="text"/>	
9	Input the coefficients for each		<input type="text"/> <input type="text"/>	
	section of the wall:		<input type="text"/> <input type="text"/>	
	Convective coefficient	h	<input type="text"/> D <input type="text"/>	$1/h$
	or length of conductive path	x	<input type="text"/> \uparrow <input type="text"/>	
	and conductive coefficient	k	<input type="text"/> E <input type="text"/>	x/k
10	Go to step 9 for next input*		<input type="text"/> <input type="text"/>	
11	Calculate overall heat transfer		<input type="text"/> <input type="text"/>	
	coefficient		<input type="text"/> C <input type="text"/>	U
12	To calculate another overall		<input type="text"/> <input type="text"/>	
	coefficient, go to step 2		<input type="text"/> <input type="text"/>	

* Press **RTN** to restart a calculation.

Program Listing I

STEP	KEY ENTRY	KEY CODE	COMMENTS	STEP	KEY ENTRY	KEY CODE	COMMENTS
001	*LBLA	21 11	Initialize				
002	Pi	16-24					
003	ST06	35 06		060			
004	CLX	-51					
005	ST08	35 08					
006	R↓	-31					
007	X≠Y	-41					
008	ST07	35 07	Idle				
009	X≠Y	-41					
010	GTOA	22 11					
011	*LBL1	21 01					
012	RTN	24	Add convective factor	070			
013	*LBLA	21 11					
014	X	-35					
015	1/X	52					
016	ST+8	35-55 08					
017	RTN	24					
018	*LBLB	21 12					
019	1/X	52	Add conductive factor	080			
020	X≠Y	-41					
021	RCL7	36 07					
022	X≠Y	-41					
023	ST07	35 07					
024	=	-24					
025	LN	32					
026	X	-35	Calculate U				
027	2	02					
028	=	-24					
029	ST-8	35-45 08					
030	GTO1	22 01					
031	*LBLC	21 13					
032	RCL8	36 08					
033	1/X	52	Add convective factors	090			
034	RCL6	36 06					
035	X	-35					
036	ST04	35 04					
037	RTN	24					
038	*LBLD	21 14					
039	1	01					
040	X≠Y	-41	Add conductive factors	100			
041	*LBLE	21 15					
042	1	01					
043	ST06	35 06					
044	CLX	-51					
045	ST08	35 08					
046	R↓	-31					
047	GTOE	22 15					
048	*LBL2	21 02					
049	RTN	24					
050	*LBLE	21 15					
051	X≠Y	-41					
052	=	-24					
053	*LBLD	21 14					
054	1/X	52		110			
055	ST+8	35-55 08					
056	GTO2	22 02					

REGISTERS									
0	1	2	3	4 U	5	6 1 or π	7 Used	8 ΣR	9
S0	S1	S2	S3	S4	S5	S6	S7	S8	S9
A	B	C	D	E	I				

FLAGS		SET STATUS		
		FLAGS	TRIG	DISP
0		ON OFF	DEG <input checked="" type="checkbox"/>	FIX <input checked="" type="checkbox"/>
1		0 <input type="checkbox"/> 1 <input checked="" type="checkbox"/>	GRAD <input type="checkbox"/>	SCI <input type="checkbox"/>
2		2 <input type="checkbox"/> 3 <input checked="" type="checkbox"/>	RAD <input type="checkbox"/>	ENG <input type="checkbox"/>
3		3 <input type="checkbox"/>		n <u>2</u>

Program Description I

Program Title Steady State Conductive Heat Transfer, Heat Load and Logarithmic Mean Temperature Difference

Contributor's Name Hewlett-Packard

Address 1000 N.E. Circle Blvd.

City Corvallis **State** Oregon **Zip Code** 97330

Program Description, Equations, Variables

Given any Three variables

$$(Q, U, A\Delta t_m) \text{ OR } (Q, W, C_p \Delta t)$$

The Program Computes the Fourth Variables:

$$Q = UA\Delta t_m, \quad U = \frac{Q}{A\Delta t_m}, \dots \text{ etc.}$$

$$Q = WC_p\Delta t, \quad C_p = \frac{Q}{W\Delta t}, \dots \text{ etc}$$

Given Temperature Conditions

$$(T_1, T_2, t_1 \text{ \& } t_2), (t_1 \text{ \& } t_2) \text{ or } (T_1 \text{ \& } T_2)$$

The Program Computes:

$$\text{OR } \Delta t_m = \frac{\Delta_2 - \Delta_1}{\ln(\Delta_2 / \Delta_1)}$$

$$\Delta t = (t_2 - t_1), (T_2 - T_1).$$

To combine these three basic heat transfer equations will increase the flexibility and speed of heat transfer design.

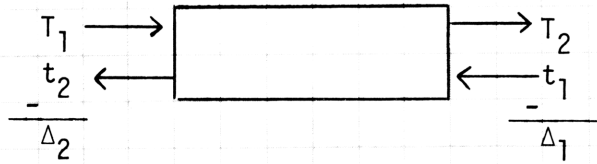
Operating Limits and Warnings

This program has been verified only with respect to the numerical example given in *Program Description II*. User accepts and uses this program material AT HIS OWN RISK, in reliance solely upon his own inspection of the program material and without reliance upon any representation or description concerning the program material.

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Program Description II

Sketch(es)



Sample Problem(s) Determine U of an existing exchanger for the following data
(Represented by the Above Sketch):

$W = 247,000$ lb per hr	Flow Rate
$C_p = 0.585$ Btu/(lb)(°F)	Heat Capacity of the Fluid
$T_1 = 410^\circ\text{F}$	Hot Inlet Temperature
$T_2 = 150^\circ\text{F}$	Hot Outlet Temperature
$t_1 = 100^\circ\text{F}$	Cold Inlet Temperature
$t_2 = 347.5^\circ\text{F}$	Cold Outlet Temperature
$A = 9390$ sq ft	Heat Transfer Area

Solution(s)

$Q = 37,568,700$ Btu/hr	Duty
$\Delta t_m = 56^\circ\text{F}$	Mean Temperature Difference
$U = 71.42$ Btu/(hr)(°F)(sq ft)	Heat Transfer coefficient

Keystrokes:

247000[B] 0.585[C] 410[ENT+] 150[—] [D] 0[A]
410[ENT+] 347.5[ENT+] 150[ENT+] 100[E]
[D] 9390[C] 0[B]

Output:

37568700(Q)
56.02 (Δt_m)
71.42 (U)

Reference(s) McAdams, W.H., Heat Transmission, McGraw-Hill Book Co.

This program is a translation of the HP-65 Users' Library program #00648A submitted by Yu Tsung Pet.

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[illegible]

97 Program Listing I

STEP	KEY ENTRY	KEY CODE	COMMENTS	STEP	KEY ENTRY	KEY CODE	COMMENTS		
001	*LBLA	21 11	Compute the Q	057	*LBL0	21 00	$T_1 - t_2 = \Delta_2$ $T_2 - t_1 = \Delta_1$ $\Delta_2 = \Delta_1 = \Delta t_m$		
002	0	00		058	-	-45			
003	X \neq Y	-41		059	R \uparrow	16-31			
004	X \neq Y?	16-32		060	X=Y?	16-33			
005	GT01	22 01		061	R/S	51			
006	RCL2	36 02		062	X \neq Y?	16-35			
007	RCL3	36 03		063	X \neq Y	-41			
008	RCL4	36 04		064	ST05	35 05			
009	X	-35		065	ST06	35 06			
010	X	-35		066	R \downarrow	-31			
011	ST01	35 01	Compute the U or W	067	ST-5	35-45 05			
012	RTN	24		068	ST \div 6	35-24 06			
013	*LBLB	21 12		069	RCL5	36 05			
014	0	00		070	RCL6	36 06			
015	X \neq Y	-41		071	LN	32			
016	X \neq Y?	16-32		072	\div	-24			
017	GT02	22 02		073	ST04	35 04			
018	RCL1	36 01		074	RTN	24			
019	RCL3	36 03		075	*LBL1	21 01			
020	RCL4	36 04		076	ST01	35 01			
021	X	-35	Compute the A or C	077	R/S	51			
022	\div	-24		078	*LBL2	21 02			
023	ST02	35 02		079	ST02	35 02			
024	RTN	24		080	R/S	51			
025	*LBLC	21 13		081	*LBL3	21 03			
026	0	00		082	ST03	35 03			
027	X \neq Y	-41		083	R/S	51			
028	X \neq Y?	16-32		084	*LBL4	21 04			
029	GT03	22 03		085	ST04	35 04			
030	RCL1	36 01		086	R/S	51			
031	RCL2	36 02							
032	RCL4	36 04							
033	X	-35							
034	\div	-24	090						
035	ST03	35 03							
036	RTN	24							
037	*LBLD	21 14							
038	0	00							
039	X \neq Y	-41							
040	X \neq Y?	16-32							
041	GT04	22 04							
042	RCL1	36 01							
043	RCL2	36 02							
044	RCL3	36 03							
045	X	-35	100						
046	\div	-24							
047	ST04	35 04							
048	RTN	24							
049	*LBLE	21 15							
050	-	-45							
051	R \downarrow	-31	Compute Δt_m or Δt from T_1, T_2, t_1 or t_2						
052	X \neq Y?	16-32							
053	GT00	22 00							
054	R \uparrow	16-31		110					
055	ST04	35 04							
056	RTN	24							
SET STATUS									
FLAGS				TRIG		DISP			
ON OFF									
0 <input type="checkbox"/> <input checked="" type="checkbox"/>				DEG <input checked="" type="checkbox"/>		FIX <input checked="" type="checkbox"/>			
1 <input type="checkbox"/> <input checked="" type="checkbox"/>				GRAD <input type="checkbox"/>		SCI <input type="checkbox"/>			
2 <input type="checkbox"/> <input checked="" type="checkbox"/>				RAD <input type="checkbox"/>		ENG <input type="checkbox"/>			
3 <input type="checkbox"/> <input checked="" type="checkbox"/>						n <u>2</u>			
REGISTERS									
0	1 Q	2 U, W	3 A, C _p	4 $\Delta t_m, \Delta t$	5 Δ_1	6 Δ_2	7	8	9
S0	S1	S2	S3	S4	S5	S6	S7	S8	S9
A		B		C		D		E	

Program Description I

Program Title Sun Altitude, Azimuth, Solar Pond Absorption
Contributor's Name Hewlett-Packard
Address 1000 N.E. Circle Blvd.
City Corvallis **State** Oregon **Zip Code** 97330

Program Description, Equations, Variables

Given: Index of refraction of pond fluid; latitude; number of days after spring equinox, and number of hours before or after solar noon;

the program computes: the sun's altitude h.

$$h = \sin^{-1}(\cos l \cos d \cos t + \sin l \sin d)$$

l - latitude in decimal degrees

d = sun's declination = $23.45 \sin D$

D = (no. of days after spring equinox)(0.9856 $\frac{\text{degrees}}{\text{day}}$)

t = (no. of hours before or after solar noon);

the sun's azimuth A.

$$A = \cos^{-1}\left(\frac{\cos i \sin l - \sin d}{\cos l \sin i}\right)$$

$i = 90 - h$;

the fraction of solar radiation striking the pond surface
which will penetrate the pond surface, E,

$$\text{Fraction } E = 2n(a^2 + b^2) \cos i \cos r$$

$$a = \frac{1}{\cos r + n \cos i} \text{ where } r = \sin^{-1}\left(\frac{\sin i}{n}\right)$$

$$b = \frac{1}{\cos i + n \cos r}$$

n = index of refraction of pond fluid

(refs: Smithsonian Physical Tables, 9th rev. Ed. & Weinberger, H., Solar energy, v8, n2,
1954 (p 729) 1964 (pp 45-56)

OPERATING LIMITS AND WARNINGS

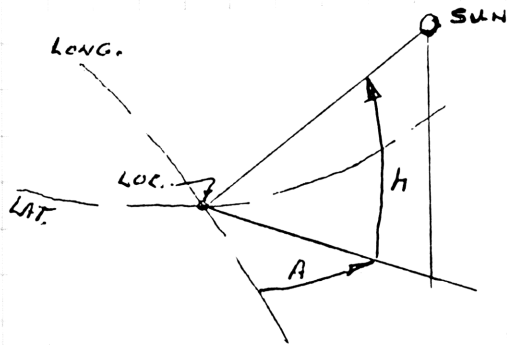
Does not compute azimuth at latitude of 90 degrees.

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Program Description II

Sketch(es)



Sample Problem(s)

Find the sun's altitude, azimuth, and the fraction of the sun's radiation which will penetrate the surface of a solar pond under the following circumstances:

Index of refraction of pond fluid $n = 1.33$

Latitude $l = 46.00$

Days after spring equinox = 68

Hours before solar noon = 4

Solution(s)

$h = 35.99$ degrees

$A = 84.41$ degrees

$E = 0.96$

Keystrokes:

23.45[STO][1] .9856[STO][2] 1.33[STO][3]

46[A] 68[B] 4[C] ----->

[R/S] ----->

[R/S] ----->

Outputs:

35.99

84.41

0.96

Reference(s)

Smithsonian Physical Tables, 9th rev. Ed., 1954, (p 729)

Weinberger, H., Solar Energy, vol 8, no. 2, 1964 (pp 45-56)

This program is a translation of the HP-65 Users' Library program #00683A submitted by Robert J. Zaworski.

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[illegible]

97 Program Listing I

STEP	KEY ENTRY	KEY CODE	COMMENTS	STEP	KEY ENTRY	KEY CODE	COMMENTS
001	*LBLA	21 11		057	GSBE	23 15	
002	COS	42		058	RCL6	36 06	
003	ST04	35 04	cos l in 4	059	RCL7	36 07	
004	LSTX	16-63		060	RCL3	36 03	
005	RTN	24	stops with latitude	061	GSBE	23 15	
006	*LBLB	21 12	in x	062	+	-55	
007	RCL2	36 02		063	RCL3	36 03	
008	X	-35		064	X	-35	
009	SIN	41		065	2	02	
010	RCL1	36 01		066	X	-35	
011	X	-35		067	RCL6	36 06	
012	ST05	35 05	d in 5	068	X	-35	
013	RTN	24	stops with decl.in x	069	RCL7	36 07	
014	*LBLC	21 13		070	X	-35	
015	1	01		071	RTN	24	
016	5	05		072	*LBLD	21 14	
017	X	-35		073	COS'	16 42	
018	COS	42		074	SIN	41	
019	ST08	35 08		075	RTN	24	
020	RCL4	36 04		076	*LBLE	21 15	
021	X	-35		077	X	-35	
022	RCL5	36 05		078	+	-55	
023	COS	42		079	ENT1	-21	
024	X	-35		080	X	-35	
025	RCL4	36 04		081	1/X	52	
026	GSBD	23 14		082	RTN	24	
027	RCL5	36 05		083	R/S	51	
028	SIN	41					
029	ST07	35 07					
030	X	-35					
031	+	-55					
032	ST06	35 06					
033	SIN'	16 41	Stops with alt. in x	090			
034	R/S	51					
035	RCL6	36 06					
036	RCL4	36 04					
037	GSBD	23 14					
038	X	-35					
039	RCL7	36 07					
040	-	-45					
041	RCL4	36 04					
042	=	-24					
043	RCL6	36 06					
044	GSBD	23 14		100			
045	=	-24					
046	COS'	16 42	Stops with azimuth				
047	R/S	51	in x				
048	RCL6	36 06					
049	GSBD	23 14					
050	RCL3	36 03					
051	=	-24					
052	SIN'	16 41					
053	COS	42					
054	ST07	35 07	cos r in 7	110			
055	RCL6	36 06					
056	RCL3	36 03					

SET STATUS			
FLAGS	TRIG	DISP	
ON OFF		DEG	<input checked="" type="checkbox"/>
0 <input type="checkbox"/> <input checked="" type="checkbox"/>		SCI	<input type="checkbox"/>
1 <input type="checkbox"/> <input checked="" type="checkbox"/>	GRAD	ENG	<input type="checkbox"/>
2 <input type="checkbox"/> <input checked="" type="checkbox"/>	RAD	n	2
3 <input type="checkbox"/> <input checked="" type="checkbox"/>			

REGISTERS								
0	1 23.45	2 20.9856 deg/day	3 n (index of refrac	4 cos l	5 d declination	6 cos i	7 sin d/cos r	8 cos t
S0	S1	S2	S3	S4	S5	S6	S7	S8
A	B	C	D	E	I			

Program Description I

Program Title Total Daily Amount of Solar Radiation

Contributor's Name Hewlett-Packard

Address 1000 N.E. Circle Blvd.

City Corvallis

State Oregon

Zip Code 9733

Program Description, Equations, Variables This program determines the total amount of solar radiation received by a horizonatal surface of unit area during one calendar day. The result is expressed as equivalent hours of direct sunshine if sun were stationary and directly overhead. Also computes length of daylight and accumulates total radiation in R7 for successive calculations. Input variables are latitude, L, suns declination (from nautical almanac) in decimal degrees.

Day Length = $24 \theta / \pi$, θ expressed in radians

$\theta = \text{Arc cos } (-\sin L \sin D / \cos L \cos D)$

Total Radiation = $2 \int_0^{\theta} \sin H d\theta$

= $(\sin L \sin D) \theta + \cos L \cos D \sin \theta$

Operating Limits and Warnings

North latitudes and declinations are entered as positive values south as negative values.

The value $90-L+D$ must be greater than zero.

Equations assume surface level with horizon and ignores atmospheric refraction and assume cloudless sky.

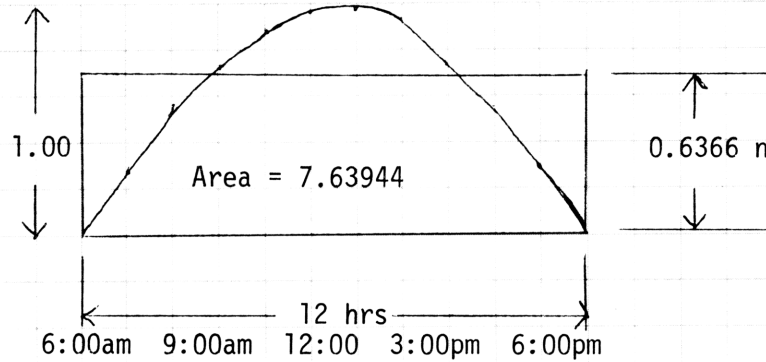
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Program Description II

Sketch(es)

Sample (1)



Sample Problem(s)

- (1) Equator, March 21 $L = 0$, $D = 0$
- (2) North Pole, June 21, $L = 90^\circ$, $D = +23.45^\circ$
- (3) Cupertino, CA 95014, September 15, 1974
 $L = 37.32^\circ$, $Dec = +2.93^\circ$

Solution(s)

- (1) DHY length = 12.00 hrs Total Rad = 7.6394 hrs
- (2) " " = 24.00 " " = 9.5508
- (3) " " = 12 hrs, 17 min, 53 sec. Total Rad = 6.4439

Keystrokes:

Outputs:

(1) 0[E] 0[A] 0[B] -----	12.0000
[C] -----	7.6394
(2) 0[E] 90[A] 23.45[B] -----	24.0000
[C] -----	9.5508
(3) 0[E] 37.32[A] 2.93[B] -----	12.1753
[C] -----	6.4439

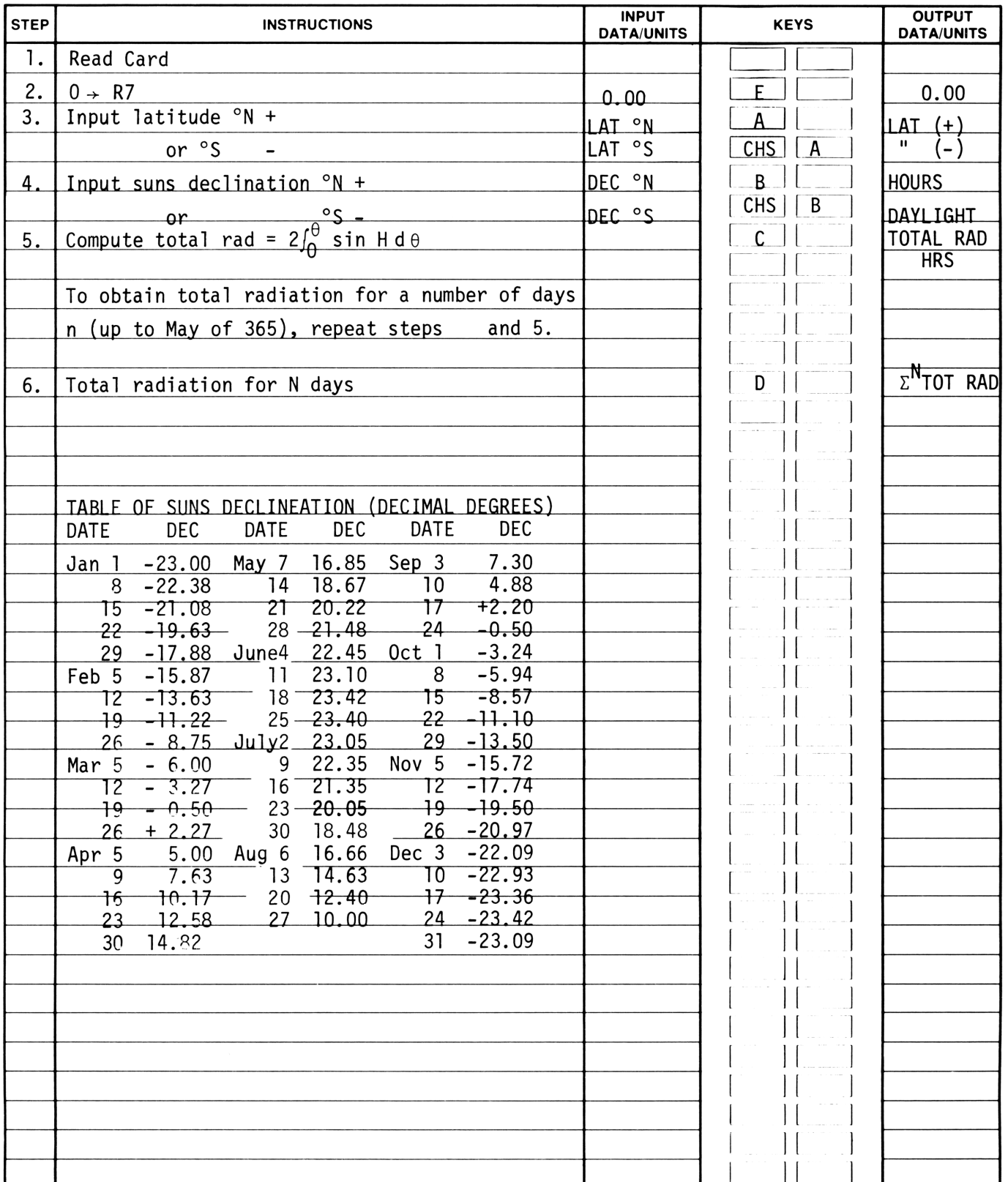
Reference(s)

- (1) The Nautical Almanac, U.S. Naval Observatory Purchase from Superintendent of Documents, Washington D.C., 20402.
- (2) American Practical Navigation, Bowditch U.S. Naval Oceanographic Office, pg 531, Also chapter XIV.
- (3) Britannica Atlas

This program is a translation of the HP-65 Users' Library program 00996A

submitted by Robert B. Egbert.

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STEP	KEY ENTRY	KEY CODE	COMMENTS	STEP	KEY ENTRY	KEY CODE	COMMENTS		
001	*LBLA	21 11	Enter latitude & store in R-1	057	ENT↑	-21	Display sums accumulated in R-7		
002	DSP4	-63 04		058	RCL7	36 07			
003	ST01	35 01		059	+	-55			
004	R/S	51		060	ST07	35 07			
005	*LBLB	21 12		061	R↓	-31			
006	ST02	35 02		062	R/S	51			
007	DEG	16-21		063	*LBLD	21 14			
008	RCL1	36 01		064	RCL7	36 07			
009	SIN	41		065	R/S	51			
010	RCL2	36 02		066	*LBL E	21 15			
011	SIN	41	Enter sums dec- lination and compute no. of hours of sunshine	067	0	00	Stores zero in R-7 for new series of calculations Limits of integra- tion for midnight s or case		
012	x	-35		068	ST07	35 07			
013	ST03	35 03		069	R/S	51			
014	RCL1	36 01		070	*LBL1	21 01			
015	COS	42		071	2	02			
016	RCL2	36 02		072	4	04			
017	COS	42		073	ST05	35 05			
018	x	-35		074	R/S	51			
019	ST04	35 04		This takes care of midnight sun					
020	X≠Y?	16-35							
021	GT01	22 01							
022	RCL3	36 03							
023	RCL4	36 04							
024	=	-24							
025	CHS	-22							
026	COS↑	16 42							
027	7	07							
028	.	-62							
029	5	05	Converts degrees (θ) to length of day in hr, min,sec						
030	=	-24							
031	ST05	35 05							
032	→HMS	16 35							
033	R/S	51							
034	*LBLC	21 13							
035	RCL5	36 05							
036	PI	16-24							
037	x	-35							
038	2	02							
039	4	04	θ is converted to radians to perform the integration & result is converted to hours						
040	=	-24							
041	ST06	35 06							
042	RAD	16-22							
043	RCL3	36 03							
044	RCL6	36 06							
045	x	-35							
046	RCL4	36 04							
047	RCL6	36 06							
048	SIN	41							
049	x	-35				SET STATUS			
050	+	-55					FLAGS TRIG DISP		
051	2	02						ON OFF DEG <input checked="" type="checkbox"/> FIX <input checked="" type="checkbox"/> 1 <input type="checkbox"/> <input checked="" type="checkbox"/> GRAD <input type="checkbox"/> SCI <input type="checkbox"/> 2 <input type="checkbox"/> <input checked="" type="checkbox"/> RAD <input type="checkbox"/> ENG <input type="checkbox"/> 3 <input type="checkbox"/> <input checked="" type="checkbox"/> n—2	
052	4	04							
053	x	-35							
054	PI	16-24							
055	=	-24							
056	ENT↑	-21							
REGISTERS									
0	1 LAT	2 DEC OF SUN	3 Sin Lx Sin D	4 cos Lx cos D	5 Σ HOURS OF SUNSHINE	6 Σ RADIANS	7 Σ sin Hdθ	8	9
S0	S1	S2	S3	S4	S5	S6	S7	S8	S9
A		B		C		D		E	

Program Description I

Program Title Temperature or Concentration Profile For A Semi-Infinite Solid

Contributor's Name Hewlett-Packard

Address 1000 N.E. Circle Blvd.

City Corvallis

State Oregon

Zip Code 97330

Program Description, Equations, Variables

Many physical situations in heat and mass transfer may be solved within engineering tolerances by assuming an infinite geometry.

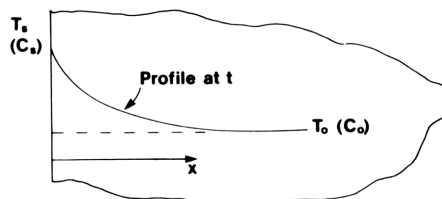


Figure 1.

In Figure 1 an infinitely thick wall initially at temperature T_0 or concentration C_0 is subject to a constant surface potential T_s or C_s . At a later time t , the internal profile will have been altered by the transport of heat or mass. This program computes values of temperature T or concentration C at time t for specified distances x from the outer surface.

Operating Limits and Warnings

This solution is exact for infinite configurations with constant cross sectional areas. However, finite geometries where the argument of the error function is greater than two will yield little or no error. This means transfer in finite bodies such as plates may be predicted until the effects of the step are felt on the far side. Also, geometries such as cylinders may be studied if the depth of penetration is small compared to the radius.

The routine used by this program will resolve error functions with arguments less than 4.5. For larger arguments, the value of the error function is set to 1.0.

This program has been verified only with respect to the numerical example given in *Program Description II*. User accepts and uses this program material AT HIS OWN RISK, in reliance solely upon his own inspection of the program material and without reliance upon any representation or description concerning the program material.

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Program Description I

Program Title Temperature or Concentration Profile For A Semi- Infinite Solid

Contributor's Name Hewlett-Packard

Address 1000 N.E. Circle Blvd.

City Corvallis

State Oregon

Zip Code 97330

Program Description, Equations, Variables

Equations:

$$T = (T_0 - T_s) \operatorname{erf} \left(\frac{x}{2 \sqrt{\frac{k}{\rho c_p} t}} \right) + T_s^*$$

where

k is thermal conductivity of the material;

ρ is the density of the material;

c_p is the specific heat of the material;

$k/\rho c_p$ is also known as the diffusivity of heat α .

Similarly, for mass transfer

$$C = (C_0 - C_s) \operatorname{erf} \left(\frac{x}{2 \sqrt{Dt}} \right) + C_s^*$$

where

D is the mass diffusivity.

*erf is the error function.

Operating Limits and Warnings

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Program Description II

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Sketch(es)

Example 1:

A large steel transmission shaft is case hardened by diffusion of carbon. The initial carbon concentration is 0.10% and the surface concentration is brought to 1.20% almost instantly. What is the carbon concentration at 1.0 mm (1×10^{-3} m) after 15 hours (54000 seconds), if the diffusivity of carbon in steel is taken to be 1.6×10^{-11} m²/s?

Keystrokes

See Displayed

1.6 **EEX** **CHS** 11 **↑** 1 **↑** 1 **A** 1.2 **↑** .1 **B** 54000
C **EEX** **CHS** 3 **D** → 0.59%

Example 2:

A furnace wall is at a constant 55°F. When the furnace is turned on the inside wall temperature is raised to 2000°F. How long will it take to raise the outside wall temperature 1°F?

$$k = 0.67 \text{ Btu/hr-ft-}^\circ\text{F}$$

$$\text{Thickness} = 1.5 \text{ feet}$$

$$c = 0.2 \text{ Btu/lb } ^\circ\text{F}$$

$$\rho = 150 \text{ lb/ft}^3$$

Keystrokes

See Displayed

An iterative solution is required since t is not a program output.
 Guess 5.0 hours for t.

.67 **↑** 150 **↑** .2 **A** 2000 **↑** 55 **B** 5 **C** 1.5 **D** → 57.92°F

Guess 4.0

Noting that x is stored in register 8.

4.0 **C** **RCL** **8** **D** → 55.75°F

Guess 4.2

4.2 **C** **RCL** **8** **D** → 56.04°F


Guess 4.18

4.18 **C** **RCL** **8** **D** → 56.01°F

Noting that t is stored in register 7.

RCL **7** **f** **→H.MS** → ≈4 hr. 10 min.

User Instructions

SEMI-INFINITE SOLID				
	$k \leftrightarrow \rho \leftrightarrow C_p$ ($D \leftrightarrow t \leftrightarrow 1$)	$T_s \leftrightarrow T_o$ ($C_s \leftrightarrow C_o$)	t	$x \leftrightarrow T(C) \quad a \leftrightarrow \text{erf}(a)$

STEP	INSTRUCTIONS	INPUT DATA/UNITS	KEYS	OUTPUT DATA/UNITS
1	Enter program		<input type="text"/> <input type="text"/>	
2	To compute the error function		<input type="text"/> <input type="text"/>	
	of an argument go to step 8.		<input type="text"/> <input type="text"/>	
3	Input:		<input type="text"/> <input type="text"/>	
	Conductivity	k	<input type="text"/> <input type="text"/>	k
	<i>then</i> density	ρ	<input type="text"/> <input type="text"/>	ρ
	<i>then</i> specific heat	C_p	<input type="text"/> <input type="text"/>	α
	<i>or</i> heat (or mass) diffusivity	α (D)	<input type="text"/> <input type="text"/>	α (D)
	<i>then</i> 1.00	1	<input type="text"/> <input type="text"/>	1.00
	<i>then</i> 1.00	1	<input type="text"/> <input type="text"/>	α (D)
4	Input:		<input type="text"/> <input type="text"/>	
	Surface temperature (con-		<input type="text"/> <input type="text"/>	
	centration)	T_s (C_s)	<input type="text"/> <input type="text"/>	T_s (C_s)
	<i>then</i> initial temperature		<input type="text"/> <input type="text"/>	
	(concentration)	T_o (C_o)	<input type="text"/> <input type="text"/>	T_s (C_s)
5	Input time	t	<input type="text"/> <input type="text"/>	t
6	Input distance from surface		<input type="text"/> <input type="text"/>	
	and calculate temperature		<input type="text"/> <input type="text"/>	
	<i>or</i> concentration	x	<input type="text"/> <input type="text"/>	T (C)
7	For new case go to step 2, 3, or		<input type="text"/> <input type="text"/>	
	4 and change inputs. For new		<input type="text"/> <input type="text"/>	
	time go to step 5. For new x go		<input type="text"/> <input type="text"/>	
	to step 6.		<input type="text"/> <input type="text"/>	
8	Input argument and compute		<input type="text"/> <input type="text"/>	
	error function	a	<input type="text"/> <input type="text"/>	erf(a)

97 Program Listing I

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STEP	KEY ENTRY	KEY CODE	COMMENTS	STEP	KEY ENTRY	KEY CODE	COMMENTS		
001	*LBLA	21 11	Store constants as α or D	057	+	-55			
002	X	-35		058	X*Y?	16-32			
003	÷	-24		059	GT01	22 01			
004	ST06	35 06		060	2	02			
005	RTN	24		061	X	-35			
006	*LBLB	21 12		062	Pi	16-24			
007	ST04	35 04		063	JX	54			
008	X*Y	-41		064	÷	-24			
009	ST05	35 05		065	RCL2	36 02			
010	RTN	24		066	2	02			
011	*LBLC	21 13	Store time	067	÷	-24			
012	ST07	35 07		068	e ^x	33			
013	RTN	24		069	÷	-24			
014	*LBLD	21 14		070	RTN	24			
015	ST08	35 08		071	*LBL0	21 00			
016	2	02		072	1	01			
017	÷	-24		073	RTN	24			
018	RCL6	36 06							
019	RCL7	36 07							
020	X	-35		Calculate temp. or concentration given x					
021	JX	54							
022	÷	-24							
023	GSBE	23 15							
024	RCL4	36 04	080						
025	RCL5	36 05							
026	-	-45							
027	X	-35							
028	RCL5	36 05							
029	+	-55							
030	RTN	24							
031	*LBL E	21 15							
032	ST01	35 01							
033	4	04							
034	.	-62		090					
035	5	05							
036	X*Y?	16-35							
037	GT06	22 00							
038	R4	-31							
039	ENT1	-21							
040	X	-35							
041	2	02							
042	X	-35							
043	ST02	35 02							
044	1	01		100					
045	ST03	35 03							
046	RCL1	36 01							
047	*LBL1	21 01							
048	RCL2	36 02							
049	RCL3	36 03							
050	2	02	Evaluate the error function						
051	+	-55							
052	ST03	35 03							
053	÷	-24							
054	RCL1	36 01		110					
055	X	-35							
056	ST01	35 01							
SET STATUS									
FLAGS				TRIG		DISP			
0	<input type="checkbox"/>	<input checked="" type="checkbox"/>	ON OFF	DEG	<input checked="" type="checkbox"/>	FIX	<input checked="" type="checkbox"/>		
1	<input type="checkbox"/>	<input checked="" type="checkbox"/>		GRAD	<input type="checkbox"/>	SCI	<input type="checkbox"/>		
2	<input type="checkbox"/>	<input checked="" type="checkbox"/>		RAD	<input type="checkbox"/>	ENG	<input type="checkbox"/>		
3	<input type="checkbox"/>	<input checked="" type="checkbox"/>				n	2		
REGISTERS									
0	1Part. Sum	2 2a ²	3 2n + 1	4 T ₀ (C ₀)	5 T _s (C _s)	6 α	7 t	8 x	9 Used
S0	S1	S2	S3	S4	S5	S6	S7	S8	S9
A	B	C	D	E	I				

Program Description I

Program Title Transient Temperature Distribution In A Semi-Infinite Solid With Convection Boundary Condition
Contributor's Name Hewlett-Packard
Address 1000 N.E. Circle Blvd.
City Corvallis **State** Oregon **Zip Code** 97330

Program Description, Equations, Variables

Given the data set:

x = Depth from surface

α = Thermal diffusivity

k = Thermal conductivity

h = Heat transfer coefficient

θ = Time

The program computes the following factor \bar{x}

$$\bar{x} = \text{ERF} \left(\frac{x}{2\sqrt{\alpha\theta}} \right) + \left[\text{EXP} \left(\frac{hx}{k} + \frac{h^2\alpha\theta}{k^2} \right) \right] \left[1 - \text{ERF} \left(\frac{x}{2\sqrt{\alpha\theta}} + \frac{h\sqrt{\alpha\theta}}{k} \right) \right]$$

where ERF = Error function

EXP = Exponential

The user must then manually compute the desired temperature $T(x, \theta)$, according to:

$$T(x, \theta) = T_{\infty} + (T_i - T_{\infty}) \bar{x}$$

where T_{∞} = Sink temperature

T_i = Initial solid temperature

Operating Limits and Warnings

This program has been verified only with respect to the numerical example given in *Program Description II*. User accepts and uses this program material AT HIS OWN RISK, in reliance solely upon his own inspection of the program material and without reliance upon any representation or description concerning the program material.

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Program Description II

Sketch(es)

Sample Problem(s)

For the data set:

$$x = 10^{-2} \text{ cm.}$$

$$\theta = 10^{-1} \text{ cm.}$$

$$\alpha = 7.141 \times 10^{-3} \text{ cm}^2 \text{ sec}^{-1}$$

$$k = 6.322 \times 10^{-3} \text{ cal cm}^{-1} \text{ sec}^{-1} \text{ } ^\circ\text{C}^{-1}$$

$$h = 6.0 \times 10^{-1} \text{ cal cm}^{-2} \text{ sec}^{-1} \text{ } ^\circ\text{C}^{-1}$$

Solution(s) The program computes the value:

$$\bar{x} = 0.3973$$

$$\text{for } T_i = 1050^\circ\text{C and } T_\infty = 450^\circ\text{C}$$

$$T(x, \theta) = T_\infty + (T_i - T_\infty) \bar{x} = 688.40^\circ\text{C}$$

Keystrokes:

Outputs:

1[EEX][CHS] 2[STO][4] 1[EEX][CHS] 1[STO][5] 7.141[EEX][CHS] 3[STO][6]

6[EEX][CHS] 1[STO] [7] 6.322[EEX][CHS] 3[STO][8]

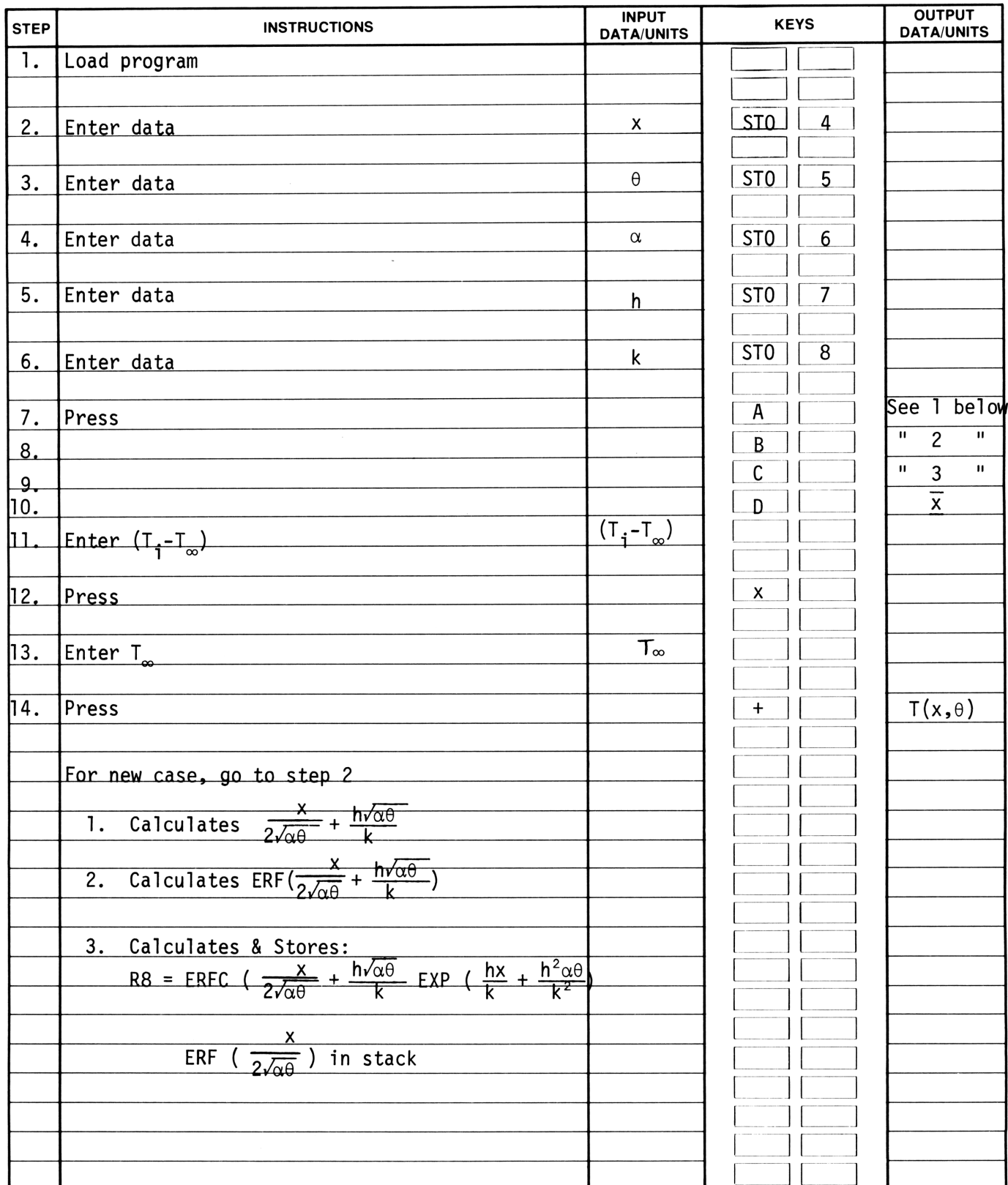
[A][B][C][D] -----> 0.3973

1050[ENT+] 450[-][x] 450[+] ----->688.40

Reference(s)

Hockman, J.P. Heat Transfer Third Edition pgs. 91-96 McGraw Hill, 1972

This program is a translation of the HP-65 Users' Library program #01472A submitted by John S. Wasylyr.



Program Listing I

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STEP	KEY ENTRY	KEY CODE	COMMENTS	STEP	KEY ENTRY	KEY CODE	COMMENTS
001	*LBLA	21 11		057	*LBLC	21 13	Calc
002	RCL8	36 08		058	1	01	$\text{ERFC}\left(\frac{x}{2\sqrt{\alpha\theta}} + \frac{h\sqrt{\alpha\theta}}{k}\right)$
003	ST=7	35-24 07	Calc (h/k) Sto R-7	059	X*Y	-41	
004	RCL6	36 06		060	-	-45	
005	STX5	35-35 05	Calc ($\alpha\theta$) Sto R-5	061	ST08	35 08	Store R-8
006	RCL4	36 04		062	RCL7	36 07	
007	RCL7	36 07		063	RCL5	36 05	
008	X	-35		064	JX	54	
009	ST06	35 06	Calc (hx/k) Sto R-6	065	X	-35	
010	RCL4	36 04		066	X ²	53	Calc $\left(\frac{hx}{k} + \frac{h^2x\theta}{k^2}\right)$
011	2	02		067	RCL6	36 06	EXP
012	=	-24		068	+	-55	
013	RCL5	36 05		069	e ^x	33	
014	JX	54		070	RCL8	36 08	Calc (EXP)(ERFC)
015	=	-24		071	X	-35	Sto R-8
016	ST04	35 04	Calc $(4/2\sqrt{\alpha\theta})$ Sto R-4	072	ST08	35 08	
017	RCL7	36 07		073	RCL4	36 04	Calc. ERF $\left(\frac{x}{2\sqrt{\alpha\theta}}\right)$
018	RCL5	36 05		074	GT08	22 12	
019	JX	54		075	RTN	24	
020	X	-35		076	*LBLD	21 14	Calc
021	+	-55	Calc $(x/2\sqrt{\alpha\theta} + h\sqrt{\alpha\theta}/k)$	077	RCL8	36 08	ERF + (EXP)(ERFC)
022	RTN	24		078	+	-55	
023	*LBLB	21 12		079	RTN	24	
024	ST01	35 01					
025	ENT↑	-21	Calc				
026	X	-35	ERF $(x/2\sqrt{\alpha\theta} + h\sqrt{\alpha\theta}/k)$				
027	2	02					
028	X	-35					
029	ST02	35 02					
030	1	01					
031	ST03	35 03					
032	RCL1	36 01					
033	*LBL1	21 01					
034	RCL2	36 02					
035	RCL3	36 03					
036	2	02					
037	+	-55					
038	ST03	35 03					
039	=	-24					
040	RCL1	36 01					
041	X	-35					
042	ST01	35 01					
043	+	-55					
044	X*Y?	16-32					
045	GT01	22 01					
046	2	02					
047	X	-35					
048	Fi	16-24					
049	JX	54					
050	RCL2	36 02					
051	2	02					
052	=	-24					
053	e ^x	33					
054	X	-35					
055	=	-24					
056	RTN	24					

REGISTERS									
0	1 Used	2 Used	3 Used	4 x	5 α	6 θ	7 h	8 k	9 Used
S0	S1	S2	S3	S4	S5	S6	S7	S8	S9
A	B	C	D	E	I				

SET STATUS					
FLAGS		TRIG		DISP	
ON	OFF	DEG	<input checked="" type="checkbox"/>	FIX	<input checked="" type="checkbox"/>
0 <input type="checkbox"/>	<input checked="" type="checkbox"/>	GRAD	<input type="checkbox"/>	SCI	<input type="checkbox"/>
1 <input type="checkbox"/>	<input checked="" type="checkbox"/>	RAD	<input type="checkbox"/>	ENG	<input type="checkbox"/>
2 <input type="checkbox"/>	<input checked="" type="checkbox"/>			n	
3 <input type="checkbox"/>	<input checked="" type="checkbox"/>				

Program Description I

Program Title	Conservation of Energy		
Contributor's Name	Hewlett-Packard		
Address	1000 N.E. Circle Blvd.		
City	Corvallis	State	Oregon
		Zip Code	97330

Program Description, Equations, Variables	
	<p>These cards convert kinetic energy, potential energy and pressure-volume work to energy. Card 1 is for English units while Card 2 is for SI or metric units. Energy is stored as a running total. When a zero is displayed, pressing the B, C, D or E keys will cause the running total to be converted to an equivalent velocity, height, pressure or energy per unit mass. The cards may be used in a large number of fluid flow problems, where velocity, elevation and pressure change along the path of flow.</p>
Operating Limits and Warnings	
	<p>Downstream values should be input as negatives. However, when an output is called for, the calculator displays the relative value with no regard to upstream or downstream location.</p> <p>Flashing zeros will result when the total energy sum stored in register 8 is negative and an attempt is made to calculate velocity.</p>

This program has been verified only with respect to the numerical example given in *Program Description II*. User accepts and uses this program material AT HIS OWN RISK, in reliance solely upon his own inspection of the program material and without reliance upon any representation or description concerning the program material.

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Program Description I

Program Title

CONSERVATION OF ENERGY

Contributor's Name Hewlett-Packard

Address 1000 N.E. Circle Blvd.

City Corvallis

State Oregon

Zip Code 97330

Program Description, Equations, Variables

$$\frac{v_1^2}{2} + gz_1 + \frac{P_1}{\rho} + \frac{E_1}{\dot{m}} = \frac{v_2^2}{2} + gz_2 + \frac{P_2}{\rho} + \frac{E_2}{\dot{m}}$$

where

v is the fluid velocity;

z is the height above a reference datum;

P is the pressure;

E is an energy term which could represent inputs of work or friction losses (negative value);

g is the acceleration of gravity;

ρ is the fluid density;

\dot{m} is the mass flow rate (assumed to be unity);

subscripts 1 and 2 refer to upstream and downstream values respectively.

Operating Limits and Warnings

This program has been verified only with respect to the numerical example given in *Program Description II*. User accepts and uses this program material AT HIS OWN RISK, in reliance solely upon his own inspection of the program material and without reliance upon any representation or description concerning the program material.

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Program Description II

Sketch(es)

Sample Problem(s)

Example 1:

A water tower is 100 feet high. What is the zero flow rate pressure at the base? The density of water is 62.4 lb/ft^3 .

Keystrokes

See Displayed

Using card 1

62.4 **A** 100 **C** **D** \longrightarrow 43.33 psig

If water is flowing out of the tower at a velocity of 10 ft/sec, what is the static pressure?

10 **CHS** **B** **D** \longrightarrow 42.66 psig

What is the maximum frictionless flow velocity which could be achieved with the 100 foot tower?

62.4 **A** 100 **C** **B** \longrightarrow 80.21 ft/sec

If 10000 pounds of water are pumped to the top of the tower every hour, at a velocity of 20 ft/sec, with a frictional pressure drop of 2 psi, how much power is needed at the pump?

62.4 **A** 20 **B** 2 **D** 100 **C** **E** \longrightarrow 0.14 Btu/lb

10000 \longrightarrow 1424.29
(Btu/hr)

Solution(s)

Reference(s)

The diagram illustrates a hydraulic jump in a channel. The flow transitions from a supercritical state (1) to a subcritical state (2). The upstream velocity is 3 m/s and the downstream velocity is 15 m/s. The channel bed has a step down of 3.7 m.

Figure 1.

735 **A** 3 **B** 3.7 **C** 15 **CHS** **B** **D** \longrightarrow -52710.82
(Nt/m²)

• -52710.82
(Nt/m^2)

Program Description II

Sketch(es)

Sample Problem(s)

Example 3:

A reservoir's level is 25 meters above the discharge pond. Assuming 85% power generation efficiency, how much power can be generated with a flow rate of 20 m³/s?

$$\rho = 1000 \text{ kg/m}^3$$

Keystrokes

See Displayed

Using card 2

1000 A 25 C E	→	245.17 (joule/kg)
.85 X	→	208.39 (joule/kg)
20 ↑ 1000 X	→	20000 (kg/s)
X	→	4167826.25 (watts)


Solution(s)

Reference(s)


User Instructions

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CONSERVATION OF ENERGY-ENGLISH


 $\rho(\text{START})$ (lb/ft³)
 v (ft/sec)
 z (ft)
 P (psi)
 E (Btu)

CONSERVATION OF ENERGY-SI


 $\rho(\text{START})$ (kg/m³)
 v (m/s)
 z (m)
 P (N/m²)
 E (J/kg)

STEP	INSTRUCTIONS	INPUT DATA/UNITS	KEYS	OUTPUT DATA/UNITS
1	For English units (pounds, feet, seconds, Btus), enter		<input type="text"/> <input type="text"/>	
	Card 1. for SI units		<input type="text"/> <input type="text"/>	
	(kilograms, meters, seconds, watts), enter Card 2		<input type="text"/> <input type="text"/>	
2	Input fluid density	ρ	A <input type="text"/>	g
3	Input the following (negative values are downstream values):		<input type="text"/> <input type="text"/>	
	Fluid velocity	v	B <input type="text"/>	0.00
	Height from reference datum	z	C <input type="text"/>	0.00
	Pressure	P	D <input type="text"/>	0.00
	Energy input	E	E <input type="text"/>	0.00
4	Repeat step 3 for all input values		<input type="text"/> <input type="text"/>	
5	Calculate the unknown:		<input type="text"/> <input type="text"/>	
	Fluid velocity	0.00	B <input type="text"/>	v
	Height from reference datum	0.00	C <input type="text"/>	z
	Pressure	0.00	D <input type="text"/>	P
	Energy	0.00	E <input type="text"/>	E
6	For new case go to step 2, or		<input type="text"/> <input type="text"/>	
	store 0.00 in register 8 and go		<input type="text"/> <input type="text"/>	
	to step 3.		<input type="text"/> <input type="text"/>	

97 Program Listing I

STEP	KEY ENTRY	KEY CODE	COMMENTS	STEP	KEY ENTRY	KEY CODE	COMMENTS
001	*LBLA	21 11	Store ρ and constants	057	GT01	22 01	Energy
002	ST04	35 04		058	RCL8	36 08	
003	CLX	-51		059	RCL7	36 07	
004	ST08	35 08		060	=	-24	
005	7	07		061	RCL4	36 04	
006	7	07		062	X	-35	
007	8	08		063	RCL6	36 06	
008	.	-62		064	=	-24	
009	1	01		065	RTN	24	
010	6	06		066	*LBLB	21 15	
011	ST05	35 05	Velocity	067	ENT↑	-21	Summation
012	3	03		068	RCL5	36 05	
013	2	02		069	X	-35	
014	.	-62		070	RCL6	36 06	
015	1	01		071	X	-35	
016	7	07		072	0	00	
017	ST06	35 06		073	X#Y?	16-32	
018	RTN	24		074	GT01	22 01	
019	*LBLB	21 12		075	RCL8	36 08	
020	ENT↑	-21		076	RCL5	36 05	
021	ABS	16 31	Height	077	=	-24	
022	X	-35		078	RCL6	36 06	
023	2	02		079	=	-24	
024	=	-24		080	RTN	24	
025	0	00		081	*LBL1	21 01	
026	X#Y?	16-32		082	R↓	-31	
027	GT01	22 01		083	ST+8	35-55 08	
028	RCL8	36 08		084	0	00	
029	2	02		085	RTN	24	
030	X	-35					
031	JX	54	Pressure				
032	RTN	24					
033	*LBLC	21 13					
034	ENT↑	-21		090			
035	RCL6	36 06					
036	X	-35					
037	0	00					
038	X#Y?	16-32					
039	GT01	22 01					
040	RCL8	36 08					
041	RCL6	36 06					
042	=	-24					
043	RTN	24					
044	*LBLD	21 14		100			
045	ENT↑	-21					
046	1	01					
047	4	04					
048	4	04					
049	ST07	35 07					
050	X	-35					
051	RCL4	36 04					
052	=	-24					
053	RCL6	36 06					
054	X	-35		110			
055	0	00					
056	X#Y?	16-32					

REGISTERS

0	1	2	3	4	5	6	7	8	9
				ρ	778.16	g	144	ΣE	Used
S0	S1	S2	S3	S4	S5	S6	S7	S8	S9
A	B	C	D	E	I				

97 Program Listing II

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STEP	KEY ENTRY	KEY CODE	COMMENTS	STEP	KEY ENTRY	KEY CODE	COMMENTS
001	*LBLA	21 11	Store and gravity constant	057	R↓	-31	Summation
002	STO4	35 04		058	ST+8	35-55 08	
003	CLX	-51		059	0	00	
004	STO8	35 08		060	RTN	24	
005	9	09					
006	.	-62					
007	0	08					
008	0	00					
009	6	06					
010	6	06					
011	5	05					
012	STO6	35 06	Velocity	180			
013	RTN	24					
014	*LBLB	21 12					
015	ENT↑	-21					
016	ABS	16 31					
017	x	-35					
018	2	02					
019	=	-24					
020	0	00					
021	X#Y?	16-32		190			
022	GT01	22 01					
023	RCL8	36 08	Height				
024	2	02					
025	x	-35					
026	JX	54					
027	RTN	24					
028	*LBLC	21 13					
029	ENT↑	-21					
030	RCL6	36 06					
031	x	-35		200			
032	0	00					
033	X#Y?	16-32					
034	GT01	22 01	Pressure				
035	RCL8	36 08					
036	RCL6	36 06					
037	=	-24					
038	RTN	24					
039	*LBLD	21 14					
040	ENT↑	-21					
041	RCL4	36 04		210			
042	=	-24					
043	0	00					
044	X#Y?	16-32					
045	GT01	22 01	Energy				
046	RCL8	36 08					
047	RCL4	36 04					
048	x	-35					
049	RTN	24					
050	*LBLE	21 15					
051	ENT↑	-21					
052	0	00		220			
053	X=Y?	16-33					
054	RCL8	36 08					
055	RTN	24					
056	*LBL1	21 01					

LABELS					FLAGS	SET STATUS				
A	B	C	D	E	0	FLAGS		TRIG	DISP	
p	v	z	P	E		ON	OFF			
a	b	c	d	e	1	0	<input type="checkbox"/>	<input checked="" type="checkbox"/>	DEG <input checked="" type="checkbox"/>	FIX <input checked="" type="checkbox"/>
0	1	2	3	4	2	1	<input type="checkbox"/>	<input checked="" type="checkbox"/>	GRAD <input type="checkbox"/>	SCI <input type="checkbox"/>
5	6	7	8	9	3	2	<input type="checkbox"/>	<input checked="" type="checkbox"/>	RAD <input type="checkbox"/>	ENG <input type="checkbox"/>
						3	<input type="checkbox"/>	<input checked="" type="checkbox"/>		n-2

NOTES

Hewlett-Packard Software

In terms of power and flexibility, the problem-solving potential of the Hewlett-Packard line of fully programmable calculators is nearly limitless. And in order to see the practical side of this potential, we have several different types of software to help save you time and programming effort. Every one of our software solutions has been carefully selected to effectively increase your problem-solving potential. Chances are, we already have the solutions you're looking for.

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Business Decisions
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Navigation

Users' Library

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Each of the Books, containing up to 15 programs without cards, is priced at \$10.00, a savings of up to \$35.00 over single copy cost.

The Users' Library Solutions Books will compliment our other applications of software and provide you with a valuable new tool for program solutions.

Options/Technical Stock Analysis
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Real Estate Investment
Taxes
Home Construction Estimating
Marketing/Sales
Home Management
Small Business
Antennas
Butterworth and Chebyshev Filters
Thermal and Transport Sciences
EE (Lab)
Industrial Engineering
Aeronautical Engineering
Control Systems
Beams and Columns
High-Level Math
Test Statistics
Geometry
Reliability/QA

Medical Practitioner
Anesthesia
Cardiac
Pulmonary
Chemistry
Optics
Physics
Earth Sciences
Energy Conservation
Space Science
Biology
Games
Games of Chance
Aircraft Operation
Aviation
Calendars
Photo Dark Room
COGO-Surveying
Astrology
Forestry

ENERGY CONSERVATION

This is a book with heat transfer calculations for everyday use. Programs include basic heat transfer calculations in conduction, convection, and radiation.

AIR COOLING SYSTEM DESIGN

BLACK BODY THERMAL RADIATION

ECONOMIC INSULATION THICKNESS

HEAT TRANSFER THROUGH COMPOSITE CYLINDERS AND
WALLS

STEADY STATE COND. HEAT TRANS., HEAT LOAD &
LOGARITHMIC MEAN

SUN ALTITUDE, AZIMUTH, SOLAR POND ABSORPTION

TOTAL DAILY AMOUNT OF SOLAR RADIATION

TEMPERATURE OR CONCENTRATION PROFILE FOR A
SEMI-INFINITE SOLID

TRANSIENT TEMPERATURE DISTRIBUTION IN A SEMI-
INFINITE SOLID WITH CONVECTION BOUNDARY CONDITION

CONSERVATION OF ENERGY

